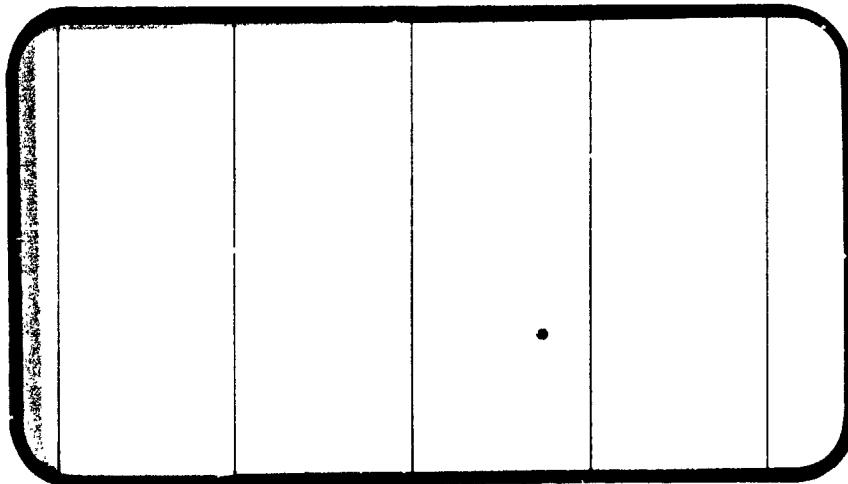




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AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER  
HOUSTON, TEXAS

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SPACE DIVISION  CHRYSLER  
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EFFECTS OF AIR BREATHING ENGINE PLUMES  
ON SSV ORBITER SUBSONIC  
WING PRESSURE DISTRIBUTION (OA57B)

VOLUME 2 of 2

By

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Rockwell International

Prepared under NASA Contract Number NAS9-13247

by

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for

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EFFECTS OF AIR BREATHING ENGINE PLUMES  
ON SSV ORBITER SUBSONIC  
WING PRESSURE DISTRIBUTION (OA57B)

By T. Soard\*

ABSTRACT

The data presented in this report were obtained during wind tunnel tests of a 0.0405-scale model of the -89B Ferry Configuration of the Space Shuttle Vehicle Orbiter. These tests were conducted in the Rockwell International Low Speed Wind Tunnel (NAAL) during the time period of September 18 to September 23, 1973. NASA Space Shuttle test designation was OA57B.

The primary test objective was to investigate orbiter wing pressure distributions resulting from nacelle plumes above and below the wing. Three six-engine nacelle configurations were tested. One configuration had a twin-podded nacelle mounted above each wing and the others had one mounted below each wing. Both had a centerline twin-podded nacelle mounted below the wing. Wing pressure distribution was determined by locating static pressure bugs on the upper and lower surfaces of the left wing. Pressure bugs were also located on the upper and lower surfaces of the body flap and on the B<sub>12</sub> afterbody fairing when it was installed. Base and balance cavity pressures were recorded and a strain gage instrumented beam in the right wing measured elevon hinge moments and normal forces.

Testing was conducted at 3 ground plane heights ( $h/b = 0.039, 0.125,$

\* Rockwell International

and 0.286), with 4 engine pressure ratios ( $F_{TN}/P_{so} = 0, 1.0, 1.3$ , and  $1.5$ ), with elevon deflections of  $0^\circ$  and  $\pm 15^\circ$ , and with body flap deflections of  $-18^\circ$ ,  $0^\circ$ , and  $+20^\circ$ . The nominal angle of attack range was  $-4^\circ$  to  $+20^\circ$  with an angle of sideslip of  $0^\circ$ . A Mach number of 0.200 was maintained throughout the test.

The model was mounted on a 2.5-inch diameter dummy balance using the W-1052-5 sting and W-1092-A-2 adapter, locating the center of rotation at the trailing edge of the root chord.

This report is presented in two volumes. Volume 1 contains the data figures and Volume 2 contains the tabulated source data.

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**SCHEDULE OF COEFFICIENTS PLOTTED:**

- (A) CMWING, CNWING versus ELEVON
- (B) CHNE, CNE versus ALPHA
- (C) CPBF versus ALPHA
- (D) CMWING, CNWING, CNFLAP, CMFLAP, CNTOTL, CMTOTL versus ALPHA
- (E) CAB, CPBAVE versus ALPHA
- (F) CNABDY, CNABDY, CMTOTL, CNTOTL versus ALPHA
- (G) CP versus 2Y/B
- (H) CP versus X/C
- (I) DELCP versus X/C
- (J) CP versus X/L

## NOMENCLATURE

<u>Symbol</u>	<u>SADSAC Symbol</u>	<u>Definition</u>
A		area of influence, ft <sup>2</sup>
b	BREF	reference span, inches
C <sub>A</sub>	CA	axial-force coefficient, axial force/qS
C <sub>HM</sub>	CHME	hinge-moment coefficient, hinge moment/qS <sub>E</sub> ̄C <sub>E</sub>
C <sub>N</sub>	CN	normal-force coefficient, normal force/qS
C <sub>m</sub>	CLM	pitching-moment coefficient, pitching moment/qS̄C
C <sub>p</sub>	CP	pressure coefficient, P <sub>sl</sub> - P <sub>s0</sub> /q
C.P.		center of pressure
GN <sub>2</sub>		nitrogen gas
XCP		distance from MRP to C.P. on X axis, inches
̄c	LREF	reference length, inches
h <sub>WTE</sub> /b	H/B	ratio of height of wing trailing edge above ground plane to reference span
l		distance from MCR to static pressure tap
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis
P		pressure, psf
q	Q(PSF)	dynamic pressure, psf
S	SREF	reference area, ft <sup>2</sup>
T		temperature, °R

NOMENCLATURE (Continued)

$\alpha$	ALPHA	angle of attack, degrees
$\beta$	BETA	angle of sideslip, degrees
$\gamma$		ratio of specific heats, $\gamma = 1.4$ for air
$\Lambda_{LE}$		leading edge sweep angle, degrees
	CWING	wing pitching moment coefficient
	CNWING	wing normal force coefficient
$\delta_e$	ELEVON	elevon, surface deflection angle positive deflection, trailing edge down; degrees
$\Delta C_p$	DELCP	incremental pressure coefficient
$C_{A_b}$	CAB or CABL	base force coefficient, $\frac{\text{base force}}{qS} = A_b (p_b - p_0) / qS$
$C_{N_E}$	CNE	elevon normal force coefficient
	CPBF	body flap pressure coefficient
	CPBAVE	average base pressure coefficient
	CNTOTL	total normal force coefficient
	CMTOTL	total pitching moment coefficient
$C_p$	CP	pressure coefficient $(p_l - p_0) / q$
$x/c$	X/C	local chordwise position/wing chord length
$Y/b/2$	2Y/B or n	local spanwise position/wing semi-span
	X/L	local chordwise position/reference length
	CNFLAP	body flap normal coefficient
	CMFLAP	body flap pitching moment coefficient
	CNABDY	afterbody normal force coefficient
	CMABDY	afterbody pitching moment coefficient

NOMENCLATURE (Concluded)

$P_{TN}/P_o$	PTN/P	total nozzle pressure/free stream pressure
$R_e$	RN/L	Reynold's number, RN/L, millions per foot
$\delta_{BF}$	BDFLAP	flap, surface deflection angle, positive
$l_B$		body length, inches
	CPSB	base pressure coefficient
	CPSAB	side afterbody pressure coefficient
	CPBAB	bottom afterbody pressure coefficient

SUBSCRIPTS

B	base
BC	balance cavity
E	elevon
HM	hinge moment
i	index, value at station i
L	lower surface
l	local
N	nozzle
o	freestream
p	pressure
s	static
T	total
U	upper

## CONFIGURATIONS INVESTIGATED

The model tested was a 0.0405-scale representation of the Rockwell International-89B Space Shuttle Orbiter. The basic model was of the blended wing-body design utilizing a double delta wing ( $75^\circ/45^\circ \Lambda_{LE}$ ), full span elevons (unswept hingeline), and a canopy. To complete the ferry configuration, air breathing engine nacelles were tested in locations above and below the wing as per SS-A00028. Jet plumes were simulated by exhausting compressed  $\text{GN}_2$  from all nacelles.

All model components were per VL70-00089B configuration except for the fuselage lines from station 1307 aft and the engine nacelle groupings and locations.

The orbiter model was constructed of wood and aluminum and was mounted on the Task Corporation 2.5-inch Mk IX dummy balance. The following nomenclature was used to designate the various model components:

Component	Description
B <sub>16</sub>	-89B fuselage
B <sub>12</sub>	-89B fuselage with base fairing
C <sub>5</sub>	-89B canopy
F <sub>1</sub>	Body flap, ATP baseline
J <sub>40</sub>	Air breathing propulsion system consisting of a twin podded nacelle below each wing and a twin podded nacelle on the lower fuselage centerline
J <sub>41</sub>	Same as J <sub>40</sub> except engines under wings extended aft an additional 90 inches full scale

- J<sub>42</sub> Air breathing propulsion system consisting of a twin podded nacelle above each wing and a twin podded nacelle on the lower fuselage centerline
- W<sub>87</sub> -89B double delta wing ( $75^\circ/45^\circ \Lambda_{LE}$ )
- E<sub>18</sub> Elevon, full span split, used with wing W<sub>87</sub>

## TEST FACILITY

The North American Aerodynamics Laboratory (NAAL) 7.75 x 11.0-Foot Wind Tunnel is a continuous flow, closed circuit, single return type tunnel capable of speeds up to 200 miles per hour. The test section is vented to atmospheric pressure and is 7.75 x 11 feet wide by 12 feet in length. Power is supplied by a 1250 horsepower nacelle mounted synchronous motor driving a 19 foot, seven blade, laminated birch propeller. The airspeed is controlled by varying the degree of coupling between the motor and propeller by means of a magnetic clutch. A damping screen and honeycomb section in the settling chamber upstream from the contraction cone (ratio 7.53 to 1) minimizes turbulence in the test section. The NAAL Wind Tunnel has been in operation since June 1943 and calibrations are available over a wide range of test conditions.

Tests may be conducted using a variety of mounting systems, e.g., a single strut, double strut, sting strut, reflection plane, cable suspension, and two dimensional wall. Aerodynamic data may be measured by a planar type external balance. An automatic data Acquisition System is used to collect, multiplex, digitize, and record 50 channels of force and/or pressure data on magnetic tape. This data is then rapidly reduced and plotted using automatic data processing equipment and an automatic digital plotter.

## DATA REDUCTION

Since only pressure distribution and elevon hinge data were required during the test period, the model was mounted on the Task Mk IX 2.5" dummy balance. Therefore no corrections were made for balance and sting deflection. But standard facility corrections for blockage were applied as required. A base and balance cavity axial force coefficient was calculated as presented below:

$$C_{A_{BC}} = - \left( \frac{P_{BC} - P_{SO}}{q} \right) \left( \frac{A_{BC}}{S} \right)$$

and:

$$C_{A_{Bl}} = - \left( \frac{P_B - P_{SO}}{q} \right) \left( \frac{A_B}{S} \right), P_B = 1/5 (P_{B1} + \dots + P_{B5})$$

Elevon hinge moments and normal force coefficients were calculated in the following manner:

$$C_{HM} = \frac{HM_E}{qS_E c_E}$$

where:  $HM_E$  = (gage output)x(calib. factor), in-lbs

$S_E$  = Elevon area,  $\text{ft}^2$

$c_E$  = M.A.C. of elevon, in

and:

$$C_{N_E} = \frac{HM_E}{qS(c_E/2)}$$

$S$  = Reference area,  $\text{ft}^2$

## DATA REDUCTION (Continued)

Static pressure coefficients were calculated as shown below:

$$c_{p_i} = \left( \frac{P_{s_i} - P_{s_0}}{q} \right)$$

where:

$P_{s_i}$  = Local static pressure, psf

$P_{s_0}$  = Tunnel static pressure, psf

$q$  = Tunnel dynamic pressure, psf

Wing  $i_{MAX} = 30$

Body flap  $i_{MAX} = 3$

Afterbody  $i_{MAX} = 3$

$$c_N = \sum_{i=1}^{i_{MAX}} (c_{p_{i_u}} - c_{p_{i_l}}) \frac{A_{p_i}}{S}$$

where:  $c_{p_{i_u}}$  = Upper surface pressure coefficient  
 $c_{p_{i_l}}$  = Lower surface pressure coefficient  
 (assume afterbody  $c_{p_{i_u}} = 0.0$ )

$c_{p_{i_L}}$  = Lower surface pressure coefficient

$A_{p_i}$  = Area of influence,  $\text{ft}^2$ , (see tables IV and V for values)

$S$  = Reference area,  $\text{ft}^2$

and:

$$c_m = \sum_{i=1}^{i_{MAX}} c_{N_i} \frac{l_i}{\bar{c}}$$

DATA REDUCTION (Continued)

where:

$l_i$  = Distance of tap from moment reference point, in, see tables IV and V

$\bar{c}$  = Reference length, in

and:

$$x_{CP} = \frac{C_m}{C_N} (\bar{c})$$

and:

$$C_N \text{ Total} = C_{N_{\text{Wing}}} + C_{N_{\text{Body flap}}} + C_{N_{\text{Afterbody}}}$$

and:

$$C_m \text{ Total} = C_{m_{\text{Wing}}} + C_{m_{\text{Body flap}}} + C_{m_{\text{Afterbody}}}$$

and:

$$C_m/C_N \text{ Total} = C_m/C_{N_{\text{Wing}}} + C_m/C_{N_{\text{Body flap}}} + C_m/C_{N_{\text{Afterbody}}}$$

The following reference dimensions and constants were used in data reduction.

Basic constants:

Symbol	Definition	Model Scale
S	Reference area, ft <sup>2</sup>	4.412
b	Reference span, in	37.935
$\bar{c}$	Reference length, in	19.230
$S_E$	Elevon reference area, ft <sup>2</sup>	0.336
$\bar{c}_E$	Elevon reference length, in	3.44

DATA REDUCTION (Concluded)

A <sub>N</sub>	Nacelle nozzle exit area, ft <sup>2</sup>	0.01278
XMRP	Moment reference point on X axis, in	43.598
XMRP	Moment reference point on Y axis, in	0.0
ZMRP	Moment reference point on Z axis, in	-0.405

TABLE I.

TABLE II.

TEST: DA578 (WAAL 7/3) DATA SET/RUN NUMBER COLLATION SUMMARY

DATA SET IDENTIFIER	CONFIGURATION	SCHED.	PARAMETERS/VALUES						NO. OF RUNS	MACH NUMBERS .20
			$\alpha$	$\beta$	$\gamma_{1b}$	$\delta_f$	$\delta_e$			
ENV 011	B16 C5F,J40 W/87E/8	10	0	VAR	037	-18	0		1	1
002		10	VAR							2
003		A	1.0							3
004			1.3							4
005			1.5	▼						5
006		B	1.0	.125						6
007			1.3							7
008			1.5	▼						8
009			1.5	0						9
010			1.3							10
011			1.0	▼						11
012			1.5	.286						12
013			1.0							13
014			1.3	▼						14
015			1.0	-18						15
016			1.5							16
017			1.3							17
018			1.0	▼						18
1										
7										
13										
19										
25										
31										
37										
43										
49										
55										
61										
67										
75.7										

EFFICIENTS

$$\alpha(A) = 10, 15, 20$$

$$\alpha(B) = -4, 0, 5, 10, 15, 20$$

\* DEPENDENT VARIABLES ARE LISTED IN THE TABULATED SOURCE DATA (VOLUME 2).

TABLE II. — Continued.

TEST : #A578 (MAC 7/3)

## DATA SET/RUN NUMBER COLLATION SUMMARY

DATE: SEPT 18-23, 1973

TABLE II. — Continued.

TEST: #A578 (WEEK 7/3)

## DATA SET/BIN NUMBER CORRELATION SUMMARY

DATE : SEPT 18-23, 1973

\*DEPENDENT VARIABLES ARE LISTED IN THE REGULATED SOURCE DATA (VOLUME 2).

$$\begin{array}{l} \alpha(A) = 10, 15, 20 \\ \alpha(B) = -4, 0, 5, 10 \end{array}$$

$$\begin{aligned}\alpha(A) &= 10/15, 20 \\ \alpha(B) &= -4, 0, 5, 10, 15, 20\end{aligned}$$

OFFICE

TABLE II. - Continued.

TEST: **AA578 (NAAL 7/3)** DATA SET/RUN NUMBER COLLATION SUMMARY DATE: **SEPT 18-23, 1973**

\* DEPENDENT VARIABLES ARE LISTED IN THE TABULATED SOURCE DATA (VOLUME 2).

$$\begin{aligned}\alpha(A) &= 10, 15, 20 \\ \alpha(B) &= -4, 0, 5, 10\end{aligned}$$

**a or b**

CONFERENCE

15 APRIL 1945

TABLE II. - Concluded.

TEST : 0A57B (NAAL 7/3)

## DATA SET/RUN NUMBER COLLATION SUMMARY

$$\begin{aligned}\alpha(4) &= 10, 13, 28 \\ \alpha(3) &= -4, 0, 5, 10, 15, 20\end{aligned}$$

100

\*DEPENDENT VARIABLES ARE LISTED IN THE TABULATED SOURCE DATA (VOLUME 2).

TABLE III. - MODEL DIMENSIONAL DATA

MODEL COMPONENT: BODY - B12 BodyGENERAL DESCRIPTION: Fuselage, 2A configuration with base fairing,  
lightweight orbiter per Rockwell lines VL70-000103

Scale Model = 0.0405

DRAWING NUMBER: SS-A00102  
VL70-000103

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length - in.	1705.0	69.05250
Max. Width - in.		
Max. Depth - in.	248.0	10.0440
Fineness Ratio	6.875	6.875
Area - $\text{Ft}^2$		
Max. Cross-Sectional	355.278	0.58274
Planform		
Wetted		
Base		

TABLE III. - Continued.

MODEL COMPONENT: BODY - Bl6

GENERAL DESCRIPTION: -89B Fuselage

Scale Model = 0.0405

DRAWING NUMBER: VI72-000089

DIMENSIONS:

	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length - in.	<u>1328.3</u>	<u>53.796</u>
Max. Width - in.	<u>          </u>	<u>          </u>
Max. Depth - in.	<u>248.0</u>	<u>10.044</u>
Fineness Ratio - in.	<u>5.35605</u>	<u>5.35605</u>
Area - Ft <sup>2</sup>		
Max. Cross-Sectional	<u>355.278</u>	<u>0.583</u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>

TABLE III. - Continued.

MODEL COMPONENT: CANOPY - C5GENERAL DESCRIPTION: -89B CanopyScale Model = 0.0405

DRAWING NUMBER

VL70-000092DIMENSION:FULL SCALEMODEL SCALELength (Sta Fwd Bulkhead) 391.0Max Width (T.E. Bulkhead) 560.0Max Depth (WPZ<sub>o</sub> = 421.922 to Z<sub>o</sub> = 500)   Fineness Ratio   Area   Max Cross-Sectional   Planform   Wetted   Base   Sta. Fwd. Bulkhead, Fus. Sta 391.00 15.836Sta. T.E. , Fus. Sta. 560. 22.680

TABLE III. - Continued.

MODEL COMPONENT:	<u>Body Flap - F1</u>	
GENERAL DESCRIPTION:	<u>Body Flap Located on Lower Aft Portion of Fuselage Trailing Edge</u>	
Scale Model = .0405		
DRAWING NUMBER:	<u>VL70-000003A</u>	
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Length in.	<u>236.54</u>	<u>9.580</u>
Flap L.E. Fus. Sta. in.	<u>1528.30</u>	<u>61.896</u>
Flap T.E. Fus. Sta. in.	<u>1650.56</u>	<u>66.848</u>
Span in.	<u>236.54</u>	<u>9.580</u>
Area ft <sup>2</sup>		
Max. Cross-Sectional		
Planform	<u>199.75</u>	<u>0.328</u>
Wetted		
Base		

TABLE III. - Continued.

MODEL COMPONENT: ELEVON E-18GENERAL DESCRIPTION: Unswept hingeline elevon used on Wing W67Scale Model = 0.0405DRAWING NUMBER: VL70-000093

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area - FT <sup>2</sup>	<u>205.52</u>	<u>0.337</u>
Span (equivalent) - IN.	<u>353.34</u>	<u>14.310</u>
Inb'd equivalent chord	<u>114.78</u>	<u>4.649</u>
Outb'd equivalent chord	<u>55.00</u>	<u>2.228</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>.208</u>	<u>.208</u>
At Outb'd equiv. chord	<u>.400</u>	<u>.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>0.00</u>	<u>0.00</u>
Tailing Edge	<u>-10.24</u>	<u>-10.24</u>
Hingeline	<u>0.00</u>	<u>0.00</u>
Area Moment (Normal to hinge line) - FT <sup>3</sup>	<u>1548.07</u>	<u>2.539</u>
Product of Area Moment		

TABLE III. - Continued.

MODEL COMPONENT: AIR BREATHING PROPULSION SYSTEM - J<sub>40</sub>GENERAL DESCRIPTION: One Twin-podded nacelle under each wing plus one  
bottom centerline twin-podded nacelle.

Scale: 0.0405

DRAWING NUMBER: SS-A00028

DIMENSIONS:	FULL SCALE	MODEL SCALE
Length - In.	<u>231.580</u>	<u>9.379</u>
Max. Width - In.	<u>52.000</u>	<u>2.106</u>
Max. Depth - In.	<u>52.000</u>	<u>2.106</u>
Fineness Ratio		
Area		
Max Cross-Sectonal - IN. <sup>2</sup>	<u>2123.717</u>	<u>3.483</u>
Nozzle - In. <sup>2</sup>	<u>1122.327</u>	<u>1.841</u>
Wetted		
Base		
Forebody Nose	Wing	Q
Model Station - In.	<u>38.88</u>	<u>38.88</u>
Water Line - In.	<u>10.328</u>	<u>9.577</u>
Thrust Line	OUTB'D	INB'D
Butt Line-In	<u>11.117</u>	<u>8.567</u>
Incidence - D.g.	<u>+4°</u>	<u>+4°</u>

TABLE III. - Continued.

MODEL COMPONENT: AIR BREATHING PROPULSION SYSTEM - J<sub>41</sub>

GENERAL DESCRIPTION: One twin-podded nacelle above each wing plus one bottom centerline twin podded nacelle. Engines 1, 2, 5, and 6 have been extended 90 inches full-scale.

SCALE: 0.0405

DRAWING NUMBER: SS-A00028

## DIMENSIONS:

	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length - In.	<u>321.58</u>	
Max. Width	<u>52.000</u>	<u>2.106</u>
Max. Depth	<u>52.000</u>	<u>2.106</u>
Fineness Ratio		
Area		
Max. Cross-Sectional - In. <sup>2</sup>	<u>2123.717</u>	<u>3.843</u>
Nozzle - In. <sup>2</sup>	<u>1122.327</u>	<u>1.841</u>
Wetted		
Base		
Forebody Nose	<u>Wing</u>	<u>C</u>
Model Station - In.	<u>38.86</u>	<u>38.86</u>
Water Line - In.	<u>15.15</u>	<u>9.577</u>
Thrust Line	<u>Out'd</u>	<u>Inb'd</u>
Butt Line - In.	<u>11.117</u>	<u>8.567</u>
Incidence - Deg.	<u>0°</u>	<u>+ 4°</u>

TABLE III. - Continued.

MODEL COMPONENT: AIR BREATHING PROPULSION SYSTEM - J42

GENERAL DESCRIPTION: One twin-podded nacelle above each wing plus one bottom centerline twin-podded nacelle.

SCALE: 0.0405

DRAWING NUMBER: SS-A00028

DIMENSIONS:	FULL SCALE	MODEL SCALE
Length	<u>231.580</u>	<u>9.379</u>
Max. Width	<u>52.000</u>	<u>2.106</u>
Max. Depth	<u>52.000</u>	<u>2.106</u>
Fineness Ratio	—	—
Area	—	—
Max. Cross-Sectional - In. <sup>2</sup>	<u>2123.717</u>	<u>3.843</u>
Nozzle - In. <sup>2</sup>	<u>1122.327</u>	<u>1.841</u>
Wetted	—	—
Base	—	—
Forebody Nose	WING	Q
Model Station - In.	<u>38.88</u>	<u>38.88</u>
Water Line - In.	<u>15.15</u>	<u>9.577</u>
Thrust Line	Outb'd	Inb'd
Butt Line - In.	<u>11.117</u>	<u>8.567</u>
Incidence - Deg.	<u>0°</u>	<u>+ 4°</u>

TABLE III. - Concluded.

MODEL COMPONENT: WING, W87

GENERAL DESCRIPTION: Double delta wing ( $75^\circ / 45^\circ$  ALE)

SCALE MODEL: 0.0405

DRAWING NUMBER: VL70-000093

## DIMENSIONS:

	FULL-SCALE	MODEL SCALE
<b>TOTAL DATA</b>		
Area - ft <sup>2</sup>	2689.38	4.411
Planform		
Wetted		
Span - equivalent	77.17	3.125
Aspect Ratio	2.214	2.214
Rate of Taper	1.176	1.176
Taper Ratio	0.209	0.209
Dihedral Angle, deg. @ X/C=75.33%	3.861	3.861
Incidence Angle, degrees	3.000	3.000
Aerodynamic Twist, degrees	--	--
Toe-In Angle	--	--
Cant Angle	--	--
Sweep-Back Angles, degrees		
Leading Edge	44.873	44.873
Trailing Edge	-10.242	-10.242
0.25 Element Line	35.050	35.050
Chords:		
Root (Wing Sta. 0.0)	690.19	27.953
Tip (equivalent)	144.30	5.844
MAC	476.76	19.309
Fus. Sta. of .25 MAC	1136.12	46.013
W.P. of .25 MAC	289.44	11.722
B.L. of .25 MAC	181.03	7.330
Airfoil Section		
Root	--	--
Tip	--	--
<b>EXPOSED DATA</b>		
Area - ft <sup>2</sup>	1746.89	2.865
Span (equivalent) - ft.	59.16	2.396
Aspect Ratio	2.004	2.004
Taper Ratio	0.256	0.256
Chords - in.		
Root	562.77	22.792
Tip	144.30	5.844
MAC	394.81	15.990
Fus. Sta. of .25 MAC	1185.17	47.000
W.P. of .25 MAC	291.56	11.808
B.L. of .25 MAC	250.54	10.147
Leading Edge Cuff		
Planform Area - ft <sup>2</sup>	121.42	0.199
L.E. Intersects Fus. @ Sta.	560.00	22.860
L.E. Intersects Wing @ Sta.	1035.00	41.918

TABLE IV. - WING PRESSURE CONSTANTS AND LOCATIONS

$c_{p_i}$	x/c	$\eta$	$A_{p_i}/S$	$l_i/\bar{c}$
i=1	15	0	.0732	.282
7	15	33.4	.0592	.019
13	15	52.0	.0355	-.131
19	15	66.3	.0355	-.246
25	15	87.6	.0236	-.418
2	30	0	.0489	-.065
8	30	33.4	.0395	-.122
14	30	52.0	.0236	-.247
20	30	66.3	.0223	-.332
26	30	87.6	.0157	-.460
3	45	0	.0489	-.152
9	45	33.4	.0395	-.263
15	45	52.0	.0236	-.363
21	45	66.3	.0223	-.419
27	45	87.6	.0157	-.502
4	60	0	.0489	-.369
10	60	33.4	.0395	-.404
16	60	52.0	.0236	-.479
22	60	66.3	.0223	-.505
28	60	87.6	.0157	-.544
5	75	0	.0489	-.586
11	75	33.4	.0395	-.545
17	75	52.0	.0236	-.595
23	75	66.3	.0223	-.592
29	75	87.6	.0157	-.586
6	90	0	.0489	-.803
12	90	33.4	.0395	-.686
18	90	52.0	.0236	-.711
24	90	66.3	.0223	-.679
30	90	87.6	.0157	-.628

Note: Values identical for upper and lower surface

TABLE V. - BODY FLAP AND AFTERBODY PRESSURE CONSTANTS

Body Flap Pressure Constants and Locations

Location	$A_{p_i}/S$	$l_i/\bar{c}$
1	.0170	-1.0399
2	.0160	-1.0399
3	.0170	-1.0399

Note: Values identical for upper and lower surface

Afterbody Pressure Constants and Locations\*

Location	$A_{p_i}/S$	$l_i/\bar{c}$
1	.0479	-1.030
2	.0479	-1.191
3	.0479	-1.373

- Note:
1. Values identical for side and lower surfaces
  2. Assume that all  $C_p$  upper = 0.0 for afterbody only
  3. Side not utilized for integration

\* Afterbody installed during runs 71 and 72 only

**Notes:**

- Positive directions of force coefficients, moment coefficients, and angles are indicated by arrows
- For clarity, origins of wind and stability axes have been displaced from the center of gravity

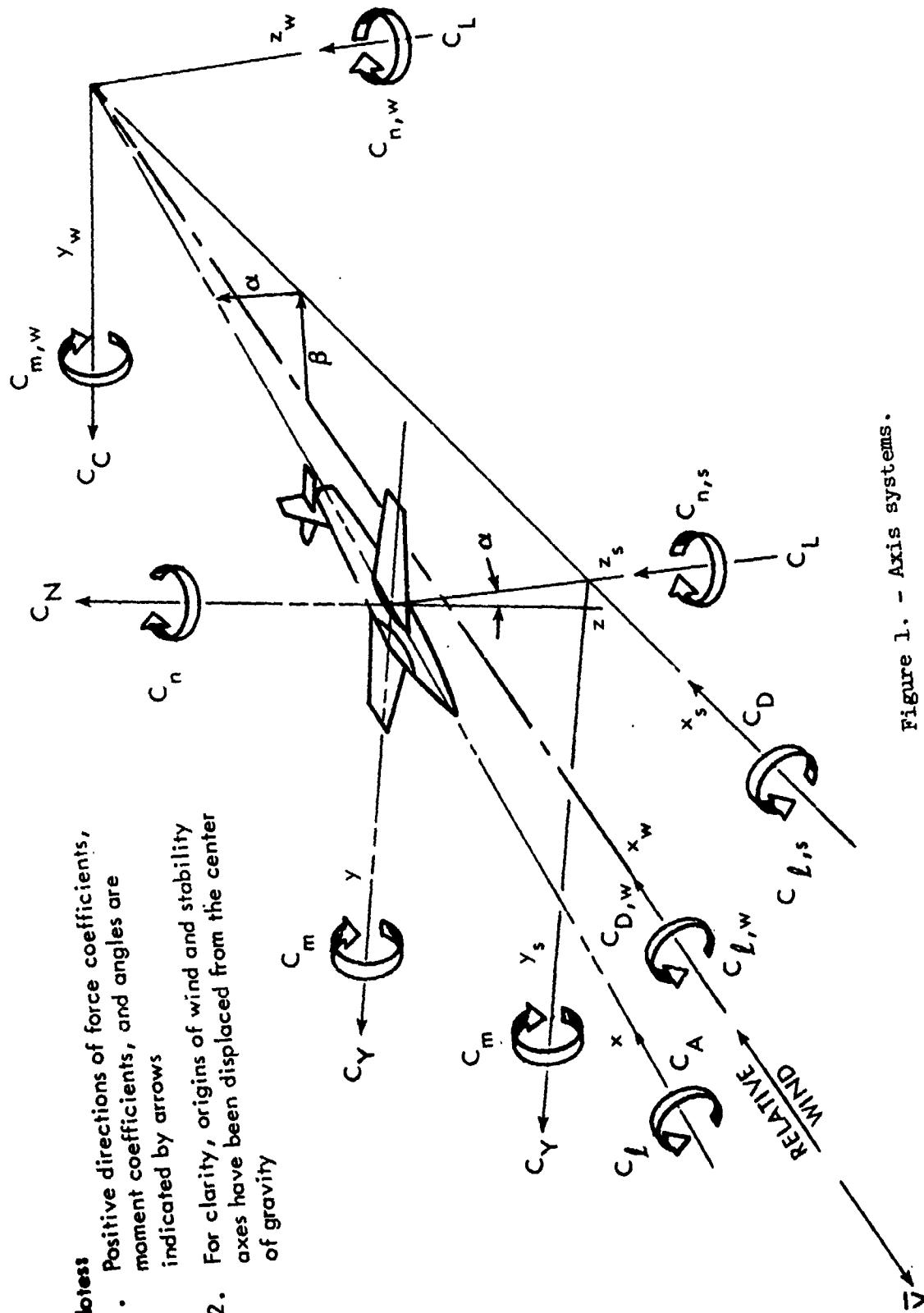
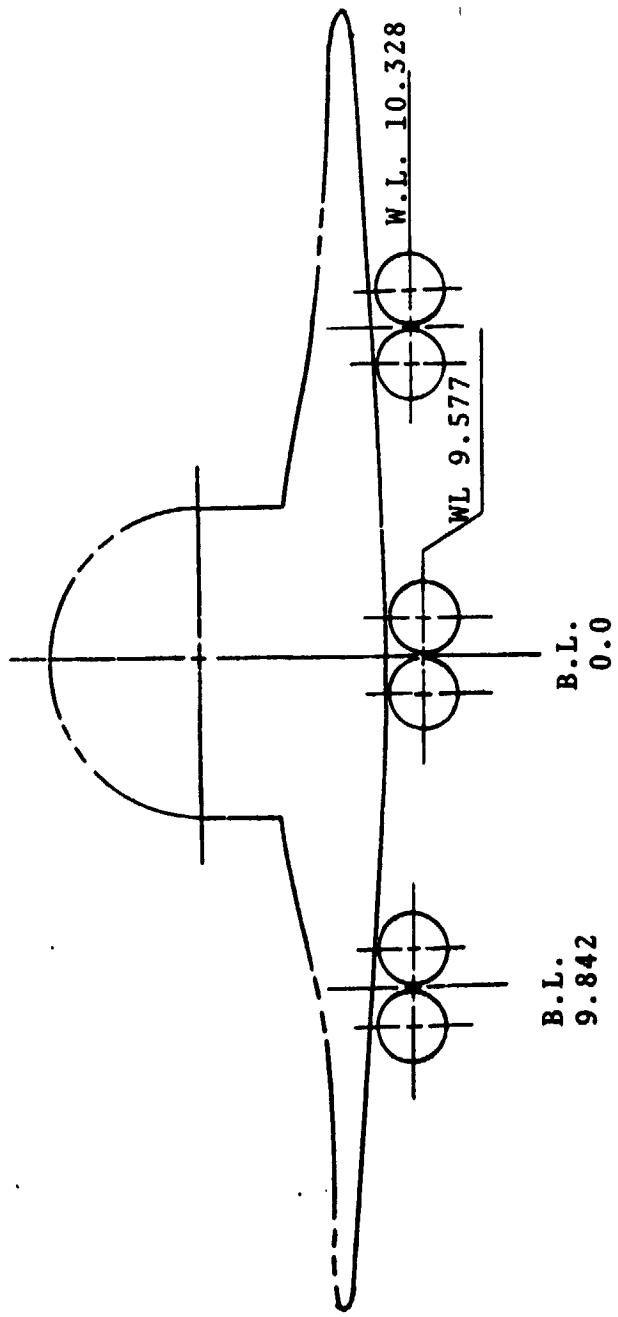


Figure 1. - Axis systems.

J40

J41 - As Shown With 90-Inch (Full Scale) Insert In Engines 1, 2, 5, and 6.

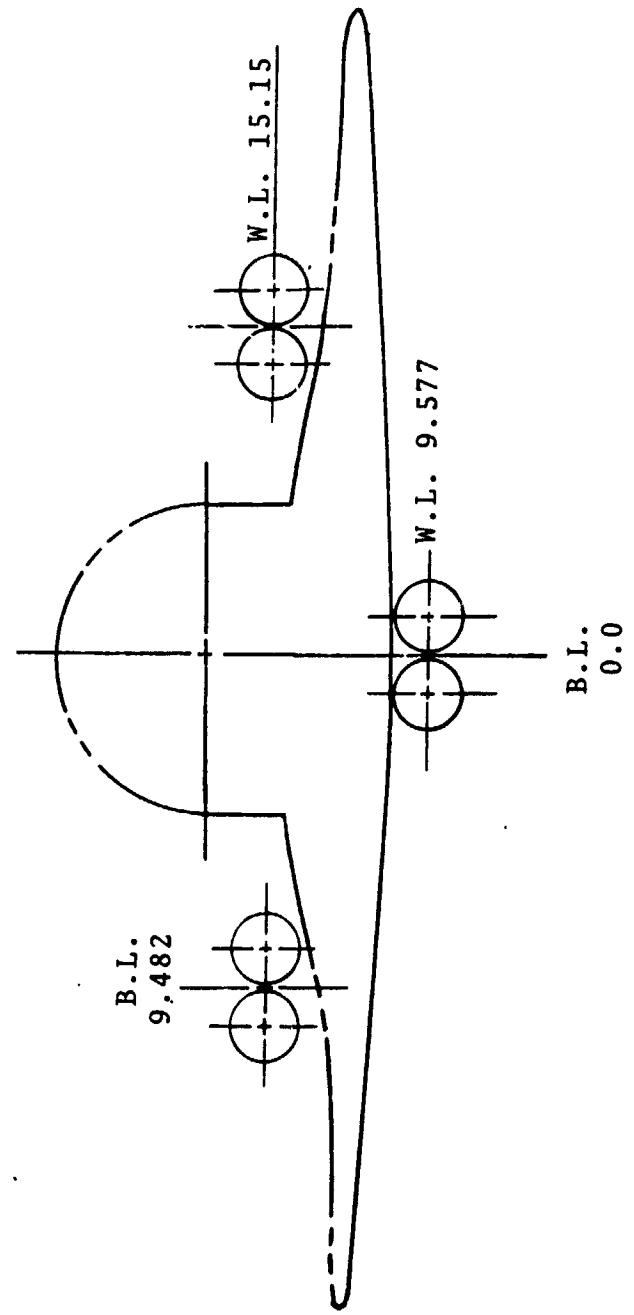


41

Nose at M.S. 38.88 (6 PLCS)

- a. J40 and J41 nacelle configuration  
Figure 2. - Model sketches.

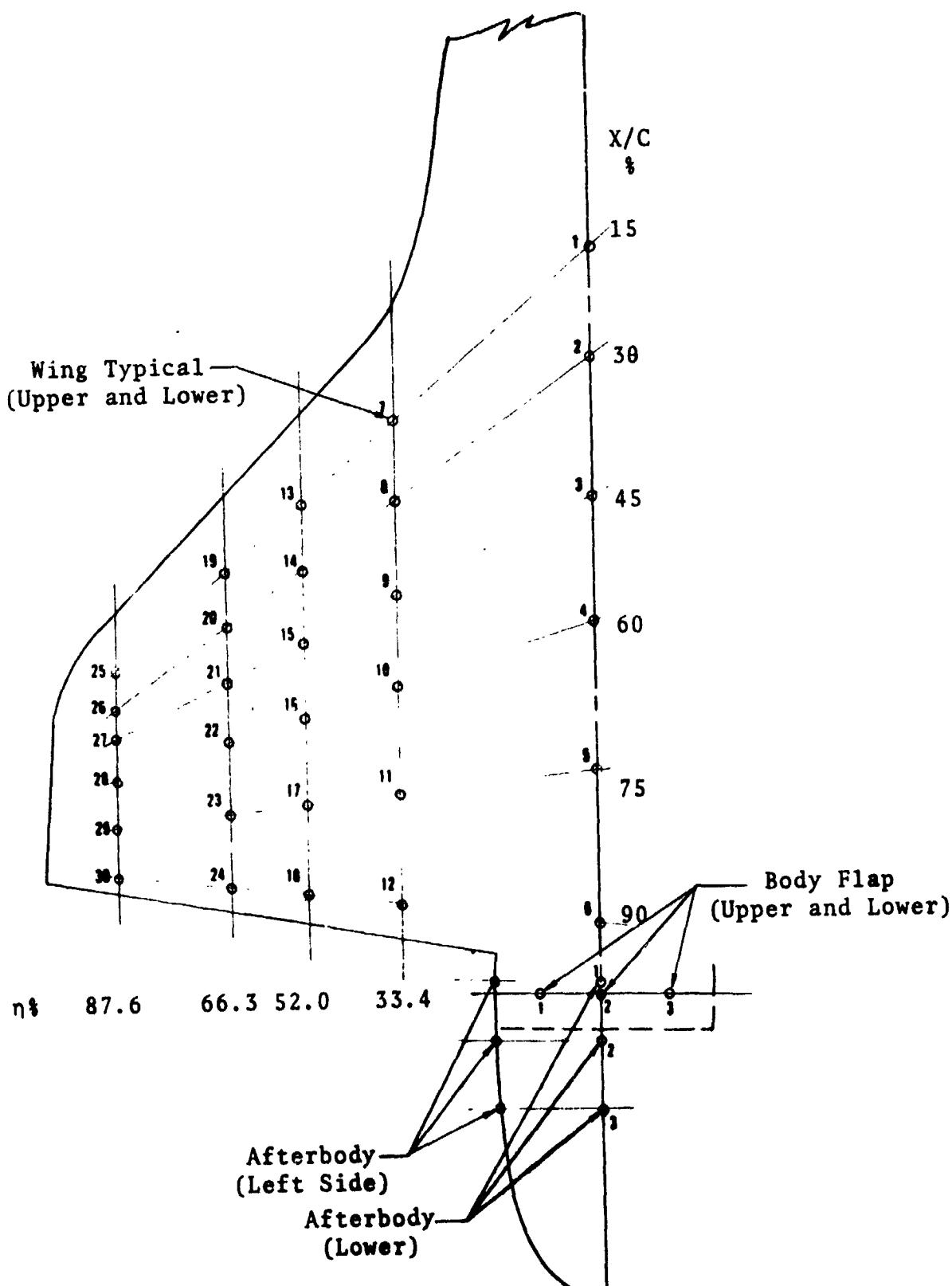
J42



42

Nose at M.S. 38.88 (6 PLCS)

- b. J42 nacelle configuration.  
Figure 2. - Continued.



c. Pressure bug locations.

Figure 2. - Concluded.

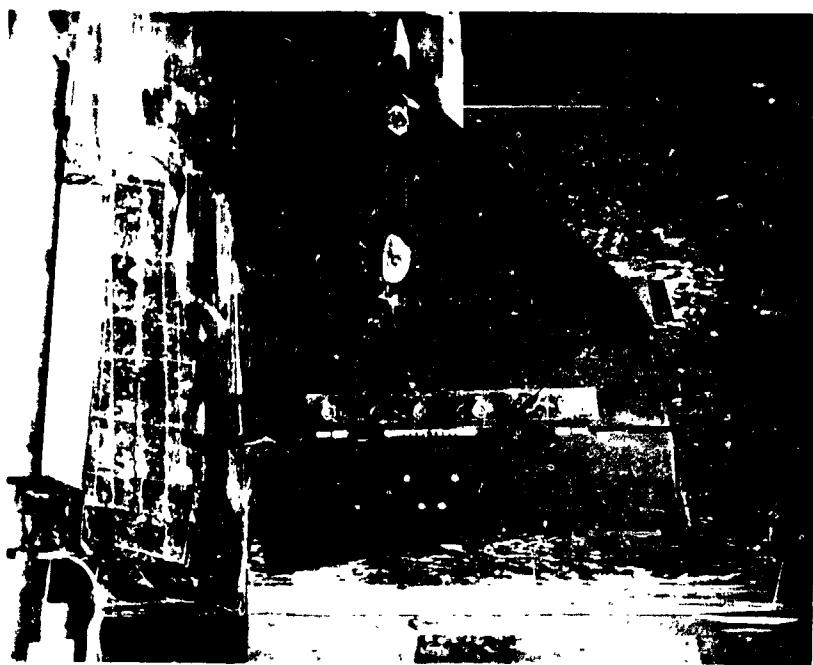


a. Model Installation.



b. Pressure bug installation,  
upper surface of left wing.

Figure 3. - Model Photographs.

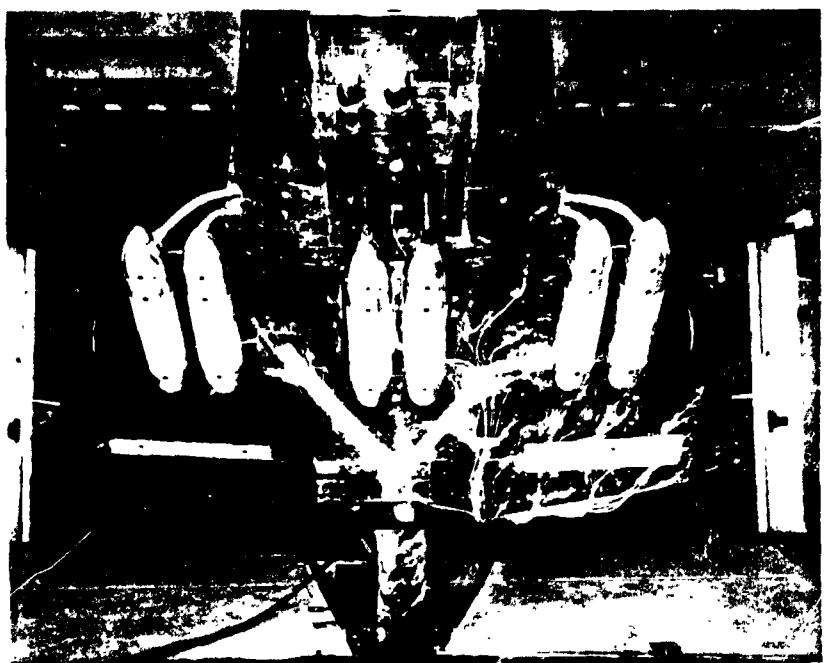


c. Right elevon showing strain gage instrumented beam.

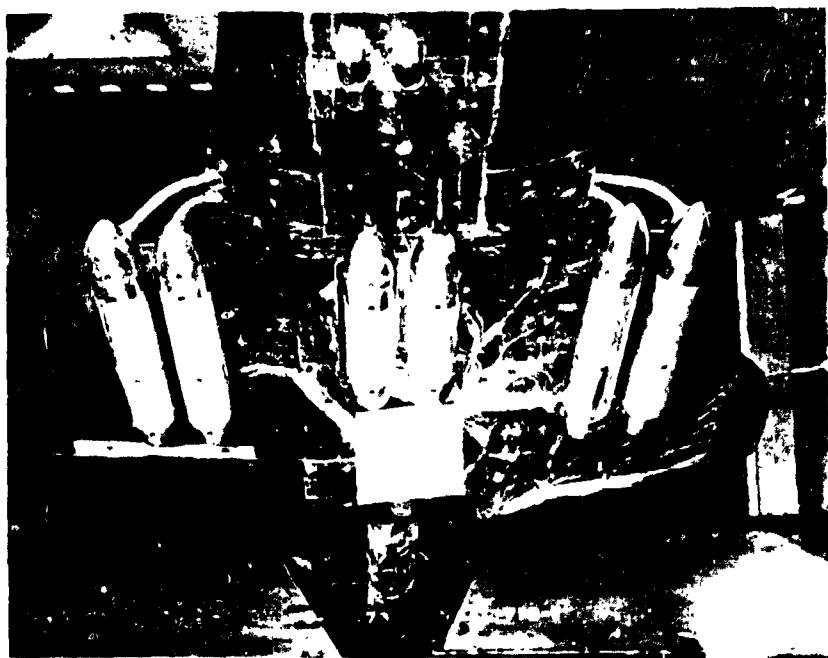


d. Pressure bug location, upper surface of body flap.

Figure 3. - Continued.

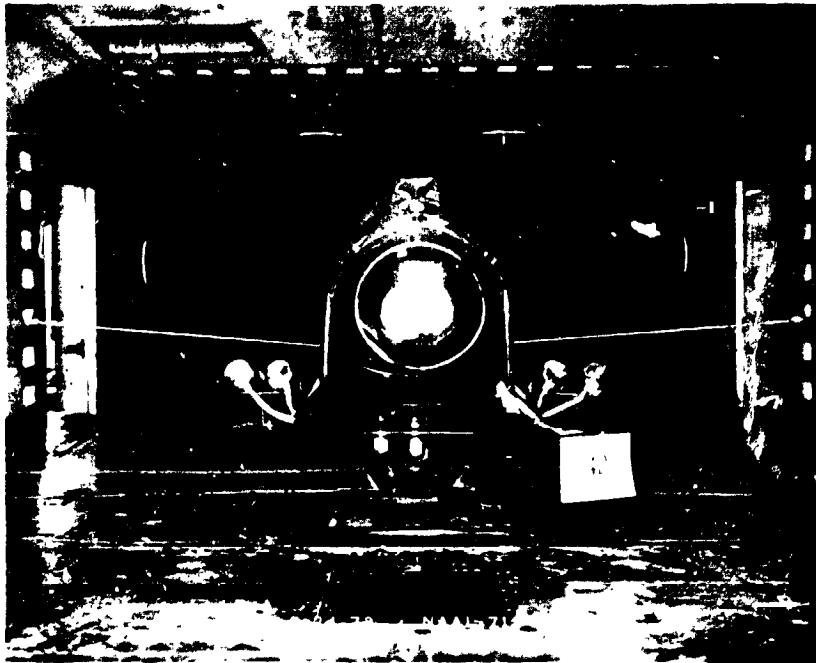


e. J<sub>40</sub> configuration of ABPS.

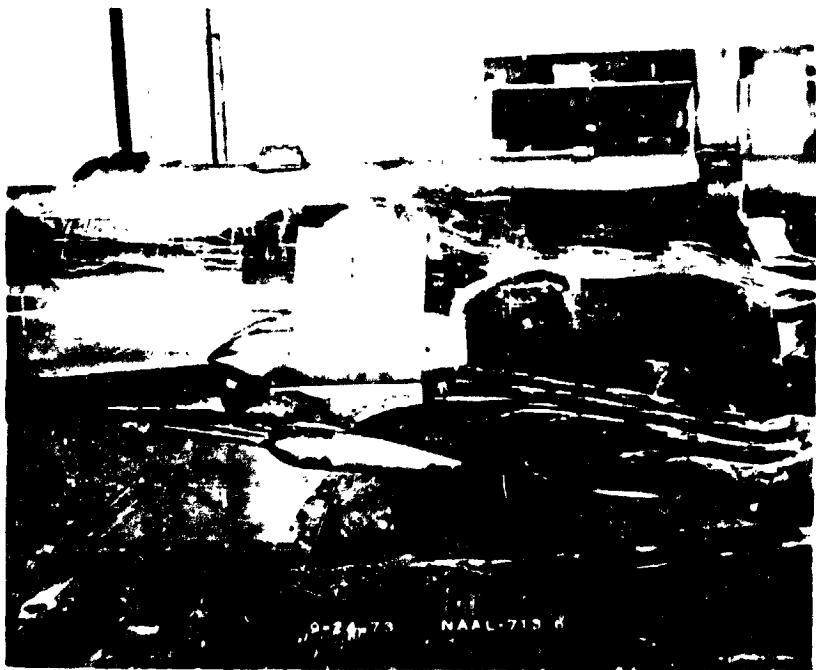


f. J<sub>41</sub> configuration of ABPS.

Figure 3. - Continued.

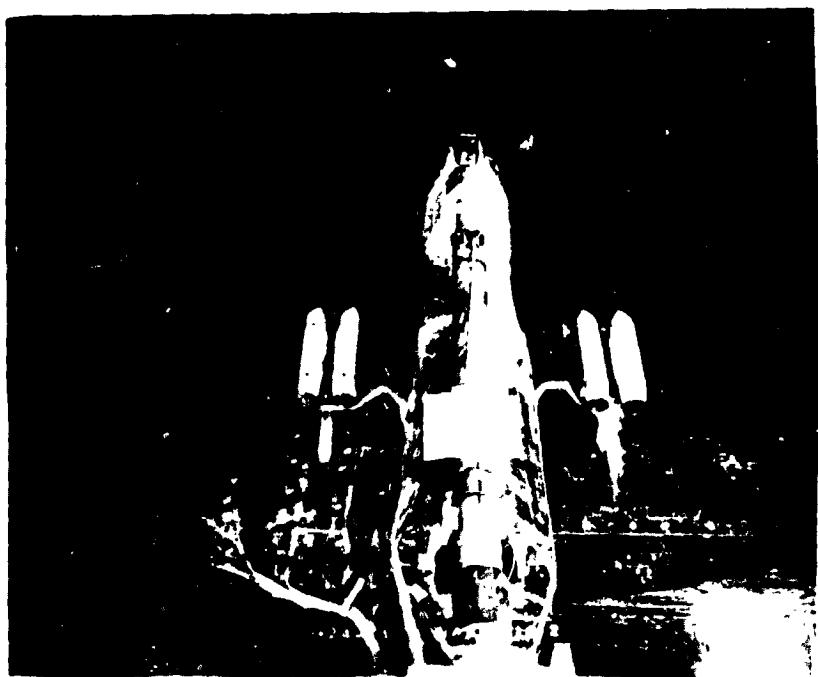


g. J<sub>42</sub> configuration of ABPS.



h. J<sub>42</sub> left wing.

Figure 3. - Continued.



i. J<sub>42</sub> top view.

Figure 3. - Concluded.

## TABULATED SOURCE DATA

Tabulation of plotted data are available on request from Data Management Services.

DATE ON CCT 74

TABULATED SOURCE DATA - C57B

FACE 1

C57B (MAIL 713) 816 CS F1 JAO W87 E18

## REFERENCE DATA

SPEC = 4.4120 Sq.FT. DEP = 43.5980 IN.  
 USEF = 19.2300 IN. DEP = .0000 IN.  
 BDEF = 37.9350 IN. DEP = -.4050 IN.  
 SCALE = .0405

RUN NO. SV U RAVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

## REFERENCE DATA

MACH ALPHA CAB1 CAB2 OME CPBC CPS1 CPS2 CPS4 CPS5  
 .165 9.9000 .00375 .01911 -.01000 .01000 -.18600 -.21900 -.23000 -.23000  
 .165 15.0000 .00501 .01914 -.00100 .01200 -.19000 -.19600 -.20000 -.20000  
 .165 20.0000 .00799 .02478 -.18200 .02700 -.25000 -.25700 -.26000 -.26000  
 .165 GRADIENT .00026 .00021 -.01382 .00216 -.00157 .00451 .00357 -.00200 -.00200

C57B (MAIL 713) 816 CS F1 JAO W87 E18 (DIV) (UA NCV 73)

## REFERENCE DATA

SPEC = 4.4120 Sq.FT. DEP = 43.5980 IN.  
 USEF = 19.2300 IN. DEP = .0000 IN.  
 BDEF = 37.9350 IN. DEP = -.4050 IN.  
 SCALE = .0405

RUN NO. SV U RAVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

## REFERENCE DATA

MACH ALPHA CAB1 CAB2 OME CPBC CPS1 CPS2 CPS4 CPS5  
 .165 9.9000 .00361 .01916 -.01500 .01200 -.18800 -.19100 -.20000 -.20000  
 .165 15.0000 .00392 .01934 -.00700 .01400 -.19200 -.20100 -.20700 -.20700  
 .165 19.0000 .00593 .02092 -.19300 .03000 -.27600 -.25900 -.26300 -.26300  
 .165 GRADIENT .00022 .00024 -.01638 .00235 -.00059 .00000 .00074 .00412 .00074

C57B (MAIL 713) 816 CS F1 JAO W87 E18 (DIV) (UA NCV 73)

## REFERENCE DATA

SPEC = 4.4120 Sq.FT. DEP = 43.5980 IN.  
 USEF = 19.2300 IN. DEP = .0000 IN.  
 BDEF = 37.9350 IN. DEP = -.4050 IN.  
 SCALE = .0405

RUN NO. SV U RAVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

## REFERENCE DATA

MACH ALPHA CAB1 CAB2 OME CPBC CPS1 CPS2 CPS4 CPS5  
 .165 9.9000 .00365 .01849 -.01600 .00800 -.17500 -.19700 -.22500 -.22500  
 .165 14.9000 .00667 .01865 -.10000 .01500 -.18500 -.16900 -.20400 -.20400  
 .165 19.9000 .02553 .02840 -.21100 .03200 -.27100 -.25900 -.28700 -.28700  
 .165 GRADIENT .00028 .00023 -.01740 .00260 -.00060 .00160 .00160 .00074 .00074

(DIV) (UA NCV 73)

## PARAMETRIC DATA

BETA = .000 E08 F = 1.000  
 M/B = .039 E08 F = -18.000  
 ELEVN = .000

(DIV) (UA NCV 73)

BETA = .000 E08 F = 1.000  
 M/B = .039 E08 F = -18.000  
 ELEVN = .000

BETA = .000 E08 F = 1.000  
 M/B = .039 E08 F = -18.000  
 ELEVN = .000

## TABULATED SOURCE DATA - CA57B

CA57B (INAL 713) B16 C5 F1 J40 W47 E18 (IDVA36) (LA NCV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 RREF = 37.9350 IN. ZREF = -.4060 IN.  
 SCALE = .04125

RUN NO. 6/0 RNL = 1.20 GRADIENT INTERVAL = -4.00 16.00  
 MACH ALPHA CAB1 C4E OPE OPBS1 OPBS2 OPBS3 OPBS4 OPBS5  
 .165 -3.900 .00245 .01843 .07100 -.01000 -.17600 -.20700 -.21600 -.22700 -.24000 -.13400  
 .165 .000 .00930 .01724 .02800 -.00800 -.17100 -.20300 -.21300 -.20700 -.22500 -.16700  
 .165 4.900 .00492 .01606 .04100 -.00600 -.15900 -.19100 -.19900 -.19400 -.20200 -.15300  
 .165 10.000 .00510 .01639 .03200 .00200 -.16500 -.18300 -.18200 -.18700 -.18700 -.16800  
 .165 14.900 .00816 .01688 .04800 .00700 -.16700 -.17500 -.17500 -.17600 -.17600 -.14300  
 .165 19.900 .00812 .02444 -.13000 .01900 -.25900 -.25700 -.25700 -.25800 -.25800 -.29500  
 GRADIENT -.00002 -.00038 -.00623 .00189 .00248 .00177 .00289 .00346 .00384 .00384 .00384

CA57B (INAL 713) B16 C5 F1 J40 W47 E18 (IDVA37) (LA NCV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 RREF = 37.9350 IN. ZREF = -.4060 IN.  
 SCALE = .04125

RUN NO. 7/0 RNL = 1.20 GRADIENT INTERVAL = -4.00 16.00  
 MACH ALPHA CAB1 C4E OPE OPBS1 OPBS2 OPBS3 OPBS4 OPBS5  
 .165 -3.900 .00261 .01632 .09700 -.01400 -.18100 -.21100 -.21800 -.22200 -.23200 -.19900  
 .165 .000 .00269 .01755 .08220 -.01200 -.17100 -.21200 -.21200 -.21200 -.21200 -.19400  
 .165 4.900 .00498 .01573 .05900 -.00800 -.16100 -.18100 -.19200 -.19200 -.19200 -.16300  
 .165 9.900 .00447 .01624 .01403 -.00200 -.14400 -.18200 -.19300 -.19300 -.19300 -.17600  
 .165 14.900 .00612 .01639 -.02400 .00400 -.16500 -.17300 -.18400 -.18400 -.18400 -.20400  
 .165 19.900 .00712 .02252 -.11900 .01800 -.23000 -.22600 -.23600 -.23600 -.23600 -.23300  
 GRADIENT -.00004 -.00011 -.00673 .0097 .00121 .00200 .00139 .00139 .00139 .00139 .00139

## PARAMETRIC DATA

BETA = .000 PTF = 1.300  
 H/B = .125 EPLAF = -18.000  
 ELEVN = .000

BETA = .000 PTF = 1.300  
 H/B = .125 EPLAF = -18.000  
 ELEVN = .000



DATE 01 OCT 74

## TABULATED SOURCE DATA - CA57B

FACE 3

CA57B (MAAL 713) B16 CS F1 J40 W87 E18 (INVADB) ( USA NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 8/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	CME	CNE	CPBC	CP81	CP82	CP83	CP84	CP85
.165	-4.000	.00683	.01900	.11400	-.01700	-.18800	-.21900	-.22600	-.23300	-.24800	-.19300
.165	.000	.00632	.01773	.09700	-.01400	-.17200	-.20100	-.21700	-.21300	-.23500	-.17800
.165	4.900	.00460	.00460	.06600	-.01000	-.14800	-.17500	-.19100	-.17600	-.20500	-.13700
.165	9.900	.00380	.01373	.02700	-.00400	-.12500	-.15000	-.16200	-.17400	-.17800	-.14300
.165	14.900	.00516	.01674	.12200	.00300	-.16700	-.19200	-.20300	-.21100	-.21700	-.16600
.165	19.900	.00774	.02348	.11600	.01700	-.25000	-.25100	-.25900	-.29300	-.23800	-.29700
GRAIDENT		-.00006	-.00017	-.00712	.00105	.00185	.00214	.00238	.00171	.00245	.00189

CA57B (MAAL 713) B16 CS F1 J40 W87 E18 (INVADB) ( USA NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 9/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	CME	CNE	CPBC	CP81	CP82	CP83	CP84	CP85
.165	-3.900	.02682	.01621	.11000	-.01600	-.18800	-.23900	-.23300	-.21800	-.23000	-.14600
.165	.000	.02511	.01597	.08600	-.01400	-.16500	-.21400	-.21700	-.19400	-.20500	-.11900
.165	4.900	.00820	.00820	.06300	-.00900	-.26500	-.15900	-.25600	-.21000	-.17800	-.19100
.165	9.900	.00493	.01571	.02300	-.00300	-.15800	-.19300	-.19800	-.18300	-.21200	-.13800
.165	14.900	.02659	.01791	.02600	.00400	-.18000	-.20600	-.22300	-.21600	-.25400	-.15600
.165	19.900	.00814	.02445	-.12200	.01800	-.26300	-.27600	-.30000	-.31100	-.31300	-.23800
GRAIDENT		-.00002	-.00001	-.00729	.00108	.00034	.00177	.00178	.00225	.00126	.00183

## REFERENCE DATA

BETA = .000 FTNU/F = 1.500  
 H/B = .125 EDFLAF = .000  
 ELEVON = .000

## REFERENCE DATA

BETA = .000 FTNU/F = 1.500  
 H/B = .125 EDFLAF = .000  
 ELEVON = .000

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XNRP = 43.5980 IN.  
 LREF = 19.2300 IN. YNRP = .0000 IN.  
 BREF = 37.9350 IN. ZNRP = -.4050 IN.  
 SCALE = .04105

RUN NO. 10V 0 RVAL = 1.20 GRADIENT INTERVAL = -4.0V 16.0U

MACH	ALPHA	CABC	CAB1	CHE	CNE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-4.000	.00592	.01182	.09300	-.01400	-.19100	-.25500	-.23900	-.21300	-.22710J	-.1350J
.165	.000	.00525	.01591	.08000	-.01200	-.16900	-.26000	-.21200	-.18900	-.20000	-.1370J
.165	4.900	.00496	.01530	.06000	-.00700	-.16000	-.26500	-.20500	-.18000	-.19800	-.1210J
.165	9.900	.00568	.01741	.01000	-.00100	-.18300	-.21600	-.22000	-.20000	-.23300	-.1550J
.165	14.900	.00665	.02083	.04000	.00600	-.21500	-.24100	-.25700	-.24000	-.23300	-.1950J
.165	19.900	.00860	.02621	.13100	.02000	-.27800	-.28300	-.30400	-.32300	-.34500	-.2450J
.165	GRADIENT	.00004	.00017	.00709	.00108	-.00139	-.00255	-.00145	-.00146	-.00317	-.00311

CA57B (NAAL 713) B16 C5 F1 J40 L87 E18 (DVA11) (18 Nov 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XNRP = 43.5980 IN.  
 LREF = 19.2300 IN. YNRP = .0000 IN.  
 BREF = 37.9350 IN. ZNRP = -.4050 IN.  
 SCALE = .04105

RUN NO. 11V 0 RVAL = 1.20 GRADIENT INTERVAL = -4.0V 16.0U

MACH	ALPHA	CABC	CAB1	CHE	CNE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-4.000	.00575	.01185	.08000	-.01000	-.18600	-.23100	-.23700	-.21700	-.22900	-.1660J
.165	.000	.00522	.01665	.06600	-.00800	-.17100	-.21400	-.21800	-.19600	-.23800	-.1440J
.165	4.900	.00476	.01522	.03900	-.00600	-.15000	-.20200	-.19500	-.17600	-.19200	-.1260J
.165	9.900	.00538	.01766	.02000	.00000	-.17600	-.21400	-.22100	-.20400	-.23600	-.1650J
.165	14.900	.00642	.01940	.05200	.00800	-.20700	-.23000	-.24300	-.23100	-.25900	-.1790J
.165	19.900	.00846	.02502	.14400	.02200	-.27300	-.27800	-.31100	-.31300	-.32400	-.2470J
.165	GRADIENT	.00003	.00037	.02626	.00093	-.00102	-.00002	-.00039	-.00104	-.00158	-.00117

REFERENCE DATA

BETA = .000 FTN/F = 1.000  
 H/B = .125 EDFL/AF = .000  
 ELEV/N = .000

REFERENCE DATA

BETA = .000 FTN/F = 1.000  
 H/B = .125 EDFL/AF = .000  
 ELEV/N = .000

REFERENCE DATA

BETA = .000 FTN/F = 1.000  
 H/B = .125 EDFL/AF = .000  
 ELEV/N = .000



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## TABULATED SOURCE DATA - CASTE

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CASTB (NAAI 713) B16 C5 F1 J40 W87 E18

(RDVA12) ( USA NAV 73 )

## REFERENCE DATA:

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 12/ 0 RML = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	C+E	OE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4
.165	-4.000	.00506	.01543	.09000	-.01300	-.16300	-.19600	-.20200	-.18900	-.21200
.165	.000	.00463	.01432	.08000	-.01200	-.15000	-.18200	-.19100	-.16600	-.14900
.165	4.900	.00409	.01294	.06100	-.00900	-.13200	-.17300	-.17300	-.14600	-.10700
.165	9.900	.00473	.01551	.02700	-.00400	-.15200	-.19500	-.20300	-.17800	-.13000
.165	15.000	.00549	.01762	.00900	.00100	-.17700	-.21000	-.22300	-.21600	-.15300
.165	19.900	.00775	.02337	.08900	.01300	-.25100	-.27900	-.30900	-.29400	-.20100
	GRADIENT	.00002	.00012	.00528	.00076	-.00072	-.00078	-.00120	-.00149	-.00196

CASTB (NAAI 713) B16 C5 F1 J40 W87 E18

(RDVA13) ( USA NAV 73 )

## REFERENCE DATA:

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 13/ 0 RML = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	C+E	OE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4
.165	-4.000	.00499	.01631	.05500	-.00600	-.16100	-.21200	-.21400	-.19600	-.21200
.165	.000	.00493	.01539	.04700	-.03700	-.15900	-.19900	-.20100	-.19500	-.13700
.165	4.900	.00480	.01495	.03700	-.00600	-.15500	-.19700	-.19800	-.17500	-.11900
.165	9.900	.00525	.01654	.00500	.00000	-.17000	-.21000	-.21600	-.19300	-.14200
.165	14.900	.00628	.01904	.03200	.00800	-.20300	-.22800	-.24500	-.23300	-.16900
.165	20.000	.00816	.02407	.10800	.01600	-.26400	-.28900	-.30700	-.30500	-.21500
	GRADIENT	.00006	.00014	-.00457	.00070	-.00204	-.00095	-.00163	-.00166	-.00244

## PARAMETRIC DATA

BETA = .000  
 H/B = .286  
 ELEV/N = .000

BETA = .000  
 H/B = .286  
 ELEV/N = .000

BETA = .000  
 H/B = .286  
 ELEV/N = .000

BETA = .000  
 H/B = .286  
 ELEV/N = .000

DATE 01 OCT 74

TABULATED SOURCE DATA - CA57B

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(CA57B) (108 REV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 EREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 14/ 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00  
 MACH CABC CAB1 CH-E ONE CPBC CPBS1 CPBS2 CPBS3 CPBS4 CPBS5  
 -4.000 .00630 .01559 .07500 -.01100 -.17100 -.20900 -.18900 -.19400 -.11700  
 -.165 .00463 .01448 .06600 -.01000 -.14900 -.18700 -.19300 -.18600 -.11400  
 .165 4.900 .00485 .01464 .04800 -.00700 -.15700 -.19700 -.19900 -.18600 -.11100  
 .165 9.900 .00482 .01572 .01600 -.00200 -.15700 -.19900 -.20300 -.20000 -.13600  
 .165 14.900 .00575 .01844 -.02100 .00300 -.18600 -.21800 -.24200 -.22300 -.15800  
 .165 20.000 .00805 .02447 -.09700 .01400 -.26000 -.29700 -.31100 -.31100 -.21200  
 GRADIENT .00003 .00015 -.00511 .00076 -.00036 -.00036 -.00180 -.00171 -.02246 -.00223

(CA57B) (108 REV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 EREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 15/ 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00  
 MACH CABC CAB1 CH-E ONE CPBC CPBS1 CPBS2 CPBS3 CPBS4 CPBS5  
 -3.900 .00506 .01605 .05900 -.00900 -.16300 -.21200 -.18800 -.20500 -.12700  
 -.165 .00503 .01595 .04900 -.00700 -.16200 -.20400 -.19000 -.19000 -.13300  
 .165 4.900 .00462 .01497 .03900 -.00600 -.14900 -.19600 -.19400 -.19000 -.12200  
 .165 10.000 .00509 .01672 .00700 -.00100 -.16400 -.21800 -.21800 -.22300 -.14500  
 .165 14.900 .00534 .01955 -.03000 .00400 -.22600 -.22700 -.25000 -.24700 -.16900  
 .165 19.900 .00805 .02443 .010400 .01500 -.26000 -.29100 -.31600 -.31400 -.25200  
 GRADIENT .00006 .00017 -.00466 .00064 -.00186 -.00050 -.00188 -.00283 -.0263 -.0212

(CA57B) (108 REV 73)

## PARAMETRIC DATA

BETA = .000 H-B = .286 EFLAP = .000  
 ELEVON = .000

BETA = .000 H-B = .286 EFLAP = .000  
 ELEVON = .000

(CA57B) (108 REV 73)

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DATE 01 OCT 74

TABULATED SOURCE DATA - CA57B

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(DVA16) (08 NOV 73)

## REFERENCE DATA

SREF = 4.4120 SD.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 16/ 0 RVAL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CABC	CAB1	CME	CPE	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00496	.01697	.010100	-.01500	-.16000	-.18800	-.20700	-.22400	-.17800
.165	.000	.00480	.01542	.08000	-.01300	-.15500	-.17400	-.18600	-.19500	-.16700
.165	4.300	.00451	.01514	.07100	-.01000	-.14600	-.16800	-.17900	-.19800	-.16600
.165	9.900	.00436	.01529	.03800	-.00500	-.14100	-.16600	-.18600	-.19300	-.16500
.165	15.000	.00494	.01644	.02000	-.00100	-.16000	-.17300	-.18700	-.21400	-.18300
.165	19.900	.00575	.02157	.07500	.01100	-.21800	-.21700	-.25900	-.27100	-.25200
	GRADIENT			.00002	-.00528	.00080	.00266	.00076	.00344	.00355

(DVA17) (08 NOV 73)

## REFERENCE DATA

SREF = 4.4120 SD.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 17/ 0 RVAL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CABC	CAB1	CME	CPE	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00223	.01705	.08500	-.01200	-.16500	-.18600	-.20200	-.22000	-.18400
.165	.000	.00477	.01575	.07600	-.01100	-.15400	-.17600	-.19300	-.20300	-.17200
.165	4.300	.00436	.01465	.05800	-.00800	-.14100	-.16100	-.17400	-.18300	-.16200
.165	9.900	.00450	.01501	.02500	-.00300	-.14500	-.15600	-.17100	-.18100	-.17800
.165	14.900	.00512	.01708	.02900	-.00100	-.16500	-.17800	-.19900	-.21700	-.18500
.165	20.000	.00533	.02259	.08100	.01200	-.22700	-.24100	-.28400	-.29300	-.25600
	GRADIENT			-.00001	-.00007	.00072	.00030	.00072	.00331	.00319

(DVA16) (08 NOV 73)

## PARAMETRIC DATA

BETA = .000 FTN/P = 1.500  
 H/B = .286 BDF/LAP = -18.000  
 ELEV/N = .000

## PARAMETRIC DATA

BETA = .000 FTN/P = 1.370  
 H/B = .286 BDF/LAP = -18.000  
 ELEV/N = .000

(DVA17) (08 NOV 73)

## PARAMETRIC DATA

BETA = .000 FTN/P = 1.370  
 H/B = .286 BDF/LAP = -18.000  
 ELEV/N = .000

DATE 01 OCT 74

## TABULATED SOURCE DATA - CA578

PAGE 4

CA578 (MAAL 713) B16 C5 F1 J4U W87 E18

(CDAV18) ( USA NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 18/ 0 RFL/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	ONE	ONE	CP8C	CP8S1	CP8S2	CP8S3	CP8S4	CP8S5
.165	-3.900	.00216	.01753	.06100	-.00900	-.16700	-.19400	-.21100	-.21200	-.23000	-.19500
.165	.000	.00632	.01640	.06300	-.00400	-.17200	-.18900	-.19400	-.20300	-.22100	-.18200
.165	4.900	.00451	.01580	.04300	-.00600	-.14600	-.18000	-.19400	-.18700	-.20300	-.17300
.165	9.900	.00482	.01651	.01100	-.00100	-.15600	-.18900	-.19700	-.23400	-.24900	-.19400
.165	14.900	.00534	.01724	.02300	-.02300	-.17300	-.18900	-.21200	-.22100	-.22100	-.18200
.165	19.900	.00738	.02270	.09400	-.01400	-.23900	-.21000	-.26000	-.26300	-.27900	-.25400
GRADIENT				.00002	-.00446	.00066	.00006	.00031	-.00013	.00045	.00061

(CDAV19) ( USA NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 19/ 0 RFL/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	ONE	ONE	CP8C	CP8S1	CP8S2	CP8S3	CP8S4	CP8S5
.165	-3.900	.00891	.02507	.08100	-.01200	-.28200	-.32100	-.32800	-.34300	-.34300	-.25200
.165	.000	.00930	.02593	.07000	-.01000	-.30100	-.33900	-.32500	-.35600	-.35600	-.25500
.165	4.900	.00866	.02349	.06100	-.00700	-.28000	-.29300	-.28200	-.32300	-.32300	-.21200
.165	9.900	.00936	.02657	.01600	-.00200	-.29300	-.31600	-.34500	-.37400	-.37400	-.24600
.165	14.900	.00940	.02783	.02700	.00400	-.30400	-.31000	-.34200	-.36600	-.36600	-.29100
.165	19.900	.01165	.03319	.11100	.01700	-.37700	-.39600	-.40300	-.44900	-.44900	-.35400
GRADIENT				.00002	-.00573	.00065	-.00051	.00090	-.00133	.00061	-.00136

(CDAV19) ( USA NAV 73 )

## PARAMETRIC DATA

BETA = .000  
 H/B = -.286  
 ELEV/N = .000

BETA = .000  
 H/B = -.286  
 ELEV/N = .000

DATE OF CCT 74

## TABULATED SOURCE DATA - CA57B

PAGE 9

CA57B (NAAL 713) B16 C5 F1 J4U W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 21/0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CME	ONE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00838	.02334	.06900	-.01000	-.27100	-.29800	-.22800	-.33400	-.21600
.165	.000	.00868	.02118	.06000	-.00900	-.28100	-.30300	-.23400	-.33500	-.23700
.165	4.300	.00871	.02441	.06100	-.00600	-.28100	-.28900	-.25700	-.34400	-.24700
.165	9.900	.00879	.02288	.06600	-.00100	-.29400	-.31900	-.24200	-.35000	-.23400
.165	14.900	.00876	.02281	.05600	.00600	-.31500	-.32300	-.33700	-.34500	-.29100
.165	19.900	.01104	.03177	.11800	.01800	-.35700	-.38300	-.39700	-.41500	-.29100
GRADIENT		.00007	.00024	.00560	.00081	-.00214	-.00125	-.00237	-.00251	-.00339

CA57B (NAAL 713) B16 C5 F1 J4U W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 21/0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CME	ONE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.02115	.05300	-.00800	-.24200	-.27100	-.26600	-.22100	-.28900	-.19600
.165	.000	.02074	.04400	-.00600	-.23100	-.28500	-.26300	-.22400	-.27100	-.19700
.165	5.000	.02128	.03600	-.00400	-.25400	-.27900	-.27800	-.23400	-.27300	-.19400
.165	10.000	.02052	.02387	.00000	-.20300	-.23100	-.23100	-.20500	-.32400	-.22100
.165	15.000	.02569	.02572	-.00400	-.05600	-.26100	-.23700	-.28700	-.35300	-.24600
.165	20.000	.01264	.02958	-.12000	.01800	-.30100	-.35900	-.38500	-.38000	-.25500
GRADIENT		.00018	.00026	-.00602	.00072	-.00259	-.00160	-.00343	-.00346	-.00345

(EVA20) ( 3A NCV 73 )

## PARAMETRIC DATA

BETA = .000 FTN/F = 1.300  
 H/B = .296 EDFLAF = 20.000  
 ELEV/N = .000

BETA = .000 FTN/F = 1.300  
 H/B = .296 EDFLAF = 20.000  
 ELEV/N = .000

(EVA21) ( 3A NCV 73 )

## PARAMETRIC DATA

BETA = .000 FTN/F = 1.300  
 H/B = .296 EDFLAF = 20.000  
 ELEV/N = .000

BETA = .000 FTN/F = 1.300  
 H/B = .296 EDFLAF = 20.000  
 ELEV/N = .000

BETA = .000 FTN/F = 1.300  
 H/B = .296 EDFLAF = 20.000  
 ELEV/N = .000

(DIA NCV 73)

(DIA NCV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREFP = 43.5980 IN.  
 UREF = 19.2300 IN. YREFP = .0000 IN.  
 EREF = 37.9350 IN. ZREFP = -.4050 IN.  
 SCALE = .0025

RUN NO. 22/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CAB3	CAB4	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00035	.02569	.00000	-.01400	-.30200	-.32000	-.33200	-.27000	-.35300	-.24000
.165	.00000	.00089	.02929	.00000	-.01200	-.31900	-.36200	-.36400	-.31500	-.40200	-.27400
.165	.00000	.00064	.00069	.00000	-.00700	-.34400	-.37400	-.36800	-.34900	-.41600	-.29500
.165	5.000	.01033	.02973	.00000	.00000	-.33400	-.33400	-.38100	-.33000	-.40200	-.23900
.165	9.300	.01049	.03083	.00000	-.06200	-.33900	-.34600	-.33800	-.39700	-.37400	-.36200
.165	14.900	.01171	.03389	.00000	-.15700	-.37900	-.38900	-.39300	-.44000	-.43700	-.33800
.165	20.000	.00000	.00022	-.00245	.00123	-.00163	-.00240	-.00265	-.00262	-.00274	-.00265
GRADIENT											

(DIA NCV 73) B16 C5 F1 JAU W87 E18 (DIA NCV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREFP = 43.5980 IN.  
 UREF = 19.2300 IN. YREFP = .0000 IN.  
 EREF = 37.9350 IN. ZREFP = -.4050 IN.  
 SCALE = .0025

RUN NO. 23/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CAB3	CAB4	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00039	.02455	.00100	-.01200	-.29000	-.30600	-.31400	-.26800	-.32500	-.23300
.165	.00000	.00092	.02672	.00500	-.01000	-.31700	-.34400	-.33800	-.28100	-.35800	-.25300
.165	4.900	.01012	.02826	.03400	-.00600	-.32700	-.33600	-.35100	-.31100	-.38500	-.23100
.165	9.900	.01061	.02937	.02100	.00100	-.34400	-.36000	-.38300	-.32500	-.39700	-.23500
.165	15.000	.01137	.03122	.07700	.01100	-.33500	-.35100	-.34200	-.35100	-.40100	-.35100
.165	19.900	.01174	.03370	.16600	.02500	-.34000	-.35900	-.39400	-.43100	-.43200	-.33800
GRADIENT											

(DIA NCV 73) B16 C5 F1 JAU W87 E18 (DIA NCV 73)

## REFERENCE DATA

BETA = .000 FTNUF = 1.300  
 H/B = .125 EDLAF = 20.000  
 ELEVON = .000

(DIA NCV 73)

(DIA NCV 73)





DATE 01 OCT 74

TABULATED SOURCE DATA - C4578

FADE 12

## REFERENCE DATA

SREF = 4.4120 Sq.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 267 0 RFL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CABC	CAB1	CAB2	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	9.900	.03777	.02303	-.03203	.02400	-.25100	-.26400	-.24500	-.26200	-.23500
.165	14.900	.00845	.02517	-.13000	.01900	-.27300	-.27100	-.29000	-.34100	-.32500
.165	20.000	.01026	.02993	-.23400	.03500	-.33200	-.32100	-.36000	-.39500	-.34900
	GRADIENT	.00014	.00043	-.01960	.00000	-.00440	-.00140	-.00100	-.01580	-.01450

C4578 (NAAI 713) B16 CS F1 J4U W87 E18 (F0VA26) ( USA NEW 73 )

## REFERENCE DATA

SREF = 4.4120 Sq.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 277 0 RFL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CABC	CAB1	CAB2	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	9.900	.00662	.00007	-.01300	.00100	-.21400	-.23500	-.23300	-.23900	-.25900
.165	14.900	.00696	.02111	-.10200	.01500	-.22500	-.23400	-.25500	-.27700	-.27700
.165	19.900	.00893	.02701	-.19800	.03000	-.29400	-.28200	-.30700	-.34100	-.34100
	GRADIENT	.00007	.00021	-.01700	.00200	-.00220	.00120	-.00200	-.00350	-.00350

C4578 (NAAI 713) B16 CS F1 J4U W87 E18 (F0VA27) ( USA NEW 73 )

## PARAMETRIC DATA

BETA = .000 FINVF = 1.000  
 N/B = .125 EDFLAF = .000  
 ELEV/N = .000

BETA = .000 FINVF = 1.000  
 N/B = .125 EDFLAF = .000  
 ELEV/N = .000

INSULATED SOURCE DATA - CAST8

Case 39 (mai 21-1) 016 CE E1 140 147 E19

(FONDAZIONE) ( UNI-NCR 75 )

REFERENCE DATA

<b>SCALE</b>	=	<b>.0005</b>
<b>SECF</b>	=	<b>4.4120 52.FT.</b>
<b>LEEF</b>	=	<b>19.2350 IN.</b>
<b>BRDF</b>	=	<b>37.9350 IN.</b>
<b>BET</b>	=	<b>.000</b>
<b>H/B</b>	=	<b>.125</b>
<b>ELEVN</b>	=	<b>15.000</b>
<b>PNTP</b>	=	<b>.125</b>
<b>EDFLAP</b>	=	<b>-18.000</b>

RUN NO.	20° D	1.20	GRADIENT INTERVAL =	-4.0V 16.00
CAB1	0.4E	0.4E	CPBC	CPBS1
.02298	-.08700	.01300	-.24300	-.25000
.02293	-.11700	.01700	-.21000	-.21800
.02284	-.14000	.02100	-.17200	-.18900
.01756	-.16200	.02700	-.16000	-.17500
.533	-.18400	.03600	-.18100	-.18500
.01571	-.23600	.04800	-.28000	-.28500
.497	-.25699	.04800	-.28000	-.28500
.561	-.32000	.04800	-.28000	-.28500
.02597	-.32000	.04800	-.28000	-.28500
.965	-.32000	.04800	-.28000	-.28500
.00034	-.00759	.01168	.03357	.03394
111	-.00034	.01168	.03357	.03394

1108 VASUDEVAN

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$$\begin{array}{lcl} \text{PTMF} & = & 1.37 \\ \text{EDLAF} & = & -18.07 \end{array}$$

**43.5980** IN.  
**.0000** IN.  
**-.4060** IN.

4120 FT.  
2300 IN.  
.9350 IN.

	$\text{CF}_{\text{S}}$	$\text{CF}_{\text{E}}$	$\text{CF}_{\text{S+E}}$
1. $\text{CF}_{\text{S}}$	- .27500	- .27500	- .27500
2. $\text{CF}_{\text{E}}$	- .24920	- .24920	- .24920
3. $\text{CF}_{\text{S+E}}$	- .21100	- .21100	- .21100
4. $\text{CF}_{\text{S}}$	- .21100	- .21100	- .21100
5. $\text{CF}_{\text{E}}$	- .22100	- .22100	- .22100
6. $\text{CF}_{\text{S+E}}$	- .20200	- .20200	- .20200
7. $\text{CF}_{\text{S}}$	- .20200	- .20200	- .20200
8. $\text{CF}_{\text{E}}$	- .36300	- .36300	- .36300
9. $\text{CF}_{\text{S+E}}$	- .34300	- .34300	- .34300
10. $\text{CF}_{\text{S}}$	- .34300	- .34300	- .34300
11. $\text{CF}_{\text{E}}$	.42261	.42261	.42261
12. $\text{CF}_{\text{S+E}}$	.40333	.40333	.40333

	CHE	CNE	CPAC	CPBS
11	.07500	.01100	-.24500	-.25600
33316	-.07500	.01400	-.21000	-.21700
33346	-.08600	.01300	-.16300	-.17700
725	-.12600	.01300	-.16300	-.16200
652	-.16900	.02500	-.18200	-.19400
867	-.21800	.03300	-.19900	-.20700
36598	-.30500	.04600	-.29700	-.30700
2	-.07500	.01116	.01228	.01228

CABC	.0077
ALPHA	
-4.900	.0066
.000	.0056
4.900	.0056
9.900	.0056
14.900	.0056
19.900	.0056

DATE 01 OCT 74

TABULATED SOURCE DATA - CA57B

FACE 14

CA57B (NAIL 713) B16 C5 F1 J4U M87 E18 (DIVA31) (13A NEW 73)

## REFERENCE DATA

SQFT = 4.4120 Sq.FT.  
 LREF = 19.2300 IN.  
 RREF = 37.9350 IN.  
 SCALE = .0475

RUN NO. 30 / 0 RFLV = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

NOCH	ALPHA	CAB1	CAB2	CAB3	CAB4	CAB5	CAB6	CAB7	CAB8	CAB9	CAB10	CAB11	CAB12	CAB13	CAB14	CAB15
.165	-3.300	.01662	.01126	-.06700	.01000	-.21400	-.22900	-.24700	-.24600	-.23900	-.23400	-.23400	-.23400	-.23400	-.23400	-.23400
.165	.01000	.01639	.01995	-.05700	.01300	-.19700	-.21300	-.21300	-.21300	-.21300	-.21300	-.21300	-.21300	-.21300	-.21300	-.21300
.165	4.300	.03497	.01808	-.11900	.01800	-.16100	-.17800	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900
.165	9.300	.03485	.01634	-.16200	.02400	-.15700	-.15900	-.17200	-.17200	-.17200	-.17200	-.17200	-.17200	-.17200	-.17200	-.17200
.165	14.300	.03483	.01905	-.20500	.03100	-.23400	-.23100	-.21100	-.21100	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900
.165	19.300	.03482	.02871	-.29200	.04400	-.32200	-.32200	-.31900	-.31900	-.31900	-.31900	-.31900	-.31900	-.31900	-.31900	-.31900
GRADIENT		-.00016	-.00016	-.00042	.00112	.00118	.00193	.00193	.00193	.00193	.00193	.00193	.00193	.00193	.00193	.00193

CA57B (NAIL 713) B16 C5 F1 J4U M87 E18 (DIVA31) (13A NEW 73)

## REFERENCE DATA

SQFT = 4.4120 Sq.FT.  
 LREF = 19.2300 IN.  
 RREF = 37.9350 IN.  
 SCALE = .0475

RUN NO. 31 / 0 RFLV = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

NOCH	ALPHA	CAB1	CAB2	CAB3	CAB4	CAB5	CAB6	CAB7	CAB8	CAB9	CAB10	CAB11	CAB12	CAB13	CAB14	CAB15
.165	10.000	.00537	.01768	-.35300	.08300	-.17400	-.17900	-.19500	-.19500	-.22400	-.22400	-.22400	-.22400	-.22400	-.22400	-.22400
.165	15.000	.00515	.01598	-.40100	.06100	-.19000	-.19000	-.19000	-.19000	-.21700	-.21700	-.21700	-.21700	-.21700	-.21700	-.21700
.165	19.900	.00514	.02646	-.46500	.07000	-.27600	-.26400	-.27900	-.27900	-.33400	-.33400	-.33400	-.33400	-.33400	-.33400	-.33400
GRADIENT		.00016	.00034	-.03960	.00160	-.00500	-.00320	-.00320	-.00320	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600

(DIVA31) (13A NEW 73)

DATE 01 OCT 74

## TABULATED SOURCE DATA - CA57B

PAGE 15

CA57B (MAAL 713) 816 CS F1 JAU 197 E18 (COVAB) (LA NCY 73)

## REFERENCE DATA

*SREF* = 4.4120 S.G.FT. *TMGP* = 43.5980 IN.  
*URF* = 19.2300 IN. *TMCP* = .0000 IN.  
*BREF* = 37.9350 IN. *ZMGP* = -.4050 IN.  
*SCALE* = .00105

RUN NO. 32/ U RVL = 1.20 GRADIENT INTERVAL = -.4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CFE	CPBC	CPBS1	CPBS2	CPBS4	CPBS5
.165	.9.920	.00534	.01684	-.26400	.14000	-.17300	-.18500	-.23900	-.19400
.165	14.900	.00594	.01369	-.31500	.04700	-.19200	-.18700	-.20400	-.23240
.165	19.940	.00392	.02460	-.39600	.40200	-.30100	-.28400	-.35600	-.35240
	GRADIENT	.000112	.00047	-.01020	.00140	-.00340	-.00240	-.00460	-.00480

CA57B (MAAL 713) 816 CS F1 JAU 197 E18 (COVAB) (LA NCY 73)

## REFERENCE DATA

*SREF* = 4.4120 S.G.FT. *TMGP* = 43.5980 IN.  
*URF* = 19.2300 IN. *TMCP* = .0000 IN.  
*BREF* = 37.9350 IN. *ZMGP* = -.4050 IN.  
*SCALE* = .00105

RUN NO. 33/ U RVL = 1.20 GRADIENT INTERVAL = -.4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CFE	CPBC	CPBS1	CPBS2	CPBS4	CPBS5
.165	10.000	.00538	.01613	-.18200	.02700	-.17400	-.18000	-.25300	-.21640
.165	14.900	.00571	.01925	-.24600	.03700	-.18400	-.18000	-.21100	-.23540
.165	21.000	.00398	.02764	-.34900	.05300	-.29000	-.25900	-.32500	-.32440
	GRADIENT	.01	.0023	-.01306	.00204	-.00234	.00200	-.00122	-.00147

## PARAMETRIC DATA

*BETA* = .000  
*H2* = .039  
*ELEVN* = 15.000

*BETA* = .000  
*H2* = .039  
*ELEVN* = 15.000

*BETA* = .000  
*H2* = .039  
*ELEVN* = 15.000

==

DATE 01 CCT 74

TABULATED SOURCE DATA - CA57B

PAGE 16

CA57B (NAAL 713) B16 C5 F1 J40 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRF = 43.5980 IN.  
 LREF = 19.2300 IN. YMRF = .0000 IN.  
 BREF = 37.9350 IN. ZMRF = -.1050 IN.  
 SCALE = .0405

RUN NO. 34 / 0 RFL = 1.20 GRADIENT INTERVAL = -4.00V 16.00

MACH	ALPHA	CABC	CME	CNE	CPBC	CPES1	CPES2	CPES3	CPES4	CPES5
.165	-3.900	.00675	.01958	-.09000	.01400	-.21800	-.20600	-.20400	-.24100	-.23500
.165	.000	.00576	.01835	-.12100	.01200	-.16600	-.19100	-.19000	-.24400	-.22900
.165	4.900	.00472	.01648	-.14100	.02100	-.15200	-.16300	-.19100	-.22100	-.24600
.165	9.900	.00497	.01624	-.17100	.02600	-.16600	-.16500	-.17400	-.22500	-.20000
.165	14.900	.00598	.01786	-.20700	.03100	-.19300	-.18400	-.19100	-.22700	-.22500
.165	19.900	.00833	.02628	-.27800	.04200	-.29200	-.30100	-.29600	-.31600	-.31900
GRADIENT		-.00005	-.00011	-.00077	.00048	.00149	.00162	.00193	.00145	.00110

CA57B (NAAL 713) B16 C5 F1 J40 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRF = 43.5980 IN.  
 LREF = 19.2300 IN. YMRF = .0000 IN.  
 BREF = 37.9350 IN. ZMRF = -.4060 IN.  
 SCALE = .0405

RUN NO. 35 / 0 RFL = 1.20 GRADIENT INTERVAL = -4.00V 16.00

MACH	ALPHA	CABC	CME	CNE	CPBC	CPES1	CPES2	CPES3	CPES4	CPES5
.165	-3.900	.01630	.01926	-.07700	.01100	-.25300	-.20100	-.26000	-.23000	-.22000
.165	.000	.00569	.01813	-.10100	.01900	-.18400	-.18400	-.19700	-.24200	-.21600
.165	4.900	.00500	.01637	-.12100	.01800	-.16100	-.16900	-.19700	-.22400	-.19500
.165	9.900	.00681	.01821	-.15100	.02000	-.18800	-.19200	-.19200	-.24600	-.21800
.165	14.900	.00892	.01859	-.19000	.02800	-.20700	-.19100	-.19700	-.25200	-.21600
.165	19.900	.00922	.02724	-.26300	.04000	-.29800	-.30400	-.30400	-.33600	-.32000
GRADIENT		-.00001	-.00001	-.00082	.00088	.00035	.00032	.00019	.00041	.00030

CA57B (NAAL 713) B16 C5 F1 J40 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRF = 43.5980 IN.  
 LREF = 19.2300 IN. YMRF = .0000 IN.  
 BREF = 37.9350 IN. ZMRF = -.4060 IN.  
 SCALE = .0405

RUN NO. 36 / 0 RFL = 1.20 GRADIENT INTERVAL = -4.00V 16.00

MACH	ALPHA	CABC	CME	CNE	CPBC	CPES1	CPES2	CPES3	CPES4	CPES5
.165	-3.900	.01630	.01926	-.07700	.01100	-.25300	-.20100	-.26000	-.23000	-.22000
.165	.000	.00569	.01813	-.10100	.01900	-.18400	-.18400	-.19700	-.24200	-.21600
.165	4.900	.00500	.01637	-.12100	.01800	-.16100	-.16900	-.19700	-.22400	-.19500
.165	9.900	.00681	.01821	-.15100	.02000	-.18800	-.19200	-.19200	-.24600	-.21800
.165	14.900	.00892	.01859	-.19000	.02800	-.20700	-.19100	-.19700	-.25200	-.21600
.165	19.900	.00922	.02724	-.26300	.04000	-.29800	-.30400	-.30400	-.33600	-.32000
GRADIENT		-.00001	-.00001	-.00082	.00088	.00035	.00032	.00019	.00041	.00030

(REV A34) (08 NOV 73)

## PARAMETRIC DATA

BETA = .000 PTNF = 1.500  
 H/B = .236 EDFLAF = -18.000  
 ELEVON = 15.000

(REV A35) (08 NOV 73)

## PARAMETRIC DATA

BETA = .000 PTNF = 1.370  
 H/B = .236 EDFLAF = -19.000  
 ELEVON = 15.000

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TABULATED SOURCE DATA - CAS7B

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CAS7B (NAAL 713) B16 C5 F1 J40 W87 E18 (DVA36) ( US NCV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 EREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 36/ 0 RVAL = 1.20 GRADIENT INTERVAL = -4.00' 16.00

MACH	ALPHA	CAB1	CAB2	ONE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.01939	.06700	.01000	-.20300	-.20400	-.23000	-.25100	-.25600	-.22100
.165	.000	.00628	.01884	.00600	-.01300	-.18100	-.19100	-.23100	-.23900	-.21300
.165	5.000	.00562	.01758	.01700	-.16300	-.17900	-.20600	-.22000	-.22900	-.20000
.165	9.900	.00560	.01783	.02200	-.14700	-.18100	-.18600	-.20100	-.21600	-.24400
.165	15.000	.00609	.01842	.02700	-.17900	-.19700	-.20000	-.19500	-.20400	-.22900
.165	19.900	.00777	.02852	.02500	-.03400	-.31600	-.32300	-.31700	-.34700	-.35000
GRADIENT	-.00001	-.00004	-.00599	.00399	.00017	.00022	.00210	.00068	.00068	.00018

CAS7B (NAAL 713) B16 C5 F1 J40 W87 E18 (DVA37) ( US NCV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 EREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 37/ 0 RVAL = 1.20 GRADIENT INTERVAL = -4.00' 16.00

MACH	ALPHA	CAB1	CAB2	ONE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00386	.01236	.03500	-.12000	-.13300	-.12900	-.15800	-.15600	-.15000
.165	.000	.00339	.01067	.01900	-.03300	-.10900	-.12400	-.11900	-.13300	-.13100
.165	4.900	.00333	.01050	.01700	-.02700	-.10700	-.12500	-.11900	-.12500	-.13200
.165	9.900	.00361	.01077	.01600	-.02500	-.11300	-.11900	-.13400	-.13600	-.13400
.165	14.900	.00391	.01307	.01600	-.02300	-.12600	-.13500	-.16200	-.16900	-.16400
.165	19.900	.00602	.01948	.01010	-.01500	-.19400	-.20000	-.23000	-.23600	-.23600
GRADIENT	.00001	.00004	-.00432	.00067	-.00022	-.00021	-.00105	-.00101	-.00244	.00018

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TABULATED SOURCE DATA - CA578

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CONFIDENTIAL DATA

CA57B (NAAU 713) B16 C5 F1 J40 W37 E18

(FDVA3H) (LH NEW 73 )

PASSENGER DATA

REFERENCE DATA		PARAMETRIC DATA	
CASTB (MAAL 713)	B16 C5 F1 J40 M47 E18	GHTA	PTN/P
SEEF = 4.4120 SQ.FT.	1MFP = 43.5940 IN.	H/B = .000	1.300
LREF = 19.2300 IN.	1MFP = .0000 IN.	EDFLAF = .236	-18.000
REF = 37.9350 IN.	2MFP = -.4050 IN.	ELEVON = -15.000	
SCALF = .0000			

#### **PASSENGER DATA**

CA57B (NAAU 713) 816 C3 F1 JAU w87 E18 (DOVA39) (US NAV 73 )

REFERENCE DATA

PHARMACEUTICAL DATA

<b>SRCF</b>	<b>=</b>	<b>4,4121</b>	<b>Sq.FT.</b>	<b>YMRP</b>	<b>=</b>	<b>43,5980</b>	<b>IN.</b>	<b>PTR/F</b>	<b>=</b>	<b>1,220</b>
<b>LRCF</b>	<b>=</b>	<b>19,2300</b>	<b>IN.</b>	<b>YMRP</b>	<b>=</b>	<b>.0000</b>	<b>IN.</b>	<b>BDFLAF</b>	<b>=</b>	<b>-18,000</b>
<b>BRCF</b>	<b>=</b>	<b>37,9350</b>	<b>IN.</b>	<b>ZNRP</b>	<b>=</b>	<b>-.4050</b>	<b>IN.</b>	<b>ELEV/N</b>	<b>=</b>	<b>-15,000</b>
								<b>BETA</b>	<b>=</b>	<b>.000</b>
								<b>H/B</b>	<b>=</b>	<b>.286</b>

PHARMACEUTICAL DATA

ACh	ALPHA	CABC	CAB1	CH-E	O-E	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPES
-3.900	.00450	.01432	.21000	.03200	-.14500	-.15700	-.16700	-.17500	-.18500	-.19500	-.1649
.165	.000	.01257	.19200	-.02900	-.12500	-.13300	-.15400	-.15900	-.15400	-.13900	-.1394
.165	.000	.00388	.01200	.16200	-.02000	-.11500	-.12800	-.14700	-.14700	-.15500	-.1394
.165	4.900	.00357	.01200	.16200	-.02200	-.11300	-.13200	-.15300	-.16300	-.16200	-.1414
.165	10.000	.00350	.01277	.14700	-.02000	-.15000	-.16600	-.17700	-.18900	-.18700	-.1644
.165	14.900	.00465	.01501	.13200	-.02000	-.15000	-.16600	-.17700	-.18900	-.18700	-.1644
.165	19.900	.00645	.02060	.06300	-.03900	-.20800	-.22300	-.24900	-.26400	-.25600	-.2210
.165	GRADIENT	.00000	.00003	.00000	.000421	.00065	-.00003	-.00044	-.00045	-.00074	.00031

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TABULATED SOURCE DATA - CA578

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CA578 (NAAI 713) B16 C5 F1 J40 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 43/ 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	CHE	CPE	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00714	.02093	.32000	-.104800	-.23100	-.20000	-.24500	-.24600	-.26100
.165	.0000	.00676	.01712	.26800	-.04000	-.18600	-.20400	-.19000	-.20600	-.20900
.165	4.900	.00483	.01427	.21200	-.03200	-.15600	-.15300	-.15300	-.18600	-.17900
.165	9.900	.00438	.01533	.19000	-.02800	-.14200	-.15300	-.17500	-.19200	-.16900
.165	14.900	.00531	.01679	.16900	-.02500	-.17100	-.16900	-.18400	-.21000	-.18900
.165	20.000	.00665	.02186	.09900	-.01500	-.21500	-.2, 400	-.26500	-.23400	-.22700
GRADIENT		-.000010	-.00021	-.00792	.00121	.00334	.00359	.00256	.00170	.00217

CA578 (NAAI 713) B16 C5 F1 J40 W87 E18 (RDVA43) ( 08 NCV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 44/ 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CABC	CAB1	CHE	CPE	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-4.000	.00669	.01938	.27100	-.04100	-.21600	-.21900	-.22000	-.23500	-.22500
.165	.0000	.00555	.01637	.23400	-.03600	-.17900	-.17900	-.18400	-.20600	-.19300
.165	4.900	.00467	.01518	.18500	-.02800	-.15100	-.16500	-.18600	-.18600	-.18400
.165	9.900	.00493	.01656	.16900	-.02500	-.15800	-.17600	-.19700	-.20400	-.21100
.165	14.900	.00540	.01745	.14700	-.02200	-.17400	-.19100	-.20500	-.22300	-.18900
.165	19.900	.00752	.02307	.07800	-.01100	-.24900	-.25600	-.27000	-.28900	-.24500
GRADIENT		-.00006	-.00007	-.00661	.00102	.00211	.00114	.00027	.00046	.00155

REFERENCE DATA

BETA = .0000 FT/P = 1.300  
 H/B = .125 EDFLAP = -18.000  
 ELEVON = -15.000

REFERENCE DATA

BETA = .0000 FT/P = 1.300  
 H/B = .125 EDFLAP = -18.000  
 ELEVON = -15.000





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TABULATED SOURCE DATA - CA578

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## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4LSU IN.  
 SCALE = .0405

RUN NO. 47/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CAB3	CAB4	CAB5	CAB6	CAB7	CAB8	CAB9	CAB10	CAB11	CAB12	CAB13	CAB14	CAB15	CAB16	CAB17	CAB18
.165	-4.000	.00675	.02096	-.06900	.01000	-.21400	-.22100	-.25000	-.26500	-.25800	-.24300	-.24300	-.22600	-.23300	-.23300	-.23300	-.23300	-.23300	-.23300
.165	.000	.00251	.01482	-.08000	.01300	-.18100	-.21000	-.23000	-.23000	-.23000	-.23000	-.23000	-.23000	-.23000	-.23000	-.23000	-.23000	-.23000	-.23000
.165	4.900	.00494	.01714	-.12700	.01900	-.16000	-.17900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900	-.19900
.165	9.900	.00498	.01689	-.11700	.02600	-.16100	-.18300	-.19400	-.19400	-.19400	-.19400	-.19400	-.19400	-.19400	-.19400	-.19400	-.19400	-.19400	-.19400
.165	14.900	.00498	.02000	-.22100	.03300	-.21200	-.21200	-.26300	-.26300	-.26300	-.26300	-.26300	-.26300	-.26300	-.26300	-.26300	-.26300	-.26300	-.26300
.165	19.900	.00492	.02745	-.29400	.04500	-.29400	-.31000	-.30400	-.30400	-.30400	-.30400	-.30400	-.30400	-.30400	-.30400	-.30400	-.30400	-.30400	-.30400
GRADIENT	-	.00003	-.00007	-.00015	.00124	-.00028	.00028	.00028	.00028	.00028	.00028	.00028	.00028	.00028	.00028	.00028	.00028	.00028	.00028

CA578 (NAAL 713) B16 C5 F1 J41 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 48/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CAB3	CAB4	CAB5	CAB6	CAB7	CAB8	CAB9	CAB10	CAB11	CAB12	CAB13	CAB14	CAB15	CAB16	CAB17	CAB18
.165	.000	.00490	.01654	-.26400	.04000	-.15200	-.15700	-.18800	-.18800	-.18800	-.21000	-.22300	-.21000	-.21000	-.21000	-.21000	-.21000	-.21000	-.21000
.165	4.900	.00602	.01945	-.32600	.04900	-.19400	-.23600	-.23700	-.23700	-.23700	-.25500	-.25200	-.25200	-.25200	-.25200	-.25200	-.25200	-.25200	-.25200
.165	9.900	.00936	.02825	-.40700	.05200	-.30200	-.28100	-.31100	-.31100	-.31100	-.35700	-.35700	-.35700	-.35700	-.35700	-.35700	-.35700	-.35700	-.35700
GRADIENT	-	.00045	.00118	-.01445	.00222	-.01457	.01253	-.01253	-.01253	-.01253	-.01523	-.01523	-.01523	-.01523	-.01523	-.01523	-.01523	-.01523	-.01523

(NAAL 713) B16 C5 F1 J41 W87 E18

## PARAMETRIC DATA

BETA = .000 PTNF = 1.300  
 H/B = .039 EDLAF = -18.000  
 ELEVN = 15.000

(NAAL 713) B16 C5 F1 J41 W87 E18

## PARAMETRIC DATA

BETA = .000 PTNF = 1.300  
 H/B = .039 EDLAF = -18.000  
 ELEVN = 15.000

REFERENCE DATA							(DVA49) ( DA NCV 73 )						
SREF	4.4120 SQ.FT.	X4GP	=	43.5980 IN.			B16	C5	F1	J41	W47	E16	PARAMETRIC DATA
UREF	19.2300 IN.	Y4GP	=	.0000 IN.									BETA = .000 FTN/P = 1.000
BREF	37.9350 IN.	Z4GP	=	-.4050 IN.									H/B = .0339 EDF LAF = -18.000
SCALE	.0405												ELEV/N = 15.000
REFERENCE DATA							(DVA50) ( DA NCV 73 )						
MACH	ALPHA	CABC	CAB1	O4E	O4E	O4E	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5	PARAMETRIC DATA
.165	.0000	.00619	.01748	-.18000	.12700	-.16000	-.17200	-.19000	-.20600	-.21700	-.23500	-.25300	BETA = .000 FTN/P = 1.000
.165	4.9000	.00613	.01569	-.24600	.03700	-.19500	-.20100	-.23300	-.25600	-.24600	-.22600	-.34600	H/B = .0339 EDF LAF = -18.000
.165	9.9000	.00632	.02837	-.33500	.09100	-.26900	-.25900	-.27800	-.34000	-.33500	-.33400	-.33400	ELEV/N = 15.000
GRADIENT	.00032	.00090	.01566	.01243	.01122	.00280	.00122	.00798	.01193	.01193	.01193	.01193	(DVA50) ( DA NCV 73 )
REFERENCE DATA							(DVA50) ( DA NCV 73 )						
MACH	ALPH4	CABC	CAB1	O4E	O4E	O4E	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5	PARAMETRIC DATA
.165	-4.000	.00634	.02126	-.07800	.01100	-.21100	-.22200	-.21700	-.21000	-.21000	-.21000	-.24400	BETA = .000 FTN/P = 1.000
.165	.0000	.00636	.01768	-.10300	.01500	-.17300	-.17100	-.17100	-.20600	-.20600	-.21700	-.21700	H/B = .0339 EDF LAF = -18.000
.165	4.9000	.00514	.01698	-.13300	.02000	-.16600	-.16600	-.17300	-.19400	-.19400	-.21400	-.21400	ELEV/N = 15.000
.165	9.9000	.00542	.01710	-.16800	.02500	-.17500	-.18200	-.17400	-.22400	-.22400	-.23900	-.23900	(DVA50) ( DA NCV 73 )
GRADIENT	.00613	.01912	.02070	.03100	.03100	.01900	.02000	.02000	.02700	.02700	.03200	.03200	ELEV/N = 15.000
.165	14.9000	.03932	.02743	-.22900	.04200	-.30100	-.30900	-.30900	.03200	.03200	.33400	.33400	(DVA50) ( DA NCV 73 )
GRADIENT	-.00031	-.00009	-.00673	.00105	.00040	.00041	.00226	.00110	.00110	.00110	.00126	.00126	(DVA50) ( DA NCV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 PEF = 37.9350 IN. ZREF = -.4060 IN.  
 SCALE = .0425

RUN NO. 51/0 RVAL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CABC	CAB1	CAB2	CAB3	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-4.000	.00610	.01934	-.07700	.01100	-.19700	-.20600	-.23000	-.26100	-.24500	-.22700
.165	.000	.00631	.01775	-.09600	.01400	-.17200	-.17900	-.20100	-.24000	-.22700	-.21400
.165	4.900	.00479	.01728	-.12900	.01900	-.15500	-.18100	-.21300	-.24900	-.26200	-.19000
.165	9.900	.00529	.01765	-.16500	.02500	-.17100	-.18000	-.19000	-.24000	-.21700	-.21100
.165	14.900	.00530	.01967	-.19800	.03000	-.20300	-.21500	-.21000	-.26700	-.23700	-.22900
.165	19.900	.00336	.02770	-.27100	.04100	-.30300	-.31400	-.30400	-.34200	-.33000	-.33800
GRADIENT	.00001	-.00000	-.00654	.00103	-.00032	-.00032	-.00032	-.00102	-.00034	-.00032	-.00030

CA57B (NVAL 713) B16 C5 F1 J41 W87 E18 (DVA511) (L8 NCV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 PEF = 37.9350 IN. ZREF = -.4060 IN.  
 SCALE = .0425

RUN NO. 52/0 RVAL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CABC	CAB1	CAB2	CAB3	CPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-4.000	.00619	.01720	.07100	-.01000	-.16400	-.18700	-.19400	-.21900	-.21700	-.19600
.165	.000	.00444	.01538	.56200	-.03900	-.14300	-.16600	-.18100	-.23400	-.20000	-.14300
.165	4.900	.00402	.01534	.04800	-.00700	-.13000	-.14500	-.16000	-.20100	-.19100	-.16900
.165	9.900	.00450	.01608	.01500	-.02000	-.14500	-.16000	-.18400	-.21900	-.20200	-.17900
.165	14.900	.00652	.01719	-.02100	.00300	-.17800	-.18300	-.19500	-.22500	-.21500	-.19600
.165	19.900	.00734	.02210	-.03700	.01300	-.23700	-.24200	-.23900	-.28400	-.27300	-.26400
GRADIENT	.00002	.00001	-.00488	.00070	-.00071	.00035	-.00035	-.0023	-.00061	-.00056	.00003

(DVA521) (L8 NCV 73)

## PARAMETRIC DATA

BETA = .000 PTW/P = 1.000  
 H/B = .286 EDPLAF = -18.000  
 ELEV/N = 15.000

(DVA521) (L8 NCV 73)

## PARAMETRIC DATA

BETA = .000 PTW/F = 1.333  
 H/B = .286 EDPLAF = -18.000  
 ELEV/N = .000

(DVA521) (L8 NCV 73)

## PARAMETRIC DATA

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## TABULATED SOURCE DATA - CA578

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CA578 (INAL 713) B16 C5 F1 J41 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 53V 0 RFL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CAB1	CAB2	O/E	OPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	-3.900	.00625	.01777	.05900	-.00900	-.17000	-.19800	-.22000	-.22100	-.19700
.165	.000	.00463	.01676	.04900	-.00700	-.14900	-.18100	-.19600	-.21300	-.20900
.165	4.900	.00431	.01577	.03900	-.00600	-.13900	-.16100	-.18900	-.20500	-.18700
.165	9.900	.00479	.01673	.03800	-.00100	-.15500	-.18000	-.19700	-.21200	-.19800
.165	14.900	.00532	.01768	.02900	.00400	-.17200	-.18800	-.20400	-.23300	-.18800
.165	19.900	.00778	.02304	.03400	.01400	-.25200	-.24900	-.25900	-.28200	-.19700
GRADIENT	.00001	-.00000	-.00461	.00084	-.00029	.00038	.00015	-.00049	.00018	-.00006

CA578 (INAL 713) B16 C5 F1 J41 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 54V 0 RFL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	CAB1	CAB2	O/E	OPBC	CPBS1	CPBS2	CPBS3	CPBS4	CPBS5
.165	9.900	.00623	.01768	.05100	.00100	-.16900	-.18000	-.20500	-.23500	-.21900
.165	15.000	.00564	.01690	.03600	.01400	-.16200	-.19300	-.22300	-.25600	-.23400
.165	19.900	.00840	.02556	.02100	.03000	-.27200	-.24700	-.28600	-.33700	-.28600
GRADIENT	.00008	.00024	-.01867	.00255	-.00265	-.00255	-.00353	-.00392	-.00294	-.00118

(RDVA53) (OR NCV 73)

(RDVA54) (OR NCV 73)

(RDVA55) (OR NCV 73)

DATE 01 OCT 74

TABULATED SOURCE DATA - CA578

FACE 26

(08 NOV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 55/0 RVL = 1.20 GRADIENT INTERVAL = -4.00/16.00

MACH	ALPHA	CAB1	CAB2	ONE	OPBC	OPBS1	OPBS2	OPBS3	OPBS4	OPBS5
.165	9.9000	.00543	.01774	-.01200	.00100	-.17500	-.17700	-.23700	-.23700	-.23700
.165	14.9000	.00559	.01893	-.09400	.01400	-.19000	-.18500	-.22100	-.24600	-.24600
.165	19.9000	.00576	.02831	-.19800	.03000	-.28300	-.25600	-.24600	-.33100	-.31700
GRADIENT	.0405	.00013	.0024	-.01641	.00260	-.00100	-.00160	-.00381	-.00440	-.00440

(08 NOV 73) B16 C5 F1 J41 W87 E18 (08 NOV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 56/0 RVL = 1.20 GRADIENT INTERVAL = -4.00/16.00

MACH	ALPHA	CAB1	CAB2	ONE	OPBC	OPBS1	OPBS2	OPBS3	OPBS4	OPBS5
.165	10.0000	.00514	.01725	-.00000	.00100	-.16600	-.17900	-.19700	-.21500	-.19900
.165	15.0000	.00535	.01874	-.07900	.01200	-.19200	-.18500	-.21800	-.23500	-.21300
.165	20.0000	.00578	.02484	-.18100	.02700	-.28300	-.24000	-.26300	-.31400	-.34600
GRADIENT	.0405	.00016	.00300	-.01420	.02221	-.00521	-.00121	-.03421	-.03521	-.03521

(08 NOV 73)

## PARAMETRIC DATA

BETA = .000  
 H/B = .039  
 ELEV/N = .000

BETA = .000  
 H/B = .039  
 ELEV/N = .000

BETA = .000  
 H/B = .039  
 ELEV/N = .000

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = .9.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 57 / 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/16.00

MACH	ALPHA	CAB1	CAB2	OFE	OFC	OPBS1	OPBS2	OPBS3	OPBS4	OPBS5
.165	-3.900	.02155	.02150	-.01300	-.22000	-.23500	-.24100	-.27300	-.26700	-.25300
.165	.000	.00580	.01896	.07400	-.01100	-.16800	-.20800	-.24500	-.23400	-.22100
.165	4.300	.00521	.01720	.05300	-.01200	-.16800	-.17500	-.20800	-.22200	-.21300
.165	9.900	.01675	.01675	.01500	.00200	-.15400	-.17900	-.19500	-.22200	-.23400
.165	15.000	.01645	.01645	.02600	.02500	-.16100	-.17100	-.18600	-.22100	-.21300
.165	19.900	.02766	.02763	-.11100	.01700	-.24700	-.24500	-.25600	-.29500	-.29000
GRADIENT	-	-.00010	-.0024	-.02633	.00395	.00312	.00329	.00255	.00255	.00255

C57B (INAL 713) B16 CS F1 J41 W87 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = .9.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 58 / 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/16.00

MACH	ALPHA	CAB1	CAB2	OFE	OFC	OPBS1	OPBS2	OPBS3	OPBS4	OPBS5
.165	-3.900	.02047	.02020	-.01100	-.20200	-.23300	-.24300	-.25300	-.25300	-.22300
.165	.000	.00570	.01911	.05700	-.00600	-.18400	-.21100	-.22200	-.24200	-.23800
.165	4.300	.00482	.01723	.04100	-.00600	-.15600	-.18800	-.21600	-.21300	-.19400
.165	10.000	.00605	.01687	.03200	.00000	-.16300	-.17600	-.19800	-.22000	-.21000
.165	15.000	.00612	.01752	-.04500	.03600	-.16500	-.16600	-.20100	-.23900	-.22200
.165	19.900	.00797	.02357	-.12300	.01800	-.25800	-.25400	-.26100	-.28400	-.23300
GRADIENT	-	-.00016	-.00017	-.00009	.00008	.00193	.00350	.00103	.00192	.00121

C57B (INAL 713) B16 CS F1 J41 W87 E18

## REFERENCE DATA

BETA = .000 FTNU/F = 1.000  
 H/B = .125 EDFLAP = -14.125  
 ELEVN = .000

(FDV57) (C57 NEW 73)

(FDV57) (C57 NEW 73)

(FDV57) (C57 NEW 73)



DATE 01 OCT 74

## TABULATED SOURCE DATA - CA578

FACE 2

CA578 (INAL 713) 816 CS F1 J42 W47 E18

(DVAR1) (US NAV 73)

## REFERENCE DATA

SQEF = 4.4120 SQ.FT. XREF = 43.5000 IN.  
 LREF = 19.2300 IN. TMRF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0005

RUN NO. 6170 RAVL = 1.20 GRADIENT INTERVAL = -4.00/16.00

## PARAMETRIC DATA

BETA = .0174  
 H2 = .125  
 ELEVON = .000

CAB1 ONE OFC CPES1 CPES2 CPES3 CPES4  
 .01675 .02400 .01200 -.16300 -.11700 -.19200 -.21100 (.P-SS)  
 .00604 .02473 .01595 .07000 -.01300 -.15300 -.17500 -.19300 (.P-SS)  
 .00000 .00473 .01595 .07000 -.01300 -.15300 -.17500 -.19300 (.P-SS)  
 .00000 .00452 .01532 .03800 -.00600 -.13200 -.15300 -.17600 (.P-SS)  
 .00000 .00453 .01549 .03400 .01200 -.14600 -.15300 -.17600 (.P-SS)  
 .00000 .00453 .01549 .03400 .01200 -.14600 -.15300 -.17600 (.P-SS)  
 .00000 .00452 .01624 .03600 .02800 -.16500 -.17900 -.19300 (.P-SS)  
 .00000 .00452 .01624 .03600 .02800 -.16500 -.17900 -.19300 (.P-SS)  
 .00000 .00452 .02146 .01600 .01600 -.22600 -.25600 -.28100 (.P-SS)  
 .00000 .00452 .02146 .01600 .01600 -.22600 -.25600 -.28100 (.P-SS)  
 .00000 .00452 .02734 .00000 .00000 .00000 .00000 .00000 (.P-SS)

CA578 (INAL 713) 816 CS F1 J42 W47 E18

(DVAR2) (US NAV 73)

## REFERENCE DATA

SQEF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. TMRF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4600 IN.  
 SCALE = .0005

RUN NO. 6270 RAVL = 1.20 GRADIENT INTERVAL = -4.00/16.00

## PARAMETRIC DATA

BETA = .0174  
 H2 = .125  
 ELEVON = .000

CAB1 ONE OFC CPES1 CPES2 CPES3 CPES4  
 .01436 .01900 .00000 -.15900 -.17000 -.16700 -.16900 (.P-SS)  
 .00691 .03378 .01600 -.10500 -.34000 -.15500 -.15100 -.15700 (.P-SS)  
 .00000 .02917 .01600 -.10500 -.34000 -.15500 -.15100 -.15700 (.P-SS)  
 .00000 .03288 .016900 .02800 -.09900 -.08400 -.12600 -.14700 (.P-SS)  
 .00000 .03378 .01720 .02800 -.09900 -.08400 -.12600 -.14700 (.P-SS)  
 .00000 .03378 .01720 .02800 -.09900 -.08400 -.12600 -.14700 (.P-SS)

DATE 01 OCT 74

## TABULATED SOURCE DATA - CA57B

FACE 30

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 UREF = 19.2330 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .045E

## REFERENCE DATA

	RUN NO.	63/ 0	RVL =	1.20	GRADIENT INTERVAL = -4 U/ 16.00			
MACH	ALPHA	CAB1	ONE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4
.165	9.900	.00466	.01496	-.11700	.00200	-.15000	-.16800	-.17400
.165	14.900	.00438	.01387	-.10100	.01500	-.14100	-.15100	-.15500
.165	19.900	.00607	.01612	-.18700	.02800	-.16400	-.16400	-.17500
	GRADIENT			-.00022	-.01640	.00260	.00180	-.19900
						.00340	.00120	-.19400

CA57B (NAAL 713) B16 C5 F1 J42 W47 E18

(FDVA63) ( 3A NEW 73 )

## REFERENCE DATA

	RUN NO.	64/ 0	RVL =	1.20	GRADIENT INTERVAL = -4.U/ 16.00			
MACH	ALPHA	CAB1	ONE	CPBC	CPBS1	CPBS2	CPBS3	CPBS4
.165	9.900	.00473	.01519	-.02100	.00300	-.15400	-.16700	-.17900
.165	15.000	.00513	.01611	-.09700	.01400	-.16600	-.17700	-.18800
.165	19.930	.00790	.02504	-.17700	.02700	-.25500	-.26700	-.28800
	GRADIENT			.00007	.00018	.00216	.00235	-.29700
						-.00216	-.00137	-.30700

(FDVA64) ( 3A NEW 73 )

PARAMETRIC DATA

	BETA = .000	FTN/F = 1.300
SREF	H/B = .339	DOFLAF = -1.4 (.4.)
UREF	ELEVON = .000	

	BETA = .000	FTN/F = 1.300
SREF	H/B = .339	DOFLAF = -1.4 (.4.)
UREF	ELEVON = .000	

	BETA = .000	FTN/F = 1.300
SREF	H/B = .339	DOFLAF = -1.4 (.4.)
UREF	ELEVON = .000	

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 65/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CAB3	CAB4	CAB5	CAB6	CAB7	CAB8	CAB9	CAB10	CAB11	CAB12	CAB13	CAB14	CAB15	CAB16	CAB17	CAB18
.165	-3.900	.02650	.01605	.01300	-.01100	-.01700	-.16400	-.17000	-.16700	-.15300	-.12900	-.13600	-.13700	-.13500	-.13200	-.12900	-.12800	-.12700	
.165	.000	.00475	.01571	.07500	-.01100	-.01100	-.15300	-.15300	-.15300	-.15300	-.12900	-.13600	-.13700	-.13500	-.13200	-.12900	-.12800	-.12700	
.165	4.900	.00356	.00359	.01433	.04300	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	
.165	9.900	.00346	.00346	.01220	.01220	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	
.165	14.900	.00346	.00346	.01270	.01270	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	-.02300	
.165	19.900	.00346	.00346	.01268	.01268	-.12900	-.09000	-.09000	-.09000	-.09000	-.11600	-.11600	-.11600	-.11600	-.11600	-.11600	-.11600	-.11600	
	GRADENT			-.00010	-.00010	-.00733	-.00733	-.00733	-.00733	-.00733	-.00112	-.00326	-.00326	-.00326	-.00326	-.00326	-.00326	-.00326	

CA57B (NANL 713) B16 C5 F1 J42 W37 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 66/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CAB3	CAB4	CAB5	CAB6	CAB7	CAB8	CAB9	CAB10	CAB11	CAB12	CAB13	CAB14	CAB15	CAB16	CAB17	CAB18
.165	1.000	.00506	.01630	.08800	-.01300	-.01300	-.16300	-.17200	-.17200	-.15300	-.14500	-.14500	-.14500	-.14500	-.14500	-.14500	-.14500	-.14500	
.165	.000	.00449	.01555	.07400	-.01100	-.01100	-.15100	-.15100	-.15100	-.15100	-.12400	-.12400	-.12400	-.12400	-.12400	-.12400	-.12400	-.12400	
.165	4.900	.00384	.00449	.01426	.04400	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	-.00600	
.165	9.900	.00385	.00449	.01327	.03500	-.00300	-.00300	-.00300	-.00300	-.00300	-.12400	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900
.165	14.900	.00413	.00413	.01367	.05000	-.00700	-.00700	-.00700	-.00700	-.00700	-.13900	-.15200	-.15200	-.15200	-.15200	-.15200	-.15200	-.15200	-.15200
.165	19.900	.00477	.00477	.01598	.012300	-.01800	-.01800	-.01800	-.01800	-.01800	-.15400	-.16600	-.16600	-.16600	-.16600	-.16600	-.16600	-.16600	-.16600
	GRADENT			-.00005	-.00016	-.00728	-.00108	-.00108	-.00108	-.00108	-.00165	.00164	.00164	.00164	.00164	.00164	.00164	.00164	.00164

CA57B (NANL 713) B16 C5 F1 J42 W37 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 67/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	CAB1	CAB2	CAB3	CAB4	CAB5	CAB6	CAB7	CAB8	CAB9	CAB10	CAB11	CAB12	CAB13	CAB14	CAB15	CAB16	CAB17	CAB18
.165	1.000	.00506	.01630	.08800	-.01300	-.01300	-.16300	-.17200	-.17200	-.15300	-.14500	-.14500	-.14500	-.14500	-.14500	-.14500	-.14500	-.14500	
.165	.000	.00449	.01555	.07400	-.01100	-.01100	-.15100	-.15100	-.15100	-.15100	-.12400	-.12400	-.12400	-.12400	-.12400	-.12400	-.12400	-.12400	
.165	4.900	.00384	.00449	.01426	.04400	-.00600	-.00600	-.00600	-.00600	-.00600	-.12400	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900
.165	9.900	.00385	.00449	.01327	.03500	-.00300	-.00300	-.00300	-.00300	-.00300	-.12400	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900	-.13900
.165	14.900	.00413	.00413	.01367	.05000	-.00700	-.00700	-.00700	-.00700	-.00700	-.13900	-.15200	-.15200	-.15200	-.15200	-.15200	-.15200	-.15200	-.15200
.165	19.900	.00477	.00477	.01598	.012300	-.01800	-.01800	-.01800	-.01800	-.01800	-.15400	-.16600	-.16600	-.16600	-.16600	-.16600	-.16600	-.16600	-.16600
	GRADENT			-.00005	-.00016	-.00728	-.00108	-.00108	-.00108	-.00108	-.00165	.00164	.00164	.00164	.00164	.00164	.00164	.00164	.00164

CA57B (NANL 713) B16 C5 F1 J42 W37 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

BETA = .000 F1/F = .000 EDF/LAF = .000  
 H/B = .125 EDF/LAF = .125 ELEV/N = .000

CPES3 CPES4 CPES5  
 CPES3 CPES4 CPES5





DATE 01 OCT 74

TABULATED SOURCE DATA - CA573

FACE 34

CA57B (NAAI 713) B12 C5 J42 W87 F18

(EDVA71) (US NAV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 UREF = 19.2300 IN. YMRF = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 71/0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	ONE
.165	-4.000	.07100
.165	.000	.06500
.165	4.900	-.03900
.165	10.000	-.01200
.165	14.900	-.03100
.165	19.900	-.08500
GRADIENT		.01400
		-.00530
		.00076

(EDVA72) (US NAV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 UREF = 19.2300 IN. YMRF = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 72/0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	ONE
.165	-3.900	.06900
.165	.000	.05700
.165	4.900	-.03900
.165	10.000	-.01100
.165	14.900	-.03600
.165	19.900	-.08700
GRADIENT		.00541
		.00078

(EDVA72) (US NAV 73)

## PARAMETRIC DATA

BETA = .000  
 H/B = .236 ELEVN = .000

(EDVA72) (US NAV 73)

## PARAMETRIC DATA

BETA = .000  
 H/B = .236 ELEVN = .000

(EDVA72) (US NAV 73)



CA57B (NAAI 713) B16 C5 F1 J40 W87 E18 (RDV237) (LA NEW 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 6/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	Q(PSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPEBF1	CPEBF2	CPEBF3	WE
.165	-3.900	60.80000	2116.39999	-.03500	-.19400	-.18700	-.23000	-.36400	-.26200	.12800
.165	.000	61.10000	2115.79999	-.004100	-.18800	-.17400	-.25800	-.35000	-.21400	.12500
.165	4.300	61.50000	2115.68701	-.06500	-.18300	-.15700	-.26300	-.34100	-.20500	.12100
.165	10.000	61.80000	2115.89999	-.02300	-.17800	-.17400	-.25900	-.38500	-.22400	.11400
.165	14.900	62.40000	2115.79999	.03700	-.19100	-.17500	-.27100	-.53200	-.27100	.11600
.165	19.900	63.20000	2117.39999	-.06100	-.29200	-.30200	-.30200	-.74000	-.34400	.11500
GRADIENT	.06201	-.02162	.00350	.00032	.00046	-.00169	-.00739	-.00169	-.00065	-.00065

CA57B (NAAI 713) B16 C5 F1 J40 W87 E18 (RDV237) (LA NEW 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 7/ 0 RNL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	Q(PSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPEBF1	CPEBF2	CPEBF3	WE
.165	-3.900	60.30000	2117.10001	-.07700	-.19800	-.19600	-.25200	-.51700	-.33900	.12900
.165	.000	61.30000	2117.10001	-.06200	-.19500	-.18900	-.25300	-.49900	-.33100	.12600
.165	4.900	61.80000	2116.89999	-.08400	-.17800	-.15700	-.23400	-.43100	-.26400	.12100
.165	9.900	62.10000	2117.20001	-.03200	-.16400	-.16800	-.24000	-.59900	-.29500	.11800
.165	14.900	62.90000	2117.79999	.04800	-.18300	-.17000	-.23300	-.75000	-.32300	.11600
.165	19.900	62.40000	2117.79999	-.08700	-.26900	-.27100	-.26400	-.34400	-.11500	-.00065
GRADIENT	.12519	.03248	.00613	.00087	.00150	-.00219	-.01011	-.00242	-.00065	-.00065

## PARAMETRIC DATA

BETA = .000 PTBF = 1.000  
 H/B = .125 EDLAF = -18.000  
 ELEVN = .000

## PARAMETRIC DATA

BETA = .000 PTBF = 1.000  
 H/B = .125 EDLAF = -18.000  
 ELEVN = .000

(RDV205) (DR NEV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0005

## PARAMETRIC DATA

MACH	ALPHA	QPSF	FSTAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	-4.000	60.70000	2118.29999	-.07800	-.21300	-.29800	-.50000
.165	.000	61.20000	2118.10001	-.10800	-.20400	-.28600	-.42800
.165	4.900	61.30000	2116.70001	-.08500	-.17100	-.27400	-.40700
.165	9.900	61.40000	2116.29999	.01400	-.14700	-.13900	-.26500
.165	14.900	61.50000	2117.00000	.07100	-.18900	-.17400	-.31200
.165	19.900	64.00000	2118.50000	-.39900	-.29000	-.23400	-.36300
GRADIENT	.04981	.09123	.00899	.00220	.00218	.00008	-.03917

RUN NO. 8/ U RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

## PARAMETRIC DATA

MACH	ALPHA	QPSF	FSTAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	-4.4120 SQ.FT. XREF = 43.5980 IN.						
LREF = 19.2300 IN. YREF = .0000 IN.							
BREF = 37.9350 IN. ZREF = -.4050 IN.							
SCALE = .0005							

RUN NO. 9/ U RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

## PARAMETRIC DATA

MACH	ALPHA	QPSF	FSTAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	-3.930	60.80000	2118.20001	-.15100	-.23000	-.17200	-.20200
.165	.000	61.30000	2117.89999	-.13400	-.23900	-.14600	-.15600
.165	4.900	62.30000	2117.79999	-.10700	-.15400	-.19100	-.13700
.165	9.900	62.10000	2117.10001	-.13700	-.18900	-.14400	-.06100
.165	14.900	62.40000	2117.29999	-.17100	-.23400	-.17700	-.02900
.165	19.900	63.60000	2118.50000	-.28000	-.30700	-.29100	-.01900
GRADIENT	.08304	-.05452	-.00101	.00141	-.00020	.01035	.03991

RUN NO. 9/ U RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

## PARAMETRIC DATA

MACH	ALPHA	QPSF	FSTAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	-4.4120 SQ.FT. XREF = 43.5980 IN.						
LREF = 19.2300 IN. YREF = .0000 IN.							
BREF = 37.9350 IN. ZREF = -.4050 IN.							
SCALE = .0005							

RUN NO. 9/ U RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

## REFERENCE DATA

SREF = 4.4120 S2.FT.  
LREF = 19.2300 IN.  
BREF = 37.9350 IN.  
SCALE = .0436

(CA578) (NAAL 713)

CA578 B16 C5 F1 J40 W47 E18

## PARAMETRIC DATA

BETA = .000  
H/B = .125  
ELEVN = .000

(REV210) (NAAL 73 )

FTH/F = 1.300  
EDFLAF = .000

RUN NO. 10/ 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	3(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-4.000	60.70000	2117.89999	-.13500	-.23500	-.17100	-.16300	-.12700	-.10000
.165	.000	60.80000	2117.50000	-.16000	-.20200	-.15200	-.15200	-.10200	-.05600
.165	4.900	61.70000	2117.39999	-.15700	-.19400	-.13700	-.10700	-.05500	-.01100
.165	9.900	62.40000	2117.60001	-.16400	-.22100	-.16600	-.04400	-.02300	-.05100
.165	14.900	62.80000	2117.79999	-.20300	-.22300	-.22300	-.01100	-.02000	-.04200
.165	19.900	63.30000	2118.29999	-.29300	-.30300	-.30300	-.04000	-.06200	-.09300
.165	GRADIENT	.12212	-.000399	-.00293	-.00341	-.00246	-.0018	-.00744	-.00465

(CA578) (NAAL 713)

CA578 B16 C5 F1 J40 W47 E18

## PARAMETRIC DATA

SREF = 4.4120 S2.FT.  
LREF = 19.2300 IN.  
BREF = 37.9350 IN.  
SCALE = .0436

(CA578) (NAAL 73 )

BETA = .000  
H/B = .125  
ELEVN = .000

(REV211) (NAAL 73 )

RUN NO. 11/ 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	3(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-4.000	60.30000	2116.39999	-.14900	-.23000	-.17400	-.14600	-.15400	-.03300
.165	.000	60.30000	2116.10001	-.15200	-.20800	-.15600	-.12100	-.11800	-.03200
.165	4.900	61.90000	2115.70001	-.15700	-.18800	-.13100	-.09100	-.07800	-.01300
.165	9.900	61.70000	2116.10001	-.16400	-.21200	-.17100	-.04600	-.04600	-.04200
.165	14.900	62.10000	2116.70001	-.19300	-.23300	-.20100	-.01600	-.00600	-.06700
.165	19.900	62.60000	2117.39999	.29000	-.28700	-.30100	-.00100	-.00100	-.07400
.165	GRADIENT	.10463	.01409	-.00212	-.00299	-.00152	.00733	.00775	-.00778

(CA578) (NAAL 73 )

(REV212) (NAAL 73 )

RUN NO. 12/ 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	3(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-4.000	60.30000	2116.39999	-.14900	-.23000	-.17400	-.14600	-.15400	-.03300
.165	.000	60.30000	2116.10001	-.15200	-.20800	-.15600	-.12100	-.11800	-.03200
.165	4.900	61.90000	2115.70001	-.15700	-.18800	-.13100	-.09100	-.07800	-.01300
.165	9.900	61.70000	2116.10001	-.16400	-.21200	-.17100	-.04600	-.04600	-.04200
.165	14.900	62.10000	2116.70001	-.19300	-.23300	-.20100	-.01600	-.00600	-.06700
.165	19.900	62.60000	2117.39999	.29000	-.28700	-.30100	-.00100	-.00100	-.07400
.165	GRADIENT	.10463	.01409	-.00212	-.00299	-.00152	.00733	.00775	-.00778



CA57B (NAAI 713) B16 CS F1 J40 W47 E18

(RDV12) ( USA NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4000 IN.  
 SCALE = .0405

RUN NO. 12/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q(FSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPFBF1	CPFBF2	CPFBF3
.165	-4.000	60.10000 2119.89999	-1.15600	-1.18500	-1.14500	-1.17300	-1.07700	-1.05700	.29400
.165	.000	60.90000 2119.29999	-1.15300	-1.18000	-1.13200	-1.15700	-1.05200	-1.04400	.24600
.165	4.900	60.80000 2118.29999	-1.13500	-1.16300	-1.10800	-1.11000	-1.02400	-1.02200	.23200
.165	9.900	61.10000 2119.00000	-1.14300	-1.19200	-1.14100	-1.16600	-1.03400	-1.02700	.21790
.165	15.000	61.50000 2119.70001	-1.16000	-1.21500	-1.16200	-1.19100	-1.05100	-1.03600	.21390
.165	19.900	63.20000 2119.10001	-1.26000	-1.27500	-1.27100	-1.16000	-1.05800	-1.01600	.21770
GRADIENT	.07623	-.05464	.000001	-.000157	-.000093	.000003	-.000114	.000493	-.000004

CA57B (NAAI 713) B16 CS F1 J40 W47 E18

(RDV13) ( USA NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = .5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4000 IN.  
 SCALE = .0405

RUN NO. 13/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q(FSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPFBF1	CPFBF2	CPFBF3
.165	-4.000	60.60000 2118.29999	-1.14900	-.20100	-1.14600	-.16100	-.10700	-.04100	.29400
.165	.000	60.90000 2117.79999	-1.14800	-.18200	-1.13900	-.13500	-.08600	-.02300	.24600
.165	4.900	61.30000 2117.50000	-1.15300	-.18400	-.13000	-.11600	-.05500	.23200	.23200
.165	9.900	61.40000 2117.50000	-1.14900	-.20500	-.15400	-.10400	-.03400	.21400	.21400
.165	14.900	62.70000 2117.49999	-1.17600	-.23700	-.18600	-.09500	-.02700	.12300	.21790
.165	20.000	63.00000 2118.29999	-1.26600	-.23700	-.27100	-.09700	-.03700	.10400	.21790
GRADIENT	.01749	-.02172	-.000117	-.000194	-.000205	.000384	-.000442	.000364	-.000004

(RDV13) ( USA NAV 73 )

## REFERENCE DATA

BETA = .000 FTWF = 1.000  
 H/B = .286 DFLAP = .000  
 ELEVON = .000

BETA = .000 FTWF = 1.000  
 H/B = .286 DFLAP = .000  
 ELEVON = .000

BETA = .000 FTWF = 1.000  
 H/B = .286 DFLAP = .000  
 ELEVON = .000

TABULATED SOURCE DATA - CAST78

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REFERENCE DATA

S&F	=	4,4120 Sq.FT.	XRP	=	43,5980 IN.
U&F	=	19,2300 IN.	XRP	=	.0000 IN.
B&F	=	37,9350 IN.	ZRP	=	-.4050 IN.

PASSENGER DATA

BETA	=	.000	FITVF	=	1.750
HVB	=	.246	EDFLAF	=	.000
TELEVCN	=	.000			

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	RUN NO	147 U	FNU/L =	1.20	GRADIENT INTERVAL =	-4.000	16.00
MAC-1	ALPHA	3(PSF)	FSTAT	OPTF1	OPTF2	OPTF3	OPTF4
.165	-4.000	61.00000	2119.39999	-.16000	-.20000	-.14700	-.15400
.165	.000	61.10000	2118.89999	-.14400	-.19100	-.13100	-.12500
.165	4.900	61.60000	2118.79999	-.15800	-.18800	-.13100	-.11400
.165	9.800	61.80000	2118.50000	-.14800	-.19500	-.14500	-.09400
.165	14.900	62.70000	2118.50001	-.17100	-.22000	-.18100	-.09100
.165	20.000	63.30000	2119.39997	-.26400	-.28100	-.28100	-.10500
.165	GRADIENT	10.6661	10.66614	-.000068	-.000062	-.000072	-.000017

CONFERENCE DATA

<b>STEF</b>	<b>=</b>	<b>4.4120</b>	<b>Sq.FT.</b>	<b>3040F</b>	<b>=</b>	<b>43.5980</b>	<b>IN.</b>
<b>UCEF</b>	<b>=</b>	<b>19.230</b>	<b>IN.</b>	<b>146F</b>	<b>=</b>	<b>.0000</b>	<b>IN.</b>
<b>BCEF</b>	<b>=</b>	<b>37.9350</b>	<b>IN.</b>	<b>246F</b>	<b>=</b>	<b>-4.050</b>	<b>IN.</b>

(Rev. 15)

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PARAMETIC DATA

SREF	=	4.4120	SQ.FT.	XREF =	43.5980 IN.
UREF	=	19.2300	IN.	YREF =	.0000 IN.
BREF	=	37.9350	IN.	ZREF =	-.4050 IN.
SCALE	=	.0435			
RUN NO.	=	15	0	RVAL =	1.00
				GRADIENT INTERVAL =	-4.00V 16.00

<i>TA</i>	<i>XMRP</i>	=	43,590 IN.
	<i>YMRP</i>	=	.0000 IN.
	<i>ZMRP</i>	=	-.4050 IN.

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## REFERENCE DATA

**SREF** = 4.4120 SQ.FT.  
**URF** = 19.2300 IN.  
**BRF** = 37.9350 IN.  
**SCALE** = .0405

RUN NO. 16/ 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH ALPHA Q(FSF) PSTAT CPTBF1 CPTBF2 CPTBF3 CPTBF1 CPTBF2 CPTBF3  
 .165 -3.900 60.50000 2115.60000 -.08700 -.17700 -.23700 -.27800 -.24900  
 .165 .000 61.60000 2114.70001 -.05700 -.17200 -.23100 -.26500 -.24100  
 .165 4.900 62.20000 2114.60001 -.01800 -.16500 -.14900 -.22800 -.21900  
 .165 9.900 62.40000 2114.60001 -.01000 -.17000 -.16100 -.24600 -.23200  
 .165 15.000 62.60000 2114.10001 -.03500 -.19000 -.16800 -.24600 -.21900  
 .165 19.900 63.30000 2114.60001 -.01300 -.25200 -.24100 -.24600 -.24900  
 GRADIENT -.19267 -.03973 -.00500 -.00003 -.00003 -.00003 -.00002 -.00002 -.00002

CA57B (NVAL 713) B16 CS F1 JAU WAT E18

(REV17) ( L&amp; N&amp;V 73 )

## REFERENCE DATA

**SREF** = 4.4120 SQ.FT.  
**URF** = 19.2300 IN.  
**BRF** = 37.9350 IN.  
**SCALE** = .0405

RUN NO. 17/ 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH ALPHA Q(FSF) PSTAT CPTBF1 CPTBF2 CPTBF3 CPTBF1 CPTBF2 CPTBF3  
 .165 -3.900 60.70000 2114.60001 -.06700 -.18800 -.21900 -.26700 -.31400  
 .165 .000 61.20000 2114.20001 -.05600 -.17300 -.22300 -.26100 -.35100  
 .165 4.900 61.40000 2113.60001 -.12000 -.15900 -.15100 -.25400 -.33200  
 .165 9.900 62.20000 2113.70001 -.02400 -.16900 -.15600 -.20700 -.33600  
 .165 14.900 62.60000 2113.79999 -.04500 -.19800 -.17700 -.22400 -.38200  
 .165 20.000 62.90000 2114.29999 -.02200 -.27100 -.25900 -.28900 -.34100  
 GRADIENT .19369 -.04305 -.00643 -.00041 -.00022 -.00011 -.00046 -.00024 -.00011

PARAMETRIC DATA

**BETA** = .000  
**MB** = .246  
**ELEVON** = .000

**BETA** = .000  
**MB** = .246  
**ELEVON** = .000

**BETA** = .000  
**MB** = .246  
**ELEVON** = .000

PARAMETRIC DATA

**BETA** = .000  
**MB** = .246  
**ELEVON** = .000

**BETA** = .000  
**MB** = .246  
**ELEVON** = .000

**BETA** = .000  
**MB** = .246  
**ELEVON** = .000

**BETA** = .000  
**MB** = .246  
**ELEVON** = .000

DATE OF CCT 74

TABULATED SOURCE DATA - C4378

FADE 42

C4378 (NML 713) 816 CS F1 J40 W47 E18

## REFERENCE DATA

**SOF** = 4,4120 SQ.FT. **MGP** = 43,5380 IN.  
**LGF** = 19,2300 IN. **TGF** = .0000 IN.  
**HGF** = 37,9350 IN. **ZGF** = -.4050 IN.  
**SCALE** = .0005

RUN NO. 18/ U ROLL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	OPTF1	OPTF2	OPTF3	OPTF4	OPTF5
ALPHA	.01PSF	PSAT	OPTF1	OPTF2	OPTF3
.165	-3.900	60,00000 2113,70001	-0,05200	-1,19300	-1,16100
.165	.000	61,70000 2113,69999	-.06600	-.17200	-.24500
.165	4,500	61,30000 2113,00000	-.13500	-.17800	-.19300
.165	9,900	61,70000 2113,00000	-.02300	-.18900	-.19400
.165	14,300	63,10000 2112,89999	-.02900	-.23900	-.19600
.165	19,300	62,90000 2113,70001	-.03200	-.27800	-.18200
GRADIENT	.38643	-.05229	.01341	-.00397	-.00071

C4378 (NML 713) 816 CS F1 J40 W47 E18

## REFERENCE DATA

**SOF** = 4,4120 SQ.FT. **MGP** = 43,5380 IN.  
**LGF** = 19,2300 IN. **TGF** = .0000 IN.  
**HGF** = 37,9350 IN. **ZGF** = -.4050 IN.  
**SCALE** = .0005

RUN NO. 19/ U ROLL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	OPTF1	OPTF2	OPTF3	OPTF4	OPTF5
ALPHA	.01PSF	PSAT	OPTF1	OPTF2	OPTF3
.165	-3.900	60,40000 2113,20001	-.51200	-.25700	-.24900
.165	.000	61,60000 2113,39999	-.33000	-.27500	-.29600
.165	4,900	61,40000 2111,70001	-.32400	-.26720	-.26600
.165	9,900	62,40000 2112,20001	-.36000	-.28400	-.28500
.165	14,300	62,70000 2111,79999	-.39200	-.32600	-.29500
.165	19,300	63,70000 2113,50000	-.51300	-.34800	-.36100
GRADIENT	.11247	-.03411	-.00394	-.00307	-.00024

(C4378) (NML 73)

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## REFERENCE DATA

**SOF** = 4,4120 SQ.FT. **MGP** = 43,5380 IN.  
**LGF** = 19,2300 IN. **TGF** = .0000 IN.  
**HGF** = 37,9350 IN. **ZGF** = -.4050 IN.  
**SCALE** = .0005

RUN NO. 18/ U ROLL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	OPTF1	OPTF2	OPTF3	OPTF4	OPTF5
ALPHA	.01PSF	PSAT	OPTF1	OPTF2	OPTF3
.165	-3.900	60,00000 2113,70001	-.05200	-.17700	-.16100
.165	.000	61,70000 2113,69999	-.06600	-.17200	-.24500
.165	4,500	61,30000 2113,00000	-.13500	-.17800	-.19300
.165	9,900	61,70000 2113,00000	-.02300	-.18900	-.19400
.165	14,300	63,10000 2112,89999	-.02900	-.23900	-.23500
.165	19,300	62,90000 2113,70001	-.03200	-.27800	-.24300
GRADIENT	.38643	-.05229	.01341	-.00397	-.00071

(C4378) (NML 73)

DATE 01 OCT 74

TABULATED SOURCE DATA - C578

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C578 (MAAL 713) B16 CS F1 JAU 1617 E18

## REFERENCE DATA

$S_{EF}$  = 4.4120 SLT.  
 $U_{EF}$  = 19.2300 IN.  
 $B_{EF}$  = 37.9350 IN.  
 $SCALE$  = .0435

RUN NO. 21/0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MAOH	ALPHA	4IFS1	FSTAT	CPTF1	CPTF2	CPTF3	CPTF4	CPTF5
.165	-3.900	60.40000 2112.39999	-.29600	-.25200	-.25900	-.26400	-.21500	.29000
.165	.000	60.00000 2112.20001	-.91300	-.25700	-.27900	-.06700	.25100	.24600
.165	4.900	61.30000 2111.7001	-.31400	-.28400	-.28900	-.19000	.25700	.25600
.165	9.900	62.10000 2111.29999	-.35500	-.28300	-.24773	-.19400	.27200	.27400
.165	14.900	62.30000 2111.79991	-.43100	-.31900	-.29500	-.23100	.29500	.33600
.165	19.900	62.70000 2112.29999	-.48600	-.33000	-.33400	-.29100	.30480	.27700
GRADIENT		.10380	-.04787	-.00887	-.00335	.01167	.01144	.0345

C578 (MAAL 713) B16 CS F1 JAU 1617 E18

## REFERENCE DATA

$S_{EF}$  = 4.4120 SLT.  
 $U_{EF}$  = 19.2300 IN.  
 $B_{EF}$  = 37.9350 IN.  
 $SCALE$  = .0435

RUN NO. 21/0 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MAOH	ALPHA	4IFS1	FSTAT	CPTF1	CPTF2	CPTF3	CPTF4	CPTF5
.165	-3.900	60.30000 2111.50000	-.28900	-.23200	-.22200	-.05200	.15500	.17200
.165	.000	60.30000 2111.10001	-.76207	-.23700	-.21200	-.03600	.07600	.17400
.165	5.000	61.30000 2111.72399	-.28400	-.25100	-.22600	.10700	.11400	.21300
.165	10.000	62.20000 2110.39999	-.32600	-.26000	-.23300	.16400	.12500	.27300
.165	15.000	62.20000 2110.70001	-.37100	-.27900	-.24500	.20000	.17100	.27400
.165	20.000	62.60000 2111.73999	-.44300	-.31600	-.29800	.21900	.11400	.27700
GRADIENT		.10618	-.03690	-.00430	-.00218	.01223	.01137	.01634

(50VZ1) ( 24 NCV 73 )

## PARAMETRIC DATA

BETA	= .000	PIN/P	= 1.300
H/B	= .246	EDFLAF	= 20.000
ELEVON	= .000		

(50VZ1) ( 24 NCV 73 )

## PARAMETRIC DATA

BETA	= .000	PIN/P	= 1.000
H/B	= .239	EDFLAF	= 20.000
ELEVON	= .000		

(50VZ1) ( 24 NCV 73 )

## PARAMETRIC DATA

(UD NCV 73)

(UDV222) (UD NCV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMGP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4080 IN.  
 SCALE = .0405

RUN NO. 22/ 0 RFLN = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPBF1	CPBF2	CPBF3	H/B
.165	-3.900	61.20000 2113.50000	-.35400	-.30000	-.26900	-.06100	.21.30	.12200	
.165	.000	61.60000 2113.70001	-.35100	-.32200	-.31800	.02900	.19800	.19400	.12500
.165	5.000	62.40000 2113.10001	-.39200	-.32500	-.33400	.17200	.28600	.30300	.12100
.165	9.900	62.40000 2112.50000	-.42700	-.31600	-.34100	.32200	.40800	.40000	.11800
.165	14.900	62.20000 2113.10001	-.44500	-.36900	-.35600	.45600	.52000	.47600	.11600
.165	20.000	63.80000 2114.10001	-.50800	-.37300	-.40900	.51500	.61900	.53200	.11500
GRADIENT	.05793	-.04234	-.00548	-.00276	-.00407	.02907	.01803	.01923	-.00065

CA578 (UDL 713) B16 C5 F1 J4D W87 E18

(UDV223) (UD NCV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMGP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4080 IN.  
 SCALE = .0405

RUN NO. 23/ 0 RFLN = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPBF1	CPBF2	CPBF3	H/B
.165	-3.900	60.90000 2113.39999	-.32600	-.28400	-.26000	-.05710	.06000	.14000	.12800
.165	.000	61.10000 2112.60001	-.34000	-.30000	-.30300	.06200	.20200	.22000	.12500
.165	4.900	61.40000 2112.60001	-.36800	-.31000	-.31000	.20600	.27800	.30600	.12100
.165	9.900	62.30000 2112.79999	-.42100	-.32700	-.34100	.32400	.37500	.37100	.11800
.165	15.000	62.70000 2113.39999	-.45300	-.35600	-.35400	.44300	.49700	.45600	.11600
.165	19.900	63.50000 2114.10001	-.48700	-.37900	-.40000	.52500	.59600	.50900	.11500
GRADIENT	.10132	-.00674	-.00706	-.00359	-.00469	.02839	.02184	.01637	-.00065

(UDV224)

(UD NCV 73)

(UDV225)

(UD NCV 73)

(UDV226)

(UD NCV 73)

(UDV227)

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DATE 01 OCT 74

TABULATED SOURCE DATA - CA57B

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CA57B (NAAL 713) 816 CS F1 J40 W47 E18

(RDV24) ( OA NEW 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 24 / 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q(PSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	OPTBF4	H/B
.165	-3.00	60.20000	2112.20001	-.29300	-.24000	-.20200	-.05000	.11100
.165	.000	60.80000	2111.79999	-.28800	-.25600	-.24700	.08000	.14300
.165	4.930	61.40000	2111.89999	-.32600	-.27900	-.27100	.17600	.13200
.165	10.000	61.80000	2111.79999	-.39100	-.28800	-.28200	.25000	.21700
.165	14.900	62.40000	2112.00000	-.40300	-.30200	-.28600	.31300	.30100
.165	19.900	62.70000	2113.10001	-.47600	-.31000	-.34800	.34400	.35200
	GRADIENT	.00013	-.00756	-.00686	-.00326	-.00418	.01912	.01247

CA57B (NAAL 713) 816 CS F1 J40 W47 E18

(RDV25) ( OA NEW 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 25 / 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q(PSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	OPTBF4	H/B
.165	9.900	61.80000	2113.70001	-.33900	-.31800	-.27700	.14200	.61100
.165	14.900	62.40000	2114.50000	-.38100	-.38600	-.34300	.55100	.44600
.165	19.900	62.90000	2115.50000	-.45800	-.38600	-.39600	1.14500	2.43900
	GRADIENT	.12000	.16000	-.00840	-.01360	-.01320	.10180	.16120

## PARAMETRIC DATA

SREF	H/B	ELEVN	F TN/F	F DN/F
.000	.125	.000	.11100	.12500
			.07600	.14300
			.13200	.24800
			.21700	.28500
			.30100	.35200
			.36200	.35900
			.01912	.01247
			.01567	.01745

## PARAMETRIC DATA

SREF	H/B	ELEVN	F TN/F	F DN/F
.000	.125	.000	.11100	.12500
			.07600	.14300
			.13200	.24800
			.21700	.28500
			.30100	.35200
			.36200	.35900
			.01912	.01247
			.01567	.01745

DATE 01 CCT 74

TABULATED SOURCE DATA - CA578

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CA578 (FINAL 713) B16 C5 F1 J4U W87 E18

(RDV26) (DR NCV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 26/0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	3(PSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPEBF1	CPEBF2	CPEBF3	H/B
.165	9.900	62.30000	2113.6000	-.39700	-.29700	-.24000	.01200	.27900	.07000	.03200
.165	14.900	62.60000	2114.6000	-.33400	-.34300	-.29500	.32900	.76200	.47400	.03000
.165	20.000	62.80000	2115.89999	-.40000	-.41100	-.37700	.69700	1.24500	.56200	.12900
GRADIENT	.04000	.02000	-.00880	-.00880	-.01100	.06340	.08660	.02840	-.00440	

CA578 (FINAL 713) B16 C5 F1 J4U W87 E18

(RDV27) (DR NCV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 27/0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	3(PSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPEBF1	CPEBF2	CPEBF3	H/B
.165	9.900	62.00000	2113.2100	-.26400	-.24400	-.21400	.01200	.04600	-.03200	.03200
.165	14.900	62.10000	2113.29999	-.28500	-.26500	-.23900	.15400	.27200	.15800	.13700
.165	19.900	63.20000	2114.70001	-.32700	-.33600	-.31100	.29700	.52000	.35100	.32900
GRADIENT	.02000	.02000	-.00420	-.00420	-.00500	.02540	.04520	.03200	-.00440	

## PARAMETRIC DATA

BETA	= .000	FTR/F = 1.300
H/B	= .125	EDFLAF = .000
ELEVON	= .000	

## PARAMETRIC DATA

BETA	= .000	FTR/F = 1.000
H/B	= .125	EDFLAF = .000
ELEVON	= .000	

REFERENCE DATA				PARAMETRIC DATA			
SREF	4.4120 SQ.FT.	XMRP	= 43.5980 IN.	B16 CS F1	J40 W87 E18	BETA	.000
UREF	19.2300 IN.	YMRP	= .0000 IN.			H/B	.12800
BREF	37.9350 IN.	ZMRP	= -.4050 IN.			EDFLAP	-.18000
SCALE	.0405					ELEVON	15.000
RUN NO. 28/0 RNU/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00							
MACH	ALPHA	Q(PSF)	FSTAT	OPTBF1	OPTBF2	OPTBF3	CFBF3
.165	-4.000	60.50000	2114.79999	-.32400	-.27000	-.26800	-.35600
.165	.000	61.30000	2114.50000	-.27300	-.24000	-.23100	-.33200
.165	4.900	61.90000	2114.29999	-.28600	-.20400	-.19500	-.28000
.165	9.900	62.30000	2114.60001	-.20100	-.20700	-.18400	-.27900
.165	14.900	62.70000	2115.10001	-.19600	-.23200	-.18900	-.27300
.165	19.900	63.70000	2116.39999	-.29100	-.28200	-.33600	-.24900
GRADIENT	.11228	.01596	.00683	.01223	.00423	.00471	.00234
RUN NO. 29/0 RNU/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00							
MACH	ALPHA	Q(PSF)	FSTAT	OPTBF1	OPTBF2	OPTBF3	CFBF3
.165	-4.000	60.70000	2114.79999	-.31000	-.26700	-.26000	-.35900
.165	.000	61.20000	2114.29999	-.26400	-.23900	-.23100	-.33800
.165	4.900	61.40000	2113.79999	-.22600	-.19200	-.19700	-.33500
.165	9.900	62.20000	2114.20001	-.19100	-.19700	-.17500	-.32500
.165	14.900	63.00000	2114.79999	-.21900	-.24500	-.22900	-.31500
.165	19.900	63.40000	2115.60001	-.30400	-.32800	-.37300	-.28200
GRADIENT	.11780	-.00020	.02630	.00169	.03327	.02210	-.01177
RUN NO. 29/0 RNU/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00							
MACH	ALPHA	Q(PSF)	FSTAT	OPTBF1	OPTBF2	OPTBF3	CFBF3
.165	-4.000	60.70000	2114.79999	-.31000	-.26700	-.26000	-.35900
.165	.000	61.20000	2114.29999	-.26400	-.23900	-.23100	-.33800
.165	4.900	61.40000	2113.79999	-.22600	-.19200	-.19700	-.33500
.165	9.900	62.20000	2114.20001	-.19100	-.19700	-.17500	-.32500
.165	14.900	63.00000	2114.79999	-.21900	-.24500	-.22900	-.31500
.165	19.900	63.40000	2115.60001	-.30400	-.32800	-.37300	-.28200
GRADIENT	.11780	-.00020	.02630	.00169	.03327	.02210	-.01177

REFERENCE DATA				PARAMETRIC DATA			
SREF	4.4120 SQ.FT.	XMRP	= 43.5980 IN.	B16 CS F1	J40 W87 E18	BETA	.000
UREF	19.2300 IN.	YMRP	= .0000 IN.			H/B	.12800
BREF	37.9350 IN.	ZMRP	= -.4050 IN.			EDFLAP	-.18000
SCALE	.0405					ELEVON	15.000
RUN NO. 29/0 RNU/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00							
MACH	ALPHA	Q(PSF)	FSTAT	OPTBF1	OPTBF2	OPTBF3	CFBF3
.165	-4.000	60.70000	2114.79999	-.31000	-.26700	-.26000	-.35900
.165	.000	61.20000	2114.29999	-.26400	-.23900	-.23100	-.33800
.165	4.900	61.40000	2113.79999	-.22600	-.19200	-.19700	-.33500
.165	9.900	62.20000	2114.20001	-.19100	-.19700	-.17500	-.32500
.165	14.900	63.00000	2114.79999	-.21900	-.24500	-.22900	-.31500
.165	19.900	63.40000	2115.60001	-.30400	-.32800	-.37300	-.28200
GRADIENT	.11780	-.00020	.02630	.00169	.03327	.02210	-.01177

(GOV230) ( LR NCV 73 )

PARAMETRIC DATA

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMAP = 43.5980 IN.  
 LREF = 19.2300 IN. YMAP = .0000 IN.  
 BREF = 37.9350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

RUN NO. 3010 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALPHA	P(SF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPBF1	CPBF2	CPBF3	H/B
.165	-3.900	60.6000	2113.10001	-.29100	-.24300	-.23600	-.33500	-.42000	-.24600	.12400
.165	.000	61.10000	2113.00000	-.25900	-.22400	-.21400	-.32700	-.44300	-.23500	.12500
.165	4.900	61.50000	2112.89999	-.23700	-.19800	-.19600	-.32100	-.52700	-.21400	.12100
.165	9.900	62.10000	2112.89999	-.20000	-.19300	-.17300	-.29400	-.54800	-.16900	.11800
.165	14.900	62.40000	2113.60001	-.22900	-.26000	-.21300	-.30200	-.62300	-.17600	.11600
.165	19.900	63.00000	2114.79999	-.34300	-.35200	-.31100	-.36100	-.73800	-.25700	.11500
GRADIENT	.09654	.01986	.00379	-.00017	.00176	.00234	-.00392	.00435	-.00465	

(GOV231) ( LR NCV 73 )

PARAMETRIC DATA

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMAP = 43.5980 IN.  
 LREF = 19.2300 IN. YMAP = .0000 IN.  
 BREF = 37.9350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

RUN NO. 3010 RNL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALPHA	P(SF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	CPBF1	CPBF2	CPBF3	H/B
.165	10.000	62.10000	2119.29999	-.23700	-.21200	-.20800	-.60900	-.59200	-.34400	.03200
.165	15.000	63.30000	2119.50000	-.21000	-.23400	-.21900	-.55400	-.58400	-.41200	.03000
.165	19.900	63.10000	2119.79999	-.31900	-.31400	-.31200	-.48500	-.38800	-.51000	.02900
GRADIENT	.24000	.04000	.00080	-.00040	-.00340	.01000	.00120	-.01360	-.00040	

(GOV231) ( LR NCV 73 )

PARAMETRIC DATA

CA578 (NAAL 713)

B16 C5 F1

J40 W7 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 32/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	STAT	CPTBF1	CPTBF2	CPTBF3	CPEBF1	CPEBF2	CPEBF3
MACH	ALPHA	.2(PSF)	-22300	-.20500	-.19000	-.33900	-.45700
.165	9.900	61.50000	2118.20000	-.22900	-.20900	-.34500	-.44700
.165	14.900	62.50000	2118.50000	-.24000	-.22600	-.37500	-.44600
.165	19.900	63.10000	2119.79999	-.34300	-.33800	-.37500	-.43700
GRADIENT	.20000	.06000	-.00340	-.00340	-.00340	-.00340	-.01360

(RDV232) ( DR ACV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 33/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	STAT	CPTBF1	CPTBF2	CPTBF3	CPEBF1	CPEBF2	CPEBF3
MACH	ALPHA	.2(PSF)	-.22000	-.21100	-.19900	-.34000	-.49900
.165	10.000	62.40000	2117.50000	-.23400	-.23700	-.31000	-.42400
.165	14.900	62.50000	2117.60001	-.32700	-.32200	-.36900	-.51700
.165	21.000	62.50000	2119.00000	-.02046	-.02286	-.00163	.0142-

(RDV233) ( DR ACV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 33/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	STAT	CPTBF1	CPTBF2	CPTBF3	CPEBF1	CPEBF2	CPEBF3
MACH	ALPHA	.2(PSF)	-.22000	-.21100	-.19900	-.34000	-.49900
.165	10.000	62.40000	2117.50000	-.23400	-.23700	-.31000	-.42400
.165	14.900	62.50000	2117.60001	-.32700	-.32200	-.36900	-.51700
.165	21.000	62.50000	2119.00000	-.02046	-.02286	-.00163	.0142-

(RDV233) ( DR ACV 73 )

DATE 01 OCT 74

## TABULATED SOURCE DATA - CA57B

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## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 EREF = 57.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 36 / 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Z(FSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	OPTBF4	H/B
.165	-3.900	60.90000	2120.00000	-.29100	-.22500	-.35300	-.24800	-.27800
.165	.000	61.10000	2119.50000	-.23900	-.21100	-.34700	-.25100	-.27400
.165	4.900	61.50000	2118.60001	-.19900	-.18500	-.32500	-.22400	-.25000
.165	9.900	61.90000	2118.50000	-.21800	-.18500	-.33700	-.23200	-.27900
.165	14.900	62.30000	2118.89999	-.22800	-.20300	-.32600	-.21600	-.22300
.165	19.900	62.80000	2119.70001	-.32700	-.34700	-.37500	-.27100	-.27800
GRADIENT	.07602	-.06361	.03295	.00356	.00126	.00133	.00226	.00357

CA57B (NML 713) B16 CS F1 JAU W87 E18

(RDV234) ( LR NEW 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 EREF = 57.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 36 / 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Z(FSF)	PSTAT	OPTBF1	OPTBF2	OPTBF3	OPTBF4	H/B
.165	-3.900	60.70000	2119.20001	-.26400	-.22500	-.32500	-.23600	-.25800
.165	.000	60.70000	2118.79999	-.23900	-.21200	-.31100	-.22500	-.23000
.165	4.900	61.30000	2118.10001	-.23500	-.19100	-.29700	-.21600	-.23400
.165	9.900	62.90000	2119.20001	-.25200	-.21300	-.31900	-.21600	-.22400
.165	14.900	62.20000	2118.60001	-.23600	-.24000	-.32400	-.21900	-.22900
.165	19.900	62.80000	2119.60001	-.33100	-.35600	-.36400	-.23200	-.27400
GRADIENT	.11004	-.01545	.03285	-.00073	.00046	.00084	.00087	.00065

(RDV235) ( LR NEW 73 )

## PARAMETRIC DATA

BETA	= .000	F1NUF	= 1.500
H/B	= .286	EDFLAP	= -13.000
ELEVON	= 15.000		



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TABULATED SOURCE DATA - C457B

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C457B (NAAL 713)

(RDV236) ( US NAV 73 )

## REFERENCE DATA

SQFT = 4,4120 SQ.FT. XREFP = 43.5980 IN.  
 LBDF = 19.2300 IN. YREFP = .0000 IN.  
 BREF = 37.9350 IN. ZREFP = -.4050 IN.  
 SCALE = .0405

RUN NO. 36/0 RFL = 1.20 GRADIENT INTERVAL = -4.00 16.00

	ALPHA	ZPSF	PSAT	OPTF1	OPTF2	OPTF3	OPBF1	OPBF2	OPBF3	H/B
.165	-3.900	60.40000	2118.39999	-.27200	-.22200	-.21300	-.29500	-.21900	-.22300	.29300
.165	.000	61.00000	2118.00000	-.24000	-.21300	-.20500	-.28300	-.21300	-.22200	.24600
.165	5.000	61.50000	2117.39999	-.24500	-.19600	-.18900	-.28000	-.19400	-.20500	.24200
.165	9.900	61.70000	2117.50000	-.24400	-.23400	-.19300	-.29400	-.19100	-.18200	.27900
.165	15.000	62.40000	2117.79999	-.23500	-.23200	-.21200	-.28300	-.20100	-.18100	.27800
.165	19.900	63.10000	2118.79999	-.33600	-.31300	-.34500	-.27300	-.25100	-.25100	.27700
GRADIENT	.089811	-.03434	.00141	-.00229	.00225	.00225	.00119	.00261	-.00164	

C457B (NAAL 713) 816 CS F1 J40 W87 E18

(RDV237) ( US NAV 73 )

## REFERENCE DATA

SQFT = 4,4120 SQ.FT. XREFP = 43.5980 IN.  
 LBDF = 19.2300 IN. YREFP = .0000 IN.  
 BREF = 37.9350 IN. ZREFP = -.4050 IN.  
 SCALE = .0405

RUN NO. 37/0 RFL = 1.20 GRADIENT INTERVAL = -4.00 16.00

	ALPHA	ZPSF	PSAT	OPTF1	OPTF2	OPTF3	OPBF1	OPBF2	OPBF3	H/B
.165	-3.900	60.70000	2112.39999	-.14200	-.15100	-.14600	-.21400	-.22700	-.26900	.04200
.165	.000	61.20000	2112.79999	-.13700	-.12900	-.12600	-.25700	-.25200	-.27600	.03800
.165	4.900	62.10000	2112.60221	-.13800	-.13200	-.13200	-.25600	-.25600	-.27400	.03400
.165	9.900	62.00000	2112.50222	-.13800	-.13700	-.12900	-.26900	-.27000	-.27400	.03200
.165	14.900	62.30000	2112.20221	-.15700	-.15700	-.14100	-.28900	-.27900	-.26400	.03000
.165	19.900	63.40000	2112.39999	-.22100	-.23400	-.22900	-.32700	-.31800	-.29600	.02900
GRADIENT	.08315	-.03591	-.00068	-.002249	.00014	-.002335	-.000562	.00223	-.000363	

PARAMETRIC DATA

BETA = .000 FTN/P = 1.000  
 H/B = .286 EDF/LAP = -18.000  
 ELEVON = 15.000

BETA = .000 FTN/P = 1.000  
 H/B = .286 EDF/LAP = -18.000  
 ELEVON = 15.000

BETA = .000 FTN/P = 1.000  
 H/B = .286 EDF/LAP = -18.000  
 ELEVON = 15.000

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TABULATED SOURCE DATA - CA578

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CA578 (MAIL 713) B16 CS F1 .460 W87 E18 (RDV-39) ( US NAV 73 )

## REFERENCE DATA

SREF = 4.4120 Sq.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 387 0 RVAL = 1.20 GRADIENT INTERVAL = -4.00/16.00

MACH	ALPHA	Z(PF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-3.900	60.70000	2113.20001	-.14900	-.15700	-.15100	-.21800	-.28600	-.27800
.165	.000	61.00000	2112.50000	-.13600	-.13800	-.13600	-.26000	-.27000	-.26700
.165	5.000	61.70000	2112.29999	-.14000	-.13500	-.13400	-.26800	-.27800	-.26600
.165	10.000	62.20000	2112.29999	-.14200	-.14600	-.14100	-.27500	-.27900	-.27100
.165	14.900	62.20000	2112.10001	-.16900	-.18100	-.16100	-.27800	-.27300	-.25700
.165	19.900	63.40000	2112.50000	-.22300	-.23300	-.20600	-.30000	-.31400	-.28500
GRADIENT	.0405	.0405	-.04920	-.000101	-.00125	-.00067	-.00276	.00033	-.00063

CA578 (MAIL 713) B16 CS F1 .460 W87 E18 (RDV-39) ( US NAV 73 )

## REFERENCE DATA

SREF = 4.4120 Sq.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 387 0 RVAL = 1.20 GRADIENT INTERVAL = -4.00/16.00

MACH	ALPHA	Z(PF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-3.900	60.40000	2111.89999	-.16400	-.16100	-.15100	-.20000	-.26600	-.25000
.165	.000	60.80000	2111.70001	-.14600	-.14700	-.13600	-.22000	-.25100	-.21600
.165	4.900	61.60000	2111.20001	-.14000	-.13800	-.12400	-.23400	-.25100	-.22400
.165	10.000	61.70000	2111.20001	-.14500	-.14900	-.13600	-.24500	-.25300	-.22900
.165	14.900	62.50000	2111.20001	-.17500	-.17700	-.15600	-.25100	-.25700	-.22700
.165	19.900	62.70000	2111.89999	-.23400	-.25000	-.22700	-.26800	-.23400	-.24300
GRADIENT	.0405	.0405	-.03939	-.000361	-.000366	-.00027	-.00284	.00030	-.00062

## PARAMETRIC DATA

BETA = .000 PTVF = 1.000  
 M/B = .286 EDLAF = -18.000  
 ELEVN = -15.000

BETA = .000 PTVF = 1.000  
 M/B = .286 EDLAF = -18.000  
 ELEVN = -15.000

BETA = .000 PTVF = 1.000  
 M/B = .286 EDLAF = -18.000  
 ELEVN = -15.000

## PARAMETRIC DATA



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## TABULATED SOURCE DATA - CA57B

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CASTB (MAIL 713) 816 CS F1 ADJ WAT E18

(RDV243) (LGR NAV 73)

## REFERENCE DATA

**SREF** = 4.4120 SQ.FT. **DREF** = 43.5980 IN.  
**LREF** = 19.2300 IN. **TREF** = .0000 IN.  
**BREF** = 37.9350 IN. **ZREF** = -.4050 IN.  
**SCALE** = .0405

RUN NO. 43/0 RVL = 1.20 GRADIENT INTERVAL = -4.0V 16.0

	ALPHA	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-3.9000	60.50000 2112.2001	-24200	-23600	-30400	-30700	-34100	-12810
.165	-1.0000	60.50000 2112.29999	-13900	-20500	-19400	-28300	-32700	-12510
.165	4.9000	61.70000 2110.70001	-17700	-17000	-17200	-30900	-32300	-12110
.165	9.9000	62.50000 2111.20001	-17100	-16900	-16800	-32900	-32500	-11420
.165	14.9000	62.80000 2111.79549	-18000	-19900	-17900	-35000	-31300	-11620
.165	20.0000	63.20000 2111.62224	-23700	-24600	-35100	-33500	-31510	-11510
<b>GRADIENT</b>	.13362	-.01593	.00297	.01238	.00285	-.00170	.00175	-.00124

CA57B (MAIL 713) 816 CS F1 ADJ WAT E18

(RDV244) (LGR NAV 73)

## REFERENCE DATA

**SREF** = 4.4120 SQ.FT. **DREF** = 43.5980 IN.  
**LREF** = 19.2300 IN. **TREF** = .0000 IN.  
**BREF** = 37.9350 IN. **ZREF** = -.4050 IN.  
**SCALE** = .0405

RUN NO. 44/0 RVL = 1.20 GRADIENT INTERVAL = -4.0V 16.0

	ALPHA	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-4.0000	60.90000 2111.60001	-23000	-22000	-28600	-29300	-31200	-12410
.165	.0000	60.80000 2111.20001	-19200	-19500	-21100	-29200	-31300	-12610
.165	4.9000	61.10000 2111.10001	-17100	-17900	-17400	-30400	-32400	-12110
.165	9.9000	62.30000 2111.39999	-17900	-18600	-17800	-33700	-33100	-11310
.165	14.9000	62.80000 2111.10001	-18300	-22200	-18800	-32600	-29400	-11610
.165	19.9000	63.40000 2111.79999	-26900	-25700	-33900	-35100	-32400	-11510
<b>GRADIENT</b>	.11264	-.01593	.00216	.00127	.00157	-.00321	.00233	-.00365

PARAMETRIC DATA

**BETA** = .000 **H8** = .125 **ELEVN** = -15.000

RUN NO. 44/0 RVL = 1.20 GRADIENT INTERVAL = -4.0V 16.0

	ALPHA	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-3.9000	60.50000 2112.2001	-24200	-23600	-30400	-30700	-34100	-12810
.165	-1.0000	60.50000 2112.29999	-13900	-20500	-19400	-28300	-32700	-12510
.165	4.9000	61.70000 2110.70001	-17700	-17000	-17200	-30900	-32300	-12110
.165	9.9000	62.50000 2111.20001	-17100	-16900	-16800	-32900	-32500	-11420
.165	14.9000	62.80000 2111.79549	-18000	-19900	-17900	-35000	-31300	-11620
.165	20.0000	63.20000 2111.62224	-23700	-24600	-35100	-33500	-31510	-11510
<b>GRADIENT</b>	.13362	-.01593	.00297	.01238	.00285	-.00170	.00175	-.00124

PARAMETRIC DATA

**BETA** = .000 **H8** = .125 **ELEVN** = -15.000

DATE ON CCT 74

TABULATED SOURCE DATA - CAS7B

FAILE 55

## REFERENCE DATA

**S<sub>EFS</sub>** = 4.4120 SF.FT.  
**U<sub>EFS</sub>** = 19.2300 IN.  
**B<sub>EFS</sub>** = 37.9350 IN.  
**SCALE** = .0406

RUN NO. 451 U RAVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	2(FSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
.165	-4.000	60.40000 2111.0000	-1.23400	-1.23100	-1.23600	-1.28400	-1.28100	.12400
.165	.000	61.10000 2110.39999	.39700	.20100	.19900	.19200	.25500	.12500
.165	4.300	61.70000 2110.10001	-1.17000	-1.17200	-1.16500	-1.29300	-1.29700	.12100
.165	9.900	61.30000 2110.20001	-1.17900	-1.18000	-1.17900	-1.26600	-1.29300	.11800
.165	14.900	62.50000 2110.70001	-1.19600	-1.21700	-1.19300	-1.26600	-1.27300	.11500
.165	19.900	63.00000 2110.79999	-1.66900	-1.25700	-1.29100	-1.25700	-1.28600	.11500
GRADIENT		.37970	-.31517	-.30476	.3104	.3192	-.30187	-.30065

CAS7B (NAAI 713) 816 CS F1 J41 W47 E18 (CPTBF) (See N.Y. 73 )

## REFERENCE DATA

**S<sub>EFS</sub>** = 4.4120 SF.FT.  
**U<sub>EFS</sub>** = 19.2300 IN.  
**B<sub>EFS</sub>** = 37.9350 IN.  
**SCALE** = .0406

RUN NO. 461 U RAVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	2(FSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
.165	-4.000	60.50000 2114.70001	-1.25900	-1.24400	-1.31700	-1.23300	-1.30000	.12400
.165	.000	60.50000 2114.29999	-1.24500	-1.21900	-1.31900	-1.26500	-1.27500	.12500
.165	4.900	61.11000 2114.10001	-1.20700	-1.18900	-1.24100	-1.23500	-1.22200	.12100
.165	9.900	61.30000 2114.39999	-1.19600	-1.18100	-1.19900	-1.21400	-1.20900	.11800
.165	14.900	62.70000 2115.00001	-1.23500	-1.22900	-1.23600	-1.23900	-1.21900	.11500
.165	19.900	63.00000 2116.00001	-1.31100	-1.29200	-1.28700	-1.27900	-1.21400	.11500
GRADIENT		.12281	.31623	.31153	.31579	.31533	.31533	.31533

CAS7B (NAAI 713) 816 CS F1 J41 W47 E18 (CPTBF) (See N.Y. 73 )

## PARAMETRIC DATA

DETA	=	.000	F1N1	=	1.300
VB	=	.125	DFCAF	=	-18.000
ELEVN	=	15.000			
CPBF2					
CPBF3					
CPBF4					
CPBF5					

DETA	=	.000	F1N1	=	1.300
VB	=	.125	DFCAF	=	-14.000
ELEVN	=	15.000			
CPBF2					
CPBF3					
CPBF4					
CPBF5					

CA578 (MAIL 713) B16 CS F1 J41 W37 E18

(DYN47) (100 REV 73)

## REFERENCE DATA

**SQFT** = 4,4120 Sq.FT.  
**LF** = .0000 IN.  
**UF** = 19.2300 IN.  
**BDF** = 37.9350 IN.  
**SCALE** = .00005

## PARAMETRIC DATA

**ZETA** = .000 FTNUF = 1.000  
**VB** = .125 DDFNUF = -1.000  
**ELEVON** = 15.000

RUN NO. 477 U RVEL = 1.20 GRADIENT INTERVAL = -4.ULV 16.ULV

	PSTAT	OPTBF1	OPTBF2	OPTBF3	OPTBF4	OPTBF5
ALPHA	.0(FSF)	-1.200	-1.200	-1.200	-1.200	-1.200
1.65	60.7000 2113.79999	-24500	-24500	-24500	-24500	-24500
.165	60.8000 2113.29999	-24600	-24600	-24600	-24600	-24600
.165	61.4000 2113.29999	-21400	-23000	-19500	-28600	-22300
.165	61.9000 2113.50000	-22200	-23600	-14600	-21700	-13600
.165	62.4000 2114.20000	-24500	-24700	-22600	-21600	-24600
.165	62.9000 2115.00000	-31600	-32900	-30000	-28400	-27900
GRADIENT	.000012	.00151	.00012	.00047	.00031	.00031
ANGLE	.02287					

CA578 (MAIL 713) B16 CS F1 J41 W37 E18

(DYN47) (100 REV 73)

## REFERENCE DATA

**SQFT** = 4,4120 Sq.FT.  
**LF** = .0000 IN.  
**UF** = 19.2300 IN.  
**BDF** = 37.9350 IN.  
**SCALE** = .00005

## PARAMETRIC DATA

**ZETA** = .000 FTNUF = 1.000  
**VB** = .039 DDFNUF = -1.000  
**ELEVON** = 15.000

RUN NO. 488 U RVEL = 1.20 GRADIENT INTERVAL AL = -4.ULV 16.ULV

	PSTAT	OPTBF1	OPTBF2	OPTBF3	OPTBF4	OPTBF5
ALPHA	.0(FSF)	-1.18500	-1.18200	-1.18200	-1.18600	-1.18600
1.65	60.5000 2116.20000	-1.19300	-1.22300	-1.24200	-1.24200	-1.27500
.165	60.7000 2116.60000	-1.22300	-1.33300	-1.34200	-1.34200	-1.34500
.165	61.1000 2117.69999	-1.32400	-1.33300	-1.34200	-1.34200	-1.34700
GRADIENT	.00007	.01325	.01436	.01618	.01618	.01618
ANGLE	.017212					

CA578 (MAIL 713) B16 CS F1 J41 W37 E18

(DYN47) (100 REV 73)

## PARAMETRIC DATA

**ZETA** = .000 FTNUF = 1.000  
**VB** = .039 DDFNUF = -1.000  
**ELEVON** = 15.000



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## TABULATED SOURCE DATA - CA578

FACE 57

CA578 (NAAL 713)

B16 C5 F1 J41 W47 E18

(6DVZ49) (LA NIV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 49/0 TINV/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH ALPHA Q(PSF) PSTAT OPTBF1 OPTBF2 OPTBF3 OPTBF1 OPTBF2 OPTBF3  
 .165 .000 64.0000 2115.29999 -.21200 -.22400 -.19100 -.17600 -.23100 -.18700  
 .165 4.900 65.2000 2115.70011 -.22800 -.23200 -.21500 -.19100 -.20400 -.20600  
 .165 9.900 65.1000 2116.89999 -.29500 -.30500 -.30500 -.27700 -.33900 -.33200  
 GRADIENT .051227 .16188 -.010840 -.010226 -.01154 -.011223 -.011788 -.011993 -.012261

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5540 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

CA578 (NAAL 713) B16 C5 F1 J41 W47 E18 (6DVZ50) (LA NIV 73)

## PARAMETRIC DATA

BETA = .000 PTNU/F = 1.000  
 H2 = .039 EDFLAF = -18.000  
 ELEVON = 15.000

RUN NO. 50/0 TINV/L = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH ALPHA Q(PSF) PSTAT OPTBF1 OPTBF2 OPTBF3 OPTBF1 OPTBF2 OPTBF3  
 .165 -4.000 61.80000 2113.89999 -.29500 -.23900 -.31200 -.26700 -.26700 -.26900  
 .165 .000 61.00000 2112.39999 -.24500 -.20300 -.26300 -.24800 -.24800 -.25000  
 .165 4.900 61.50000 2112.00000 -.20200 -.19600 -.21000 -.23700 -.23700 -.24000  
 .165 9.900 62.10000 2112.00000 -.23700 -.19800 -.18400 -.27400 -.27400 -.27900  
 .165 14.900 62.20000 2112.29999 -.25700 -.23000 -.21600 -.32500 -.32500 -.27400  
 .165 19.900 62.60000 2113.20001 -.32200 -.33500 -.29900 -.31200 -.31200 -.27700  
 GRADIENT .04175 -.07236 .00165 .00127 .00117 .00117

## PARAMETRIC DATA

BETA = .000 PTNU/F = 1.000  
 H2 = .245 EDFLAF = 19.000  
 ELEVON = 15.000

BETA = .000 PTNU/F = 1.000  
 H2 = .245 EDFLAF = 19.000  
 ELEVON = 15.000

CA57B (NAAI 713) B16 C5 F1 J41 W47 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. X4RP = 43.5980 IN.  
 LREF = 19.2300 IN. Y4RF = .0000 IN.  
 BREF = 37.9350 IN. Z4RP = -.4050 IN.  
 SCALE = .0405

RUN NO. 51 / 0 ROLL = 1.20 GRADIENT INTERVAL = -4.00V 16.00

MACH	ALPHA	Z(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5	CPTBF6
.165	-4.000	60.90000	2112.20001	-.22600	-.22400	-.22700	-.24200	-.22900	.23600
.165	.000	60.70000	2111.29999	-.22800	-.22600	-.21500	-.22600	-.22200	.23600
.165	4.900	61.30000	2111.00000	-.25000	-.19600	-.19300	-.20500	-.21400	.24200
.165	9.900	61.70000	2111.20001	-.23700	-.20600	-.19300	-.20300	-.20700	.24200
.165	14.900	62.20000	2111.60001	-.25900	-.23600	-.21900	-.21900	-.19300	.27900
.165	19.900	62.90000	2112.50000	-.32700	-.34100	-.30000	-.31700	-.29000	.27700
	GRADIENT	.07646	-.02502	.00001	-.300X2	.00024	.00013	.00333	-.00064

CA57B (NAAI 713) B16 C5 F1 J41 W47 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. X4RF = 43.5980 IN.  
 LREF = 19.2300 IN. Y4CF = .0000 IN.  
 BREF = 37.9350 IN. Z4RP = -.4050 IN.  
 SCALE = .0405

RUN NO. 52 / 0 ROLL = 1.20 GRADIENT INTERVAL = -4.00V 16.00

MACH	ALPHA	Z(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5	CPTBF6
.165	-4.000	60.30000	2110.70001	-.22100	.19500	-.19400	-.23500	-.22900	.23600
.165	.000	60.90000	2110.20001	-.19000	-.17700	-.17900	-.23200	-.23000	.23600
.165	4.900	61.10000	2109.70001	-.16900	-.16700	-.16900	-.21600	-.21700	.24200
.165	9.900	61.80000	2110.10001	-.16900	-.17900	-.17500	-.22400	-.23300	.24200
.165	14.900	62.50000	2110.39999	-.21200	-.20100	-.19400	-.23500	-.23200	.27800
.165	19.900	63.00000	2110.60001	-.27600	-.26100	-.25600	-.27400	-.26100	.27700
	GRADIENT	.11116	-.01305	.00032	-.00036	-.00005	-.02018	-.00053	-.00064

(RDY251) ( USA NAV 73 )

## PARAMETRIC DATA

BETA = .000 FTNF = 1.300  
 R/B = .286 DFLAF = -18.000  
 ELEVCN = 15.000

(RDY252) ( USA NAV 73 )

## PARAMETRIC DATA

BETA = .000 FTNF = 1.300  
 R/B = .286 DFLAF = -18.000  
 ELEVCN = .000

(RDY253) ( USA NAV 73 )

## PARAMETRIC DATA

BETA = .000 FTNF = 1.300  
 R/B = .286 DFLAF = -18.000  
 ELEVCN = .000



DATE 01 OCT 74

TABULATED SOURCE DATA - C4578

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## REFERENCE DATA

SREF = 4.4120 SA.FT. XMRP = 43.5980 IN.  
 URF = 19.2300 IN. YMRP = .0000 IN.  
 BRF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 53V 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	-3.900	60.40000	2109.89999	-.22500	-.19800	-.19300	-.20600	-.22900	-.19700
.165	.000	61.10000	2109.39999	-.19300	-.18600	-.18300	-.20400	-.22300	-.19400
.165	4.900	61.70000	2109.29999	-.19200	-.17900	-.17100	-.19000	-.22400	-.20000
.165	9.900	62.40000	2109.29999	-.19700	-.19000	-.17700	-.21100	-.22400	-.19200
.165	14.900	62.20000	2109.29999	-.21300	-.20400	-.19500	-.22100	-.22500	-.17900
.165	19.900	62.90000	2110.20001	-.28400	-.27600	-.27100	-.26300	-.26600	-.22200
	GRADIENT	.10199	-.02632	.00126	-.00039	-.00030	-.00013	.00031	-.00065

## REFERENCE DATA

SREF = 4.4120 SA.FT. XMRP = 43.5980 IN.  
 URF = 19.2300 IN. YMRP = .0000 IN.  
 BRF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 54V 0 RAVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3
.165	9.900	61.90000	2109.00000	-.21100	-.19900	-.19800	-.35400	-.46100	-.36700
.165	15.000	62.40000	2109.29999	-.22300	-.21600	-.20400	-.37500	-.54500	-.41900
.165	19.900	62.80000	2110.20001	-.30300	-.30300	-.30400	-.30000	-.48800	-.44100
	GRADIENT	.09804	.09803	-.00235	-.00333	-.00118	-.00412	-.01647	-.01062

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF1	CPTBF2	CPTBF3

## REFERENCE DATA

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TABULATED SOURCE DATA - CA57B

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CASTE (NAAL 713)

B16 C5 F1 J41 W47 E18

(FDV255) (1.2 NCV 73 )

## REFERENCE DATA

SREF = .4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 55/ U RVEL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALPHA	Z (PSF)	PSIAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	9.300	62.30000 2108.10001	-21500	-19500	-37000	-46500	-37300
.165	14.900	63.10000 2109.50001	-22200	-21500	-36700	-42700	-42700
.165	19.900	63.30000 2110.70001	-32900	-30400	-31700	-34900	-32900
GRADIENT	.6000	.08004	-.00140	-.00340	-.01240	-.01360	-.01360

CASTFB (NAAL 713) B16 C5 F1 J41 W47 E18

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 56/ U RVEL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALPHA	Z (PSF)	PSIAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	10.000	61.90000 2108.10001	-21600	-20100	-18600	-29300	-33100
.165	15.000	62.20000 2108.89999	-23400	-21500	-20300	-23600	-32200
.165	20.000	62.80000 2109.39999	-29700	-28800	-29900	-21000	-30600
GRADIENT	.06100	.16000	-.00360	-.00280	-.00340	.01140	-.00180

(FDV256) (1.2 NCV 73 )

## REFERENCE DATA

SREF = .4120 SQ.FT. XREF = 43.5980 IN.  
 LREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 56/ U RVEL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALPHA	Z (PSF)	PSIAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	10.000	61.90000 2108.10001	-21600	-20100	-18600	-29300	-33100
.165	15.000	62.20000 2108.89999	-23400	-21500	-20300	-23600	-32200
.165	20.000	62.80000 2109.39999	-29700	-28800	-29900	-21000	-30600
GRADIENT	.06100	.16000	-.00360	-.00280	-.00340	.01140	-.00180

(FDV256) (1.2 NCV 73 )

## REFERENCE DATA

	ALPHA	Z (PSF)	PSIAT	CPTEF1	CPTEF2	CPTEF3	CPTEF4
.165	10.000	61.90000 2108.10001	-21600	-20100	-18600	-29300	-33100
.165	15.000	62.20000 2108.89999	-23400	-21500	-20300	-23600	-32200
.165	20.000	62.80000 2109.39999	-29700	-28800	-29900	-21000	-30600
GRADIENT	.06100	.16000	-.00360	-.00280	-.00340	.01140	-.00180

(FDV256) (1.2 NCV 73 )

(EDV 57) ( DA NAV 73 )

## REFERENCE DATA

	CAS7B (NAAL 713)	B16 CS F1	J41	W37 E18	
SREF =	4.4120 SQ.FT.	XREF =	43.5980 IN.		
UREF =	19.2300 IN.	YREF =	.0000 IN.		
BREF =	37.9350 IN.	ZREF =	-.4050 IN.		
SCALE =	.0405				

RUN NO. 577 0 RVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPEBF1	CPEBF2	CPEBF3
.165	-3.900	60.50000	2107.29999	-.26900	-.25100	-.24400	-.27100	-.27900	-.29300
.165	.000	61.00000	2106.39999	-.22900	-.22200	-.21300	-.24700	-.25100	-.26300
.165	4.900	61.80000	2106.60001	-.21200	-.19500	-.18400	-.25700	-.27200	-.29300
.165	9.900	62.20000	2106.50000	-.20600	-.19300	-.18200	-.25800	-.26300	-.28200
.165	15.000	62.20000	2106.70001	-.21000	-.19400	-.18000	-.23200	-.25900	-.27100
.165	19.900	63.20000	2107.50000	-.20200	-.20700	-.20100	-.28000	-.30200	-.25800
GRADIENT	.09543	-.02109	.00289	.00299	.00299	.00299	.00147	.00055	.00430

(EDV 57) ( DA NAV 73 )

## REFERENCE DATA

	CAS7B (NAAL 713)	B16 CS F1	J41	W37 E18	
SREF =	4.4120 SQ.FT.	XREF =	43.5980 IN.		
UREF =	19.2300 IN.	YREF =	.0000 IN.		
BREF =	37.9350 IN.	ZREF =	-.4050 IN.		
SCALE =	.0405				

(EDV 57) ( DA NAV 73 )

## PARAMETRIC DATA

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPEBF1	CPEBF2	CPEBF3
.165	-3.900	60.30000	2105.70001	-.24600	-.23600	-.22400	-.26700	-.26700	-.23600
.165	.000	60.60000	2105.39999	-.22400	-.21700	-.20400	-.25400	-.25400	-.23600
.165	4.900	61.10000	2105.10001	-.20600	-.19600	-.18500	-.24800	-.25600	-.22100
.165	10.000	61.90000	2105.39999	-.20200	-.19100	-.18200	-.23400	-.23400	-.19700
.165	15.000	62.30000	2105.89399	-.21100	-.19900	-.19200	-.23400	-.23400	-.18700
.165	19.900	62.70000	2105.50000	-.20600	-.20200	-.20200	-.24200	-.24200	-.21200
GRADIENT	.13693	.00987	.00174	.00204	.00174	.00198	.00171	.00171	.00065

(EDV 57) ( DA NAV 73 )

## PARAMETRIC DATA

(EDV 57) ( DA NAV 73 )

DATE 01 OCT 74

## TABULATED SOURCE DATA - CA57B

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CA57B (MAAL 713)

(FDVZ59) ( J.R. NEV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 5910 RNL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPBF3	CPBF2	CPBF1	BETA
.165	-4.000	61.10000	2112.50000	-.21200	-.21600	-.20100	-.27100	-.25400	-.23000	.12400
.165	.000	61.10000	2111.89999	-.19300	-.18900	-.17600	-.28100	-.25400	-.23200	.12500
.165	4.900	61.90000	2111.89999	-.18300	-.17500	-.17100	-.29300	-.26500	-.23500	.12100
.165	9.900	61.40000	2110.89999	-.16000	-.15900	-.14300	-.29500	-.24200	-.21800	.11800
.165	14.900	62.50000	2111.39999	-.15200	-.15300	-.13800	-.27700	-.24100	-.20900	.11600
.165	19.900	63.10000	2112.20001	-.15100	-.15500	-.14100	-.32100	-.24700	-.21100	.11500
GRADIENT	.06537	-.06635	.00328	.003324	.003331	-.00061	.00032	.000325	-.00032	-.000325

CA57B (MAAL 713) B16 C5 F1 J42 W47 E18

(FDVZ60) ( J.R. NEV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 6010 RNL = 1.20 GRADIENT INTERVAL = -4.00 16.00

MACH	ALPHA	Q(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPBF3	CPBF2	CPBF1	BETA
.165	-4.000	60.70000	2111.50000	-.20300	-.20600	-.19400	-.28600	-.26500	-.23300	.12400
.165	.000	60.60000	2111.29999	-.18600	-.18900	-.17700	-.28200	-.26100	-.23300	.12500
.165	4.900	61.50000	2111.20001	-.16700	-.16600	-.16200	-.33000	-.27600	-.21400	.12100
.165	9.900	61.80000	2111.20001	-.16500	-.17400	-.15800	-.33000	-.26600	-.21900	.11800
.165	14.900	62.50000	2111.70001	-.17400	-.17100	-.15700	-.34300	-.29100	-.22300	.11600
.165	19.900	63.50000	2112.20001	-.19500	-.19600	-.18600	-.35300	-.31600	-.24500	.11500
GRADIENT	.00146	.00146	.00161	.00161	.00174	.00192	-.00124	-.00121	-.00146	-.00165

(FDVZ60) ( J.R. NEV 73 )

H/B = .125 EDFLAF = -.13.000  
 ELEVCN = .000

H/B = .125 EDFLAF = -.13.000  
 ELEVCN = .000

H/B = .125 EDFLAF = -.13.000  
 ELEVCN = .000



CA57B (NAAL 713)

B16 C5 F1 J42 W47 E18

(RDVZ61) ( 3A NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 UREF = 19.2300 IN. YMRF = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 61/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Z(PSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
.165	-4.000	60.30000	2110.60001	-.20000	-.19100	-.19800	-.26400	-.26100
.165	'000	60.90000	2110.50000	-.19000	-.19500	-.19300	-.28300	-.26400
.165	4.900	61.40000	2110.10001	-.16600	-.17800	-.18600	-.29500	-.27400
.165	9.400	62.30000	2110.60001	-.18400	-.18900	-.18200	-.31300	-.26900
.165	14.900	62.70000	2111.10001	-.19400	-.19800	-.18600	-.38300	-.27900
.165	19.900	62.80000	2111.79999	-.26100	-.27100	-.25500	-.42300	-.33900
	GRADIENT	.12985	.02419	.00033	-.00018	.00081	-.00566	-.00119

(RDVZ62) ( 3A NAV 73 )

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMRP = 43.5980 IN.  
 UREF = 19.2300 IN. YMRF = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0405

RUN NO. 62/ 0 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Z(PSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
.165	9.900	62.10000	2111.50000	-.18700	-.19400	-.18000	-.68700	-.53000
.165	14.900	62.40000	2111.20001	-.27500	-.52800	-.13200	-.13800	-.13100
.165	19.900	63.50000	2112.00000	-.13500	-.13600	-.15200	-.38100	-.75900
	GRADIENT	.06000	-.06000	-.01760	-.06800	.00960	.15980	.07930

(RDVZ63) ( 3A NAV 73 )

## PARAMETRIC DATA

BETA = .000 PTV/P = 1.000  
 V/B = .125 EDFLAF = -18.000  
 ELEV.N = .000

H/E

BETA = .000 PTV/P = 1.000  
 V/B = .125 EDFLAF = -18.000  
 ELEV.N = .000

(RDVZ64) ( 3A NAV 73 )

## PARAMETRIC DATA

BETA = .000 PTV/P = 1.540  
 V/B = .039 EDFLAF = -18.000  
 ELEV.N = .000

H/E

BETA = .000 PTV/P = 1.540  
 V/B = .039 EDFLAF = -18.000  
 ELEV.N = .000

(RDVZ65) ( 3A NAV 73 )

## PARAMETRIC DATA

(DVZ63) (18 NOV 73)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

RUN NO. 6310 RFL = 1.20 GRADIENT INTERVAL = -4.00V 16.0U

MACH	ALPHA	P(FSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
.165	9.900	62.27000 2111.11000	-.18000	-.19200	-.17800	-.53600	-.49100	-.35700
.165	14.900	62.80000 2111.39999	-.16700	-.17600	-.17200	-.46900	-.50500	-.41100
.165	19.900	63.40000 2112.83999	-.19100	-.20400	-.20300	-.50200	-.51100	-.41000
GRADIENT	.12000	.36000	.02660	.02320	.00120	.01340	-.01480	-.01160

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XREF = 43.5980 IN.  
 UREF = 19.2300 IN. YREF = .0000 IN.  
 BREF = 37.9350 IN. ZREF = -.4150 IN.  
 SCALE = .0405

RUN NO. 6410 RFL = 1.20 GRADIENT INTERVAL = -4.00V 16.0U

MACH	ALPHA	P(FSF)	PSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
.165	9.900	61.70000 2110.20001	-.16700	-.19200	-.17900	-.41100	-.34100	-.26200
.165	15.000	62.60000 2110.79999	-.18000	-.20100	-.19600	-.35000	-.37200	-.29300
.165	19.900	62.90000 2112.20001	-.28600	-.29400	-.34200	-.35100	-.41400	-.25700
GRADIENT	.17647	.11767	.00020	-.00176	-.00333	.01196	-.01627	-.01133

## PARAMETRIC DATA

BETA	V2	F1N/F	F1L/F	CPTBF3	CPTBF2	CPTBF1	CPTBF5	CPTBF4
.039	.039	.039	.039	-.35700	-.49100	-.53600	-.35700	-.49100
ELEV/N	.040	.040	.040	-.35700	-.46900	-.46900	-.35700	-.46900

(DVZ64) (18 NOV 73)

## PARAMETRIC DATA

BETA	V2	F1N/F	F1L/F	CPTBF3	CPTBF2	CPTBF1	CPTBF5	CPTBF4
.039	.039	.039	.039	-.35700	-.37200	-.41400	-.35700	-.37200
ELEV/N	.040	.040	.040	-.35700	-.35100	-.41400	-.35700	-.35100

(DVZ64) (18 NOV 73)

REFERENCE DATA

SREF = 4.4120 S.L.F.T. XREF = 43.5980 IN.  
 UGF = 19.2300 IN. YREF = .0000 IN.  
 EGF = 37.9350 IN. ZREF = -.4050 IN.  
 SCALE = .0405

PARAMETRIC DATA

(C4578) (USA NAV 73)

CAS7B (INAL 713) B16 CS F1 J42 W87 E18

(C4578)

RUN NO. 65/ 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALP-4A	Q(FSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
MACH	.165	61.60000	2116.70001	-1.23000	-1.39400	-1.29900	-1.24800	-1.25100
	.165	61.60000	2116.50000	-1.18600	-1.19500	-1.19600	-1.20000	-1.23500
	.165	61.30000	2116.20001	-1.17900	-1.17900	-1.16300	-1.23400	-1.25200
	.165	62.10000	2116.29999	-1.16400	-1.16400	-1.14900	-1.31400	-1.24100
	.165	62.10000	2115.89999	-1.15600	-1.15600	-1.14000	-1.31400	-1.23200
	.165	62.40000	2115.89999	-1.15000	-1.15000	-1.14000	-1.29400	-1.23600
	.165	63.00000	2116.75001	-1.15200	-1.15400	-1.15100	-1.38600	-1.24500
GRADIENT	.09128	-.04621	.00355	.02299	.00347	.00333	.00339	.00337

REFERENCE DATA

(C4578) (USA NAV 73)

PARAMETRIC DATA

(C4578)

CAS7B (INAL 713) B16 CS F1 J42 W87 E18

(C4578)

RUN NO. 66/ 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALP-4A	Q(FSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
MACH	.165	61.10000	2116.39999	-1.23200	-1.19200	-1.19200	-1.26200	-1.25400
	.165	60.70000	2116.20001	-1.18000	-1.18700	-1.18700	-1.26300	-1.25500
	.165	61.20000	2116.00000	-1.16300	-1.16300	-1.16300	-1.26400	-1.26700
	.165	61.90000	2115.89999	-1.15300	-1.16300	-1.15100	-1.32000	-1.26100
	.165	62.50000	2115.89999	-1.15300	-1.16900	-1.15600	-1.32400	-1.26900
	.165	63.50000	2116.89999	-1.15700	-1.19400	-1.15400	-1.40100	-1.31500
GRADIENT	.08357	-.02692	.02231	.01448	.01251	.01148	.00396	.00337

REFERENCE DATA

(C4578) (USA NAV 73)

PARAMETRIC DATA

(C4578)

CAS7B (INAL 713) B16 CS F1 J42 W87 E18

(C4578)

RUN NO. 66/ 0 RVL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

	ALP-4A	Q(FSF)	FSTAT	CPTBF1	CPTBF2	CPTBF3	CPTBF4	CPTBF5
MACH	.165	61.60000	2116.70001	-1.23000	-1.39400	-1.29900	-1.24800	-1.25100
	.165	61.60000	2116.50000	-1.18600	-1.19500	-1.19600	-1.20000	-1.23500
	.165	61.30000	2116.20001	-1.17900	-1.17900	-1.16300	-1.23400	-1.25200
	.165	62.10000	2115.89999	-1.16400	-1.16400	-1.14900	-1.31400	-1.24100
	.165	62.10000	2115.89999	-1.15600	-1.15600	-1.14000	-1.31400	-1.23200
	.165	62.40000	2115.89999	-1.15000	-1.15000	-1.14000	-1.29400	-1.23600
	.165	63.00000	2116.75001	-1.15200	-1.15400	-1.15100	-1.38600	-1.24500
GRADIENT	.09128	-.04621	.00355	.02299	.00347	.00333	.00339	.00337

DATE UI CCI 74

VALIDATED SECURITY DATA - CASES

FACE

CA57B (MM 713)

SQ.FT.	=	4,412	SQ.FT.	=	365P	=	43,5980	IN.
L.FT.	=	19,2300	IN.	14P	=	3,000	IN.	
H.FT.	=	37,9350	IN.	2P	=	-4,050	IN.	

PRACTICAL DATA

卷之三

BETA	=	.001	FITN F	=	1.000
H/B	=	.124	DFLAF	=	-1.124
ELEVON	=	.001			

GRADIENT INTERVALS

卷之二

MACH	ALPHA	$\Delta(FSF)$	FSTAT	OPTF1	OPTF2	OPTF3	CFEF1	CFEF2	CFEF3	CFEF4
.165	-4.000	60.30000	2115.50000	-1.19700	-2.24000	-1.19500	-2.26200	-2.26500	-2.33500	-1.12400
.165	.000	61.00000	2115.60000	-2.25600	-1.18900	-1.19700	-2.20100	-2.24900	-2.35100	-1.12500
4.900	61.10000	2115.29999	-1.17900	-1.17100	-1.18000	-1.18200	-2.21000	-2.23400	-2.34700	-1.12100
.165	9.000	61.00000	2115.29999	-1.17400	-1.18300	-1.16000	-1.31300	-2.25500	-2.35700	-1.11400
.165	14.900	61.00000	2115.29999	-1.18600	-2.20000	-1.19300	-1.38600	-2.29300	-2.45100	-1.11600
.165	19.900	63.00000	2116.29999	-2.21100	-2.24100	-1.41300	-3.14000	-2.38600	-2.50600	-1.11500
GRADIENT	.13459	.00663	.00111	.00024	.00068	.00026	.00023	.00023	.00023	.00023

AIA EVIDENCE

卷之三

FATIGUE DATA	
ETIA	= .003
~VB	= .246
ELEVON	= .000

卷之三

	<i>ALPHA</i>	<i>3IPSF</i>	<i>FSTAT</i>	<i>GPTEF1</i>	<i>GPTEF2</i>	<i>GPTEF3</i>	<i>GPEEF1</i>	<i>GPEEF2</i>	<i>GPEEF3</i>
MACH	-3.940	60.20000	2117.70001	-17000	-15800	-16200	-29100	-24200	-29400
	.165	.000	61.70000	2117.70001	-16300	-15500	-16300	-21100	-23600
	.165	.49000	61.80000	2117.10001	-15400	-14200	-14200	-20400	-22200
	.165	9.900	61.70000	2116.39999	-13600	-13200	-12900	-27700	-21300
	.165	15.900	63.10000	2117.10001	-15900	-15600	-15100	-31700	-28500
	.165	19.900	62.90000	2117.10001	-16100	-17200	-16200	-36400	-26500
GRADIENT	.11948	-1.65200	.000096	.00045	.00128	.00194	.00047	.00047	.00046

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DATE OF CCT 74

TABULATED SOURCE DATA - CAS7B

FACE 67

CAS7B INAL 713) 816 CS F1 J42 W47 E18

(WCV-7,J) (N,V-7,J)

## REFERENCE DATA

SREF = 4,4120 SQ.FT. XREF = 43,5980 IN.  
 UREF = 19,2300 IN. YREF = .0000 IN.  
 BREF = 37,9350 IN. ZREF = -.4460 IN.  
 SCALE = .0405

RUN NO. 65/0 RAVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

	OPTF1	FSTAT	OPTF2	OPTF3	OPTF4	OPTF5
MAOH	ALPHA	2.0(FSF)	-1.16200	-1.16700	-1.26600	-1.24700
.165	-3.200	60.50000 2117.10001	-1.16900	-1.15100	-1.15600	-1.23400
.165	.000	60.70000 2116.79999	-1.15900	-1.14200	-1.14900	-1.24200
.165	5.000	61.40000 2116.50000	-1.15000	-1.15000	-1.14300	-1.25200
.165	10.000	61.70000 2116.50000	-1.14800	-1.15000	-1.14500	-1.24900
.165	14.900	63.00000 2116.70001	-1.16100	-1.17500	-1.16600	-1.26000
.165	19.900	63.00000 2117.10001	-1.15200	-1.19500	-1.18100	-1.25500
GRADIENT	.13483	-.02237	.00064	-.00357	.00035	-.00034

CAS7B INAL 713) 816 CS F1 J42 W47 E18

(WCV-7,J) (N,V-7,J)

## REFERENCE DATA

SREF = 4,4120 SQ.FT. XREF = 43,5980 IN.  
 UREF = 19,2300 IN. YREF = .0000 IN.  
 BREF = 37,9350 IN. ZREF = -.4450 IN.  
 SCALE = .0406

RUN NO. 70/0 RAVL = 1.20 GRADIENT INTERVAL = -4.00 16.00

	OPTF1	FSTAT	OPTF2	OPTF3	OPTF4	OPTF5
MAOH	ALPHA	4(FSF)	-1.17100	-1.16600	-1.17400	-1.25800
.165	-4.000	60.50000 2116.29999	-1.17300	-1.15700	-1.15600	-1.25000
.165	.000	61.20000 2115.89999	-1.15900	-1.15900	-1.15900	-1.24200
.165	4.900	61.30000 2115.70001	-1.15900	-1.16500	-1.15900	-1.24400
.165	9.800	61.70000 2115.50000	-1.16100	-1.16500	-1.15000	-1.24100
.165	14.900	62.20000 2115.70001	-1.17200	-1.18200	-1.17400	-1.24700
.165	19.900	63.10000 2116.39999	-1.22000	-1.23400	-1.22300	-1.23400
GRADIENT	.02574	-.13274	-.00012	-.00007	-.00009	-.00003

## PARAMETRIC DATA

DETA = .000 FTR/F = 1.000  
 VB = .246 CFFAF = -14.000  
 ELEVON = .000

DETA = .000 FTR/F = 1.000  
 VB = .246 CFFAF = -14.000  
 ELEVON = .000

DETA = .000 FTR/F = 1.000  
 VB = .246 CFFAF = -14.000  
 ELEVON = .000

DETA = .000 FTR/F = 1.000  
 VB = .246 CFFAF = -14.000  
 ELEVON = .000

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TABULATED SOURCE DATA - CAS7B

FADE = 1

CAS7B (MAA) 713) B12 C5 J42 W17 F18

(DOV71) (LAD N.E. 73 )

## REFERENCE DATA

SQFT = 4.4120 SQ.FT. MGRF = 43.5900 IN.  
 LREF = 19.2300 IN. TMRF = .0000 IN.  
 BREF = 37.9350 IN. ZRPF = -.4050 IN.  
 SCALE = .0405

RUN NO. 7170 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q (PSF)	PSTAT	OPBAE1	OPBAE2	OPBAE3	OPSAE1	OPSAE2	OPSAE3
.165	-4.000	60.60000 2117.70001	-27000	-.01300	-.00100	-.17400	-.15600	-.14600	.29400
.165	.000	60.60000 2116.70001	-25300	-.01600	.01200	-.16200	-.13500	-.12700	.24400
.165	4.000	61.00000 2116.39999	-23400	-.01600	.01200	-.17600	-.12900	-.12200	.24200
.165	10.000	61.40000 2116.39999	-24200	-.01200	.01800	-.19200	-.12100	-.11700	.27900
.165	14.000	62.00000 2116.39999	-25500	-.01600	.00900	-.20900	-.13600	-.13400	.26200
.165	19.000	63.00000 2117.50000	-26300	-.01600	.00600	-.25400	-.16100	-.15400	.27700
GRADIENT		.10400	-.16879	.00003	.00286	.00065	-.13134	-.13131	-.13131

CAS7B (MAA) 713) B12 C5 J42 W17 F18

(DOV72) (LAD N.E. 73 )

## REFERENCE DATA

SQFT = 4.4120 SQ.FT. MGRF = 10.3800 IN.  
 LREF = 19.2300 IN. TMRF = .0000 IN.  
 BREF = 37.9350 IN. ZRPF = -.4050 IN.  
 SCALE = .0405

RUN NO. 7270 RFL = 1.20 GRADIENT INTERVAL = -4.00/ 16.00

MACH	ALPHA	Q (PSF)	PSTAT	OPBAE1	OPBAE2	OPBAE3	OPSAE1	OPSAE2	OPSAE3
.165	-3.900	60.80000 2116.10001	-22100	-.017800	-.01600	-.24500	-.13500	-.11600	.24900
.165	.000	61.20000 2115.79999	-22300	-.06400	-.00500	-.23600	-.13600	-.12400	.24400
.165	4.900	61.30000 2115.60001	-22800	-.05700	-.00600	-.24000	-.13500	-.12500	.24200
.165	10.900	61.80000 2115.10001	-21500	-.03500	.00200	-.21900	-.12300	-.10500	.27900
.165	14.900	63.00000 2115.29999	-22500	-.03700	.00100	-.21100	-.14200	-.13400	.27100
.165	19.900	62.90000 2116.20001	-23600	-.06400	-.02300	-.21500	-.19300	-.19300	.27700
GRADIENT		.10566	-.14801	-.00006	.00232	.00085	.00439	-.00032	-.00032

## PARAMETRIC DATA

DETA = .000 RFL = 1.20 FTWF = 1.000  
 HFB = .246 ELEVN = .000

## PARAMETRIC DATA

DETA = .000 RFL = 1.20 FTWF = 1.000  
 HFB = .246 ELEVN = .000

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TABLED SOURCE DATA - DATA

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## CAST-B 810CSF1 JAO MATE10 WING LOWER SURFACE

(ADN03) (112 NOV 73)

## REFERENCE DATA

SPAN =	4.1120 E+11.	XMAP =	45.9840 IN.	BETA =	.000
LEOF =	16.3300 IN.	YMAP =	.0000 IN.	H/B =	.039
SWF =	37.9340 IN.	ZMAP =	-.4050 IN.	ELEV0 =	.000
SCALE =	.0405				

MACH 1 (1) = .165 ALPH1 (1) = 10.000 RFL = 1.2000 MACH = .165

## SECTION 111WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
2170						
.000	.4235	.4637	.2055	.3617	.3032	.2050
.334	.4581	.1421	.3194	.3656	.2176	.1505
.520	.0744	.2504	.2491	.1368	.1633	.0940
.663	.4404	.2136	.0313	-.0344	-.2121	.0566
.873	.3555	.0444	-.1042	-.2215	-.2967	-.1729

MACH 1 (1) = .165 ALPH1 (2) = 15.025 RFL = 1.2000 MACH = .165

## SECTION 111WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
2170						
.000	.4696	.5120	.3975	.4732	.4400	.3272
.334	.5130	.2332	.3762	.4509	.3264	.2465
.520	.1267	.2002	.2358	.2986	.2462	.1239
.663	.3885	.2035	.1307	.0411	-.0223	.0490
.873	.3914	.1456	-.0132	-.1668	-.2720	-.2194

MACH 1 (1) = .165 ALPH1 (3) = 20.045 RFL = 1.2000 MACH = .165

## SECTION 111WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
2170						
.000	.5463	.3797	.4373	.5460	.5332	.4411
.334	.5806	.3505	.4001	.5203	.3964	.2977
.520	.2536	.3474	.2776	.2301	.2452	.0622
.663	.6268	.3929	.2240	.1755	-.0603	-.0019
.873	.4687	.2674	.1383	-.1086	-.2070	-.2202

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TABULATED SOURCE DATA - CASTA

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CASTA 816CSF1 JAO WTE10 WING LOWER SURFACE

(RDYLD1) (12 NOV 73)

## REFERENCE DATA

	CASTA	CASTA	CASTA	CASTA	CASTA	CASTA
BDF	4.4120 04.FT.	.2048P	=	43.5940 IN.		
LDF	19.2200 IN.	.1948P	=	.0000 IN.		
BDF	37.9350 IN.	.2048P	=	-.4050 IN.		
SCALE	.0403					

MACH (1) = .165 ALPHA (1) = 10.000 RHL = 1.200 MACH = .165

## SECTION 11 WING

## DEPENDENT VARIABLE CP

	2/18	2/18	2/18	2/18	2/18	2/18
N/C	.0000	.42515	.46682	.2670	.4157	.4257
	.324	.47115	.2089	.4087	.4850	.3905
	.320	.1036	.2930	.3280	.2709	.2046
	.043	.3359	.2033	.0756	.1299	.0647
	.073	.3264	-.0360	-.1793	-.3281	-.3718

MACH (1) = .165 ALPHA (2) = 15.050 RHL = 1.200 MACH = .165

## SECTION 11 WING

## DEPENDENT VARIABLE CP

	2/18	2/18	2/18	2/18	2/18	2/18
N/C	.0000	.4932	.5122	.5345	.4867	.5912
	.324	.5211	.2031	.4990	.5640	.4845
	.320	.0209	.3269	.3228	.2372	.3352
	.043	.1945	.2721	.1832	.1916	.1254
	.073	.4022	-.1517	-.0198	-.1797	-.2906

MACH (1) = .165 ALPHA (3) = 19.960 RHL = 1.200 MACH = .165

## SECTION 11 WING

## DEPENDENT VARIABLE CP

	2/18	2/18	2/18	2/18	2/18	2/18
N/C	.0000	.55113	.5792	.4411	.5251	.6540
	.324	.5870	.4001	.5396	.6158	.5375
	.320	.2481	.3331	.3447	.2761	.3551
	.043	.4762	.3812	.2249	.2109	.0365
	.073	.4751	.2487	.1244	-.0467	-.1989

MACH (1) = .165 ALPHA (4) = 19.960 RHL = 1.200 MACH = .165

## PARAMETRIC DATA

BETA	=	.000	PIMP =	1.300
A/B	=	.039	BDFLAP =	-10.000
ELEVON	=	.000		

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TABLEU, TLU SOURCE DATA - DATA 18

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## OK57-B 816CSF1 JAO WTE10 WING LOWER SURFACE

(ADPLO3) (112 MON 73)

## REFERENCE DATA

SREF	=	4.4120 94.1 FT.	XMAP =	43.940 IN.
LREF	=	19.2500 IN.	XMAP =	.0000 IN.
BREF	=	37.9350 IN.	ZMAP =	-.4050 IN.
SCALE	=	.0405		

MACH (1) = .165 ALPHA (1) = .9.990 RNL = 1.200 MACH = .165

## SECTION 1 WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/8						
.000	.4417	.4247	.2446	.4083	.4810	.5551
.334	.4708	.2091	.4253	.5308	.3988	.2794
.520	.0598	.2657	.3508	.5641	.2869	.0769
.663	.2532	.1383	.0653	.1346	-.0286	.0983
.873	.3430	-.0111	-.1430	-.3145	-.3348	-.1593

MACH (1) = .165 ALPHA (2) = 15.000 RNL = 1.200 MACH = .165

## SECTION 1 WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/8						
.000	.4862	.4768	.3684	.4939	.6377	.7821
.334	.5122	.2631	.4579	.5996	.5763	.4753
.520	-.1683	.2477	.3263	.2296	.4424	.2353
.663	-.3663	.2002	.1686	.1673	.0146	.1603
.873	.4041	.1359	-.0194	-.1677	-.2713	-.1636

MACH (1) = .165 ALPHA (3) = 20.000 RNL = 1.200 MACH = .165

## SECTION 1 WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/8						
.000	.5549	.5352	.4227	.5570	.6832	.8993
.334	.5816	.3973	.5246	.6463	.6290	.5218
.520	.2223	.2956	.3338	.3120	.3865	.2029
.663	.4177	.4401	.2694	.2015	-.0007	.0886
.873	.4644	.2281	.0945	-.0753	-.2470	-.1614

## OA57-B B16C5F1 JAO WATE10 WING LOWER SURFACE

(ADVLOG) ( 12 NOV 73 )

## REFERENCE DATA

REF	=	4.4120 IN. FT.	XMAP =	43.5940 IN.	BETA =	= .000	PIN/P =	1.000
LREF	=	19.2300 IN.	YMAP =	.0000 IN.	H/B =	.125	BDFLAP =	-18.000
BREF	=	37.9350 IN.	ZMAP =	-.4030 IN.	ELEVON =	.000		
SCALE	=	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.980 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/18	.000	.0505	-.6191	-.6472	-.3566	-.2643	-.2104
	.334	-.3923	-.9395	-.4980	-.3047	-.2871	-.1878
	.520	-.5676	-.3627	-.2302	-.2594	-.2514	-.1367
	.663	-.4604	-.2460	-.2331	-.2352	-.3514	-.0729
	.873	-.0944	-.2504	-.2622	-.3371	-.2803	-.0264

MACH ( 1 ) = .165 ALPHA ( 2 ) = .010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/18	.000	.1488	-.2918	-.4548	-.1765	-.1773	-.1440
	.334	-.0981	-.6042	-.3613	-.1323	-.2474	-.1301
	.520	-.3822	-.1831	-.0732	-.1309	-.1688	-.0756
	.663	-.1258	-.0134	-.0557	-.0992	-.2908	-.0565
	.873	.2236	-.0864	-.1400	-.2242	-.1759	-.0247

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.980 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/18	.000	.2433	-.0118	-.2213	.0029	-.0405	-.0553
	.334	.1528	-.2772	-.0920	.0049	-.1339	-.0742
	.520	-.3014	-.0099	.0611	-.0768	-.1873	-.1518
	.663	.2120	.2250	.1267	.0392	-.2123	-.0326
	.873	.3703	.0748	-.0144	-.1407	-.1607	-.0700



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TABULATED SOURCE DATA - QAS7B

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CA57-B B16CSF1 J40 WTE18 WING LOWER SURFACE

(ADVL07) ( 12 NOV 73 )

## REFERENCE DATA

SREF	=	4.4120 82.FT.	XMAP	=	43.5940 IN.
LREF	=	19.2300 IN.	TMAP	=	.0000 IN.
LREF	=	37.6350 IN.	TMAP	=	-.4050 IN.
SCALE	=	.0003			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.960 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	.0277	-.8789	-.9019	-5152	-.3134	-.1778
.334	-.5076	-1.1843	-.6515	-.3929	-.3895	-.2027
.520	-.7981	-.5926	-.3525	-.2939	-.2699	-.1103
.663	-.5619	-.2739	-.2620	-.2700	-.3807	-.1022
.873	-.1279	-.2765	-.3025	-.3420	-.2962	-.0516

MACH ( 1 ) = .165 ALPHA ( 2 ) = .005 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	.1394	-.4165	-.6015	-.2717	-.2154	-.1453
.334	-.1411	-.7149	-.3934	-.2011	-.3071	-.1560
.520	-.4972	-.2591	-.1231	-.1641	-.2036	-.0767
.663	-.1937	-.0537	-.0825	-.1224	-.3140	-.0665
.873	.1774	-.0831	-.1456	-.2249	-.1910	-.0017

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.995 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	.2430	-.0818	-.2957	-.0469	-.0533	-.0466
.334	.1361	-.3332	-.1177	-.0026	-.1485	-.0827
.520	-.2399	-.0392	.0456	-.0854	-.1966	-.1524
.663	.1610	.1847	.0850	-.0010	-.2652	-.1054
.873	.3502	.0610	-.0266	-.1367	-.1498	-.0724

## PARAMETRIC DATA

BETA	=	.000	PIN/P	=	1.300
H/B	=	.125	BDFLAP	=	-.16.000
ELEVON	=	.000			

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## TABULATED SOURCE DATA - Q4570

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		Q457-B B16CF1 J40 WATE10 WING LOWER SURFACE				(ADW.07)	
MACH ( 1 ) =	.165	ALPHA ( 4 ) =	.075	RN/L =	1.200	MACH =	.165
<b>SECTION ( 1 ) WING</b>							
DEPENDENT VARIABLE CP							
X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/B							
.000	.3657	.1623	-.1073	.1093	.0821	.0188	
.334	.2743	-.0988	.0568	.1515	.0008	.0076	
.520	-.1119	.1648	.1368	-.0200	-.1699	-.1156	
.663	.3508	.3340	.2161	.0952	.2155	.0899	
.873	.4414	.1578	.0584	-.0553	-.1742	-.0193	
MACH ( 1 ) =	.165	ALPHA ( 4 ) =	.075	RN/L =	1.200	MACH =	.165
<b>SECTION ( 1 ) WING</b>							
X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/B							
.000	.4063	.2317	.0171	.2193	.1775	.0756	
.334	.3758	.0430	.1632	.2532	.0758	.0518	
.520	-.0356	.1040	.0918	.0562	-.0276	-.0506	
.663	.2366	.4736	.3208	.1864	-.1583	-.0633	
.873	.4416	.2400	.1234	-.0020	-.1353	-.0218	
MACH ( 1 ) =	.165	ALPHA ( 4 ) =	.075	RN/L =	1.200	MACH =	.165
<b>SECTION ( 1 ) WING</b>							
X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/B							
.000	.5014	.3838	.1451	.3182	.2587	.1177	
.334	.4848	.2151	.2954	.3240	.1349	.0676	
.520	.2405	.1476	.0894	.0846	-.0313	-.1541	
.663	.6714	.5398	.3327	.1641	-.2613	-.2550	
.873	.5168	.3508	.2145	.0678	-.1067	-.0389	

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TABULATED SOURCE DATA - OA578

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OA57-B B16C5F1 J40 WATE10 WING LOWER SURFACE

(REV.06) ( 12 NOV 73 )

## REFERENCE DATA

BREF	4.4120 IN.	XMAP	=	43.5940 IN.	BETA	=	.000	PIN/P	=	1.300
LREF	.19.2500 IN.	YMAP	=	.0000 IN.	H/B	=	.125	BDFLAP	=	-16.000
BREF	.37.0150 IN.	ZMAP	=	-.4050 IN.	ELEVON	=	.000			
SCALE	.0405									

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (1) = -4.015 \quad R/V_L = 1.200 \quad \text{MACH} = .165$$

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

2/18

.000	-.0137	-1.0455	-1.1210	-.6626	-.4034	-.2177
.334	-.8249	-1.4036	-.8751	-.5101	-.4986	-.2610
.520	-.9599	-.7160	-.4753	-.3825	-.2921	-.2552
.663	-.7087	-.4180	-.3683	-.3249	-.4403	-.1630
.873	-.2055	-.3312	-.3468	-.3757	-.3262	-.0788

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (2) = -.030 \quad R/V_L = 1.200 \quad \text{MACH} = .165$$

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

2/18

.000	.1141	-.5022	-.7432	-.3907	-.2829	-.1674
.334	-.2157	-.7636	-.5381	-.2988	-.3639	-.1959
.520	-.6465	-.3612	-.2165	-.2381	-.2030	-.2503
.663	-.2986	-.1221	-.1362	-.1581	-.1800	-.1288
.873	.1132	-.1207	-.1782	-.2408	-.2162	-.0361

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (3) = 4.990 \quad R/V_L = 1.200 \quad \text{MACH} = .165$$

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

2/18

.000	.2471	-.1281	-.3698	-.1155	-.0925	-.0516
.334	-.1101	-.3791	-.1874	-.0877	-.1859	-.1132
.520	-.2722	-.0736	.0871	-.1282	-.2256	-.3190
.663	.1086	.1573	.0876	.0138	-.3037	-.1462
.873	.3518	.0643	-.0243	-.1197	-.1402	-.0972



REFERENCE DATA						PARAMETRIC DATA		
SECTION ( 1 )WING			DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500	RH/L	=	MACH = .165
MACH ( 1 ) = .165	ALPHA ( 1 ) = -3.965					BETA = .000	RIN/P = .286	(PDYL15) ( 12 NOV 73 )
LREF = 4.4120 IN.	XMAP = 43.5940 IN.					N/B = .000	BDFLAP = .000	
LREF = 19.2300 IN.	YMAP = .0000 IN.					ELEVON = .000		
BREF = 37.9350 IN.	ZMAP = -.4050 IN.							
SCALE = .0405								
MACH ( 1 ) = .165	ALPHA ( 2 ) = .015							
SECTION ( 1 )WING	DEPENDENT VARIABLE CP							
X/C	.1500	.3000	.4500	.6000	.7500	RH/L	=	MACH = .165
MACH ( 1 ) = .165	ALPHA ( 2 ) = .015							
SECTION ( 1 )WING	DEPENDENT VARIABLE CP							
X/C	.1500	.3000	.4500	.6000	.7500	RH/L	=	MACH = .165
MACH ( 1 ) = .165	ALPHA ( 3 ) = 5.000							
SECTION ( 1 )WING	DEPENDENT VARIABLE CP							
X/C	.1500	.3000	.4500	.6000	.7500	RH/L	=	MACH = .165

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TABULATED SOURCE DATA - OA37B

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MACH ( 1 ) = .165    ALPHA ( 4 ) = 10.015    RNL = 1.200    MACH = .165  
 SECTION 1 WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	.2841	.0177	-.1500	.0595	.0239	.0241
.334	.2298	-.1304	-.0550	.0763	-.0495	-.0241
.520	-.0172	.2086	.1005	-.0922	.2277	-.1562
.663	.3924	.3624	.2336	.1347	-.1110	-.0326
.873	.4446	.1796	.0538	-.0756	-.1938	-.1912

MACH ( 1 ) = .165    ALPHA ( 5 ) = 14.975    RNL = 1.200    MACH = .165

SECTION 1 WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	.3767	.1508	-.0568	.1409	.0813	.0651
.334	.3215	.0090	.0553	.1454	-.0177	-.0099
.520	.0829	.2054	-.0288	.0029	-.0582	-.0602
.663	.5638	.4657	.2737	.1671	-.0910	-.0680
.873	.4711	.2532	.1271	-.0032	-.1743	-.2642

MACH ( 1 ) = .165    ALPHA ( 6 ) = 20.000    RNL = 1.200    MACH = .165

SECTION 1 WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	.4655	.2726	.0320	.2135	.1134	.0766
.334	.4144	.1396	.1422	.1732	-.0111	-.0421
.520	.1067	.1026	-.1158	.0722	-.0550	-.1715
.663	.7377	.4981	.2751	.1720	-.1112	-.1059
.873	.4975	.2944	.1601	.0163	-.2009	-.3728

OA37-B

B16C5F1

J40

WTE10

WING LOWER SURFACE

(ADV15)

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## TABULATED SOURCE DATA - CAA578

(ADV16) ( 12 Nov 73 )

## REFERENCE DATA

BREF = 4.41200 6A. FT.  
 LREF = 10.2900 IN.  
 SREF = 37.9350 IN.  
 SCALE = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.965 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8  
 .000 .0312 -.5661 -.6763 -.3445 -.2361 -.1626  
 .334 -.3524 -.8184 -.6142 -.6437 -.3179 -.1482  
 .520 -.6637 -.4504 -.3035 -.2533 -.2780 -.1154  
 .663 -.4790 -.2366 -.2136 -.2261 -.3279 -.0780  
 .873 -.1471 -.2285 -.2844 -.3131 -.2866 -.1358

MACH ( 1 ) = .165 ALPHA ( 2 ) = .015 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8  
 .000 .1037 -.4250 -.5546 -.2497 -.1692 -.1603  
 .334 -.1616 -.7037 -.4864 -.1416 -.2687 -.1328  
 .520 -.5182 -.3310 -.1690 -.1533 -.2206 -.1043  
 .663 -.2151 -.0645 -.0831 -.1326 -.2912 -.0715  
 .873 .1125 -.0846 -.1873 -.2448 -.2556 -.1241

MACH ( 1 ) = .165 ALPHA ( 3 ) = 5.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8  
 .000 .1901 -.2326 -.3998 -.1461 -.1100 -.1216  
 .334 .0214 -.4554 -.2687 -.0502 -.2053 -.1135  
 .520 -.2151 -.1076 -.0178 -.1387 -.2632 -.2247  
 .663 .0107 .1234 .0494 -.0320 -.2490 -.0986  
 .873 .3075 .0559 -.0639 -.1638 -.2183 -.1599



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TABULATED SOURCE DATA - CRATE

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C57-B B16C5F1 J40 WATE10 WING LOWER SURFACE

(RNL=1.0)

MACH ( 1 ) = .165 ALPH ( 4 ) = 10,000 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

M/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .2714 -.0036 -.2736 -.0421 -.0475 -.0656

.334 .1734 -.2334 -.1436 .0486 -.1346 -.0752

.520 -.1329 .0629 .0424 .1384 .2940 -.2553

.663 .2837 .2716 .1686 .0812 -.1709 -.0835

.873 .4229 .1567 .0527 -.1020 -.1753 -.1844

MACH ( 1 ) = .165 ALPH ( 5 ) = 15,010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

M/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .3712 .0750 -.1544 .0685 .0326 -.0343

.334 .2891 -.0747 -.0124 .1428 -.0482 -.0245

.520 -.0241 .1699 .0829 -.0683 -.2526 -.2270

.663 .4204 .4000 .2975 .1725 -.1235 -.1190

.873 .4644 .2513 .1266 -.0204 -.1286 -.2209

MACH ( 1 ) = .165 ALPH ( 6 ) = 20,000 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

M/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .4617 .2061 -.0550 .1587 .0868 -.0318

.334 .3694 .0740 .1271 .1955 -.0263 -.0460

.520 -.0671 .1306 -.1326 -.0260 -.1284 -.2458

.663 .5636 .5401 .3625 .2330 -.1349 -.2636

.873 .5293 .3340 .2001 .0592 -.1076 -.2894

CAST-B B16CF1 J40 WATE10 WING LOWER SURFACE (ADULT) (112 NOV 73)

## REFERENCE DATA

REF	=	4,1120 5A FT.	XMAP =	43,5940 IN.	BETA =	.000	PIN/P =	1,300
LREF	=	16,2800 IN.	YMAP =	.0000 IN.	H/B =	.286	BLFLAP =	-10,000
SREF	=	37,9350 IN.	ZMAP =	-.0050 IN.	ELEVON =	.000		
SCALE =	.0405							
MACH (1) = .165		ALPHA (1) = -3.975	RNL = 1,200	MACH = .165				

## SECTION 1 WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000		
2/8								
.000	.0466	-.5521	-.5733	-.2912	-.2093	-.1474		
.334	-.3035	-.7802	-.5208	-.2153	-.2740	-.1256		
.668	-.5885	-.3067	-.2768	-.2321	-.2282	-.0781		
.000	-.4036	-.2004	-.1923	-.1917	-.2053	-.0403		
.673	-.0066	-.2135	-.2645	-.2970	-.2824	-.1321		
MACH (1) = .165		ALPHA (2) = .005	RNL = 1,200	MACH = .165				

## SECTION 1 WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000		
2/8								
.000	.1131	-.3603	-.4700	-.2953	-.1615	-.1280		
.334	-.1297	-.8060	-.4170	-.1053	-.2352	-.1086		
.668	-.3836	-.2117	-.1450	-.1342	-.1776	-.0636		
.000	-.1558	-.0340	-.0621	-.1052	-.2607	-.0635		
.673	-.1544	-.0655	-.1706	-.2330	-.2452	-.1222		
MACH (1) = .165		ALPHA (3) = 5.000	RNL = 1,200	MACH = .165				

## SECTION 1 WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000		
2/8								
.000	.2036	-.2050	-.3312	-.0042	-.0072	-.0090		
.334	.0589	-.3565	-.2303	-.0054	-.1719	-.0031		
.668	-.1928	-.0222	.0405	-.1136	-.2185	-.1582		
.000	.1275	.1680	.0801	-.0101	-.2184	-.0896		
.673	.3209	.0767	-.0347	-.1576	-.2007	-.1460		
MACH (1) = .165		ALPHA (3) = 5.000	RNL = 1,200	MACH = .165				

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TABULATED SOURCE DATA - Q1378

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(RDVL17)

Q137-8 816CSF1 J40 W7E10 WING LOWER SURFACE

MACH ( 11 ) = .165 ALPHAI ( 4 ) = .975 RNL = 1.200 MACH Z = .165

SECTION 111WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/2  
.2000 -.2000 -.0410 -.2082 .0065 -.0163 -.0483

.334 -.1994 -.1926 -.0914 .0676 .0692 -.0425

.320 -.0724 .1782 .0617 -.0638 -.2274 -.1780

.663 .3239 .3034 .1975 .0904 -.1625 -.1030

.873 .4653 .1710 .0600 -.0931 -.1666 -.1650

MACH ( 11 ) = .165 ALPHAI ( 9 ) = 15.000 RNL = 1.200 MACH Z = .165

SECTION 111WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/2  
.0000 .3792 .1008 -.1108 .0979 .0517 -.0211

.334 .3079 -.0334 .0161 .1822 -.0141 -.0202

.320 .0135 .2259 .1240 -.0253 -.1921 -.1497

.663 .4421 .4258 .3051 .1606 -.1421 -.1444

.873 .4741 .2420 .1266 -.0156 -.1418 -.2233

MACH ( 11 ) = .165 ALPHAI ( 6 ) = 20.005 RNL = 1.200 MACH Z = .165

SECTION 111WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8  
.0000 -.4663 .2373 .0023 .1606 .0946 -.0044

.334 .4003 .1240 .1623 .2214 -.0017 -.0481

.320 .0276 .1346 -.1226 .0102 -.1009 -.1923

.663 .6904 .5268 .3294 .1734 -.1941 -.3145

.873 .5265 .3319 .2101 .0537 -.1079 -.2765

CANT-B 0:ECF1 100 MPH 100000 WING-LEADER SURFACE (INFLATE) 1 12 NOV 73 1

## REFERENCE DATA

REF = 4.4185 64.91 - ROLL = 43.2840 IN.  
 LREF = 19.2000 IN. ROLL = .0000 IN.  
 BREF = 37.9350 IN. ROLL = -.4050 IN.  
 SCALE = .0005

MACH (11) = .165 ALPHAB(11) = -.4.005 ANGL = 1.800 MACH = .165

## DEPENDENT VARIABLE CP

SECTION 1 (WING)

E/C .1500 .3000 .4500 .6000 .7500 .9000 .0000

2/B .000 .0021 -.7004 -.0020 -.0000 -.0000 -.0000

2/B .000 .1913 -.3506 -.0101 -.0000 -.0000 .0000

2/B .000 .1016 -.3506 -.0000 -.0000 -.0000 .0000

2/B .000 .1517 -.3507 -.0000 -.0000 -.0000 .0000

2/B .000 .0913 -.3505 -.0000 -.0000 -.0000 .0000

MACH (11) = .165 ALPHAB(12) = -.4.005 ROLL = 1.800 MACH = .165

## DEPENDENT VARIABLE CP

SECTION 1 (WING)

E/C .1500 .3000 .4500 .6000 .7500 .9000 .0000

2/B .000 .1913 -.3506 -.0101 -.0000 -.0000 .0000

2/B .000 .1016 -.3506 -.0000 -.0000 -.0000 .0000

2/B .000 .1517 -.3507 -.0000 -.0000 -.0000 .0000

2/B .000 .0913 -.3505 -.0000 -.0000 -.0000 .0000

2/B .000 .0000 -.3506 -.0000 -.0000 -.0000 .0000

MACH (11) = .165 ALPHAB(3) = 4.950 ROLL = 1.800 MACH = .165

## DEPENDENT VARIABLE CP

SECTION 1 (WING)

E/C .1500 .3000 .4500 .6000 .7500 .9000 .0000

2/B .000 .2802 -.5318 -.0274 .0592 .0013 .1122

2/B .000 .1734 -.2224 .0011 .2224 .0699 .1122

2/B .000 .1191 -.1410 .1774 .1314 .3456 .2702

2/B .000 .2047 .2411 .1969 .1313 .3456 .2533

2/B .000 .0921 .1313 .1313 .1313 .1313 .0861

2/B .000 .1122 .1122 .1122 .1122 .1122 .1122

2/B .000 .1174 .1174 .1174 .1174 .1174 .1174

2/B .000 .1191 .1191 .1191 .1191 .1191 .1191

## PARAMETRIC DATA

BETA = .000 ROLL = 125 DTW/R = 1.000  
 M/B = .125 DTFLAP = -16.000  
 ELEVON = 15.000

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TABULATED SOURCE DATA - OA578

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## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.965 RN/L = 1.200 MACH = .165

(NDVL26)

## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.965 RN/L = 1.200 MACH = .165

DEPENDENT VARIABLE CP

(NDVL26)

## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990 RN/L = 1.200 MACH = .165

DEPENDENT VARIABLE CP

(NDVL26)

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

QAS7-B B1ECSF1 140 WTE10 WING LOWER SURFACE (RDVL20) (12 NOV 73)

## REFERENCE DATA

SREF =	4.4120 6A.FT.	XMAP =	43.1940 IN.	BETA =	.000
LREF =	19.2360 IN.	YMAP =	.0000 IN.	H/B =	.125
BREF =	37.9330 IN.	ZMAP =	-.4050 IN.	ELEVON =	15.000
SCALE =	.0405				

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
2/8							
	.000	.0703	-.6438	-.6017	-.3561	-.1708	-.13/1
	.334	-.3461	-.5293	-.3679	-.1709	-.1205	-.2512
	.520	-.6271	-.2198	-.1735	-.0894	-.1436	.1345
	.663	-.2746	-.0779	-.0535	-.0346	.1639	.0715
	.673	.0982	-.0844	-.1049	.0314	.0765	.0458

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.030 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
2/8							
	.000	.1767	-.2653	-.3576	-.1346	-.0354	-.0453
	.334	-.0365	-.1863	-.0206	.0309	.0534	-.0338
	.520	-.2398	-.0563	-.0101	.0461	.2053	.1706
	.663	.0350	.1342	.1101	.0848	.2968	.1640
	.673	.3264	.0463	.0226	.0887	.1130	-.0612

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
2/8							
	.000	.2953	.0226	-.1150	.0847	.1044	.0848
	.334	.2142	.0825	.0613	.2497	.2304	.0907
	.520	-.0474	.2204	.2250	.1369	.3218	.1473
	.663	.3562	.3563	.2393	.1794	.3864	.1743
	.673	.4510	.1277	.0954	.1156	.1319	-.0963

(ADNL 28)

## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 4 ) = 0.065 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.950 RN/L = 1.200 MACH = .165

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.960 RN/L = 1.200 MACH = .165

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .3030 .1005 .0044 .2201 .2273 .1721

.334 .3426 .2622 .1930 .3631 .3363 .2202

.520 .0672 .3511 .2669 .2223 .3731 .2420

.663 .4850 .4730 .3636 .2441 .4192 .1620

.873 .4941 .2248 .1933 .1566 .1396 .1499

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.950 RN/L = 1.200 MACH = .165

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.960 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .4673 .3098 .1597 .3152 .3083 .2132

.334 .4364 .3650 .2958 .4297 .3927 .2736

.520 -.0119 .2491 .2087 .2648 .3984 .2521

.663 .6444 .4622 .3955 .2861 .4308 .1565

.873 .5083 .3359 .2500 .2545 .2083 -.1277

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.960 RN/L = 1.200 MACH = .165

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .5473 .4208 .2499 .3927 .3553 .2367

.334 .5160 .4521 .3534 .4614 .4226 .3063

.520 .0089 .3783 .3771 .3070 .4157 .2042

.663 .5309 .4289 .3434 .2728 .3755 .0161

.873 .5424 .3836 .3061 .2879 .2413 -.1107

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.960 RN/L = 1.200 MACH = .165

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .5473 .4208 .2499 .3927 .3553 .2367

.334 .5160 .4521 .3534 .4614 .4226 .3063

.520 .0089 .3783 .3771 .3070 .4157 .2042

.663 .5309 .4289 .3434 .2728 .3755 .0161

.873 .5424 .3836 .3061 .2879 .2413 -.1107

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.960 RN/L = 1.200 MACH = .165

X/C .1500 .3000 .4500 .6000 .7500 .9000

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## TABULATED SOURCE DATA - QA378

(ADYL30) (12 NOV 73)

## REFERENCE DATA

SREF =	4.4120	64.571.	ZHAP =	43.5940 IN.	BETA =	.000	PTN/P =	1.000
LREF =	19.2300	IN.	ZHAP =	.0000 IN.	M/B =	.125	BDFLAP =	-16.000
BREF =	37.9350	IN.	ZHAP =	-.4050 IN.	ELEVON =	15.000		
SCALE =	.0405							

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/18	.000	.1046	-.4415	-.4170	-.1913	-.1409	-.1216
	.334	-.2176	-.3220	-.2466	-.0347	.0708	-.0958
	.520	-.4560	-.1335	-.0844	.0636	.0139	
	.643	-.1643	-.0055	-.0198	.0141	.1530	.0507
	.873	.1550	-.0716	-.0873	.0440	.0774	.0630

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.020 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/18	.000	.1919	-.1687	-.2586	-.0436	.0630	-.0586
	.334	.0079	-.0976	-.1128	.0821	.0463	-.0099
	.520	-.1794	.0114	.0427	.0425	.1319	.0481
	.663	-.0902	.1779	.1201	.0782	.2872	.1147
	.873	.3479	.0546	.0227	.0824	.1065	-.0799

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/18	.000	.2815	.0726	-.0600	.1237	.0740	.0269
	.334	.2298	.1242	.0690	.2362	.2009	.0877
	.520	-.0100	.2329	.2273	.0955	.2548	.0461
	.663	.3931	.4000	.2683	.1683	.3625	.1115
	.873	.4623	.1195	.1008	.1046	.1169	-.1100

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TABULATED SOURCE DATA - CASTB

QAST-B

QAST-B

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.985 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/1/8 .000 .3606 .2268 .0952 .2548 .1050 .1099

.334 .3348 .2634 .1935 .3496 .2921 .1750

.520 .1194 .3594 .2519 .1673 .2846 .0949

.663 .5926 .5187 .3130 .2148 .3777 .1442

.873 .4627 .2319 .1569 .1529 .1361 -.1613

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.980 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/1/8 .000 .4611 .3462 .1689 .3295 .2601 .1573

.334 .4376 .3601 .2816 .3915 .3086 .2194

.520 .0755 .2670 .1601 .2135 .2814 .1553

.663 .7236 .5606 .3164 .2561 .3601 .1268

.873 .5009 .3133 .2223 .2265 .1782 -.1569

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.980 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

2/1/8 .000 .5338 .4464 .2702 .3907 .2929 .1643

.334 .5246 .4861 .3671 .4275 .3045 .1929

.520 .4711 .3015 .1345 .1197 .2233 .0363

.663 .8277 .4710 .3129 .2167 .3587 .0825

.873 .5310 .3520 .2799 .2502 .1806 -.1626

(ADYL30)

## CH57-B 816CSF1 JAO WSTE10 WING LOWER SURFACE

(90VLS1) (12 NOV 73)

## REFERENCE DATA

BREF = 4.4120 IN. FT. XMAP = 43.1940 IN.  
 LREF = 19.2300 IN. YMAP = .0000 IN.  
 SREF = 37.0350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

## X/C

.1500

.3000

.4500

.6000

.7500

.9000

21/B

.000

.4777

.4623

.4442

.5623

.7552

1.0420

.334

.5353

.4576

.5906

.7471

.8439

.8594

.520

.2110

.3944

.5266

.3767

.9802

1.2512

.663

.3791

.2994

.3665

.4246

.1407

.6653

.873

.3255

.0427

-.0921

-.0101

-.0493

-.3496

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.005 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

## X/C

.1500

.3000

.4500

.6000

.7500

.9000

21/B

.000

.5477

.4847

.4757

.5969

.7814

.9927

.334

.5544

.4515

.5917

.7639

.8529

.8653

.520

-.22284

.2634

.3319

.1921

.9338

1.1403

.663

.1028

.2125

.2312

.1619

.1037

.0248

.873

.4106

.1492

.0289

.0417

.0191

-.3963

MACH ( 1 ) = .165 ALPHA ( 3 ) = 20.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

## X/C

.1500

.3000

.4500

.6000

.7500

.9000

21/B

.000

.6151

.5501

.5274

.6450

.7951

.9919

.334

.6104

.5252

.6232

.7770

.8504

.8804

.520

-.1709

.2829

.3745

.4200

.6798

.9006

.663

-.1301

.2606

.3170

.2349

.3546

.8239

.873

.4730

.2754

.1690

.1321

.0556

-.3028



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TABULATED SOURCE DATA - QAS7B

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## REFERENCE DATA

BREF	=	4.4120 3A.FT.	XNAP =	43.5940 IN.			
LREF	=	16.2500 IN.	1NAP =	.0000 IN.			
BREF	=	37.9350 IN.	2NAP =	-.4050 IN.			
SCALE	=	.0405					

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (1) = 9.990 \quad \text{RN/L} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	

2/1/8

.000	.4837	.5121	.4374	.5535	.6865	.6991
.334	.5397	.4785	.5756	.7072	.7600	.6922
.520	.2164	.4228	.4834	.3025	.7773	.7370
.663	.4754	.3376	.2848	.3772	.2292	.5646
.873	.3208	-.0849	-.1201	-.0441	-.0680	-.3319

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (2) = 14.970 \quad \text{RN/L} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	

2/1/8

.000	.5431	.5447	.4704	.5761	.7476	.7604
.334	.5627	.4641	.5749	.7228	.7603	.7137
.520	-.1155	.3094	.3769	.3621	.7643	.6151
.663	.1043	.2106	.2372	-.0825	.3919	.6724
.873	.4123	.1510	.0230	-.0071	-.0327	-.4565

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (3) = 19.980 \quad \text{RN/L} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	

2/1/8

.000	.6107	.3968	.5192	.6202	.7739	.9061
.334	.6091	.5483	.6123	.7431	.7760	.7177
.520	-.1116	.3215	.4462	.4864	.7376	.5338
.663	-.0194	.2745	.3259	.2203	.3936	.5031
.873	.4679	.2731	.1690	.1101	.0116	-.3694

f

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TABULATED SOURCE DATA - CASTS

(RDVL33) ( 12 NOV 73 )

## REFERENCE DATA

BASEF	4.4120 00. FT.	XMAP	=	43.5940 IN.			
LREF	19.2300 IN.	YMAP	=	.0000 IN.			
RREF	37.9350 IN.	ZMAP	=	-.4050 IN.			
SCALE	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.005 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .4722 .5115 .5940 .5254 .5288 .2901

.334 .5217 .4208 .4308 .5836 .5400 .5176

.520 .2075 .3842 .3976 .5337 .5463 .4116

.663 .5579 .4081 .1768 .2196 .2584 .3181

.873 .3041 -.0198 -.1505 -.3365 -.3296 -.4931

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.990 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .5300 .5341 .4523 .5446 .5458 .3767

.334 .5453 .4489 .4376 .5854 .5667 .5485

.520 -.0002 .2667 .3930 .4004 .5358 .3730

.663 .1435 .2390 .2866 .2735 .2828 .2969

.873 .4017 .1312 -.0509 -.0362 -.1766 -.4463

MACH ( 1 ) = .165 ALPHA ( 3 ) = 20.015 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .5944 .5972 .5133 .5919 .6266 .4784

.334 .5937 .5206 .5016 .6109 .6055 .5761

.520 -.0615 .3815 .5219 .4606 .5674 .3533

.663 .0888 .4179 .4271 .3475 .2812 .2311

.873 .4539 .1367 .0377 .0792 -.0202 -.3306



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TABULATED SOURCE DATA - CA37B

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(RDVL34) (112 NOV 73)

CA37-B B16C5F1 J40 WATE10 WING LOWER SURFACE

## REFERENCE DATA

	MREF = 4.4120 IN. FT.	XMAP = 43.5940 IN.	PARAMETRIC DATA
LREF = 19.2500 IN.	YMAP = .0000 IN.	BETA = .000	PTN/P = 1.500
BREF = 37.9350 IN.	ZMAP = -.4050 IN.	H/B = .286	BOFLAP = -16.000
SCALE = .0409		ELEVON = 15.000	

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.0000 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.0648	-.4958	-.3442	-.2644	-.1152	-.0962
.334	-.2846	-.4902	-.3442	-.0589	.0425	.0438
.520	-.5874	-.1516	-.1474	-.0208	.1786	.1096
.663	-.2707	-.0530	-.0301	.0127	.2203	.0859
.873	.0969	-.0741	-.0935	.0594	.0892	-.0354

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.1364	-.3447	-.4287	-.1609	-.0662	-.0854
.334	-.1077	-.3127	-.2276	.0450	.1080	-.0131
.520	-.3629	-.0761	-.0473	.0579	.2506	.1088
.663	-.0393	.1082	.0861	.0888	.3150	.1505
.873	.3097	.0368	.0035	.1027	.1245	-.0425

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.995 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.2253	-.1674	-.2878	-.0466	.0033	-.0532
.334	.0916	-.1040	-.0721	.1591	.1587	-.0073
.520	-.1615	.0659	.1153	.0634	.2946	.0397
.663	.2416	.3047	.2136	.1780	.4167	.2333
.873	.4240	.1443	.1054	.1468	.1570	-.0602

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(RDVL34)

MACH ( 1) = .165    ALPHA ( 4) = 9.895    RNL = 1.200    MACH = .165  
 SECTION ( 1) WING  
 X/C    .1500    .3000    .4500    .6000    .7500    .9000

## DEPENDENT VARIABLE CP

21/8  
 .000    .3137    -.0193    -.1639    .0606    .0707    -.0046  
 .334    .2347    .0576    .0564    .2290    .2269    .0767  
 .520    -.0323    .2871    .1992    .1152    .3093    .0539  
 .663    .4204    .4258    .3237    .2692    .4970    .5047  
 .873    .5036    .2154    .1661    .1817    .1600    -.1064  
 MACH ( 1) = .165    ALPHA ( 5) = 14.975    RNL = 1.200    MACH = .165

## DEPENDENT VARIABLE CP

SECTION ( 1) WING  
 X/C    .1500    .3000    .4500    .6000    .7500    .9000  
 21/8  
 .000    .3953    .1132    -.0748    .1447    .1345    .0079  
 .334    .3330    .1954    .1244    .2966    .2873    .1192  
 .520    .0230    .2269    .2100    .1019    .5015    .0863  
 .663    .5063    .5382    .4347    .3536    .5502    .2813  
 .873    .4944    .3210    .2398    .2463    .2074    -.1160  
 MACH ( 1) = .165    ALPHA ( 6) = 19.990    RNL = 1.200    MACH = .165

## DEPENDENT VARIABLE CP

SECTION ( 1) WING  
 X/C    .1503    .3000    .4500    .6000    .7500    .9000  
 21/8  
 .000    .4830    .2363    .0411    .2262    .1762    .0154  
 .334    .4343    .3324    .2107    .3316    .3015    .1237  
 .520    .0997    .1639    -.0154    .1090    .3756    .0889  
 .663    .7101    .6233    .4582    .3646    .5457    .1186  
 .873    .5521    .5935    .3612    .2929    .2297    -.1571

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TABULATED SOURCE DATA - OA57B

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CAST-8 B16C5F1 J40 WATER10 WING LOWER SURFACE

(NOV 30) (12 NOV 73)

## REFERENCE DATA

SREF	=	4.4120 84.871.	XMAP	=	43.5940 IN.
LREF	=	10.2300 IN.	YMAP	=	.0000 IN.
SREF	=	3.9350 IN.	ZMAP	=	-.4050 IN.
SCALE	=	.0403			

MACH (1) = .165 ALPHA (1) = -3.995 RNL = 1.200 MACH = .165

## SECTION (1)WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	.0634	-.4509	-.4564	-.1925	-.0793	-.0751
.334	-.2316	-.4167	-.2361	-.0550	.0095	-.0455
.520	-.3122	-.1015	-.0803	-.0360	.1722	.0531
.663	-.2036	.0045	-.0151	.0010	.2120	.0805
.873	.1310	-.0614	-.0796	.0393	.0890	-.0395

MACH (1) = .165 ALPHA (2) = -.020 RNL = 1.200 MACH = .165

## SECTION (1)WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	.1408	-.3060	-.3619	-.1209	-.0903	-.0544
.334	-.0786	-.2500	-.1632	.0513	.0828	-.0043
.520	-.2904	-.0476	-.0043	.0461	.2124	.0550
.663	.0174	.1551	.0846	.0778	.3062	.1333
.873	.3251	.0496	.0120	.1031	.1259	-.0494

MACH (1) = .165 ALPHA (3) = 4.960 RNL = 1.200 MACH = .165

## SECTION (1)WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	.2330	-.1251	-.2262	-.0057	.0395	-.0164
.334	.1241	-.0541	-.0130	.1593	.1492	.0168
.520	-.1261	.1404	.1516	.0647	.2612	-.0061
.663	.2979	.3440	.2248	.1651	.3924	.1468
.873	.4472	.1401	.1030	.1494	.1528	-.0677

## PARAMETRIC DATA

BETA	=	.000	PIN/P	=	1.300
H/D	=	.286	BDFLAP	=	-16.000
ELEVON	=	15.000			

## SECTION 1 (1) WING

MACH (1) = .165

ALPHA (4) = 9.885

AN/L = 1.200

MACH = .165

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 .3040 -.0021 -.1370 .0762 -.0614 -.0089

.334 .2335 .0930 .0742 .2264 .2037 .0647

.520 .0010 .2998 .2029 .1037 .2508 .0041

.663 .4464 .4474 .3068 .2173 .4319 .1643

.873 .4929 .2002 .1803 .1500 .1536 -.1338

MACH (1) = .165

ALPHA (5) = 14.970

RN/L = 1.200

MACH = .165

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 .4002 .1535 -.0246 .1760 .1538 .0416

.334 .3462 .2122 .1605 .3028 .2649 .1151

.520 -.1290 .2127 .1618 .1410 .2950 .0763

.663 .4782 .5840 .4317 .3032 .4632 .1132

.873 .5028 .3133 .2401 .2479 .2014 -.1417

MACH (1) = .165

ALPHA (6) = 19.980

RN/L = 1.200

MACH = .165

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 .4853 .2765 .0722 .2462 .1842 .0279

.334 .4428 .3549 .2991 .3222 .2619 .0876

.520 .2191 .1766 -.0177 .1157 .3014 .0259

.663 .7507 .6100 .3807 .2674 .3901 -.1308

.873 .5465 .3893 .2987 .2883 .2228 -.1596



(PROV35)

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TABULATED SOURCE DATA - CA378

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## CA37-B 810CSF1 JAI WATE16 WING LOWER SURFACE

## REFERENCE DATA

SACF	.3	4.4120 83.FT.	XMAP	z	43.5940 IN.
LREF	x	18.2500 IN.	YMAP	=	.0000 IN.
SREF	x	37.9350 IN.	ZMAP	=	.4050 IN.
SCALE	x	.0403			
MACH	( 1 ) =	.165	ALPHA ( 1 ) =	- .003	RNL =

MACH ( 1 ) = .165

ALPHA ( 1 ) = -.003

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 2 ) = -.005

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 3 ) = 4.995

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 4 ) = -.344

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 5 ) = -.1762

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 6 ) = -.1762

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 7 ) = -.1762

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 8 ) = -.1762

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 9 ) = -.1762

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 10 ) = -.1762

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

ALPHA ( 11 ) = -.1762

RNL =

MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c	.1500	.3000	.4500	.6000	.7500	.9000

MACH ( 1 ) = .165

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TABULATED SOURCE DATA - CA97B

## CAST-B 610CSF1 141 WATE10 WING LOWER SURFACE

(ADVL52)

MACH = 1.11 = .165 ALPHAL = 0.990 RHL = 1.200 MACH = .165

## SECTION (1) WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .2842 .0216 -.2538 -.0022 -.0372 .0617

.334 .2039 .0413 .0168 -.1370 -.1004 -.0449

.320 .0432 .3461 .2004 -.0726 -.1221 -.1086

.643 .3382 .4127 .1676 .0857 -.2514 -.0944

.673 .2636 .1204 .0285 -.1366 -.2961 -.2144

MACH = 1.11 = .165 ALPHAL = 1.91 = 14.990 RHL = 1.200 MACH = .165

SECTION (1) WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .3668 .1546 -.1572 .0946 .0237 .0976

.334 .3070 .1636 .1629 -.0634 -.0418 -.0371

.320 .0294 .2912 .2402 -.0323 -.0992 -.0760

.643 .3963 .4192 .2692 .1756 -.1928 -.1372

.673 .3643 .2070 .0713 -.0421 -.2480 -.2644

MACH = 1.11 = .165 ALPHAL = 1.61 = 19.990 RHL = 1.200 MACH = .165

SECTION (1) WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .4583 .2846 -.0517 .1643 .0744 .1296

.334 .4152 .2829 .2515 .0941 .0106 -.0459

.320 .0273 .2459 .2572 .0663 -.1205 -.1237

.643 .4895 .3503 .3377 .2220 -.2283 -.3198

.673 .4970 .3021 .1666 .0263 -.2545 -.3658

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TABULATED SOURCE DATA - Q370

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Q37-B 810C5F1 JAI WATE10 WING LOWER SURFACE

(RDY153) ( 12 NOV 73 )

## REFERENCE DATA

SREF	=	4.4120 84.FT.	XMRP	=	43.5940 IN.
LREF	=	10.2500 IN.	YMRP	=	.0000 IN.
BREF	=	37.9350 IN.	ZMRP	=	-.4050 IN.
SCALE	=	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

DEPENDENT VARIABLE CP

21/8					
.000	.0611	-.4080	-.4920	-.2368	-.2023
.334	-.2701	-.4353	-.2863	-.2701	-.1874
.520	-.5494	-.2143	-.0801	.1947	-.1125
.663	-.3044	-.0311	.1605	.1670	.1770
.873	-.2702	-.2267	-.3357	-.3371	.3058

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.003 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

DEPENDENT VARIABLE CP

21/8					
.000	.1277	-.2768	-.4011	-.1582	-.1660
.334	-.1065	-.2873	-.1970	-.2697	-.1504
.520	-.3013	-.0907	-.0365	-.1399	-.1343
.663	-.0721	.1246	-.0548	.1024	.0723
.873	-.0357	-.0992	-.2333	-.2812	.2908

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.985 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

DEPENDENT VARIABLE CP

21/8					
.000	.2017	-.1222	-.2371	-.0692	-.1081
.334	-.0764	-.0987	-.0883	-.2248	-.1076
.520	-.0782	.1144	.1086	-.0810	-.1554
.663	.1727	.3198	.0855	.0180	-.2720
.873	-.1523	.0276	-.1428	-.2106	-.5371

## PARAMETRIC DATA

BETA	=	.000	PTN/P	=	1.000
M/B	=	.266	BDFLAP	=	-16.000
ELEVON	=	.000			

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TABULATED SOURCE DATA - QAS7B

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## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.980 RNL = 1.200 MACH = .165

## (RDVLS3)

CA37-B B16C5F1 J41 WATE10 WING LOWER SURFACE

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.2854	.0279	-.1693	.0273	-.0371	.0048
.334	.2093	.0496	.0302	-.0173	-.0628	-.0498
.520	.0652	.3626	.1980	-.0121	-.1218	-.0778
.663	.3661	.4366	.2148	.1309	-.1082	-.0579
.873	.2616	.1240	-.0267	-.1376	-.2967	-.2210

MACH ( 1 ) = .135 ALPHA ( 5 ) = 14.985 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.3669	.1572	-.0864	.1146	.0263	.0643
.334	.3165	.1743	.1489	.0598	-.0101	-.0417
.520	.1543	.2924	.2146	.0730	-.0916	-.0586
.663	.5475	.5384	.3188	.2550	-.1142	-.0665
.873	.3975	.2190	.0837	-.0211	-.2443	-.2655

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.980 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.4479	.2755	-.0095	.1639	.0655	.0918
.334	.4056	.2794	.2438	.0956	.0111	-.0700
.520	.1163	.1960	.1886	.0648	-.1326	-.1344
.663	.5192	.5256	.3804	.2882	-.1292	-.2080
.873	.4554	.2946	.1451	.0145	-.2684	-.4011

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TABULATED SOURCE DATA - OA37B

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OA37-B 810CSF1 J41 W0TE18 WING LOWER SURFACE

(RDV155) ( 12 NOV 73 )

## REFERENCE DATA

SREF	=	4.4120 82.FT.	XNRP	=	43.5940 IN.
LREF	=	19.2300 IN.	YNRP	=	.0000 IN.
BREF	=	37.9350 IN.	ZNRP	=	-.4050 IN.
SCALE		.0455			

MACH ( 1 ) = .165 ALPHA ( 1 ) = 9.990 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500 .9000

21/B					
.000	.4806	.4686	.2695	.4550	.3631 .4560
.334	.4972	.4681	.5191	.3405	.3235 .1806
.520	.2378	.3773	.3411	.1948	.2536 .0206
.663	.2964	.3597	.0480	.0589	-.1925 -.0414
.873	.1354	-.0616	-.2499	-.3675	-.5584 -.4293

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.990 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500 .9000

21/B					
.000	.5261	.5103	.3573	.5144	.5487 .6221
.334	.5354	.5083	.5708	.4115	.4729 .3284
.520	.0853	.2396	.3505	.1779	.2957 .1219
.663	.3794	.3070	.0852	.0646	-.0590 .0194
.873	.3255	.1343	-.0246	-.0826	-.4077 -.4136

MACH ( 1 ) = .165 ALPHA ( 3 ) = 19.995 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500 .9000

21/B					
.000	.5737	.5650	.4244	.5572	.6204 .7702
.334	.5741	.5349	.6134	.4865	.5377 .3928
.520	-.1634	.2330	.4192	.1809	.3231 .1416
.663	-.0545	.2098	.1466	.1209	-.1279 -.1462
.873	.3624	.2467	.0893	-.0231	-.3593 -.4722

PARAMETRIC DATA

OA57-B B16C-5F1 J41 WTE10 WING LOWER SURFACE

(RDYL56) ( 12 NOV 73 )

## REFERENCE DATA

BREF =	4.4120 SQ.FT.	XMRP =	43.5940 IN.
LREF =	19.2300 IN.	YMRP =	.0000 IN.
BREF =	37.9350 IN.	ZMRP =	-.4030 IN.
SCALE =	.0405		

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (1) = 10.005 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B						
.000	.4737	.4601	.2664	.3941	.2147	.1451
.334	.4858	.4451	.4431	.2267	.1645	.0944
.520	.2557	.3922	.3212	.1393	.2335	.0449
.663	.3768	.4021	.0482	-.0278	.1685	-.0568
.873	.1669	-.0369	-.1659	-.2663	-.5156	-.4433

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (2) = 15.020 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B						
.000	.5114	.4964	.3630	.4656	.3760	.3353
.334	.5165	.4818	.3330	.3134	.2893	.2042
.520	.1092	.2395	.3569	.1981	.3538	.0823
.663	.4315	.3261	.1310	.1228	.0349	.0012
.873	.3410	.1611	.0048	-.0373	-.3572	-.4039

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (3) = 20.015 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B						
.000	.5755	.5623	.4332	.5329	.4935	.4718
.334	.5733	.5257	.5622	.3825	.3906	.2419
.520	-.1541	.2772	.4517	.2557	.3694	.0378
.663	-.0196	.2762	.2795	.1939	-.0338	-.0971
.873	.4067	.2666	.1069	-.0232	-.3275	-.4423

## PARAMETRIC DATA

BETA =	.000	PINP =	1.000
H/B =	.039	EDFLAP =	-10.000
ELEVON =	.000		

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TABULATED SOURCE DATA - GA370

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GA37-B B16C5F1 J41 W7E18 :LINE LOWER SURFACE

(ADVL57) ( 12 NOV 73 )

## REFERENCE DATA

BREF =	4.4120 84.FT.	XWRF =	43.5940 IN.
LREF =	19.2300 IN.	YWRF =	.0000 IN.
BREF =	37.9350 IN.	ZWRF =	-.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.980 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8						
.000	.0273	-.7682	-.9128	-.5310	-.4183	-.1166
.334	-.4465	-.7735	-.5560	-.6411	-.3829	-.2525
.520	-.7388	-.3190	-.2820	-.4200	-.3552	-.2956
.663	-.4321	-.1223	-.3065	-.3242	-.4624	-.2315
.873	-.3948	-.3189	-.4485	-.4283	-.4657	-.2429

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.005 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8						
.000	.1492	-.3324	-.6082	-.2827	-.2978	-.0526
.334	-.1202	-.3552	-.4348	-.2835	-.1578	
.520	-.3713	-.1589	-.0999	-.2439	-.2382	-.1326
.663	-.1159	.0960	-.1125	-.1576	-.3716	-.1379
.873	-.0717	-.1338	-.2658	-.3150	-.3984	-.1875

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.980 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8						
.000	.2539	-.0451	-.3350	-.0684	-.1451	.0525
.334	.1360	-.0479	-.0598	-.1985	-.1545	-.1107
.520	-.0433	.1395	.1101	-.1147	-.1669	-.0959
.663	.1665	.3219	.0712	-.0203	-.3351	-.1435
.873	.1337	.0056	-.1659	-.2355	-.3568	-.2171

## PARAMETRIC DATA

BETA =	.0000	PTN/P =	1.500
M/B =	.125	BDFLAP =	-16.000
ELEVCH =	.0000		

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MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.995 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.3454	.1405	-.1308	.0979	.0124	.1701
.334	.2689	-.1508	.1453	.0132	-.0225	-.0217
.520	.1398	.3755	.2366	-.0631	-.0177	-.0657
.663	.3476	.4245	.1694	.0969	-.2642	-.1723
.873	.2780	.1067	-.0423	-.1453	-.3160	-.2474

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.020 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.4260	.2766	-.0125	.2159	.1374	.2503
.334	.3856	.2590	.2052	.1728	.1041	.0400
.520	-.1358	.1454	.2778	.1246	.0297	-.0367
.663	.2223	.4144	.1733	.0183	-.3236	-.1836
.873	.3876	.2201	.0795	-.0031	-.2690	-.3021

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.995 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	.5069	.3960	.1174	.3156	.2156	.3394
.334	.4634	.3960	.3916	.1960	.1721	.0514
.520	.0825	.3865	.3318	.1346	.0017	-.1640
.663	.4372	.5417	.3283	.1319	-.3668	-.4680
.873	.4492	.3115	.1533	.0615	-.2567	-.3982



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TABULATED SOURCE DATA - QASTP

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CAT-7-B B16C5F1 J41 WTE10 WING LOWER SURFACE

(RDVL5B) ( 12 NOV 73 )

## REFERENCE DATA

SREF	X	4.4120 63.FT.	XMRP	=	43.5940 IN.
LREF	X	19.2300 IN.	MRP	=	.0000 IN.
BREF	X	37.9350 IN.	ZMRP	=	-.4050 IN.
SCALE	X	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.395 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 .0546 -.6458 -.6757 -.4138 -.3546 -.2025

.334 -.3633 -.6644 -.6906 -.4902 -.3431 -.1966

.320 -.5652 -.2335 -.1901 -.2977 -.2572 -.1991

.663 -.3323 -.0829 -.2285 -.2191 -.3598 -.1784

.873 -.3074 -.2694 -.3930 -.3818 -.4273 -.2145

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.005 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 .1572 -.2651 -.4700 -.2102 -.2580 -.1129

.334 -.0828 -.2904 -.3074 -.3894 -.2447 -.1406

.320 -.2660 -.1253 -.0751 -.1604 -.1850 -.1120

.663 -.0765 .1248 -.0845 -.1105 -.3271 -.134

.873 -.0366 -.1217 -.2665 -.3002 -.3687 -.11E 5

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.995 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 .2622 -.0160 -.2338 -.0246 -.1282 -.0113

.334 .1699 -.0099 -.0550 -.2111 -.1449 -.0583

.120 -.0060 .1720 .1299 -.0568 -.1338 -.0916

.663 .2167 .3451 .0984 .0472 -.2778 -.1119

.873 .1663 .0154 -.1334 -.2095 -.3465 -.2731

## PARAMETRIC DATA

	BETA	= .000	PTN/P	= 1.000
M/B	= .125		BDFLAP	= -10.000
ELEVON	= .000			

OA57-B 816CSF1 J42 WTE16 WING LOWER SURFACE

(ADVL58) (12 NOV 73)

## REFERENCE DATA

SREF =	4.4120 54.51	ZHMP =	43.5940 IN.
LREF =	19.2300 IN.	THMP =	.0000 IN.
BREF =	37.9350 IN.	ZHMP =	.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.025 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
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MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.955 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

MACH ( 1 ) = .165 ALPHA ( 4 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

MACH ( 1 ) = .165 ALPHA ( 5 ) = 4.955 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

MACH ( 1 ) = .165 ALPHA ( 6 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

MACH ( 1 ) = .165 ALPHA ( 7 ) = 4.955 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

MACH ( 1 ) = .165 ALPHA ( 8 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
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## PARAMETRIC DATA

BETA =	.000	PIN/P =	1.500
H/B =	.125	BDFLAP =	-16.000
ELEVON =	.000		

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TABULATED ACCURATE DATA - OA978

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MACH ( 1 ) = .165							CA57-B BIEGSEI 142 WATE10 WING LOWER SURFACE (RDYLS9)		
SECTION ( 1 ) WING							DEPENDENT VARIABLE CP		
X/C	.1500	.3000	.4500	.6000	.7500	.9000			
21/8									
.000	.2779	.1165	-.1621	.0604	.0089	-.0725			
.334	.2107	.2167	.1393	.0654	.0287	.0085			
.520	.1608	.2415	.2026	.0949	-.0152	.0299			
.663	.2980	.2025	.0945	.0953	-.1323	-.0240			
.673	.2646	.0990	-.0448	-.1269	-.3116	-2335			
MACH ( 1 ) = .165		ALPHA ( 5 ) = 14.980		RH/L	= 1.200	MACH = .165			
SECTION ( 1 ) WING				DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500	.9000			
21/8									
.000	.3719	.2517	-.0157	.1824	.1108	.0365			
.334	.3160	.3131	.2484	.1970	.1080	.0501			
.520	.3104	.3409	.3165	.1976	.0255	.0228			
.663	.3751	.3291	.1267	.1511	-.0547	-.0488			
.673	.3302	.1776	.0134	-.0636	-.3265	-.3692			
MACH ( 1 ) = .165		ALPHA ( 6 ) = 19.975		RH/L	= 1.200	MACH = .165			
SECTION ( 1 ) WING				DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500	.9000			
21/8									
.000	.4556	.3620	.0997	.2861	.2037	.0966			
.334	.3699	.5959	.3351	.2813	.1973	.0334			
.520	.4229	.4222	.4031	.2730	.1133	.0242			
.663	.3971	.3920	.2258	.2016	-.0143	-.0517			
.673	.3104	.2126	.0682	-.0255	-.3154	-.4639			

CA57-B B10C3F1 J42 WATE10 WING LOWER SURFACE (ADY60) ( 12 NOV 73 )

## REFERENCE DATA

WCF = 4.4120 83.571. XMAP = 43.5940 IN.  
 LREF = 19.2300 IN. YMAP = .0000 IN.  
 DREF = 37.9350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000

21/8  
 .000 .0236 -.8763 -.7536 -.4224 -.3274 -.2497  
 .334 -.3588 -.3627 -.4358 -.4070 -.3042 -.2020  
 .520 -.4650 -.3845 -.3114 -.3677 -.4392 -.1734  
 .663 -.4186 -.3299 -.4326 -.3550 -.4553 -.2200  
 .873 -.3432 -.2816 -.4090 -.4531 -.4350 -.2257

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000

21/8  
 .000 .1091 -.3166 -.5228 -.2356 -.2583 -.2463  
 .334 -.1040 -.1359 -.2177 -.2429 -.2341 -.1933  
 .520 -.2096 -.1231 -.1183 -.2176 -.3463 -.1279  
 .663 -.1071 -.1115 -.2416 -.2053 -.3612 -.1692  
 .873 -.0425 -.1238 -.2695 -.3506 -.3959 -.1495

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000

21/8  
 .000 .2009 -.0607 -.3047 -.0561 -.1195 -.1792  
 .334 .0906 -.0737 -.0160 -.0829 -.0996 -.1106  
 .520 -.0311 -.0931 -.0534 -.0380 -.2165 -.0040  
 .663 .1262 .0674 -.0510 -.0544 -.2449 -.1067  
 .873 .1263 -.0347 -.1581 -.2562 -.3555 -.2122

## PARAMETRIC DATA

BETA = .000 PTN/P = 1.300  
 K/B = -.125 BDFLAP = -.16.000  
 ELEVON = .000

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## TABULATED SOURCE DATA - CA57B

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CA57-B B16C5F1 J42 WATE10 WING LOWER SURFACE  
(MDVL80)

SECTION (1) WING DEPENDENT VARIABLE CP

MACH (1) = .165 ALPHA (4) = 9.960 RN/L = 1.200 MACH = .165

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B DEPENDENT VARIABLE CP

X/C	.000	.2829	.1047	-.1308	.0854	.0079	-.0356
	.334	.2177	.2177	.1320	.0775	.0220	-.0221
	.520	.1654	.2367	.1925	.0908	-.0689	-.0250
	.663	.2907	.2016	.0630	.0634	-.1318	-.0519
	.873	.2503	.0873	-.0495	-.1588	-.3058	-.2319

MACH (1) = .165 ALPHA (5) = 14.960 RN/L = 1.200 MACH = .165

SECTION (1) WING DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B DEPENDENT VARIABLE CP

X/C	.000	.3631	.2519	.0090	.1972	.1267	.0489
	.334	.3037	.3216	.2506	.1861	.1250	.0458
	.520	.2957	.3566	.3141	.1774	.0212	.0024
	.663	.3562	.3191	.1075	.1402	-.0508	-.0722
	.873	.3175	.1679	.0065	-.0561	-.3140	-.3214

MACH (1) = .165 ALPHA (6) = 19.960 RN/L = 1.200 MACH = .165

SECTION (1) WING DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B DEPENDENT VARIABLE CP

X/C	.000	.4640	.5678	.1245	.3011	.2106	.1502
	.334	.3957	.4669	.3465	.2828	.1890	.0784
	.520	.4262	.4213	.4073	.2558	.0839	-.0105
	.663	.4173	.3917	.2262	.1681	-.0338	-.0877
	.873	.3214	.2124	.0826	-.0212	-.3193	-.4245

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TABULATED SOURCE DATA - CA376

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CA37-N B16CSF1 J42 WATE10 WING LOWER SURFACE

(RDVLS1) ( 12 NOV 73 )

## REFERENCE DATA

BREF =	4.4120	64.51.	XMAP =	43.3940 IN.	BETA =	.000	PIN/P =	1.000
LREF =	19.2900	IN.	YMAP =	.0000 IN.	H/B =	.175	BLFLAP =	-10.000
BREF =	37.9350	IN.	ZMAP =	-.4050 IN.	ELEVON =	.000		
SCALE =	.0405							

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.030 ANVL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8	.000	.0385	-.6061	-.5692	-.3669	-.3334	-.2809
	.334	-.3176	-.3233	-.4200	-.3735	-.2651	-.3166
	.520	-.4469	-.3641	-.2991	-.3610	-.4335	-.1778
	.663	-.4037	-.3087	-.4350	-.3463	-.4456	-.2268
	.873	-.3322	-.2743	-.3964	-.4568	-.4265	-.2276

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.025 ANVL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8	.000	.1116	-.2942	-.4345	-.2033	-.2621	-.2647
	.334	-.3900	-.1042	-.2190	-.2331	-.1976	-.3598
	.520	-.2043	-.1180	-.1094	-.2228	-.3532	-.1544
	.663	-.1075	-.1096	-.2527	-.2079	-.3654	-.1855
	.873	-.0468	-.1500	-.2752	-.3646	-.3999	-.2022

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.980 ANVL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8	.000	.2103	-.0329	-.2313	-.0237	-.1278	-.1982
	.334	.1578	.0995	-.0192	-.0639	-.0791	-.3432
	.520	-.0076	.0999	.0599	-.0453	-.2160	-.1078
	.663	.1325	.0768	-.0614	-.0461	-.2532	-.1207
	.873	.1253	-.0322	-.1569	-.2643	-.3560	-.2089



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CALCULATED SURFACE DATA - LAST

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## LAST-E 816/5/51 462 WATERS WING LOWER SURFACE

(RDW 61)

MACH = 1.11 = 1.05 ALPHAS = 9.96% M/L = 1.200 MACH = .165

## SECTION 11 WING

## DEFLECTION VARIABLE CP

X/C = .1500 3000 4500 6000 7500 9000

21/E .000 .2876 .1172 -.0731 -.019 .0126 -.0434

.334 .2162 .2249 .1316 .0621 .2396 -.2328

.320 .1549 .2259 .1992 .0628 -.1024 -.0730

.663 .2756 .1883 .0386 C621 -.1716 -.0686

.672 .2417 .0657 -.1616 -.1726 -.3269 -.2321

MACH = 1.1 = .165 ALPHAS = 9.96% M/L = 1.200 MACH = .165

## SECTION 11 WING

## DEFLECTION VARIABLE CP

X/C = .1500 3000 4500 6000 7500 9000

21/E .000 2474 2492 2481 2484 2485 2486

.334 31.63 31.63 32.62 32.62 32.62 32.62

.921 254.2 32.62 32.62 32.62 32.62 32.62

.663 355.7 35.84 35.84 35.84 35.84 35.84

.672 321.9 32.19 32.19 32.19 32.19 32.19

MACH = 1.1 = .165 ALPHAS = 9.96% M/L = 1.200 MACH = .165

## SECTION 11 WING

## DEFLECTION VARIABLE CP

X/C = .1500 3000 4500 6000 7500 9000

21/E .000 45.72 45.72 45.72 45.72 45.72 45.72

.334 35.97 35.97 35.97 35.97 35.97 35.97

.921 41.45 41.45 41.45 41.45 41.45 41.45

.663 42.45 42.45 42.45 42.45 42.45 42.45

.672 35.1 35.1 35.1 35.1 35.1 35.1

C457-B B16CSF1 J42 W7E10 WING LOWER SURFACE (ADVL#2) ( 12 NOV 73 )

## REFERENCE DATA

MACH	4.0120 00.01.	XREF	Z	43.5540 IN.	BETA	.000	P/M/P	=	1.500
LREF	10.2500 1IN.	YREF	Z	.0000 IN.	M/B	.039	BDFLAP	=	-10.000
BREF	37.9350 IN.	ZREF	Z	-.4030 IN.	ELEVON	.000			
SCALE	.0015								

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000			
21/8									
.000	.3640	.3574	.1160	.2095	.3039	.4357			
.334	.3082	.3253	.5170	.3200	.2*11	.0657			
.520	.2534	.3092	.2666	.2352	.0853	.0485			
.663	.2099	.2427	.1097	.0800	-.0864	-.0118			
.873	.1520	-.0503	-.2079	-.2430	-.4433	-.3461			

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.980 RNL = 1.200 MACH = .165

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000			
21/8									
.000	.4410	.2283	.3106	.4120	-.1306	-.0245			
.334	.4549	.4286	.0579	.4122	.5110	-.4364			
.520	.4112	.4347	.2567	.4346	.3868	.690			
.663	.4154	.16.5150	.4276	.3517	12.2764	-.0310			
.873	.3463	10.2923	.4050	.2176	11.9650	.1130			

MACH ( 1 ) = .165 ALPHA ( 3 ) = 19.995 RNL = 1.200 MACH = .165

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000			
21/8									
.000	.1203	.5148	.3389	.4953	.5948	.7390			
.334	.4753	.5102	.5102	.5065	.4667	-.1644			
.520	.5224	.4924	.4801	.4042	.3477	.1702			
.663	.3259	.0086	.2962	.2844	.0337	.0491			
.873	.2309	.0269	.0191	-.0194	-.3927	-.5227			

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TABULATED SOURCE DATA - CAA57B

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CA57-B B16CSF1 J42 WSTE10 WING LOWER SURFACE (RDYL 63) 1-12 NOV 73 )

## REFERENCE DATA

BREF	=	4.4120 SQ.FT.	XLRP =	43.5940 IN.	SETA =	.000	PTN/P =	1.350
LREF	=	19.2300 IN.	YLRP =	.0000 IN.	H/B =	.039	ADFLAC =	-10.000
BREF	=	37.9350 'N.	ZLRP =	-.4050 IN.	ELEVON =	.000		
SCALE	=	.0405						

MACH ( 1 ) = .153 ALPHA ( 1 ) = 9.995 RNL = 1.200 MACH = .65

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000		
21/B								
.000	.3760	.3473	.3179	.3016	.2764	.2259		
.334	.3166	.3366	.3147	.3112	.2167	.0612		
.520	.2670	.3132	.2859	.2346	.0688	.0231		
.663	.3033	.2446	.1079	.1145	-.1113	-.0370		
.873	.1665	-.0442	-.1943	-.2328	-.4276	-.3064		

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.985 RNL = 1.200 MACH = .65

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000		
21/B								
.000	.4550	.4322	.2808	.4205	.4651	.4094		
.334	.4089	.4379	.4272	.4162	.5582	-.0731		
.520	.4138	.4148	.3989	.3310	.2229	.0682		
.663	.3435	.3403	.2176	.2106	-.0302	-.0291		
.873	.2436	.0627	-.0597	-.1365	-.4082	-.4256		

MACH ( 1 ) = .165 ALPHA ( 3 ) = 20.000 RNL = 1.200 MACH = .65

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000		
21/B								
.000	.5245	.5167	.3661	.4956	.5627	.6085		
.334	.4707	.5093	.5019	.4954	.4357	-.1321		
.520	.5217	.4863	.4711	.3813	.3534	.0657		
.663	.3648	.4072	.2794	.2670	.0127	-.1127		
.873	.2466	.0415	.0273	-.0592	-.3753	-.4552		

REFERENCE DATA							PARAMETRIC DATA				
BREF	=	4.4120 SF.FT.	WHP	=	43.5940 IN.	BETA	=	.000	PTN/P	=	1.000
LREF	=	19.2500 IN.	WHP	=	.0000 IN.	H/B	=	.039	BDFLAP	=	-10.000
BREF	=	37.0350 IN.	WHP	=	-.4050 IN.	ELEVON	=	.000			
SCALE	=	.0405									
MACH ( 1) =	.165	ALPHA ( 1) =	10.000	RNL =	1.200	MACH	=	.165			
SECTION ( 1) WING							DEPENDENT VARIABLE CP				
X/C	.1500	.3000	.4500	.6000	.7500	.9000					
2/18											
.000	.3769	.3602	.1631	.3111	.1354	.0629					
.334	.3176	.3353	.3034	.2729	.1637	.0500					
.520	.2693	.3047	.2754	.2162	.0305	-.0156					
.663	.3137	.2445	.0878	.1134	-.1233	-.0795					
.873	.1851	-.0293	-.1980	-.2182	-.3862	-.2798					
MACH ( 1) =	.165	ALPHA ( 2) =	15.010	RNL =	1.200	MACH	=	.165			
SECTION ( 1) WING							DEPENDENT VARIABLE CP				
X/C	.1500	.3000	.4500	.6000	.7500	.9000					
2/18											
.000	.4602	.4414	.3222	.4374	.3482	.2175					
.334	.4119	.4332	.4152	.3800	.2999	.0867					
.520	.4148	.4083	.2833	.3145	.1674	.0093					
.663	.3534	.3435	.2034	.2039	-.0513	-.0749					
.873	.2516	.0173	-.0374	-.1357	-.3986	-.3800					
MACH ( 1) =	.165	ALPHA ( 3) =	19.990	RNL =	1.200	MACH	=	.165			
SECTION ( 1) WING							DEPENDENT VARIABLE CP				
X/C	.1500	.3000	.4500	.6000	.7500	.9000					
2/18											
.000	.5303	.5209	.4116	.5143	.4722	.3480					
.334	.4806	.5106	.4901	.4514	.3764	-.2022					
.520	.5247	.4822	.4806	.3618	.2469	-.0031					
.663	.4061	.4053	.2929	.2333	-.0213	-.1007					
.873	.2437	.0590	.0193	-.0969	-.4019	-.4177					



CA57-B B16C5F1 J42 WOTE16 WING LOWER SURFACE (RDV160) (12 NOV 73)

## REFERENCE DATA

BREF = 4.4120 SQ.FT. XMAP = 43.5940 IN.  
 LREF = 19.2300 IN. YMAP = .0000 IN.  
 BREF = 37.0350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.030 RN/L = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B

.000	.0280	-.4253	-.5720	-.2969	-.2042	-.1918
.334	-.2480	-.2090	-.3036	-.3067	-.2179	-.1312
.520	-.3601	-.2390	-.2327	-.2514	-.3246	-.1053
.663	-.3173	-.2102	-.4630	-.2577	-.3627	-.1555
.873	-.2561	-.2113	-.3149	-.3597	-.3506	-.1747

MACH ( 1 ) = .165 ALPHA ( 2 ) = .010 RN/L = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B

.000	.0920	-.2985	-.4934	-.2050	-.1760	-.2009
.334	-.1047	-.0830	-.2072	-.2429	-.1824	-.1616
.520	-.2051	-.1208	-.1260	-.1830	-.2932	-.1056
.663	-.1061	-.0788	-.3904	-.1821	-.3268	-.1354
.873	-.0321	-.0903	-.2286	-.2878	-.3229	-.1721

MACH ( 1 ) = .165 ALPHA ( 3 ) = 5.000 RN/L = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B

.000	.1626	-.1446	-.3815	-.1005	-.1143	-.1609
.334	.0470	.0540	-.0725	-.1477	-.1200	-.1154
.520	-.0641	.0515	.0089	.0706	-.2228	-.0913
.663	.1043	.0706	-.2862	-.0722	-.2511	-.1124
.873	.1323	-.0246	-.1280	-.2082	-.1992	-.1975

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## TABULATED SOURCE DATA - GA57B

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(RDVL68)

GA57-B B16CSF1 J42 WTE18 WING LOWER SURFACE

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.960 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .2448 .0079 -.2570 .0032 -.0328 -.1017

.334 .1741 .1875 .0679 -.0369 -.0194 -.0297

.520 .0966 .1930 .1414 .0504 -.1049 .0076

.663 .2745 .2031 -.1636 .0424 .1367 -.0380

.873 .2346 .0922 -.0304 -.1291 .2594 -.1989

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.010 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .3149 .1340 -.1573 .0811 .0212 -.0440

.334 .2480 .2707 .1664 .0324 .0326 -.0231

.520 .2254 .2801 .2432 .1190 -.0465 .0088

.663 .3586 .2734 -.0783 .0971 -.0850 -.0302

.873 .3197 .1426 .0162 -.0686 -.2532 -.2897

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .4095 .2648 -.0572 .1713 .0875 .0099

.334 .3363 .3632 .2758 .1196 .0923 .0018

.520 .3641 .3766 .3362 .1949 .0049 -.0136

.663 .4333 .3652 .0160 .1608 -.0454 -.1075

.873 .3579 .2014 .0794 -.0377 -.2807 -.4287

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TABULATED SOURCE DATA - CASTB

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REFERENCE DATA							PARAMETRIC DATA		
SECTION ( 1 ) WING				DEPENDENT VARIABLE CP					
BREF = 4.4120 63.51.	XWRF = .1940 IN.	ZWRF = .0000 IN.	MACH ( 1 ) = .165	ALPHA ( 1 ) = -3.990	RH/L = 1.200	MACH = .165	BETA = .000	PIN/P = 1.300	(RDYL69) ( 12 NOV 73 )
LREF = 19.2300 IN.	(WRF = .0000 IN.	ZWRF = -.4050 IN.					H/B = .246	BLDFLAP = -16.000	
BREF = 37.9350 IN.							ELEVON = .000		
SCALE = .0405									
SECTION ( 1 ) WING							DEPENDENT VARIABLE CP		
X/C .1500 .3000 .4500 .6000 .7500 .9000									
21/B	.000 .0337 -.4097 -.5121 -.2660 -.1804 -.1867								
	.334 -.2367 -.1681 -.3051 -.2865 -.2105 -.1525								
	.520 -.3531 -.2489 -.2274 -.2560 -.3218 -.1147								
	.663 -.3145 -.2090 -.1984 -.2600 -.3461 -.1513								
	.873 -.2543 -.2097 -.3173 -.3644 -.3499 -.1783								
MACH ( 1 ) = .165	ALPHA ( 2 ) = .005	RH/L = 1.200	MACH = .165	SECTION ( 1 ) WING			DEPENDENT VARIABLE CP		
X/C .1500 .3000 .4500 .6000 .7500 .9000									
21/B	.0938 -.2863 -.4285 -.1804 -.1459 -.1727								
	.334 -.0931 -.0640 -.1934 -.2242 -.1788 -.1549								
	.520 -.1928 -.1032 -.1107 -.1828 -.2856 -.1041								
	.663 -.0937 -.0749 -.1039 -.1752 -.3078 -.1305								
	.873 -.0225 -.0865 -.2253 -.2816 -.3186 -.1677								
MACH ( 1 ) = .165	ALPHA ( 3 ) = 5.010	RH/L = 1.200	MACH = .165	SECTION ( 1 ) WING			DEPENDENT VARIABLE CP		
X/C .1500 .3000 .4500 .6000 .7500 .9000									
21/B	.1616 -.1366 -.3004 -.0869 -.0917 -.1391								
	.334 .0575 .0676 -.0655 -.1359 -.1173 -.1137								
	.520 -.0562 .0540 .0231 -.0779 -.2197 -.0738								
	.663 .1041 .0650 .0051 -.0611 -.2463 -.1132								
	.873 .1314 .0012 -.1273 -.2127 -.2950 -.1548								

DATE 08 OCT 74

TABULATED SOURCE DATA - OA57B

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OA57-B B16C5F1 J42 W87E10 WING LOWER SURFACE (RDV168)

MACH ( 1 ) = .165 ALPHA ( 4 ) = 10.010 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .2441 .0034 -.2244 .0200 -.0275 -.0914  
.334 .1866 .1828 .0598 -.0499 .0369 .0491  
.520 .0835 .1870 .1357 .0247 -.1243 -.0237  
.663 .2584 .1892 .1147 .0120 -.1610 -.0622  
.873 .2439 .0841 -.0401 -.1421 -.2612 -.1974

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.990 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .3162 .1370 -.1227 .0996 .0357 -.0206  
.334 .2595 .2773 .1746 .0398 .0315 -.0280  
.520 .2322 .2793 .2475 .1148 -.0504 -.0028  
.663 .3587 .2635 .1590 .0638 -.1176 -.0560  
.873 .3249 .1439 .0167 -.0675 -.2472 -.2659

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.975 RN/L = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 .4119 .2597 -.0218 .1860 .0883 .0190  
.334 .3421 .3604 .2732 .1149 .0717 -.0194  
.520 .3641 .3677 .3331 .1837 -.0182 -.0411  
.663 .4379 .3710 .2194 .1253 -.0986 -.1282  
.873 .3596 .1979 .0651 -.0399 -.2756 -.3922

DATE 08 OCT 74

TABULATED SOURCE DATA - DATA

PAGE 110

C417-B B16C5F1 J42 W07E18 WING LOWER SURFACE

(RDV70) (12 NOV 73)

## REFERENCE DATA

BREF =	4.4120 84.FT.	XMRP =	43.5940 IN.
LREF =	19.2300 IN.	YMRP =	.0000 IN.
BREF =	37.9350 IN.	ZMRP =	-.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.025 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B	.0720	-.4075	-.4134	-.2401	-.2080	-.2061
.334	-.2318	-.1928	-.2956	-.2902	-.2337	-.1691
.520	-.3521	-.2602	-.2278	-.2674	-.3298	-.1201
.663	-.3156	-.2116	-.2161	-.2552	-.3497	-.1544
.873	-.2551	-.2142	-.3225	-.3695	-.3518	-.1633
MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165						
SECTION ( 1 ) WING						
X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B	.0973	-.2813	-.3491	-.1525	-.1659	-.1897
.334	-.0851	-.0564	-.2002	-.2234	-.1952	-.1774
.520	-.1959	-.1046	-.1182	-.1949	-.2964	-.1335
.663	-.1026	-.0739	-.1258	-.1817	-.3135	-.1492
.873	-.0271	-.0912	-.2315	-.2963	-.3193	-.1717
MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.960 RNL = 1.200 MACH = .165						
SECTION ( 1 ) WING						
X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B	.1540	-.1289	-.2640	-.1682	-.1016	-.1538
.334	.0490	.0672	-.0768	-.1521	-.1348	-.1408
.520	-.0599	.0468	.0139	-.0979	-.2354	-.1053
.663	.0964	.0579	.0265	-.0811	-.2584	-.1432
.873	.1308	-.0083	-.1416	-.2204	-.3052	-.2046

## PARAMETRIC DATA

BETA =	.000	PTM/P = 1.000
H/B =	.286	BDFLAP = -16.000
ELEVON =	.000	

DATE 06 OCT 74

TABULATED SOURCE DATA - OA378

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## CA57-B B16C5F1 J42 WTE10 WING LOWER SURFACE (RDVL70)

MACH ( 1) = .165 ALPHA ( 4) = 9.960 RN/L = 1.200 MACH = .165

## SECTION ( 1)WING DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B						
.000	.2380	.0086	-.1669	.0337	-.0310	-.1002
.334	.1664	.1961	.0559	.0615	.0550	-.0735
.520	.0852	.1734	.1277	.0086	.1549	-.0823
.663	.2543	.1766	.0853	.0196	.1976	-.1332
.873	.2407	.0735	-.0566	-.1463	-.2723	-.2080

MACH ( 1) = .165 ALPHA ( 5) = 15.000 RN/L = 1.200 MACH = .165

## SECTION ( 1)WING DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B						
.000	.3197	.1442	-.0690	.1144	.0338	-.0295
.334	.2531	.2816	.1713	.0318	.0145	-.0501
.520	.2237	.2771	.1455	.0954	-.0724	-.0400
.663	.3610	.2614	.1493	.0768	-.1343	-.0999
.873	.3344	.1436	.0220	-.0700	-.2467	-.2644

MACH ( 1) = .165 ALPHA ( 6) = 19.995 RN/L = 1.200 MACH = .165

## SECTION ( 1)WING DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B						
.000	.4138	.2625	.0240	.1943	.0784	-.0133
.334	.3569	.3583	.2615	.0993	.0444	-.0574
.520	.3778	.3601	.3218	.1635	-.0463	-.0795
.663	.4363	.3514	.1991	.1131	-.1286	-.1539
.873	.3691	.1957	.0607	-.0430	-.2664	-.3456

CA57-B 316C5F1 JAO WTE10 WING UPPER SURFACE (NOV03) (12 NOV 73)

## REFERENCE DATA

REFERENCE DATA				PARAMETRIC DATA		
BASE = 4.6120 SQ.FT.	XHWD = 43.5940 IN.			BETA = .000	P1N/P = 1.000	
LREF = 19.2300 IN.	YHWP = .0000 IN.			H/B = .39	BDFLAP = -10.000	
BREF = 37.9350 IN.	ZHWP = -.4050 IN.			ELEVON = .000		
SCALE = .0403						

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.0001 RHL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP			
X/C	.1500	.3000	.4500
			.6000
			.7500
			.9000

21/8

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.025 RHL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP			
X/C	.1500	.3000	.4500
			.6000
			.7500
			.9000

21/8

MACH ( 1 ) = .165 ALPHA ( 3 ) = 20.045 RHL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP			
X/C	.1500	.3000	.4500
			.6000
			.7500
			.9000

21/8

MACH ( 1 ) = .165 ALPHA ( 4 ) = 25.055 RHL = 1.200 MACH = .165

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CA57-B B16C5F1 JAO W7E10 WING UPPER SURFACE (RDVU04) (112 NOV 75)

## REFERENCE DATA

SREF = 4.4120 SQ.FT. XMAP = 43.5940 IN.  
 LREF = 10.2300 IN. YMAP = .0000 IN.  
 DREF = 37.9350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

## PARAMETRIC DATA

BETA = .000 PTN/P = 1.300  
 H/B = .039 BDFLAP = -10.000  
 ELEVON = .000

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.22229	-.3653	-.2157	-.1002	-.1385	-.1994
.334	-.7488	-.5516	-.4962	-.2760	-.1400	.0046
.520	-.7143	-.5598	-.4130	-.3140	-.2057	.0357
.663	-1.1292	-.7610	-.4546	-.3033	-.2586	.0740
.873	-.9420	-.8142	-.6644	-.4688	-.2890	-.2201

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.050 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.2603	-.4115	-.2152	-.1049	-.1464	-.1960
.334	-1.2572	-.6946	-.6133	-.3285	-.1691	-.0360
.520	-.7663	-.6352	-.5397	-.4092	-.3043	-.0328
.663	-1.1655	-.7451	-.5923	-.3621	-.1514	-.0540
.873	-2.3165	-1.5493	-1.1622	-.9076	-.7753	-.5820

MACH ( 1 ) = .165 ALPHA ( 3 ) = 19.960 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3099	-.4898	-.5440	-.2146	-.3061	-.2666
.334	-1.3562	-.8566	-.8202	-.5522	-.3478	-.1362
.520	-.8023	-.6032	-.9017	-.7884	-.6237	-.2838
.663	-1.5206	-1.3613	-1.1394	-.9023	-.5712	-.5096
.873	-1.1921	-1.0703	-1.0363	-.9468	-.8656	-.8606

(REFNO5) ( 12 NOV 75 )

## REFERENCE DATA

REF	4.4120 83.FT.	XMAP	=	43.5940 IN.
LREF	.19 2300 IN.	YMAP	=	.0000 IN.
ORL	.37.9999 IN.	ZMAP	=	-.4050 IN.
SCALE	= .0405			

MACH ( 1 ) = .165    ALPHA ( 1 ) = 9.990    R/L = 1.2000    MACH = .165

## SECTION ( 1 ) WING

## DEPEN. DEPEN VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.2087	-.3564	-.2050	-.3855	-.1267	-.1929
.334	-.7368	-.5218	-.1979	-.2390	-.1179	-.02 3
.520	-.7032	-.5921	-.4290	-.2683	-.1650	.6510
.663	-.1.0915	-.7370	-.4510	-.2683	-.0407	.0757
.873	-.88632	-.7789	-.6360	-.4452	-.2560	-.1934

MACH ( 1 ) = .165    ALPHA ( 2 ) = 15.000    R/L = 1.2000    MACH = .165

## SECTION ( 1 ) WING

## DEPEN. DEPEN VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.2338	-.4C.3	-.2322	-.0947	-.1416	-.1918
.334	-.1.2339	-.6768	-.5834	-.3053	-.1720	-.0172
.520	-.7551	-.6450	-.5119	-.3807	-.2776	-.0077
.663	-.1.1446	-.7203	-.4999	-.3541	-.1372	-.0353
.873	-.1.7315	-.1.2919	-.1.C044	-.8347	-.5755	-.5358

MACH ( 1 ) = .165    ALPHA ( 3 ) = 20.000    R/L = 1.2000    MACH = .165

## SECTION ( 1 ) WING

## DEPEN. DEPEN VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3107	-.4794	-.3625	-.2213	-.2052	-.2755
.334	-.1.3380	-.6332	-.7995	-.1473	-.3403	-.1365
.520	-.7957	-.6007	-.6486	-.7725	-.6059	-.2790
.663	-.1.4242	-.1.2642	-.1.0727	-.9005	-.5530	-.5054
.873	-.1.1243	-.1.2446	-.9414	-.5032	-.5213	-.8193

DATE 08 OCT 74

TABULATED SOURCE DATA - C4570

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CA57-0 816CSF1 J40 WATE18 WING UPPER SURFACE (RDYDUE) ( 12 NOV 73 )

## REFERENCE DATA

BREF	#	4.4120 83.FT.	XMAP	=	43.5940 IN.		BETA	=	.000	PIN/P	=	1.000
LREF	#	10.2300 IN.	XMAP	=	.0000 IN.		H/B	=	.125	EDFLAP	=	-16.000
BREF	#	37.9350 IN.	ZMAP	=	-.4050 IN.		ELEVON	=	.000			
SCALE	#	.0405										
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	-3.980	RN/L =	1.200	MACH =	.165					

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	-.0570	-.1640	-.1035	-.0029	-.0567	-.1759
.334	-.1641	-.2415	-.3046	-.1770	-.0267	.0393
.520	-.3151	-.3678	-.3272	-.2067	-.0967	-1243
.663	-.3684	-.4037	-.3071	-.2155	-.0265	.1169
.873	.0854	-.3356	-.2796	-.2056	-.1019	.0089

MACH ( 1 ) = .165 ALPHA ( 2 ) = .010 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	-.0911	-.2049	-.1367	-.0331	-.0716	-.1644
.334	-.2676	-.3252	-.3453	-.1917	-.0339	.039
.520	-.4567	-.4433	-.5611	-.2171	-.0994	-.0690
.663	-.6249	-.5335	-.3676	-.2384	-.0222	.1253
.873	-.1123	-.4687	-.3612	-.2424	-.1043	-.0103

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.980 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	-.1480	2.506	-.1741	-.0672	-.0833	-.1672
.334	-.5131	-.5211	-.4412	-.2194	-.0424	.0404
.520	-.6639	-.5117	-.3744	-.2301	-.1063	-.0325
.663	-.9205	-.6871	-.4165	-.2646	-.0420	.1158
.873	-.4747	-.6350	-.4905	-.3296	-.1686	-.0169





## CA37-B 810CSF1 J40 WSTE10 WING UPPER SURFACE

(RDY000 1 12 NOV 73 )

## REFERENCE DATA

SPEC	4.4120 BA.FT	ZMAP	=	43.5940 IN.
LREF	.19.2500 IN.	INRP	=	.0000 IN.
BREF	.37.9330 IN.	PRP	=	-.4050 IN.
SCALE	.0405			

MACH ( 1 ) = .165    ALPHA ( 1 ) = -.5.000    RNL = 1.200    MACH = .165  
 SECTION ( 1 ) WING  
 R/C .1500 .3000 .4500 .6000 .7500 .9000

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 R/C .1500 .3000 .4500 .6000 .7500 .9000

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 R/C .1500 .3000 .4500 .6000 .7500 .9000

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SECTION ( 1 ) WING  
 R/C .1500 .3000 .4500 .6000 .7500 .9000

## PARAMETRIC DATA

BETA	= .000	PTN/P = 1.300
M/B	= .125	BDFLAP = -.16.000
ELEVON	= .000	

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(ADJUST)

\*197-B 51665FL 140 MPH 10 WING UPPER SURFACE

MACH ( 1 ) = .165    ALPHA ( 4 ) = 9.975    RNL = 1.000    MACH = .165

## SECTION ( 1 ) WING

DEFINENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B	.0000	-.2022	.2646	-.2508	.0778	-.1044	.1720
	.734	-.7141	-.5065	-.4773	-.2586	-.0226	.0203
	.520	-.7007	-.5697	-.4976	-.2845	-.1517	.0614
	.663	-.1.1361	-.7842	-.4779	-.3317	-.0515	.1210
	.673	-.6940	-.6021	-.6463	-.4520	-.2729	-.1511

MACH ( 1 ) = .165    ALPHA ( 5 ) = 15.000    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

DEFINENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B	.000	-.2439	-.3192	-.2288	-.0817	-.1198	.1824
	.334	-1.1957	-.5823	-.5590	-.2946	.0215	-.043C
	.520	-.7840	-.6354	-.5885	-.3295	-.2354	-.2713
	.663	-.1.2466	-.7953	-.5071	-.3366	-.1167	-.0203
	.673	-.1.6243	-.1.1487	-.9507	-.7052	-.5350	-.4457

MACH ( 1 ) = .165    ALPHA ( 6 ) = 19.975    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

DEFINENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B	.000	-.2791	-.3835	-.2835	-.1534	-.2416	.2447
	.334	-1.4732	-.7505	-.7171	-.4629	-.2339	-.1286
	.520	-.7231	-.6652	-.7785	-.6863	-.6277	-.3489
	.663	-.1.2829	-.1.2526	-.1.1095	-.1.0167	-.4932	-.4695
	.673	-.1.1125	-.1.0354	-.1.0077	-.9244	-.8615	-.8626

(ND4000) (12 NOV 73)

CAST-B B16C5F1 J40 WITE10 WING UPPER SURFACE

## REFERENCE DATA

BREF	=	4.4120 SQ.FT.	XMAP	=	43.5940 IN.
LREF	=	19.2500 IN.	YMAP	=	.0900 IN.
BREF	=	37.6350 IN.	ZMAP	=	-.4050 IN.
SCALE	=	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.015 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	,1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

2/8

	.000	-.0604	-.1934	-.0842	-.0009	-.0620	-.1766
	.334	-.1354	-.2315	-.2756	-.1193	.1662	.2116
	.520	-.2623	-.3360	-.3092	-.1473	.0818	.1138
	.663	-.2804	-.4066	-.2886	-.1949	-.0375	.0693
	.873	.1361	-.3217	-.2694	-.2024	-.1043	.0015

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.030 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	,1,00	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

2/8

	.000	-.0916	-.1901	-.1343	-.0283	-.0763	-.1745
	.334	-.2613	-.3065	-.3264	-.1509	.1899	.1924
	.520	-.4513	-.4125	-.3674	-.1742	-.1077	.0525
	.663	.3375	-.5414	-.3498	-.2319	-.0370	.0880
	.873	-.0473	-.4432	-.3448	-.2367	-.1027	-.0124

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.990 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

2/8

	.000	-.1343	-.2238	-.1527	-.0460	-.0712	-.1586
	.334	-.4616	-.4368	-.107	.1660	.1127	.1728
	.520	-.5450	-.4400	.3707	-.1700	-.1356	.0464
	.663	-.7934	-.6030	-.3708	-.2404	-.0310	.0349
	.873	-.3751	-.5643	.4516	-.2978	-.1448	.0023

SECTION ( 1 ) WING

MACH ( 1 ) = .165    ALPHA ( 4 ) = 6.980    RNL = 1.200    MACH = .165

X/C    .1500    .3000    .4500    .6000    .7500    .9000

SECTION ( 1 ) WING  
DEPENDENT VARIABLE CP

X/C	.000	.334	.520	.663	.813
	-.1894	-.4201	-.5150	-.7367	-.8048
	-.6611	-.4433	-.4604	-.4461	-.7423
	-.2163	-.2318	-.3121	-.5964	-.3954
	-.0617	-.0875	-.0469	-.3954	-.2198
	-.0772	-.0875	-.0469	-.1057	

MACH ( 1 ) = .165    ALPHA ( 5 ) = 14.950    RNL = 1.200    MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C	.000	.334	.520	.663	.813
	-.2403	-.3239	-.2113	-.0727	-.0915
	-.5901	-.5521	-.2642	-.1120	-.0310
	-.5991	-.5071	-.2702	-.1552	-.0516
	-.8710	-.5208	-.2775	-.0702	-.0422
	-.1.1714	-.9057	-.6188	-.5538	-.4120

MACH ( 1 ) = .165    ALPHA ( 6 ) = 19.965    RNL = 1.200    MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C	.000	.334	.520	.663	.813
	-.2774	-.4193	-.2893	-.1420	-.2529
	-.6965	-.6981	-.7259	-.4686	-.2154
	-.6952	-.8383	-.7004	-.5531	-.3157
	-.1.2180	-1.1660	-1.0022	-.5743	-.5172
	-.0935	-.0760	-.9963	-.8599	-.8742

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TABULATED SOURCE DATA - CAST80

CAST-8 816C5F1 J40 WATE10 WING UPPER SURFACE (RDYU19) ( 12 NOV 73 ) PAGE 130

## REFERENCE DATA

BREF =	4.4120 38.FT.	XMRP =	43.3940 IN.
LPEF =	19.2300 IN.	IMRP =	.0000 IN.
BREF =	37.9350 IN.	ZMRP =	-.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165    ALPHA ( 1 ) = -3.985    RN/L = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

21/B

	.000	-.0462	-.1546	-.0767	.0113	-.0349	-.1508
	.334	-.1299	-.2199	-.2618	-.1375	.0523	.0786
	.520	-.2959	-.3023	-.2653	-.1619	-.0005	.0986
	.663	-.3100	-.3634	-.2618	-.1700	.0817	.1296
	.873	.1544	-.3144	-.0713	-.3045	-.0691	.0210

MACH ( 1 ) = .165    ALPHA ( 2 ) = .015    RN/L = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

21/B

	.000	-.0854	-.1990	-.1297	-.0210	-.0585	-.1541
	.334	-.2681	-.3027	-.3204	-.1657	.0254	.0676
	.520	-.4574	-.3901	-.3446	-.1920	-.0762	.0840
	.663	-.1681	-.5077	-.3539	-.2082	.0412	.1209
	.873	-.0582	-.4193	-.1391	-.3514	-.0793	.0054

MACH ( 1 ) = .165    ALPHA ( 3 ) = 5.000    RN/L = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
-----	-------	-------	-------	-------	-------	-------

21/B

	.000	-.1390	-.2596	-.1678	-.0108	-.0814	-.1547
	.334	-.4015	-.4237	-.3796	-.2070	-.0259	.0512
	.520	-.5823	-.4656	-.3578	-.1970	-.0808	.0055
	.663	-.8356	-.6333	-.3661	-.2323	.0047	.1035
	.873	-.3601	-.5275	-.2407	-.4384	-.1352	-.0404



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TABULATED SOURCE DATA - C4578

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MACH ( 1 ) = .165 ALPHA ( 4 ) = 10.015 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 -.1987 -.3225 -.12080 -.0132 -.1012 -.1787

.334 -.7194 -.5254 -.4720 -.2336 -.0443 .0410

.520 -.6453 -.5375 -.4325 -.2569 -.1252 .0160

.663 -.1.1001 -.7656 -.4915 -.3176 -.0093 .1074

.873 -.8642 -.7242 -.5492 -.3914 -.1804

MACH ( 1 ) = .165 ALPHA ( 5 ) = 11.975 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 -.2493 -.3797 -.2224 -.0806 -.1140 -.2019

.334 -.1.1911 -.7298 -.5985 -.3037 -.1353 -.0002

.520 -.7705 -.5799 -.4898 -.3104 -.1194 -.135

.663 -.1.2949 -.6978 -.5389 -.3287 .1359 .02 L

.873 -.1.7249 -.1.1514 -.9130 -.6085 -.1010 -.4341

MACH ( 1 ) = .165 ALPHA ( 6 ) = 20.000 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .600 -.2718 -.4248 -.2587 -.1305 -.2096 -.2662

.334 -.1.4497 -.8623 -.7485 -.4861 -.2052 -.0894

.520 -.7342 -.6178 -.7333 -.6656 -.5057 -.2462

.663 -.1.1141 -.1.1538 -.1.1116 -.1.0045 -.5033 -.4155

.873 -.1.0323 -.1.0013 -.1.0514 -.9964 -.8934 -.6657

OA57-8 B1&amp;C5F1 JAO NOTE18 WING UPPER SURFACE

(FDVU16) (112 NOV 73 )

## REFERENCE DATA

BASE =	4.4120 83 FT.	XHAR =	43.3940 IN.	BETA =	.000	PTN/P =	.1 .150
LREF =	19.2500 IN.	XHAR =	.0000 IN.	M/B =	.286	BDFLAP =	-16.000
BREF =	37.8550 IN.	ZHAR =	-.4050 IN.	ELEVON =	.000		
SCALE =	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.985 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.0435 -.1659 -.0835 .0185 -.0345 -.1433

.334 -.1134 -.2077 -.2587 -.1316 .0127 .1027

.520 -.2780 -.2814 -.2812 .11525 -.0479 .1686

.663 -.2875 -.2815 -.2539 -.1605 .0625 .1444

.873 .1434 -.3439 -.3052 -.2158 -.0068 .0596

MACH ( 1 ) = .165 ALPHA ( 2 ) = .015 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.0804 -.2177 -.1314 -.0193 -.0600 -.1433

.334 -.2312 -.2692 -.3157 -.1596 .0289 .0867

.520 -.4421 -.3703 -.3406 -.1651 -.0483 .1455

.663 -.5520 -.4268 -.3205 -.2057 .0276 .1278

.873 -.0322 -.4867 -.3796 -.2653 -.0504 .0391

MACH ( 1 ) = .165 ALPHA ( 3 ) = 5.000 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.1463 -.2743 -.1728 -.0590 -.0883 -.1431

.334 -.4080 -.4196 -.3802 -.2253 -.0376 .0486

.520 -.5883 -.4653 -.3651 -.2185 -.0794 .0321

.663 -.6516 -.5698 -.3997 -.2425 -.0124 .0825

.873 -.5866 -.6426 -.4975 -.3177 -.1217 -.0354

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TABULATED SOURCE DATA - CA57B

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(REV 6)

MACH ( 1 ) = .165 ALPH ( 1 ) = 10.000 R/H/L = 1.200 MACH = 1.65

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

X/C	.000	.334	.520	.663	.873
	-.2174	-.3564	-.5417	-.7129	-.8117
	-.7490	-.5329	-.5417	-.61098	-.80200
	-.4113	-.2371	-.4475	-.4631	-.62028
	-.0370	.0230	-.2597	-.3132	-.4017
	.0230	.0268	-.1171	-.0664	-.1596

MACH ( 1 ) = .165 ALPH ( 1 ) = 15.010 R/H/L = 1.200 MACH = ..

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

X/C	.000	.334	.520	.663	.873
	-.2249	-.3932	-.5914	-.8104	-.9443
	-.0893	-.1528	-.1714	-.2230	-.3744
	-.1643	-.3191	-.3206	-.3862	..
	..	..	..	..	..

MACH ( 1 ) = .165 ALPH ( 1 ) = 20.000 R/H/L = 1.200 MACH = ..

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

X/C	.000	.334	.520	.663	.873
	-.2816	-.4499	-.8675	-.1365	-.3080
	-.8124	-.6675	-.7038	-.6678	-.2252
	-.4126	-.2378	-.4326	-.5590	-.0820
	-.0820	-.1493	..	-.4354	..
	..	..	..	..	..

MACH ( 1 ) = .165 ALPH ( 1 ) = 25.000 R/H/L = 1.200 MACH = ..

## CAST-8 ATC5F1 J40 WTE10 WING UPPER SURFACE

(REV017) (12 NOV 73)

## REFERENCE DATA

BREF = 4.4120 88.FT.  
 LREF = 19.2300 IN.  
 BREF = 37.930 IN.  
 SCALE = .0405

MACH + 1) = 1.65    ALPHA + 1) = -3.975    RFL = 1.2000    MACH = .165

SECTION (1) WING

x/c 1.000 .3000 .4500 .6000 .7500 .9000

2. P DEPENDENT VARIABLE CP

x/c	CP	x/c	CP	x/c	CP	x/c	CP	x/c	CP
.250	-1.9425	.1705	-1.0657	.0136	-1.5394	-.1125			
.334	-1.9184	.2986	-2.580	.1332	.5720	.0444			
.420	-1.2846	.2930	-2.877	-.1613	-.0820	.1174			
.507	-.2117	-.2974	-.2596	-.1575	.0477	.1247			
.603	-.1425	-.3534	-.3011	-.2426	-.0113	.0538			

3. CP (1) = 1.65    ALPHA (1) = .000    RFL = 1.2000    MACH = .165

SECTION (1) A(1)

x/c 1.000 .3000 .4500 .6000 .7500 .9000

4. A(1) DEPENDENT VARIABLE CP

x/c	A(1)	x/c	A(1)	x/c	A(1)	x/c	A(1)	x/c	A(1)
.250	1.2196	.1245	-.0279	-.0653	-.1408				
.334	2.114	.3006	-.3133	.1511	.0193	.0805			
.420	1.455	-.3035	-.3426	-.1955	-.0736	.1192			
.507	1.521	-.4317	-.3326	-.2104	.0213	.1155			
.603	-.0475	-.4775	-.3595	-.2850	-.0491	.0363			

5. MACH + 1) = 1.65    ALPHA (1) = 5.000    RFL = 1.3000    MACH = .165

SECTION (1) WING

x/c 1.000 .3000 .4500 .6000 .7500 .9000

6. MACH (1) DEPENDENT VARIABLE CP

x/c	MACH (1)	x/c	MACH (1)	x/c	MACH (1)	x/c	MACH (1)	x/c	MACH (1)
.250	-.14410	.2860	-.1674	-.0532	-.0947	-.1402			
.334	-.4044	-.4303	-.3969	-.2165	-.0447	.0474			
.420	-.5702	-.4601	-.3683	-.2072	-.0659	.0142			
.507	-.5478	-.5774	-.3913	-.2469	-.0119	.0832			
.603	-.3741	-.6226	-.4554	-.3439	-.1201	-.0233			

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## TABULATED SOURCE DATA - CASTB

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MACH ( 1 ) = .165    ALPHA ( 4 ) = 9.975    RNL = 1.200    MACH = .165  
 SECTION ( 1 )WING  
 X/C    .1500    .3000    .4500    .6000    .7500    .9000  
 DEPENDENT VARIABLE CP

21/8

.000    -.2012    -.3501    -.2013    -.0793    -.1089    -.1555  
 .334    -.7155    -.5309    -.4542    -.2429    -.0677    .0390  
 .520    -.6663    -.5462    -.4373    -.2606    -.1414    .0153  
 .663    -.1.0915    -.7130    -.4912    -.3134    -.0367    .0679  
 .873    -.8287    -.7872    -.6054    -.4381    -.2670    -.1696

MACH ( 1 ) = .165    ALPHA ( 5 ) = 15.000    RNL = 1.200    MACH = .165  
 SECTION ( 1 )WING  
 X/C    .1500    .3000    .4500    .6000    .7500    .9000  
 DEPENDENT VARIABLE CP

21/8

.000    -.2452    -.3992    -.2285    -.0813    -.1007    -.1773  
 .334    -.1.1622    -.7128    -.5949    -.2826    -.1367    -.0039  
 .520    -.7713    -.5927    -.4592    -.3046    -.2016    .0106  
 .663    -.1.2576    -.6095    -.5169    -.3133    -.0184    .0180  
 .873    -.1.6147    -.1.1032    -.8870    -.6636    -.5305    -.4911

MACH ( 1 ) = .165    ALPHA ( 6 ) = 20.005    RNL = 1.200    MACH = .165  
 SECTION ( 1 )WING  
 X/C    .1500    .3000    .4500    .6000    .7500    .9000  
 DEPENDENT VARIABLE CP

.000    -.2826    -.4567    -.2691    -.1318    -.1824    -.2199  
 .334    -.1.4884    -.8622    -.7030    -.4364    -.2775    -.0790  
 .520    -.7649    -.6690    -.6322    -.5564    -.4478    -.1806  
 .663    -.1.3375    -.1.3212    -.9773    -.8007    -.3692    -.2777  
 .873    -.1.3041    -.1.1653    -.1.1061    -.1.0834    -.8955    -.6798

B16C5F1 JAO WTE10 WING UPPER SURFACE  
 (ADJUST)

CA57-B

OA57-B B16C3F1 J40 WATE10 WING UPPER SURFACE

(RDVU26) ( 12 NOV 73 )

## REFERENCE DATA

REF	4,4120 SQ.FT.	XPC	Z	BETA	PIN/P	M/B	BDFLAP
LREF	19.2300 IN.	MRP	=	.0000 IN.	= .000	= .125	= -16.000
SREF	37.9350 IN.	ZMRP	=	-.4050 IN.			
SCALE	.0405						
MACH ( 1 ) = .165	ALPHA ( 1 ) = -4.073	RNL	= 1.200	MACH	= .165		

## SECTION ( 1 ) WING

## PARAMETRIC DATA

X/C	.1500	.3000	.4500	.6000	.7500	.9000	DEPENDENT VARIABLE CP
.000	-.0857	-.1939	-.1454	-.0362	-.0917	-.2169	
.334	-.2280	-.2562	-.3702	-.2991	-.3313	-.1462	
.520	-.3271	-.4080	-.3648	-.3465	-.4303	-.0751	
.663	-.4112	-.2233	-.4301	-.3942	-.6685	-.0314	
.873	.0509	-.5342	-.5908	-.4248	-.3007	-.1174	

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.020 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	DEPENDENT VARIABLE CP
.000	-.1336	-.2509	-.1944	-.0645	-.1176	-.1682	
.334	-.3646	-.3339	-.4295	-.3343	-.3273	-.1612	
.520	-.5006	-.5006	-.4376	-.3801	-.3939	-.1297	
.663	-.7046	-.3892	-.5204	-.4405	-.6431	-.0702	
.873	-.1761	-.6875	-.6628	-.5520	-.3584	-.1796	

MACH ( 1 ) = .165 ALPHA ( 2 ) = 4.95 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	DEPENDENT VARIABLE CP
.000	-.2089	-.3182	-.2362	-.1045	-.1514	-.1687	
.334	-.5587	-.5375	-.1139	-.3866	-.3404	-.1690	
.520	-.6496	-.5929	-.4732	-.3916	-.3142	-.2205	
.663	-1.0077	-.5482	-.6080	-.4801	-.6131	-.1625	
.873	-.6200	-.8793	-.8305	-.6671	-.5021	-.3352	



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TABULATED SOURCE DATA - CASTB

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CA57-B B16C5F1 J40 .3TE18 WING UPPER SURFACE (RDVU26)						
MACH ( 1 ) = .165	ALPHA ( 4 ) = 9.965	RH/L	= 1.200	MACH	= .165	
<b>SECTION ( 1 )WING</b>						
V/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3044	-.3692	-.2626	-.1306	-.1545	-.1629
.334	-.6611	-.6280	-.5684	-.4037	-.3469	-.3750
.520	-.7391	-.6196	-.5497	-.4328	-.4074	-.4918
.663	-.1.2390	-.7105	-.6703	-.5252	-.5820	-.1.382
.873	-.1.2115	-.1.1232	1.0110	-.7934	-.6620	-.5016
MACH ( 1 ) = .165	ALPHA ( 5 ) = 14.965	RH/L	= 1.200	MACH	= .165	
<b>SECTION ( 1 )WING</b>						
A/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3667	-.4293	-.3092	-.1616	-.1793	-.1924
.334	-.9853	-.7955	-.7043	-.4763	-.5609	-.2454
.520	-.8258	-.7058	-.6548	-.5226	-.5203	-.2262
.663	-.1.3011	-.7731	-.6243	-.5301	-.4412	-.2549
.873	-.7741	-.1.9679	-.1.5779	-.1.5845	-.1.1419	-.8307
MACH ( 1 ) = .165	ALPHA ( 6 ) = 19.990	RH/L	= 1.200	MACH	= .165	
<b>SECTION ( 1 )WING</b>						
A/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.4084	-.5155	-.4674	-.2890	-.3156	-.2729
.334	-.1.3077	-.4816	-.9203	-.6989	-.5174	-.5758
.520	-.8628	-.9415	-.0191	-.8642	-.6535	-.5445
.663	-.1.3979	-.1.3016	-.1.1979	-.1.1339	-.4747	-.7103
.873	-.1.2220	-.1.1672	-.1.1603	-.1.0676	-.5649	-.7879



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TABULATED SOURCE DATA - CASTE

PAGE 139

CA57-A 616C5F1 J40 WATE10 WING UPPER SURFACE (RDYU23)

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.385 R/H/L = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/6	.000	-.3097	-.3943	-.2584	-.1270	-.1694	-.1675
	.334	-.8011	-.5881	-.5928	-.5633	-.0841	-.3121
	.520	-.7778	-.6204	-.5422	-.4126	-.3637	-.1756
	.663	-.1.2686	-.6576	-.6663	-.5017	-.5472	-.1411
	.873	-.1.2758	-.1.1259	-.1.0163	-.6213	-.6403	-.5234

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.950 R/H/L = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT V, ABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/6	.000	-.3399	-.4420	-.3150	-.1710	-.2114	-.2029
	.334	-.3521	-.7743	-.7082	-.4512	-.3149	-.3660
	.520	-.8401	-.7610	-.6511	-.5134	-.5324	-.2664
	.663	-.1.3670	-.7157	-.6265	-.5036	-.4638	-.2436
	.873	-.2.3243	-.2.4098	-.1.8661	-.1.2916	-.1.2563	-.0395

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.950 R/H/L = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/6	.000	-.3637	-.4745	-.4117	-.2950	-.3528	-.2692
	.334	-.1.6295	-.8961	-.9172	-.5717	-.5421	-.1443
	.520	-.8465	-.8752	-.1.0033	-.8934	-.8475	-.5444
	.663	-.1.3650	-.3376	-.1.2427	-.1.1703	-.9479	-.6920
	.873	-.1.1975	-.1.1694	-.1.1315	-.1.0743	-.9945	-.6599

CAST-B BLOCK1 J40 WTE10 WING UPPER SURFACE

(ADVISOR) (112 NOV 75)

## REFERENCE DATA

SREF =	4.4180 83.1.	XWAP =	43.3940 IN.	BETA =	.000	PTN/P =	1.000
LREF =	19.2300 IN.	(NAP =	.0000 14.	M/B =	.125	BOFLAP =	-16.000
SREF =	37.9350 IN.	ZWAP =	-.4050 IN.	ELEVON =	15.000		
SCALE =	.0403						

MACH (1) = .165 ALPHA (1) = -4.000 RHL = 1.200 MACH = .165

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.3784 -.224 -.1160 -.0314 -.0962 -.1697

.334 -.3124 -.2703 -.3527 -.2939 -.2813 -.2876

.520 -.3609 -.4285 -.3441 -.3522 -.4261 -.1206

.663 -.4370 -.1262 -.4442 -.4145 -.7583 -.0439

.873 -.0103 -.3553 -.5975 -.4052 -.3155 -.1551

MACH (1) = .165 ALPHA (2) = -.020 RHL = 1.200 MACH = .165

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.1304 -.2920 -.1693 -.0646 -.1316 -.1635

.334 -.4519 -.3701 -.4132 -.3365 -.2843 -.2620

.520 -.3235 -.5101 -.4196 -.3860 -.3754 -.1551

.663 -.7259 -.272 -.5198 -.4542 -.694 -.0944

.873 -.1978 -.6940 -.6870 -.5207 -.3721 -.1604

MACH (1) = .165 ALPHA (3) = 4.965 RHL = 1.200 MACH = .165

## SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.2320 -.3468 -.2310 -.1072 -.1357 -.1738

.334 -.6554 -.5609 -.5413 -.3628 -.1717 -.2561

.520 -.6763 -.5843 -.4708 -.3969 -.3293 -.1679

.663 -.10266 -.6842 -.6075 -.4922 -.6904 -.1432

.873 -.6729 -.8673 -.6145 -.6478 -.5147 -.3360



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## TABULATED SOURCE DATA - CASTE

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CAST-B B16C5F1 J4D WTE10 WING UPPER SURFACE (RDVU30)							
SECTION ( 1)WING		DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.2779	-.4124	-.2568	-.1187	-.1787	-.1635
	.334	-.6582	-.3812	-.5832	-.3631	-.3021	-.2627
	.520	-.7315	-.6213	-.3416	-.4037	-.3865	-.1268
	.663	-.1.2746	-.6517	-.6824	-.4139	-.5574	-.1.477
	.873	-.1.3374	-.1.1324	-.1.0184	-.6500	-.7079	-.5437
MACH ( 1) =	.165	ALPHA ( 5) =	14.980	RH/L =	1.200	MACH =	.165
SECTION ( 1)WING		DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.3312	-.4509	-.2965	-.1604	-.1879	-.1919
	.334	-1.4802	-.8019	-.7195	-.4357	-.3365	-.3100
	.520	-.8155	-.7573	-.6706	-.5126	-.5491	-.2022
	.663	-.1.4604	-.6304	-.6543	-.5129	-.4486	-.2319
	.873	-.2.4793	-.2.3785	-.2.1294	-.1.5408	-.1.4094	-.8636
MACH ( 1) =	.165	ALPHA ( 6) =	19.980	RH/L =	1.200	MACH =	.165
SECTION ( 1)WING		DEPENDENT VARIABLE CP					
X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.3792	-.5189	-.4242	-.2621	-.3217	-.3071
	.334	-.1.9865	-.9405	-.9149	-.6436	-.5004	-.4476
	.520	-.7006	-.9326	-.9333	-.8774	-.8562	-.5194
	.663	-.1.4634	-.1.2263	-.1.1669	-.1.1213	-.9276	-.5807
	.873	-.1.1406	-.1.1078	-.1.1110	-.1.0702	-.9805	-.8526

C437-B B16C5F1 J40 W07E10 WING UPPER SURFACE

(RDWU31) (12 NOV 73)

## REFERENCE DATA

	X/W	Y/W	Z/W	XWP	YWP	ZWP	RNL	MACH	PNTAP	BDFLAP
BWF	4.4180	.63	.FT.			43.9940	.IN.		.000	.039
LAEF	18.2300	.IN.				.0000	.IN.			
BAEF	37.9350	.IN.				-.4050	.IN.			
SCALE	.0405									

MACH ( 1 ) = .165    ALPHA ( 1 ) = 10.010    RNL = 1.200    MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3226	-.4468	-.2774	-.1323	-.1753	-.1778
.334	-.7911	-.8066	-.5705	-.3732	-.3333	-.1096
.520	-.7578	-.6295	-.5235	-.4044	-.3936	-.0509
.663	-1.2336	-.6320	-.5984	-.4291	-.2169	-.1240
.873	-1.1617	-1.1023	-.9884	-.7946	-.6493	-.4723

MACH ( 1 ) = .165    ALPHA ( 2 ) = 15.005    RNL = 1.200    MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3456	-.4825	-.2764	-.1340	-.1753	-.1778
.334	-1.3409	-.8261	-.6756	-.4064	-.3051	-.2053
.520	-.8134	-.7357	-.6396	-.5078	-.5055	-.2166
.663	-1.3158	-.8674	-.6081	-.5007	-.3862	-.2585
.873	-2.6604	-1.9406	-1.3463	-1.1532	-1.3140	-.8445

MACH ( 1 ) = .165    ALPHA ( 3 ) = 20.000    RNL = 1.200    MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3679	-.5382	-.3843	-.2574	-.3128	-.2830
.334	-1.6310	-.8643	-.8596	-.6301	-.4494	-.3163
.520	-.8026	-.8205	-.9794	-.8995	-.7916	-.6912
.663	-1.6232	-1.4277	-1.2336	-1.1136	-.6547	-.7167
.873	-1.2424	-1.1774	-1.1259	-1.0503	-.9620	-.6226

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TABULATED SOURCE DATA - OA57B

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OA57-B B16C5F1 J40 WATE18 WING UPPER SURFACE

(RDVU32) ( 12 NOV 73 )

## REFERENCE DATA

REFERENCE DATA				PARAMETRIC DATA		
RREF = 4.1120 SQ.FT.	XWRF = 43.5940 IN.			BETA = .000	PTN/P = 1.300	
LREF = 19.2300 IN.	YWRF = .0000 IN.			M/B = .039	BDFLAP = -16.000	
WREF = 37.9350 IN.	ZWRF = -.4050 IN.			ELEVON = 15.000		
SCALE = .0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = 9.990 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	-.2949	-.4347	-.2680	-.1243	-.1632	-.1699
.334	-.7906	-.6067	-.5576	-.3583	-.1763	-.2019
.520	-.7315	-.6084	-.5192	-.3697	-.3723	-.0844
.663	-.1.2195	-.6239	-.5886	-.4126	-.2181	-.1191
.873	-1.1752	-1.1137	-.9603	-.8003	-.6712	-.5015

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.970 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	-.3310	-.4633	-.2782	-.1412	-.1792	-.1950
.334	-1.3108	-.3167	-.6613	-.3791	-.2138	-.2437
.520	-.7720	-.7215	-.6189	-.4650	-.4583	-.2264
.663	-1.3583	-.6781	-.5908	-.4960	-.3577	-.2236
.873	-2.6579	-2.0456	-1.6822	-1.1419	-1.2426	-.6222

MACH ( 1 ) = .165 ALPHA ( 3 ) = 19.980 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

.000	-.3644	-.5406	-.3871	-.2684	-.3265	-.2949
.334	-1.6529	-.9837	-.8704	-.6431	-.3848	-.3835
.520	-.6157	-.8115	-.9911	-.8516	-.7933	-.5117
.663	-1.6316	-1.4034	-1.12501	-1.1003	-.8940	-.7105
.873	-1.2369	-1.1905	-1.1424	-1.0551	-.9713	-.6498

## GA37-B B16CSF1 J40 W7E16 WING UPPER SURFACE

(RDVU33) (12 NOV 73)

## REFERENCE DATA

PARAMETRIC DATA						
BREF	4.4120 B3.FL.	XMAP	43.5940 IN.	BETA	.000	PIN/P = 1.000
LREF	19.2500 IN.	TMAP	.0000 IN.	H/B	.039	BDFLAP = -16.000
BREF	37.8550 IN.	TMAP	-.4030 IN.	ELEVON	15.000	
SCALE	.005					
MACH (1) = .165	ALPHA (1) = 10.005	RNL	= 1.200	MACH	= .165	
SECTION (1)WING DEPENDENT VARIABLE CP						
X/C	.1500	.3000	.4500	.6000	.7500	.9000
2/18						
.000	-.2859	-.4363	-.2642	-.1270	-.1693	-.1710
.334	-.8004	-.6148	-.5638	-.3524	-.1226	-.2122
.520	-.7578	-.6219	-.5303	-.4066	-.3936	-.0905
.663	-1.2571	-.6399	-.6207	-.4381	-.2986	-.1240
.873	-1.2319	-1.1936	-1.0360	-.8647	-.7322	-.5745
MACH (1) = .165	ALPHA (2) = 14.990	RNL	= 1.200	MACH	= .165	
SECTION (1)WING DEPENDENT VARIABLE CP						
X/C	.1500	.3000	.4500	.6000	.7500	.9000
2/18						
.000	-.3163	-.4635	-.2770	-.1403	-.1886	-.1952
.334	-1.3103	-.8299	-.6797	-.3872	-.1493	-.3226
.520	-.7679	-.6954	-.6253	-.4792	-.4105	-.2350
.663	-1.5084	-.8124	-.6597	-.5538	-.3939	-.2341
.873	-2.6405	-2.5114	-2.1367	-1.4509	-1.3606	-.8346
MACH (1) = .165	ALPHA (3) = 20.015	RNL	= 1.200	MACH	= .165	
SECTION (1)WING DEPENDENT VARIABLE CP						
X/C	.1500	.3000	.4500	.6000	.7500	.9000
2/18						
.000	-.3531	-.5431	-.1019	-.2859	-.3412	-.3000
.334	-1.6063	-.9928	-.9042	-.6627	-.4904	-.3562
.520	-.7969	-.6019	-.10599	-.9150	-.7982	-.5260
.663	-1.5016	-1.4282	-1.2469	-1.1386	-.9057	-.7172
.873	-1.2144	-1.1795	-1.1413	-1.0728	-.9617	-.6387

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TABULATED SOURCE DATA - OA57B

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OA57-B B16C5F1 J40 W07E18 WING UPPER SURFACE

(RDV034) 112 NOV 73

## REFERENCE DATA

BREF = .4120 62.FT. XMAP = 43.5940 IN.  
 1.000 = 10.2700 IN. YMAP = .0000 IN.  
 BREF = 37.9350 IN. ZMAP = -.0050 IN.  
 SCALE = .0405

MACH (+ 1) = .165 ALPHA (+ 1) = -4.000 RNL = 1.200 MACH = .165

## SECTION (-1)WING

## DEPENDENT VARIABLE CP

X/C

.1500 .3000 .4500 .6000 .7500 .9000

2/18

## SECTION (-1)WING

## DEPENDENT VARIABLE CP

.000	-.0723	-.2184	-.1362	-.0324	-.0872	-.1770
.334	-.1668	-.2735	-.2463	-.2997	-.1766	-.1177
.520	-.3060	-.4125	-.4076	-.3143	-.4185	-.0321
.663	-.3977	-.5167	-.5171	-.4154	-.3669	-.6201
.873	.0368	.0737	.0371	-.0337	-.2613	.1075

MACH (+ 1) = .165 ALPHA (+ 2) = -.010 RNL = 1.200 MACH = .165

## SECTION (-1)WING

## DEPENDENT VARIABLE CP

X/C

.1500 .3000 .4500 .6000 .7500 .9000

2/18

.000	-.1086	-.2726	-.1789	-.0500	-.1163	-.1711
.334	-.2765	-.3566	-.4009	-.2966	-.1308	-.1971
.520	-.4708	-.4911	-.4526	-.3324	-.3812	-.0750
.663	-.6695	-.2719	-.4861	-.4137	-.6046	-.0441
.873	-.1982	-.6738	-.6745	-.5157	-.3190	-.1656

MACH (+ 1) = .165 ALPHA (+ 3) = 4.995 RNL = 1.200 MACH = .165

## SECTION (-1)WING

## DEPENDENT VARIABLE CP

X/C

.1500 .3000 .4500 .6000 .7500 .9000

2/18

.000	-.1935	-.3476	-.2177	-.0636	-.1390	-.1648
.334	-.4576	-.4939	-.4900	-.3447	-.0935	-.2919
.520	-.6137	-.5648	-.4850	-.3550	-.3276	-.1280
.663	-.9675	-.4279	-.5669	-.4576	-.5953	-.0980
.873	-.5644	-.8352	-.7847	-.6203	-.4080	-.2807

PARAMETRIC DATA

BETA = .000 PTNP = 1.500  
H/B = .286 BDFLAP = -16.000  
ELEVON = 15.000

CA57-C B16C5F1 JAO W7E10 WING UPPER SURFACE  
(ADVU34)

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.995 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B

X/C	.000	.2787	.4020	.2446	-.1031	-1.589	-.1645
	.334	-.7588	-.5940	-.5525	-.3692	-.2350	-.2897
	.520	-.6826	-.6243	-.5333	-.3874	-.3337	-.1350
	.663	-.1.2391	-.6072	-.6496	-.5101	-.6415	-.0665
	.873	-.1.1065	-.0227	-.9447	-.7497	-.6024	-.5367

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.975 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B

X/C	.000	-.1299	-.4590	-.2861	-.1237	-1.468	-.2004
	.334	-1.2777	-.7651	-.7032	-.4148	-.2739	-.3304
	.520	-.8392	-.6800	-.5691	-.4361	-.4316	-.1882
	.663	-.1.3944	-.7492	-.6701	-.4633	-.4825	-.1240
	.873	-.1.9341	-.1.3714	-.1.2337	-.1.1403	-.0735	-.8608

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B

X/C	.000	-.3454	-.5194	-.3347	-.1868	-2.2293	-.2738
	.334	-1.6169	-.9704	-.8583	-.6030	-.5094	-.4507
	.520	-.7908	-.7198	-.6866	-.6137	-.6110	-.5007
	.663	-.1.2153	-.1.2011	-.1.2024	-.1.0866	-.8850	-.7311
	.873	-.1.0961	-.1.0635	-.1.0770	-.1.0866	-.9936	-.9057

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TABULATED SOURCE DATA - CASE 7B

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CASE 7-B B16C5F1 J40 WTE10 WING UPPER SURFACE

(ADVIS35) ( 12 NOV 73 )

## REFERENCE DATA

W/L	4.4120 84.1.	XREF =	43.5940 IN.	BETA =	.000	PTN/P =	1.300
UREF	10.2300 IN.	YREF =	.0000 IN.	H/B =	.286	BLFLAP =	-16.000
BREF	37.9350 IN.	ZREF =	-.0050 IN.	ELEVON =	15.000		
SCALE	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.995 ANVL = 1.200 MACH = .165

## SECTION ( 1 ) WING

M/C .1500 .3000 .4500 .6000 .7500 .9000

DEPENDENT VARIABLE CP

21/8	.000	-.0579	-2.138	-.1264	-.0170	-.0760	-.1726
	.334	-.1637	-2.940	-.3445	-.2786	-.1947	-.2061
	.520	-.3196	-.4087	-.4017	-.3157	-.4016	-.0375
	.863	-.4050	-.0900	-.0081	-.3709	-.6384	-.0116
	.873	.0205	-.3224	-.5743	-.3927	-.2640	-.1074

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.020 RH/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

M/C .1500 .3000 .4500 .6000 .7500 .9000

DEPENDENT VARIABLE CP

21/8	.000	-.1099	-2.738	-.17112	-.0632	-.11116	-.1651
	.334	-.2816	-.5392	-.4134	-.3076	-.2080	-.2276
	.520	-.4827	-.4937	-.4629	-.3437	-.3689	-.0890
	.863	-.6740	-.2418	-.4922	-.4108	-.6025	-.0555
	.873	-.1767	-.6651	-.6737	-.4981	-.3187	-.1702

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.980 RH/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

M/C .1500 .3000 .4500 .6000 .7500 .9000

DEPENDENT VARIABLE CP

21/8	.090	-.2009	-.1900	-.2175	-.0883	-.1411	-.1643
	.334	-.4613	-.5232	-.5057	-.3481	-.2224	-.2465
	.520	-.6191	-.3763	-.4837	-.3492	-.2854	-.1779
	.863	-.9628	-.4168	-.5670	-.4488	-.6107	-.1025
	.873	-.5667	-.6395	-.7950	-.6050	-.4226	-.2957

MACH ( 1 ) = .165 ALPHAS ( 4 ) = 9.985 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
<b>21/8</b>						
.000	-.2203	-.4258	-.2708	-.1229	-.1634	-.1932
.334	-.7781	-.5967	-.5637	-.3782	-.2600	-.2872
.520	-.7255	-.6547	-.5975	-.4044	-.3559	-.1630
.663	-.1.2498	-.6107	-.6683	-.3243	-.6489	-.0987
.873	-.1.1637	-.1.0559	-.9666	-.7606	-.6543	-.3622

MACH ( 1 ) = .165 ALPHAS ( 5 ) = 14.970 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
<b>21/8</b>						
.000	-.3031	-.4657	-.2732	-.1216	-.1568	-.1856
.334	-.1.2693	-.6145	-.7106	-.4191	-.2971	-.3261
.520	-.6130	-.7102	-.6533	-.4565	-.4567	-.2018
.663	-.1.0330	-.7524	-.6745	-.4718	-.4087	-.2259
.873	-.2.3841	-.1.4244	-.1.1637	-.1.0245	-.9389	-.7870

MACH ( 1 ) = .165 ALPHAS ( 6 ) = 19.980 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	.1500	.3000	.4500	.6000	.7500	.9000
<b>21/8</b>						
.000	-.3500	-.3284	-.3381	-.1950	-.2545	-.2858
.334	-.1.5864	-.9586	-.8340	-.6090	-.5259	-.4893
.520	-.7790	-.7177	-.6996	-.8193	-.8145	-.4522
.663	-.1.2039	-.1.1296	-.1.1591	-.1.0568	-.9050	-.7452
.873	-.1.0644	-.1.0606	-.1.0573	-.1.0558	-.9765	-.8985



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TABULATED SOURCE DATA - QASTB

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OA37-B B16C3F1 J41 WATE10 WING, UPPER SURFACE

(RDVUS2) (112 NOV 73)

## REFERENCE DATA

SREF =	4.4175 Sq.FT.	XMAP =	43.5940 IN.	BETA =	.000	PIN/P =	1.300
LREF =	16.2300 IN.	YMAP =	.0000 IN.	H/B =	.266	BOFLAP =	-16.000
SMAP =	37.9350 IN.	ZMAP =	-.4050 IN.	ELEVON =	.000		
SCALE =	.0403						

MACH (1) = .165 ALPHA (1) = -4.003 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.0406	-.1702	-.0888	.0171	-.0277	-.1361
	.334	-.1182	-.2026	-.2397	-.1606	-.0709	.0766
	.520	-.2424	-.3115	-.3281	-.1553	-.0901	.0693
	.663	-.3105	-.3146	-.2689	-.1531	.0368	.1341
	.873	-.2793	-.4190	-.4608	-.2544	-.1047	.0191

MACH (1) = .165 ALPHA (2) = -.005 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.0634	-.2241	-.1289	-.0143	-.0565	-.1300
	.334	-.2363	-.2955	-.2936	-.1638	-.0812	.0786
	.520	-.4122	-.4081	-.3753	-.1801	-.0958	.0782
	.663	-.5635	-.4503	-.3361	-.2160	-.0023	.1227
	.873	-.5266	-.5576	-.6131	-.2677	-.1125	.0120

MACH (1) = .165 ALPHA (3) = 4.995 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.1939	-.3578	-.2303	-.0292	-.0792	-.1306
	.334	-.3956	-.4296	-.3778	-.2313	-.1139	.0540
	.520	-.5409	-.4688	-.5831	-.1942	-.1003	-.0011
	.663	-.6358	-.5816	-.3885	-.2297	-.0354	.0825
	.873	-.6937	-.7288	-.6718	-.3406	-.1550	-.0250

## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.990 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.990 RNL = 1.200 MACH = .165

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.2781	-.4116	-.2356	-.0685	-.1018	-.1498
.334	-.8631	-.5335	-.3426	-.2125	-.1326	.0455
.520	-.8646	-.5358	-.4611	-.2626	-.1437	.0046
.663	-.1.1032	-.7283	-.4711	-.2975	-.0901	.0346
.873	-.1.2993	-.9147	-.7354	-.4318	-.2644	-.1221

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990 RNL = 1.200 MACH = .165

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.2964	-.4957	-.2703	-.0853	-.1123	-.1521
.334	-.1.1488	-.7001	-.5536	-.3180	-.1768	.0268
.520	-.7677	-.6166	-.5045	-.3162	-.1998	-.0126
.663	-.1.2760	-.8305	-.5163	-.3197	-.0461	.0768
.873	-.1.5099	-.1.0865	-.6554	-.6939	-.5432	-.3128

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990 RNL = 1.200 MACH = .165

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3662	-.5502	-.3461	-.1565	-.2226	-.2217
.334	-.1.4991	-.8668	-.6882	-.4245	-.3182	-.1024
.520	-.7662	-.6892	-.7707	-.5386	-.5517	-.2148
.663	-.1.2637	-.1.3356	-.1.0350	-.9292	-.3505	-.2953
.873	-.1.2177	-.1.1384	-.1.0962	-.1.0191	-.8341	-.5819



(WDTU52)

(100V53) (12 NOV 73)

CA57-8 816C5F1 J41 WATEL8 WING, UPPER SURFACE

## REFERENCE DATA

REF	4.4120 84.51.	XMAP	Z	43.5940 IN.	BETA	=	.000	PTN/P	=	1.000
LREF	19.2900 IN.	YMAP	=	.0000 IN.	H/B	=	.286	BDFLAP	=	-16.000
2REF	37.9350 IN.	ZMAP	=	-.0050 IN.	ELEVON	=	.000			
SCALE	.0405									
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	-.4,000	RNL =	1.200	MACH =	.165			

SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
.000	-.0464	-.1828	-.0028	.0072	-.0354	-.1483	
.334	-.1259	-.2108	-.2540	-.1656	-.0757	.0554	
.520	-.2533	-.3210	-.3967	-.1625	-.1268	.0775	
.663	-.3220	-.3267	-.2677	-.1126	-.0170	.1006	
.673	-.3102	-.4278	-.6245	-.2777	-.0997	.0104	
MACH ( 1 ) =	.165	ALPHA ( 2 ) =	-.005	RNL =	1.200	MACH =	.165

SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
.000	-.0866	-.2422	-.1390	-.0190	-.0119	-.1446	
.334	-.2369	-.3092	-.3110	-.1883	-.0912	.0494	
.520	-.4218	-.4165	-.3615	-.1637	-.1255	.0607	
.663	-.5783	-.4733	-.3365	-.1455	-.0364	.0926	
.673	-.5664	-.3717	-.7111	-.3053	-.1431	.0056	
MACH ( 1 ) =	.165	ALPHA ( 3 ) =	4.985	Z/L =	1.200	MACH =	.165

SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.1833	-.3159	-.1897	-.0602	-.1083	-.1510
.334	-.4084	-.4354	-.3629	-.2579	-.1187	.0364
.520	-.5643	-.4890	-.3676	-.2035	-.1146	.0249
.663	-.8579	-.6124	-.3910	-.1643	-.0693	.0562
.673	-.9302	-.7540	-.7433	-.3633	-.1622	-.0357



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TABULATED SOURCE DATA - CASTE

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CASTE-8 01625F1 J41 WING UPPER SURFACE

SECTION (1)WING

## AUXILIARY DATA

**SACF** = 4.4120 82.87. **ZMAP** = 43.5940 IN.  
**VREF** = 16.2300 IN. **RMAP** = .0000 IN.  
**BREF** = 37.9350 -IN. **ZMAP** = -.4050 IN.  
**SCALE** = .0405

MACH (1) = .165 ALPHA (1) = 9.990 ANL = 1.200 MACH = .165

SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

X/C	.000	-.2478	-.4233	-.2904	-.0710	-.1175	-.1690
	.334	-.7010	-.5429	-.4534	-.2717	.0169	.0168
	.520	-.7010	-.5441	-.4371	-.2897	-.3267	.0063
	.663	-.1.0635	-.7133	-.4565	-.3076	-.0677	.0676
	.873	-.1.3651	-.9426	-.7972	-.4667	-.2992	-.2165

MACH (1) = .165 ALPHA (2) = 14.990 ANL = 1.200 MACH = .165

SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8

X/C	.000	-.3113	-.5251	-.2688	-.1093	-.1564	-.1753
	.334	-.1.2323	-.7172	-.5664	-.3034	.0060	.0034
	.520	-.7158	-.6035	-.5219	-.3445	-.2297	-.0916
	.663	-.1.9536	-.7577	-.4737	-.3546	-.1394	-.0421
	.873	-.2.2639	-.1.6586	-.1.1943	-.0946	-.6619	-.3604

MACH (1) = .165 ALPHA (3) = 19.995 ANL = 1.200 MACH = .165

SECTION (1)WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

X/C	.000	-.3779	-.5764	-.3695	-.2412	-.2820	-.2541
	.334	-.1.4122	-.9022	-.7191	-.5261	-.1656	-.1538
	.520	-.7655	-.7510	-.6621	-.7240	-.6250	-.4349
	.663	-.1.7659	-.1.3425	1.1056	-.9412	-.5676	-.4110
	.873	-.1.2344	-.1.1794	-.1.1583	-.1.0396	-.8905	-.6423

(ADUSS) ( 12 NOV 73 )

PARAMETRIC DATA

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TABULATED SOURCE DATA - CA578

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CA57-B B16C5F1 J41 W7E18 WING UPPER SURFACE (ADVU56) ( 12 NOV 73 )

## REFERENCE DATA

BREF =	4,4120 SQ.FT.	XMAP =	43.5940 IN.	BETA =	.000	PTN/P =	1.000
LREF =	16.2300 IN.	YMAP =	.0000 IN.	M/B =	.039	BDFLAP =	-16.000
BREF =	37.9330 IN.	ZMAP =	-.4050 IN.	ELEVON =	.000		
SCALE =	.0405						

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (1) = 10.005 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

SECTION (1)WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B						
.000	-.2931	-.4300	-.2271	-.0795	-.1224	-.1642
.334	-.6975	-.5560	-.4497	-.2636	.0126	-.0119
.520	-.6952	-.5427	-.4434	-.2778	-.1267	-.0902
.663	-.1.0869	-.7201	-.6506	-.3987	-.0781	.0313
.873	-.1.3774	-.9539	-.8174	-.4845	-.3213	-.2465

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (2) = 15.020 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

SECTION (1)WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B						
.000	-.3307	-.5297	-.2606	-.1299	-.1603	-.1708
.334	-.1.2376	-.7529	-.5547	-.3146	-.0161	-.0049
.520	-.7214	-.5962	-.5434	-.3500	-.2190	-.1633
.663	-.1.3199	-.7871	-.7255	-.3457	-.1409	-.0504
.873	-.2.6486	-.1.6683	-.1.2715	-.8439	-.6057	-.4108

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (3) = 20.015 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

SECTION (1)WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/B						
.000	-.3674	-.5701	-.3848	-.12332	-.2773	-.2537
.334	-.1.5839	-.9062	-.7255	-.5216	-.2698	-.1295
.520	-.7532	-.7518	-.6700	-.7178	-.5997	-.4007
.663	-.1.7199	-.1.3522	-.1.1257	-.9764	-.5771	-.4503
.873	-.1.2145	-.1.1220	-.1.1390	-.1.0443	-.8952	-.6437

## TABULATED SOURCE DATA - C4978

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## REFERENCE DATA

SECTION ( 1 ) WING

BREF = 4.4120 IN. FT.    XWAP = 43.9940 IN.  
 LREF = 19.2300 IN.    YWAP = .0000 IN.  
 BREF = 37.9350 IN.    ZWAP = -.4050 IN.  
 BSCALE = .0405

MACH ( 1 ) = .165    ALPHA ( 1 ) = -3.980    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.0626 -.2050 -.1141 -.0101 -.0609 -.1786

.334 -.1553 -.2433 -.2608 -.1949 -.0338 .0701

.520 -.2913 -.3617 -.3783 -.1919 -.1558 .0205

.663 -.3646 -.3619 -.3173 -.2422 -.0450 .0713

.873 -.3334 -.4675 -.6857 -.3253 -.1510 -.0312

MACH ( 1 ) = .165    ALPHA ( 2 ) = -.005    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.1102 -.2666 -.1504 -.0135 -.0806 -.1572

.334 -.2618 -.3253 -.3246 -.2023 -.0471 .0696

.520 -.4370 -.4304 -.3933 -.2005 -.1274 .0254

.663 -.8066 -.4675 -.3682 -.2560 -.0354 .0893

.873 -.5924 -.5913 -.7330 -.3268 -.1676 -.0156

MACH ( 1 ) = .165    ALPHA ( 3 ) = 4.980    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.2417 -.3696 -.2136 -.0450 -.0844 -.1563

.334 -.5033 -.5322 -.4383 -.2355 -.0697 .0651

.520 -.5806 -.5015 -.3941 -.2305 -.1297 -.0451

.663 -.8931 -.6278 -.4243 -.2946 -.0753 .0655

.873 -.9703 -.7712 -.7663 -.3937 -.1934 -.0397

SECTION ( 1 ) WING

## PARAMETRIC DATA

BETA = .000 PTH/P = 1.300  
 H/B = .125 BDFLAP = -16.000  
 ELEVON = .00,

(ADVIS7) ( 12 NOV 73 )

## TABULATED SOURCE DATA - OA378

(ADJUST)

MACH (.1) = .165    ALPHA (.4) = 9.995    RNL = 1.200    MACH = .165  
 SECTION (1)WING

X/C    .1500    .3000    .4500    .6000    .7500    .9000  
 DEPENDENT VARIABLE CP

21/B  
 .000    -.2473    -.4054    -.2151    -.0658    -.1138    -.1585  
 .334    -.6635    -.5372    -.4485    -.2530    -.1159    .0176  
 .520    -.6844    -.5797    -.4523    -.2616    -.1654    -.0242  
 .663    -.1.1025    -.7501    -.4934    -.3309    -.0669    .0850  
 .873    -.1.3627    -.9464    -.8166    -.4780    -.2931    -.1511

MACH (.1) = .165    ALPHA (.5) = 15.020    RNL = 1.200    MACH = .165  
 SECTION (1)WING

X/C    .1500    .3000    .4500    .6000    .7500    .9000  
 DEPENDENT VARIABLE CP

21/B  
 .000    -.3168    -.4979    -.2352    -.0988    -.1390    -.1630  
 .334    -.1.1675    -.7242    -.5634    -.3194    -.1394    .0103  
 .520    -.7556    -.6208    -.5436    -.3626    -.2315    -.1297  
 .663    -.1.2417    -.7653    -.5017    -.3269    -.1226    -.0336  
 .873    -.1.7927    -.1.1398    -.9037    -.6975    -.5225    -.3365

MACH (.1) = .165    ALPHA (.6) = 19.995    RNL = 1.200    MACH = .165  
 SECTION (1)WING

X/C    .1500    .3000    .4500    .6000    .7500    .9000  
 DEPENDENT VARIABLE CP

21/B  
 .000    -.3532    -.5386    -.3470    -.1657    -.2662    -.2412  
 .334    -.5590    -.8639    -.6901    -.4511    -.3143    -.1282  
 .520    -.7611    -.7176    -.8176    -.6675    -.0067    -.3453  
 .663    -.4350    -.3646    -.1.1183    -.9637    -.4537    -.3343  
 .873    -.1.2664    -.1.1959    -.1.1952    -.1.0673    -.8986    -.6286

(ADJUST)

(ADJUST)

DATE 08 OCT 74

TABULATED SOURCE DATA - CASE 78

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CA57-B B165F1 J41 WTE10 WING UPPER SURFACE

(ADVU58) ( 12 NOV 73 )

## REFERENCE DATA

BREF =	4.4120 94.51.	XMAP =	43.5540 IN.
UREF =	.19.2300 IN.	IMAP =	.0000 IN.
BREF =	.37.9350 IN.	ZMAP =	-.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.995 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B						
.000	-.0601	-.2043	-.1002	-.0046	.0543	-.1566
.334	-.1469	-.2410	-.2700	-.1719	-.0468	.1046
.520	-.2744	-.3560	-.3693	-.1683	-.1432	.0277
.663	-.3501	-.3593	-.3064	-.2244	-.0392	.0821
.873	-.3466	-.4651	-.5987	-.3271	-.1465	.0140

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.003 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B						
.000	-.1312	-.2802	-.1571	-.0169	-.0888	-.1624
.334	-.2633	-.3352	-.3237	-.2020	-.0777	.0817
.520	-.4394	-.4412	-.4002	-.1972	-.1329	.0132
.663	-.6001	-.4950	-.3681	-.2555	-.0443	.0862
.873	-.6042	-.5988	-.7180	-.5316	-.1841	.0170

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.995 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B						
.000	-.2376	-.3766	-.2029	-.0534	-.0912	-.1571
.334	-.4962	-.5258	-.4286	-.2229	-.0819	.06C1
.520	-.5759	-.5043	-.3915	-.2152	-.1247	-.0456
.663	-.8717	-.6219	-.4158	-.2793	-.0591	.0724
.873	-.9763	-.7712	-.7627	-.3957	-.2C14	-.0544

CA37-B 016C5F1 J42 WATE16 WING UPPER SURFACE

(RDWV59) (112 NOV 73)

## REFERENCE DATA

WREF	=	4.4120 64. FT.	XMAP	=	43.3940 IN.
LREF	=	19.2300 IN.	RMAP	=	.0000 IN.
BREF	=	37.9350 IN.	ZMAP	=	-.4050 IN.
SCALE	=	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.023 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 -.0643 -.2054 -.0165 -.0435 -.0497 -.1652

.334 -.1618 -.2301 -.4575 -.2176 -.1427 .0035

.520 -.3307 -.6555 -.3370 -.2038 -.1069 -.0306

.663 -.5961 -.3967 -.3347 -.2397 -.0476 .0753

.873 -.3604 -.4847 -.7213 -.3535 -.1251 -.0156

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 -.1291 -.2709 -.0745 -.0703 -.0704 -.1499

.334 -.3263 -.3698 -.5143 -.2556 -.1484 -.0208

.520 -.6336 -.7545 -.4051 -.2372 -.1306 -.0503

.663 -.8253 -.5062 -.3681 -.2451 -.0555 .0866

.873 -.6232 -.6161 -.7784 -.3526 -.1307 -.0114

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.955 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B .000 -.2379 -.3497 -.1315 -.1068 -.1051 -.1525

.334 -.5341 -.5107 -.5880 -.5123 -.1927 .0028

.520 -.1.0468 -.9158 -.5102 -.3326 -.2034 -.0875

.663 -.1.1028 -.6665 -.4285 -.3013 -.1207 -.0198

.873 -.1.0123 -.7950 -.7856 -.4120 -.1903 -.0628



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## TABULATED SOURCE DATA - CASTS

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MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.935 RN/L = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B SECTION ( 1 ) WING

X/C	.000	.334	.667	.993	RN/L	MACH
	- .3331	- .4380	- .1420	- .1238	- .0983	= .165
	- .7449	- .6366	- .6932	- .3652	- .1798	
	- 1.2065	- 1.0423	- .5935	- .3563	- .1674	
	- 1.3419	- .7907	- .5640	- .4089	- .1092	
	- 1.3501	- .9029	- .6812	- .4536	- .3204	
					- .1694	

MACH ( 1 ) = .165 ALPHA ( 3 ) = 14.980 RN/L = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B SECTION ( 1 ) WING

X/C	.000	.334	.667	.993	RN/L	MACH
	- .4074	- .6208	- .2629	- .1500	- .1299	= .165
	- .7801	- .6940	- .7503	- .3769	- .2228	
	- 1.1689	- 1.1550	- .6938	- .4748	- .2487	
	- 1.5250	- .9984	- .7616	- .6532	- .2075	
	- 2.1748	- 1.8141	- 1.5476	- 1.2817	- .7995	
					- .4715	

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.975 RN/L = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/B SECTION ( 1 ) WING

X/C	.000	.334	.667	.993	RN/L	MACH
	- .4605	- .6337	- .3826	- .2377	- .1544	= .165
	- .8507	- .6538	- .7732	- .4168	- .2590	
	- .9538	- 1.3290	- .9670	- .8028	- .2833	
	- 2.0929	- 1.6983	- 1.3718	- 1.2640	- .6360	
	- 1.7237	- 1.5436	- 1.4965	- 1.4013	- 1.1607	
					- .8077	

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TABULATED SOURCE DATA - QASTA

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REF # 4,4120 84 FT. XMRP = 43.5940 IN.  
 LAEF # 19.2300 IN. YMRP = .0000 IN.  
 BREF # 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0403

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

SECTION ( 1 ) WING

21/8

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
	.334	-.1324	-.1963	-.4078	-.1866	-.1173	.0145
	.520	-.2933	-.3793	-.2899	-.1690	-.0914	-.0128
	.663	-.5099	-.4020	-.2976	-.2038	-.0408	.0827
	.873	-.3384	-.4680	-.7159	-.3351	-.1079	-.0056

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
	.334	-.2954	-.3366	-.4563	-.2238	-.1313	-.0128
	.520	-.5820	-.6693	-.3592	-.2078	-.1136	-.0291
	.663	-.7409	-.4803	-.3347	-.2222	-.0620	.0593
	.873	-.6032	-.5981	-.7700	-.3498	-.1238	-.0057

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

SECTION ( 1 ) WING

21/8

X/C	.0000	-.2528	-.3496	-.1431	-.0808	-.0628	-.1420
	.334	-.4920	-.4590	-.5219	-.2689	-.1559	.0009
	.520	-.9659	-.6127	-.4482	-.2777	-.1575	-.0528
	.663	-.1.0.04	-.5993	-.3877	-.2731	-.1198	-.0424
	.873	-.9668	-.7568	-.7571	-.3927	-.1760	-.0486

REFERENCE DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
	.334	-.1324	-.1963	-.4078	-.1866	-.1173	.0145
	.520	-.2933	-.3793	-.2899	-.1690	-.0914	-.0128
	.663	-.5099	-.4020	-.2976	-.2038	-.0408	.0827
	.873	-.3384	-.4680	-.7159	-.3351	-.1079	-.0056

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.2528	-.3496	-.1431	-.0808	-.0628	-.1420
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

REFERENCE DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

REFERENCE DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.2528	-.3496	-.1431	-.0808	-.0628	-.1420
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.2528	-.3496	-.1431	-.0808	-.0628	-.1420
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.2528	-.3496	-.1431	-.0808	-.0628	-.1420
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.2528	-.3496	-.1431	-.0808	-.0628	-.1420
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.2528	-.3496	-.1431	-.0808	-.0628	-.1420
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.0554	-.1087	-.0370	-.0354	-.0464	-.1531
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

PARAMETRIC DATA

SECTION ( 1 ) WING

SECTION ( 1 ) WING

X/C

X/C	.0000	-.1556	-.2812	-.0952	-.0477	-.0525	-.1422
-----	-------	--------	--------	--------	--------	--------	--------

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

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## TABULATED SOURCE DATA - CASTB

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		CAST-B B16CFF1 J42 WATE10 WING UPPER SURFACE (INDU60)				
MACH ( 1 ) =	.165	ALPHA ( 4 ) =	.9.960	RN/L	=	1.200 MACH = .165
SECTION ( 1 )WING		DEPENDENT VARIABLE CP				
X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.2914	-.4531	-.1752	-.1214	-.1322	-.1552
.334	-.7019	-.6047	-.6573	-.3310	-.1877	.0196
.520	-1.1290	-1.9357	-.5479	-.3184	-.1709	-.0118
.663	-1.2915	-.7210	-.5116	-.3549	-.0594	.0275
.873	-1.3090	-.8432	-.6346	-.4574	-.2993	-.1626
MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.960 RN/L = 1.200 MACH = .165						
SECTION ( 1 )WING		DEPENDENT VARIABLE CP				
X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.3716	-.5542	-.2347	-.1684	-.1505	-.1601
.334	-.7929	-.6459	-.7015	-.4017	-.2303	-.0499
.520	-1.1692	-1.0351	-.6491	-.4874	-.2530	-.0440
.663	-1.4170	-.8687	-.6701	-.5752	-.2388	-.1704
.873	-2.6341	-1.8729	-1.4142	-1.3280	-.6986	-.3655
MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.965 RN/L = 1.200 MACH = .165						
SECTION ( 1 )WING		DEPENDENT VARIABLE CP				
X/C	.1500	.3000	.4500	.6000	.7500	.9000
21/8						
.000	-.4303	-.6321	-.3377	-.2487	-.1970	-.1778
.334	-.8701	-.6438	-.7610	-.4345	-.3051	-.1545
.520	-.9622	-.1.2883	-.9675	-.7781	-.3112	-.1142
.663	-2.0847	-1.6222	-1.2167	-1.0124	-.6192	-.5453
.873	-1.4249	-1.3408	-1.2804	-1.1220	-.8828	-.6538

## CA57-8 B10C5F1 J42 WATE10 WING UPPER SURFACE

(RDVU61) (12 NOV 73)

## REFERENCE DATA

BREF = 4.4120 82 FT. ZMAP = 43.5940 IN.  
 LREF = 18.2500 IN. INRP = .0000 IN.  
 BREF = 37.9350 IN. ZNRF = -.4050 IN.  
 SCALE = .0405

## PARAMETRIC DATA

BETA = .000 PMP = 1.000  
 M/B = .125 BDFLAP = -.16.000  
 ELEVON = .000

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.030 ANL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.0457 -.1817 -.0184 -.0241 -.0591 -.1578

.334 -.1109 -.1658 -.3433 -.1716 -.1221 -.0112

.520 -.2643 -.4608 -.2552 -.1602 -.1015 -.0367

.863 -.4127 -.3457 -.2576 -.1959 -.0473 -.0769

.873 -.3420 -.4570 -.7269 -.3468 -.1053 -.0060

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.025 ANL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.1461 -.2698 -.0805 -.0471 -.0755 -.1518

.334 -.2699 -.2978 -.3813 -.2022 -.1333 -.0415

.520 -.5259 -.5478 -.3124 -.1958 -.1184 -.0504

.863 -.6748 -.4192 -.2934 -.2180 -.0872 -.0068

.873 -.5926 -.5614 -.7617 -.3508 -.1266 -.0123

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.960 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8 .000 -.2217 -.3417 -.1195 -.0755 -.0907 -.1560

.334 -.4641 -.4148 -.4424 -.2276 -.1522 -.0413

.520 -.8638 -.6810 -.4017 -.2298 -.1229 -.0427

.863 -.9164 -.5153 -.3535 -.2532 -.1350 -.0911

.873 -.9399 -.7209 -.7462 -.3764 -.1707 -.0439

## (REV061)

CA57-3 B16.5F1 J42 WTE16 WING UPPER SURFACE

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.965 RIV/L = 1.200 MACH = .165

SECTION ( 1 ) WING  
N/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP

	21/8	.0000	-.2126	-.4292	-.1536	-.1194	-.1421	-.1698
MACH ( 1 ) = .165	N/C	.334	-.6474	-.5217	.5560	-.3248	-.1936	-.0055
		.520	-.1.0663	-.8046	-.4741	-.3168	-.1956	-.0742
		.663	-.1.1608	-.8423	-.4206	-.3424	-.1868	-.1373
		.873	-.1.2160	-.7963	-.5980	-.4532	-.3074	-.1513

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.965 RIV/L = 1.200 MACH = .165

SECTION ( 1 ) WING  
N/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP

	21/8	.0000	-.3554	-.5162	-.2107	-.1708	-.1863	-.1929
MACH ( 1 ) = .165	N/C	.334	-.7388	-.6198	-.6518	-.3863	-.2618	-.0116
		.520	-.1.0850	-.9002	-.6120	-.4635	-.3141	-.1108
		.663	-.1.2549	-.7008	-.5665	-.5034	-.3411	-.3039
		.873	-.2.5650	-.1.6360	-.1.2415	-.1.1917	-.6144	-.4026

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.970 RIV/L = 1.200 MACH = .165

SECTION ( 1 ) WING  
N/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP

	21/8	.000	-.4110	-.5028	-.3075	-.2733	-.2456	-.2334
MACH ( 1 ) = .165	N/C	.334	-.7976	-.5939	-.7123	-.5006	-.3893	-.2.60
		.520	-.8557	-.1.1150	-.8175	-.5623	-.4557	-.2237
		.663	-.1.7325	-.1.2495	-.9462	-.8447	-.5720	-.4934
		.873	-.1.0736	-.1.0092	-.9257	-.7235	-.6325	-.5132

REFERENCE DATA						PARAMETRIC DATA		
SECTION (1) WING			SECTION (2) WING			(RDVU62) (112 NOV 73)		
SECTION (1) WING			SECTION (2) WING			SECTION (3) WING		
A/C	WING	ANGLE	A/C	WING	ANGLE	A/C	WING	ANGLE
MACH (1) = .165	ALPHA (1) = 10.000	RH/L = 1.2000	MACH = .165					
21/8								
.000	-2857	-4449	.1616	-1.1734	-1.1594			
.334	-7653	-7070	.7124	.3605	.2034	.0323		
.520	-1.2551	-1.0613	.6129	.3616	.1659	.0320		
.663	-1.4311	-1.0681	.6282	.4374	.0836	.0042		
.873	-1.4079	-1.9322	.7093	.5935	.4101	.2395		
MACH (1) = .165	ALPHA (2) = 14.980	RH/L = 1.2000	MACH = .165					
21/8								
.000	-3503	-1.8644	-1.0172	-1.1363	-1.5168	-1.3329		
.334	-3343	-1.5093	-1.6814	-1.7110	-1.1987	-1.0083		
.520	-8503	-7270	-1.2022	-1.3622	-1.2085	-1.1544		
.663	-1.2C30	-6.5174	-1.7517	-5.689270.0000	-1.0286			
.873	-1.6995-10.5136	-1.7257	-9.0000-12.9145	-1.0196				
MACH (1) = .165	ALPHA (3) = 19.995	RH/L = 1.2000	MACH = .165					
21/8								
.000	.3911	.9463	-.2343	-.2369	-.1666	-.1286		
.334	-9005	-6404	-.7524	-.4036	-.2609	-.1176		
.520	-9030	-1.3711	-1.0409	-.7803	-.2306	-.0376		
.663	-2.6173	-1.6410	-1.6032	-1.1297	-.6707	-.5041		
.873	-1.7604	-1.8420	-1.9753	-1.5269	-1.3249	-.8710		



SECTION (1) WING E16C-SF1 J-2 WATER WING UPPER SURFACE (REVURS) (12 NOV 73)

## REFERENCE DATA

**SUPER =** 4.4120 SQ.FT. **XMAP =** 43.5940 IN.  
**LALY =** 18.2500 IN. **YMAP =** .0000 IN.  
**DREF =** 37.9350 IN. **ZMAP =** -.4050 IN.  
**SCALE =** .0405

MACH ( 1 ) = .165    ALPHA ( 1 ) = 9.995    RFL/L = 1.200    MACH = .165

SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.3000	.4500	.6000	.7500	.9000
21/8					
.000	-.2675	-.4273	-.1129	-.1469	-.1736
.334	-.7441	-.6532	-.6590	-.3479	-.1968
.520	-1.1752	-.9646	-.5408	-.3465	-.1667
.663	-1.2934	-.7329	-.5431	-.4173	-.0876
.873	-1.3055	-.6714	-.6790	-.5386	-.3702

MACH ( 1 ) = .165    ALPHA ( 2 ) = 14.985    RFL/L = 1.200    MACH = .165

SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.3000	.4500	.6000	.7500	.9000
21/8					
.000	-.5312	-.5062	-.0917	-.1807	-.1823
.334	-.6374	-.6803	-.7220	-.4069	-.2476
.520	-1.1368	-1.1075	-.7265	-.5697	-.2574
.663	-1.6034	-.9084	-.8245	-.7598	-.3719
.873	-1.7537	-1.6172	-1.5072	-1.3697	-.4991

MACH ( 1 ) = .165    ALPHA ( 3 ) = 20.000    RFL/L = 1.200    MACH = .165

SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.3000	.4500	.6000	.7500	.9000
21/8					
.000	-.3991	-.3972	-.1934	-.2554	-.2063
.334	-.9032	-.8570	-.7935	-.4876	-.3561
.520	-.9612	-1.3491	-1.0629	-1.1477	-.3267
.663	-2.4687	-1.7796	-1.5941	-1.1126	-.7633
.873	-1.4753	-1.4213	-1.3564	-1.0772	-.5110

BETA = .000    P1/N/P = 1.100  
 H/B = .039    BOTFLAP = -16.000  
 ELEVON = .000

CA37-E 810CSF1 J42 WATE10 WING UPPER SURFACE

(90VLSA) (12 MCN 73)

## REFERENCE DATA

BREF = 4.4120 94.5T. TMAP = 43.5840 IN.  
 LREF = 10.2300 IN. TMAP = .0000 IN.  
 SREF = 37.9350 IN. TMAP = -.4050 IN.  
 SCALE = .0403

MACH (1) = .165 ALPHA (1) = 10.000 RHL = 1.200 MACH = .165

## SECTION 1, WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8  
 .000 -.2734 -.4083 -.0525 -.1589 -.1773 -.1688  
 .334 -.7102 -.5843 -.5650 -.3397 -.1926 .0154  
 .520 -.1.0866 -.8265 -.4967 -.3322 -.2122 -.0125  
 .663 -.1.1497 -.6118 -.4599 -.3715 -.2154 -.1690  
 .873 -.1.1484 -.7678 -.6783 -.3664 -.3137 -.1713

MACH (1) = .165 ALPHA (2) = 15.010 RHL = 1.200 MACH = .165

## SECTION 1, WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/7  
 .000 -.3217 -.4876 -.0397 -.1842 -.2041 -.2047  
 ...4 -.7967 -.6435 -.6544 -.4116 -.2879 -.0909  
 .520 -.1.0639 -.9596 -.6830 -.5184 -.3454 -.1185  
 .663 -.1.4716 -.7696 -.7253 -.6564 -.4163 -.3982  
 .873 -.1.4216 -.1.3729 -.1.2278 -.1.1221 -.7614 -.5367

MACH (1) = .165 ALPHA (3) = 19.990 RHL = 1.200 MACH = .165

## SECTION 1, WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/9  
 .000 -.3748 -.5458 -.1140 -.2645 -.2548 -.2841  
 .334 -.6335 -.5934 -.7437 -.5582 -.4589 -.2473  
 .520 -.8430 -.1.1553 -.6642 -.7112 -.5363 -.2957  
 .663 -.1.7095 -.1.3355 -.9715 -.8722 -.6541 -.6119  
 .873 -.1.0370 -.1.0092 -.9143 -.7738 -.7049 -.6341

## PARAMETRIC DATA

BETA = .000 P1N/P = 1.000  
 H/B = .039 BDFLP = -18.000  
 ELEVON = .000

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TABULATED SOURCE DATA - CA77B

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CA37-B B10CF1 J42 WATE10 WING UPPER SURFACE

(REV06) (12 NOV 73)

## REFERENCE DATA

SPCF	4.4180	.04, FT.	ZREF	43.3940 IN.
URDF	19.2300	IN.	TRFP	.0000 IN.
URDZ	37.9350	IN.	ZREFP	-.4050 IN.
SCALE	.0005			

MACH (1) = .165 ALPH (1) = -.0000 ANVL = 1.200 MACH = .165

SECTION (1)WING

L/C .1900 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	.0514	-.1674	.0194	-.0493	-.0143	-.1248
.354	-.1402	-.2005	-.4310	-.1172	-.0826	.0603
.320	-.3115	-.6111	-.2936	-.1620	-.0704	.0162
.663	-.5023	-.3697	-.2362	-.1942	-.0147	.1117
.673	-.3306	-.4455	-.7136	-.3044	-.0876	.0162

MACH (1) = .165 ALPH (2) = .010 ANVL = 1.200 MACH = .165

SECTION (1)WING

L/C .1900 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	-.1009	-.2376	-.0320	-.0714	-.0491	-.1313
.334	-.2991	-.3434	-.4617	-.2315	-.1140	.0086
.320	-.5947	-.7223	-.3776	-.2102	-.1064	.0216
.663	-.7103	-.5087	-.3441	-.2217	-.0359	.0806
.673	-.5644	-.5787	-.7786	-.3343	-.1114	.0027

MACH (1) = .165 ALPH (3) = .5,000 ANVL = 1.200 MACH = .165

SECTION (1)WING

L/C .1900 .3000 .4500 .6000 .7500 .9000

21/8 DEPENDENT VARIABLE CP

.000	-.1782	-.3330	-.0787	-.1016	-.1161	-.1243
.334	-.3011	-.4807	-.5504	-.2064	-.1539	-.0045
.320	-.6669	-.8555	-.4783	-.2649	-.1677	-.0743
.663	-.9593	-.6166	-.3903	-.2723	-.1002	-.0003
.673	-.8471	-.7420	-.7644	-.5766	-.1611	-.1390

CA57-B BLOCK1 J42 MATE16 WING UPPER SURFACE  
(ADVU68)

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.960 AN/L = 1.200 MACH = .165

SECTION ( 11WING  
X/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP21/8  
.000 -.2528 -.4110 -.1023 -.1384 -.1401 -.1274

.334 -.6634 -.6119 -.3372 -.3372 -.1617 .0519

.520 -.1.1936 -.9820 -.5893 -.3611 .1650 .0453

.863 -.1.1910 -.7304 -.5175 -.4011 -.1177 .0261

.873 -.1.3164 -.8654 -.7303 -.4530 -.2593 -.1027

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.010 AN/L = 1.200 MACH = .165

SECTION ( 11WING  
X/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP21/8  
.000 -.3216 -.4637 -.1069 -.1687 -.1457 -.1649

.334 -.7963 -.8655 -.7214 -.3606 -.2203 -.0365

.520 -.1.2506 -.1.0652 -.6347 -.4502 -.2212 -.0158

.863 -.1.4202 -.8612 -.6674 -.5064 -.1249 -.0384

.873 -.1.5964 -.1.2893 -.1.2328 -.8501 -.4902 -.2486

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.960 AN/L = 1.200 MACH = .165

SECTION ( 11WING  
X/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP21/8  
.000 -.3840 -.3738 -.1430 -.2137 -.1520 -.1663

.334 -.8617 -.6798 -.7603 -.4445 -.2602 -.1084

.520 -.1.0330 -.1.2940 -.9260 -.7676 -.3199 -.0829

.863 -.1.8729 -.4714 -.2246 -.1.1527 -.5649 -.5072

.873 -.1.5738 -.1.4445 -.1.3875 -.1.2299 -.1.0141 -.6832

## TABULATED SOURCE DATA - CA578

(ADV68) ( 12 NOV 73 )

## REFERENCE DATA

REF	2	4.4120 83 FT.	XMAP	=	43.3940 IN.		
LAT	1	19.2300 IN.	YMAP	=	.0000 IN.		
UREF	2	37.9350 IN.	ZMAP	=	-.4050 IN.		
SCALE	2	.0405					

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.990 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8	.000	-.0415	-.1587	.0273	-.0347	-.0221	-.1257
	.334	-.1192	-.1589	-.5791	-.1579	-.0676	.0534
	.520	-.2360	-.5233	-.2661	-.1379	-.0559	.0208
	.663	-.4373	-.3627	-.2696	-.1730	-.0107	.1069
	.873	-.3170	-.4312	-.7247	-.3092	-.0790	.0187

MACH ( 1 ) = .165 ALPHA ( 2 ) = .005 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8	.000	-.0923	-.2266	-.0139	-.0629	-.0501	-.1246
	.334	-.2516	-.3074	-.4239	-.1942	-.0857	.0169
	.520	-.3398	-.6263	-.3264	-.1764	-.0825	.0097
	.663	-.6240	-.4463	-.3051	-.1906	-.0351	.0807
	.873	-.5589	-.5572	-.7906	-.3267	-.1073	.0052

MACH ( 1 ) = .165 ALPHA ( 3 ) = 5.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C .1500 .3000 .4500 .6000 .7500 .9000

21/8	.000	-.1749	-.3218	-.0631	-.0997	-.1120	-.1300
	.334	-.4762	-.4401	-.4961	-.2482	-.1275	-.0048
	.520	-.9141	-.7695	-.4171	-.2515	-.1346	.0389
	.663	-.8825	-.5621	-.3516	-.2453	-.0978	.2162
	.873	-.9161	-.7102	-.7556	-.3784	-.1521	-.0352

(RDYU69)

CAST 7-B B16CSF1 J42 WATE18 WING UPPER SURFACE

MACH ( 1 ) = .165 ALPHA ( 4 ) = 10.010 RN/L = 1.200 MACH = .165

SECTION ( 1 ) WING  
X/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP21/B  
.000 -.2451 -.4115 -.0915 -.1335 -.1541 -.1416  
.334 -.6655 -.5895 -.6262 -.3230 -.1637 .0335  
.520 -.11521 -.9037 -.5283 -.3350 -.1579 .0002  
.663 -.11273 -.6849 -.4763 -.3227 -.0780 .0381  
.873 -.12745 -.8206 -.6808 -.4346 -.2538 -.1181  
MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.990 RN/L = 1.200 MACH = .165SECTION ( 1 ) WING  
X/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP21/B  
.000 -.3089 -.4715 -.0840 -.1581 -.1500 -.1675  
.334 -.7552 -.6562 -.6645 -.3632 -.2194 -.0362  
.520 -.1.1771 -.9824 -.5943 -.4252 -.2300 -.0246  
.663 -.1.3063 -.7489 -.5878 -.4779 -.1712 -.0858  
.873 -.1.5212 -.1.3000 -.1.2469 -.7988 -.4567 -.2275  
MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.975 RN/L = 1.200 MACH = .165SECTION ( 1 ) WING  
X/C .1500 .3000 .4500 .6000 .7500 .9000  
DEPENDENT VARIABLE CP21/B  
.000 -.3631 -.5705 -.0760 -.2041 -.1720 -.1697  
.334 -.8320 -.6524 -.7361 -.4469 -.3183 -.1366  
.520 -.1.0332 -.1.1933 -.9015 -.7431 -.3235 -.1201  
.663 -.1.7679 -.1.2933 -.1.1053 -.9396 -.5907 -.5177  
.873 -.1.3365 -.1.2590 -.1.2056 -.1.0459 -.0307 -.5825

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## TABULATED SOURCE DATA - CASTB

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CAST-B 816CSF1 J42 W6TE16 WING UPPER SURFACE

## REFERENCE DATA

SREF =	4.4120 80. FT.	ZHAP =	43.3940 IN.	BETA =	.000	PTM/P =	1.000
LREF =	19.2500 IN.	THAP =	.0000 IN.	M/F =	.286	BDFLAP =	-18.000
BREF =	37.9350 IN.	ZHAP =	-.4C50 IN.	ELEVON =	.000		
SCALE =	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.4.025 ANAL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.0354	-.1561	.0612	-.0394	-.0313	-.1262
	.334	-.0997	-.1460	-.3044	-.1517	-.0669	.0303
	.520	-.2243	-.4199	-.2207	-.1326	-.0744	-.0007
	.663	-.3825	-.3163	-.2386	-.1618	-.0254	.0974
	.873	-.3108	-.4183	-.7161	-.3091	-.0770	.0153

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.0790	-.2173	.0247	-.0579	-.0678	-.1295
	.334	-.2336	-.2739	-.3034	-.1683	-.0907	-.0107
	.520	-.4765	-.5144	-.2815	-.1613	-.0855	-.0202
	.663	-.5642	-.3756	-.2503	-.1874	-.0589	.0219
	.873	-.5506	-.5340	-.7708	-.3256	-.0992	.0070

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.960 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000	
21/8							
	.000	-.1699	-.3015	-.0215	-.1039	-.1098	-.1386
	.334	-.4566	-.3566	-.4163	-.2250	-.1240	-.0339
	.520	-.6302	-.6339	-.3697	-.2225	-.1187	-.0351
	.663	-.7664	-.4973	-.3283	-.2337	-.1199	-.0737
	.873	-.8706	-.6837	-.7611	-.3549	-.1525	-.0366

## PARAMETRIC DATA

(RDYUTO)

CA97-B B16C5F1 J42 W8TE10 WING UPPER SURFACE

MACH (1) = .165 ALPHA (4) = 9.950 FNL = 1.200 MACH = .165

## SECTION (1) WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
Z/R						
.000	-.2385	-.3876	-.3423	-.1415	-.1462	-.1522
.334	-.6503	-.5139	-.2659	.2998	-.1713	.1137
.520	-.10366	-.7744	-.4562	-.2556	-.1738	-.3674
.663	-.1.0099	-.6515	-.4775	-.3005	-.1588	-.1135
.873	-.1.1805	-.7701	-.6234	-.3913	-.2469	-.11130

MACH (1) = .165 ALPHA (5) = 15.30 FNL = 1.200 MACH = .165

## SECTION (1) WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
Z/R						
.000	-.2999	-.4480	-.3449	-.1555	-.1573	-.0118
.334	-.7153	-.5937	-.1783	-.3448	-.2185	-.4117
.520	-.1.1116	-.9355	-.5358	-.4095	-.2541	-.0719
.663	-.1.1772	-.6396	-.4990	-.4176	-.2845	-.2229
.873	-.1.5243	-.3426	-.1.1701	-.8442	-.5299	-.2818

MACH (1) = .165 ALPHA (6) = 19.995 FNL = 1.200 MACH = .165

## SECTION (1) WING

DEPENDENT VARIABLE CP

X/C	.1500	.3000	.4500	.6000	.7500	.9000
Z/R						
.000	-.3445	.5403	-.0750	-.1986	-.1817	-.1973
.334	-.7425	-.569	-.6742	-.4632	-.3522	-.1814
.520	-.912	-.1.0365	-.7581	-.6315	-.3974	-.1902
.663	-.1.5509	-.1.0027	-.8686	-.7567	-.5414	-.1279
.873	-.1.1169	-.1.0424	-.9651	-.8034	-.6311	-.4916



## OA37-B 810CSF1 J40 WATE10 WING TOTAL SURFACE

INDW03) ( 12 NOV 73 )

## REFERENCE DATA

LEEF = 4.4120 IN. FT. XMAP = 43.5940 IN.  
 LREF = 1.0.2300 IN. YMAP = .0000 IN.  
 ZC = 3.1.0.30 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

2/1/B .0000 .3340 .5200 .6630 .8730

X/C

-.900	.2050	.1903	.0940	.0386	-.1729
-.750	.3052	.2176	.1633	.2121	-.2967
-.600	.3917	.3659	.1368	.0344	-.2215
-.450	.2503	.3199	.2491	.0513	-.1042
-.300	.4637	.1421	.2504	.2638	.0444
-.150	.4235	.4581	.0744	.4404	.3365
.150	-.2234	-.7553	-.7158	-1.1785	-1.0151
.300	-.3770	-.5755	-.3957	-.7859	-.8503
.450	-.2303	-.5014	-.4236	-.4861	-.7052
.600	-.0957	-.2877	-.3153	-.3229	-.5207
.750	-.1456	-.1579	-.2160	-.0515	-.3484
.900	-.1989	.0020	.0319	.0836	-.2996

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.025 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

2/1/B .0000 .3340 .5200 .6630 .8730

X/C

-.900	.3272	.2463	.1239	.0490	-.2194
-.750	.4400	.3284	.2462	-.0223	-.2720
-.600	.4732	.4509	.2580	.0811	-.1868
-.450	.3575	.3762	.2958	.1307	-.0132
-.300	.5120	.2332	.2002	.2033	.1456
-.150	.4696	.5130	.1267	.3865	.3914
.150	-.2663	-.12527	-.7328	-1.3002	-2.6351
.300	-.4226	-.6571	-.6571	-.8333	-1.5604
.450	-.2533	-.6428	-.5013	-.5102	-1.1352
.600	-.1063	-.3408	-.4051	-.3510	.0670
.750	-.1646	-.2026	-.3007	-.1491	-.7556
.900	-.2015	-.0403	-.0321	-.0528	-.5976

## C157-1 816C5F1 J4 WATE18 WING TOTAL SURFACE

(6CYNW03)

MACH	1) =	.165	ALETA ( 3) =	20.045	RHL =	1.200	MACH	=	.165
SECTION ( 1) MACH					DEF	VNT V STABLE CP			
21.0	.. .900	.3340	.5200	.6650	.6721				
X/C									
- .400	.4411	.2977	.0622	-.0819	-.2202				
- .750	.3332	.3964	.2452	-.0603	-.2071				
- .500	.5460	.5203	.2321	.1755	-.1036				
- .450	.4373	.4801	.2715	.2041	.1383				
- .300	.3	.3585	.3474	.3929	.2644				
- .25	.2665	.5615	.135	.5288	.4887				
.150	-.197	-.1552	-.7552	-.4677	-.0984				
.300	-.4916	-.8539	-.8119	-.3001	-.0940				
.450	-.2443	-.8660	-.9063	-.1562	-.0510				
.575	-.3109	-.5732	-.818	-.0598	-.3487				
.750	-.2954	-.3610	-.5352	-.6286	-.2914				
.900	-.2672	-.1389	-.2028	-.518	-.8824				

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TABULATED SOURCE DATA - CASTB

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## CAST-B B16C5F1 JAO WATE10 WING TOTAL SURFACE

## REFERENCE DATA

SREF =	4.4120 93.FT.	XMRP =	43.5940 IN.	BETA =	.000	PIN/P =	1.300
LREF =	19.2300 IN.	YMRP =	.0000 IN.	H/B =	.039	BLDFLAP =	-10.000
BRPF =	37.9350 IN.	ZMRP =	-.4050 IN.	ELEVON =	.000		
SCALE =	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C							
-.900	.3428	.2140	.0724	.0712	.2090		
-.750	.4257	.3305	.2046	-.0687	-.3/18		
-.600	.4137	.4838	.2709	.1299	-.3281		
-.450	.2670	.4087	.3280	.0756	-.1793		
-.300	.4662	.2089	.2930	.2033	-.0390		
-.150	.4255	.4735	.1058	.3358	.3248		
.150	-.2229	-.7488	-.7143	-.1292	-.9420		
.300	-.3653	-.5516	-.5588	-.7610	-.8142		
.450	-.2157	-.4962	-.4330	-.4646	-.6644		
.600	-.1002	-.2780	-.3140	-.3033	-.4688		
.750	-.1385	-.1400	-.2057	-.0586	-.2890		
.900	-.1994	.0048	.0357	.0740	-.2201		

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.050 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C							
-.900	.5775	.3562	.1628	.1254	.1971		
-.750	.5912	.4845	.3352	.0504	-.2906		
-.600	.4667	.5640	.2572	.1916	-.1797		
-.450	.3545	.4390	.3228	.1822	-.0198		
-.300	.5122	.2831	.3286	.2721	.1317		
-.150	.4832	.5211	.0209	.1945	.4092		
.150	-.2605	-.12372	-.7663	-.1655	-.3165		
.300	-.4115	-.6946	-.6352	-.7431	-.5493		
.450	-.2452	-.6133	-.5397	-.5025	-.1622		
.600	-.1049	-.3205	-.4082	-.3621	-.9076		
.750	-.1464	-.1891	-.3043	-.1514	-.7753		
.900	-.1960	-.0360	-.0328	-.0540	-.5820		



		CART-8      B16C3F1 JAO WATE18 WING TOTAL SURFACE				(RDWDS) ( 12 NOV 73 )		
		REFERENCE DATA				PARAMETRIC DATA		
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	.9990	RNL =	1.200	MACH =	.165	
<b>SECTION ( 1 ) WING</b>								
21/8	.0000	.3340	.5200	.6630	.8730			
X/C								
-.900	.5551	.2794	.0769	.0983	-.1593			
-.750	.4810	.3986	.2669	.0286	-.3346			
-.600	.4083	.5308	.3641	.1346	-.3145			
-.450	.2446	.4253	.3508	.0653	-.1430			
-.300	.4247	.2091	.2657	.1383	-.0111			
-.150	.4417	.4706	.0598	.2552	.3430			
.150	-.2087	-.7368	-.7032	-.0915	-.8852			
.300	-.3564	-.5216	-.5921	-.7370	-.7889			
.450	-.2050	-.4979	-.4290	-.4510	-.6360			
.600	-.0833	-.2590	-.2883	-.2883	-.4452			
.750	-.1287	-.1179	-.1850	-.0407	-.2650			
.900	-.1929	.0233	.0510	.0757	.1954			
MACH ( 1 ) =	.165	ALPHA ( 2 ) =	.15.000	RNL =	1.200	MACH =	.165	
<b>SECTION ( 1 ) WING</b>								
21/8	.0000	.3340	.5200	.6630	.8730			
X/C								
-.900	.7661	.4753	.2355	.1603	-.1638			
-.750	.6377	.5783	.4474	.0146	-.2713			
-.600	.4939	.5996	.2296	.1673	-.1677			
-.450	.3484	.4579	.3265	.1686	-.0194			
-.300	.4768	.2631	.2477	.2002	.1359			
-.150	.4862	.5122	-.1683	-.0663	.4041			
.150	-.2338	-.12339	-.7551	-.1446	-.17313			
.300	-.4033	-.6768	-.6430	-.7203	-.2919			
.450	-.2322	-.5634	-.5119	-.4999	-.0844			
.600	-.0847	-.3053	-.3867	-.3541	-.8347			
.750	-.1416	-.1720	-.2776	-.1372	-.6759			
.900	-.1916	-.0172	-.5577	-.0399	-.5358			

CA57-B B16CSF1 J4D WATE10 WING TOTAL SURFACE (REVNO.5)						
MACH (1) = .165	ALPHA (3) = 20.000	RHO_L = 1.200	MACH = .165	DEFINITION VARIABLE CP		
SECTION 1 (1)WING						
X/C	Z/C	CP	CP	CP	CP	CP
-.900	.6985	.5218	.2029	.0886	-.1614	
-.750	.6852	.6290	.3885	-.3077	-.2470	
-.600	.5570	.6463	.3120	.2015	-.0753	
-.450	.4227	.5226	.3356	.2694	.0945	
-.300	.3352	.3973	.2956	.44C1	.22E1	
-.150	.5549	.5816	.22E1	.4177	4.644	
.150	-.5107	-.1.5380	-.79017	-1.4282	-1.1243	
.300	-.4794	-.6332	-.800C7	-1.2E+2	-1.0246	
.450	-.3626	-.7936	-.8466	-1.0757	-.9E14	
.600	-.2213	-.5473	-.7721	-.9505	-.9092	
.750	-.2862	-.3473	-.6C59	-.5630	-.821C	
.900	-.2755	-.1362	-.2752	-.5254	-.9193	

(DYNAMIC) (112 NOV 73)

## REFERENCE DATA

SURF = 4.41901 1.3 100  
 UREF = 10.23001 IN.  
 SREF = 37.93501 IN.  
 SCALE = .0405

MACH (1) = .165 ALPHA (1) = -.3,980 ANGL = 1,200 MACH = .165

CAST-B B16C3F1 140 WTE10 WING TOTAL SURFACE

## PARAMETRIC DATA

BETA = .000 PTN/P = 1.000  
 M/B = .125 BOFLAP = -10.000  
 ELEVON = .000

SECTION (1) WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900 -.2104 -.1878 -.1367 -.0729 -.0264  
 -.750 -.2643 -.2871 -.2514 -.3514 -.2403  
 -.600 -.3566 -.3047 -.2594 -.2552 -.3371  
 -.450 -.6472 -.4980 -.2902 -.2331 -.2822  
 -.300 -.6191 -.5955 -.3827 -.2460 -.2504  
 -.150 -.0505 -.3923 -.6676 -.4604 -.0944  
 .150 -.0570 -.1641 -.3151 -.3684 .0854  
 .300 -.1640 -.2415 -.3678 -.4057 -.3566  
 .450 -.1035 -.3046 -.3272 -.3071 -.2796  
 .600 -.0029 -.1770 -.2067 -.2153 -.2056  
 .750 -.0567 -.0267 -.0967 -.0263 -.1019  
 .900 -.1759 -.0393 -.1243 .1169 .0069

MACH (1) = .165 ALPHA (2) = .010 ANGL = 1,200 MACH = .165

SECTION (1) WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900 -.1440 -.13501 -.0756 -.0364 .3247  
 -.750 -.1773 -.2474 -.1644 -.2908 -.1759  
 -.600 -.1785 -.1523 -.1309 -.0992 -.2242  
 -.450 -.4548 -.3013 -.0732 -.0157 -.1450  
 -.300 -.2918 -.0642 -.1631 -.1134 -.0664  
 -.150 -.1486 -.0961 -.1822 -.1258 .2236  
 .150 -.0911 -.2676 -.4567 -.6249 -.1125  
 .300 -.2249 -.3252 -.4453 -.5155 -.4647  
 .450 -.1567 -.3455 -.3611 -.3616 -.3612  
 .600 -.0311 -.1917 -.2171 -.2384 -.2424  
 .750 -.0716 -.0335 -.5094 -.2222 -.1543  
 .900 -.1644 -.0439 -.0690 .1253 -.0103

## CASTE - TABULATED SOURCE DATA - CASTE

(RDY4006)

MACH 1.11 = .165 ALPHA = 31 = 4.980 RNL = 1.200 MACH = .165

## SECTION 1) WING

DEPENDENT VARIABLE CP

21.0 .0000 .3340 .5200 .6630 .8730

A/C

- .900	.0593	.0742	.1516	.0526	.0700
- .750	.0405	.1359	.1875	.2123	.1607
- .600	.0298	.0049	.5768	.0397	.1407
- .450	.0223	.0520	.5611	.125	.0144
- .300	.0118	.2712	.3649	.2213	.0748
- .150	.2433	.1526	.1534	.2113	.3703
.150	.1480	.9151	.5675	.9275	.4747
.300	.2556	.3211	.3117	.667	.4350
.450	.1741	.4472	.3744	.416	.4905
.600	.0672	.2194	.2301	.264	.3256
.750	.0833	.2424	.1063	.2473	.1688
.900	.1672	.0424	.0325	.1	.2199

MACH 1.1 = .165 ALPHA = 31 = 4.980 RNL = 1.200 MACH = .165

## SECTION 1) WING

DEPENDENT VARIABLE CP

21.0 .0000 .3340 .5200 .6630 .8730

A/C

- .900	.0250	.0050	.1450	.0550	.0290
- .750	.0631	.0030	.2621	.1227	.1772
- .600	.1501	.1551	.0551	.1377	.0646
- .450	.446	.0711	.1071	.2350	.0323
- .300	.1665	.0790	.1583	.3632	.1685
- .150	.3264	.2919	.1325	.4197	.4447
.150	.2744	.7175	.6827	.1	.119
.300	.1002	.5425	.5871	.7917	.2220
.450	.2067	.4866	.4503	.4902	.6066
.600	.0226	.2693	.2669	.3331	.4751
.750	.1009	.1133	.1674	.5520	.2824
.900	.1767	.5962	.5967	.1326	.1945

MACH 1.1 = .165 ALPHA = 31 = 4.980 RNL = 1.200 MACH = .165

## SECTION 1) WING

DEPENDENT VARIABLE CP

21.0 .0000 .3340 .5200 .6630 .8730

A/C

- .900	.0703	.0308	.0359	.0736	.0382
- .750	.1704	.5626	.0045	.1112	.1310
- .600	.2537	.2459	.0675	.1497	.0111
- .450	.0841	.1986	.1	.2281	.1291
- .300	.3656	.0449	.1421	.4722	.2404

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TABULATED SOURCE DATA - CASE 6

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MACH (1) = .165    ALPHA (5) = 14.000  
 SECTION (1) M10C  
 Z/R .0000 .3340 .5000 .6630 .8730

(INCHES)

SECTION (1) M10C  
 Z/R .0000 .3340 .5000 .6630 .8730

DEPENDENT VARIABLE CP

Z/R	MACH (1) = .165	ALPHA (5) = 14.000	CA97-B	B16C5F1 JAO WTE10	WING TOTAL SURFACE
-1.90	.4161	.4000	.0531	.6460	.4371
-1.50	.2471	-.7020	-.7437	-.3271	-.0487
-1.00	.3400	.6517	-.6104	.8668	-.1399
-0.50	.2233	.5773	-.4686	-.5223	-.9711
.0000	.0824	.2065	-.3483	-.3076	-.7619
.5000	.1064	.1501	-.2163	-.0663	-.6693
.9000	.1871	.0160	-.1117	.0034	-.4544

MACH (1) = .165    ALPHA (5) = 14.000  
 SECTION (1) M10C  
 Z/R .0000 .3340 .5000 .6630 .8730

DEPENDENT VARIABLE CP

Z/R	MACH (1) = .165	ALPHA (5) = 14.000	CA97-B	B16C5F1 JAO WTE10	WING TOTAL SURFACE
-1.90	.0866	.0100	-.1694	-.1338	-.1060
-1.50	.2017	.0564	-.1033	-.0846	-.1661
-1.00	.1204	.2647	.0032	.2095	.0197
-0.50	.1866	.2854	-.1123	.2815	.1584
.0000	.2675	.2230	.0843	.4163	.2743
.5000	.4875	.4623	.2063	.7724	.4988
.9000	.2916	.15174	-.7569	-.12851	-.1.0775
.2000	.4063	.8403	-.7182	-.1.1451	-.1.0298
.450	.2972	.7774	-.7639	-.1.1427	-.1.0061
.600	.1580	.5012	-.7700	-.1.0679	-.9736
.750	.2435	.3081	-.5701	-.5943	-.9245
.900	.2727	.1413	-.4139	-.4813	-.9025

REGULATED SOURCE DATA - GANTT  
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WAST-8 818CSF1 140 WATE10 WING TOTAL SURFACE

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SECTION ( 1 ) WING		SECTION ( 2 ) WING		SECTION ( 3 ) WING	
MACH ( 1 ) =	.165	MACH ( 1 ) =	.165	MACH ( 1 ) =	.165
BREF	4.4120 82. F.	XDP	=	43.5940 IN.	BETA = .000
LREF	19.2300 IN.	YDP	=	.0000 IN.	H/B = .123
BREF	37.8350 IN.	ZDP	=	-.4050 IN.	ELEVON = .000
SCALE	.0405				
DEPENDENT VARIABLE CP					
21/8	.0000	.3340	.5200	.6630	.8730
X/C					
- .900	-.1776	-.2027	-.1103	-.1022	-.0516
-.750	-.534	-.3695	-.2899	-.3807	-.2962
-.900	-.312	-.3920	-.2939	-.2700	-.3420
-.450	-.9019	-.6515	-.5325	-.3620	-.5025
-.300	-.8769	-.1843	-.5526	-.3159	-.2785
-.150	.0277	-.5076	-.7961	-.5619	-.1279
.150	-.0458	-.1364	-.2710	-.3142	.1257
.300	-.1459	-.2287	-.4102	-.3824	-.3274
.450	-.0648	-.2800	-.3034	-.2860	-.2619
.600	.0112	-.1541	-.1823	-.1900	-.1920
.750	-.0517	.0244	-.0992	-.0152	-.0933
.900	-.1665	.0529	-.0603	.1275	.0167
DEPENDENT VARIABLE CP					
21/8	.0000	.3340	.5200	.6630	.8730
X/C					
- .900	-.1453	-.1560	-.0767	-.0665	-.0017
-.750	-.2154	-.3071	-.2036	-.3140	-.1910
-.900	-.2717	-.2011	-.1641	-.1224	-.2249
-.450	-.6015	-.3934	-.1231	-.0825	-.1456
-.300	-.4165	-.7169	-.2591	-.0557	-.0631
-.150	.1394	-.1411	-.4972	-.1937	.1774
.150	-.0926	-.2547	-.1414	-.5874	-.0836
.300	-.1972	-.3245	-.4130	-.5178	-.4493
.450	-.1306	-.3421	-.3570	-.3548	-.3463
.600	-.0183	-.1753	-.2086	-.2270	-.2342
.750	-.0762	.0321	-.0894	-.0172	-.0978
.900	-.1601	.0518	-.0264	.1204	.0043

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.995 RNL = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C  
 -.100 -.0486 -.0827 -.1524 -.1034 -.0724  
 -.750 -.0533 -.1485 -.1966 -.2652 -.1498  
 -.500 -.0469 -.0026 -.0554 -.0010 -.1367  
 -.450 -.2957 -.1177 .0456 .0850 -.0266  
 -.300 -.0818 -.3352 -.0392 .1847 .0610  
 -.150 .2430 .1361 -.2399 .1650 .3502  
 .150 -.1387 -.4680 -.5620 -.8844 -.4426  
 .300 -.2378 -.4606 -.5615 -.6537 -.6171  
 .450 -.1655 -.4318 -.3737 -.4169 .4835  
 .600 -.0563 -.2069 -.2167 -.2577 -.3265  
 .750 -.0929 .0172 -.1023 -.0396 -.1635  
 .900 -.1578 .0435 -.0266 .1046 -.0198

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.975 RNL = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C  
 -.900 .0188 .0078 -.1156 -.0899 -.0193  
 -.750 .0821 .0008 -.1699 -.2155 -.1742  
 -.600 .1093 .1515 -.0200 .0952 -.0553  
 -.450 -.1073 .0569 .1368 .2161 .0584  
 -.300 .1213 -.0998 .1648 .3340 .1579  
 -.150 .3257 .2743 -.1119 .3506 .4414  
 -.150 -.2022 -.7141 -.7007 -.1361 -.8940  
 -.300 -.2846 -.5065 -.5697 -.7842 .8021  
 .450 -.2008 -.4773 -.4478 -.4779 -.6463  
 .500 -.0776 -.2586 -.2845 -.3317 -.4520  
 .750 -.1044 -.0226 -.1517 -.0515 -.2729  
 .900 -.1720 .02CJ -.0614 .1210 -.1511

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.000 RNL = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C  
 -.900 .0756 .0518 -.0606 -.1553 -.0218  
 -.750 .1775 .0758 .1276 -.1583 -.1353  
 -.600 .2193 .2112 .1552 .1254 -.0020  
 -.450 .0171 .1612 .0918 .1274 .1234  
 -.300 .2517 .0430 .1647 .4796 .2400

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MACH (1) = .165 ALPHA (5) = 15.000

(RDW007)

CA57-B B16C5F1 J40 WATE18 WING TOTAL SURFACE

## SECTION (1) WING

## DEFINENT VARIABLE CP

MACH (1) = .165	ALPHA (5) = 15.000	RNL = 19.975	RNL = 1.200	MACH = .165
21/B .0000	.3340	.5200	.5630	.8730
X/C				
-.150	.4063	.3758	-.0356	.2386
.150	-.2439	-1.1957	-.7840	-1.2466
.300	-.3392	-.6823	-.6354	-.7953
.450	-.2286	-.5890	-.5485	-.5071
.600	-.0817	-.2946	-.3295	-.3566
.750	-.1196	.0215	-.2364	-.1167
.900	-.1824	-.0430	-.0713	-.0203

MACH (1) = .165 ALPHA (5) = 15.000

(RDW007)

## SECTION (1) WING

## DEFINENT VARIABLE CP

MACH (1) = .165	ALPHA (5) = 15.000	RNL = 19.975	RNL = 1.200	MACH = .165
21/B .0000	.3340	.5200	.5630	.8730
X/C				
-.900	.1177	.0676	-.1341	-.2550
-.750	.2387	.1349	-.0313	-.2613
-.600	.3162	.3240	.0846	.1641
-.450	.1451	.2934	.0694	.3327
-.300	.3538	.2151	.1476	.5398
-.150	.3214	.4848	.2405	.7114
.150	-.2789	-.4732	-.7231	-.2829
.300	-.3835	-.7903	-.6652	-.1257
.450	-.2833	-.7171	-.7785	-.1595
.600	-.1334	-.4629	-.6863	-.0167
.750	-.2114	-.0399	-.6207	-.4932
.900	-.2447	-.1268	-.3449	-.4695

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TABULATED SOURCE DATA - CA57B

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CA57-B B16C5F1 J40 WTE18 WING TOTAL SURFACE (REV008) ( 12 NOV 73 )

## REFERENCE DATA

WREF =	4.4120 SQ.FT.	ZHAP =	43.5940 IN.	BETA =	.000	PIN/P =	1.500
W <sub>0</sub> =	.00 COUN IN.	THAP =	.0000 IN.	H/B =	.125	BDFLAP =	-16.000
DREF =	37.9350 IN.	ZHAP =	-.4050 IN.	ELEVON =	.000		
SCALE =	.0405						

MACH (.1) = .165    ALPHA (.1) = -4.015    RNL = 1.200    MACH = .165

SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.900	-.2177	-.2610	-.2552	-.1630	-.0766	
	-.750	-.4034	-.4986	-.2921	-.4403	-.3262	
	-.500	-.6828	-.5101	-.3825	-.3249	-.3757	
	-.450	-.1210	-.6751	-.4783	-.3683	-.3469	
	-.300	-.1.0435	-.1.4036	-.7160	-.4180	-.3312	
	-.150	-.0137	-.6249	-.9599	-.7087	-.2055	
	.150	-.0604	-.1.3554	-.2623	-.2804	.1361	
	.300	-.1.5334	-.2315	-.3360	-.4056	-.3217	
	.450	-.0842	-.2756	-.3092	-.2886	-.2694	
	.600	-.0009	-.1198	-.1.473	-.1.949	-.2024	
	.750	-.0620	-.1.8632	-.0918	-.0375	-.1043	
	.900	-.1.766	-.2116	-.1.158	.0893	.0015	

MACH (.1) = .165    ALPHA (.2) = -.030    RNL = 1.200    MACH = .165

SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.900	-.1.674	-.1.959	-.2503	-.1288	-.0361	
	-.750	-.2829	-.1.639	-.2030	-.3800	-.2162	
	-.500	-.3907	-.2968	-.2581	-.1.681	-.2408	
	-.450	-.7432	-.5381	-.2165	-.1.362	-.1.782	
	-.300	-.5022	-.7636	-.3612	-.1.221	-.1.207	
	-.150	-.1.141	-.2157	-.6465	-.2.998	.1.132	
	.150	-.0916	-.2615	-.4513	-.5375	-.0473	
	.300	-.1.901	-.3065	-.4125	-.5414	-.4412	
	.450	-.1.343	-.3254	-.3674	-.3.98	-.3448	
	.600	-.0283	-.1.509	-.1.742	-.2.319	-.2.367	
	.750	-.0763	-.1.699	-.1.077	-.0.370	-.1.027	
	.900	-.1.745	-.1.924	-.0.525	.0.980	-.0.0124	

## CASTE 816C5F1 J40 W7E16 WING TOTAL SURFACE

(DRAWN 08)

MACH ( 1 ) = .165    ALPHA ( 3 ) = 4.990    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C					
-.900	-.0516	-.1132	-.3190	-.1482	-.0972
-.750	-.0925	-.1859	-.2256	-.3037	-.1402
-.600	-.1155	-.0877	-.1282	-.0136	-.1197
-.450	-.3698	-.1674	.0671	.0676	.0243
-.300	-.1281	-.3791	-.0736	.1573	.0645
-.150	-.2471	-.1101	-.2722	.1098	.0518
.150	-.1543	-.4618	-.5450	-.7904	-.3751
.300	-.2236	-.4368	-.4400	.6030	-.5843
.430	-.1527	-.4107	-.3707	-.3708	-.4516
.600	-.0480	-.1680	-.1700	-.2404	-.2978
.750	-.0712	-.1127	-.1356	-.0510	-.1448
.900	-.1586	-.1728	-.0464	.0549	.0023

MACH ( 1 ) = .165    ALPHA ( 4 ) = 9.380

RNL = 1.200

MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C					
-.900	.0254	.0068	-.1723	-.1200	-.0337
-.750	.0636	-.0177	-.1791	.2105	-.1328
-.600	.0678	.1045	-.0413	.1128	-.0354
-.450	-.1421	.0376	.1572	.2221	.0807
-.300	.0869	-.1295	.1503	.3225	.1738
-.150	.3593	.2751	-.1168	.3352	.4651
.150	-.1694	-.6811	-.6335	-.1226	-.8048
.300	-.2728	-.4201	-.5160	-.7567	-.7423
.450	-.1725	-.4433	-.4664	.4461	-.5964
.600	-.0617	-.2163	-.2316	.3121	-.3954
.750	-.0772	-.0358	-.0875	.0439	-.2198
.900	-.1443	-.1018	-.0379	.0228	-.1057

MACH ( 1 ) = .165    ALPHA ( 5 ) = 14.950

RNL = 1.200

MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C					
-.900	.1141	.0751	.0421	-.1034	-.0201
-.750	.1945	.0960	-.0055	-.1739	-.1248
-.600	.1975	.2004	.0704	.2210	.0279
-.450	.0101	.1787	.1971	.3264	.1356
-.300	.2428	.0397	.2523	.4165	.2540



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TABULATED SURFACE DATA - CA378

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CA57-B B16C5F1 J40 W7E16 WING TOTAL SURFACE

(ADW06)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.950

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C					
-.150	.4194	.3815	.0331	.4281	.4689
.150	-.2403	-1.1518	-.7095	-1.1904	-1.6193
.300	-.3239	-.5901	-.5991	-.8710	-1.1714
.450	-.2113	-.5321	-.5071	-.5268	-.9057
.600	-.0727	-.2642	-.2702	-.2775	-.6188
.750	-.0815	-.1120	-.1532	-.0702	-.5538
.900	-.1732	-.0310	-.0316	.0422	-.4120

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.965 RN/L = 1.200 MACH = .145

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C					
-.900	.1466	.0649	-.1632	-.3191	-.0617
-.750	.2472	.1262	-.0462	-.2933	-.1242
-.600	.3066	.3004	.0870	.1368	.0565
-.450	.1440	.2860	.0167	.3038	.1941
-.300	.3568	.2264	.1122	.5267	.3223
-.150	.5009	.4787	.2163	.7010	.5051
.150	-.2774	-1.5065	-.7294	-1.2015	-1.0935
.300	-.4193	-.8081	-.6912	-1.2180	-1.0060
.450	-.2893	-.7299	-.8383	-1.1600	-.9963
.600	-.1420	-.4686	-.7004	-1.0022	-.9294
.750	-.2529	-.3154	-.5333	-.5733	-.8599
.900	-.2573	-.0610	-.3157	-.5172	-.8742

## CA57-B B16C5F1 J40 WATE18 WING TOTAL SURFACE

(ROW09) (112 NOV 73)

## REFERENCE DATA

BREF =	4.4120 SQ.FT.	XWRF =	43.5940 IN.	BETA =	.000	PIN/P =	1.100
LREF =	19.2350 IN.	YWRF =	.0000 IN.	H/B =	.125	BDFLAP =	.000
BREF =	37.9350 IN.	ZWRF =	-.4030 IN.	ELEVN =	.000		
SCALE =	.5405						

## PARAMETRIC DATA

MACH ( 1 ) =	.165	ALPHA ( 1 ) =	-4.000	RNL =	1.200	MACH =	.165
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## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C							
-.900	-.1568	-.2414	-.2257	-.1614	.0800		
-.750	-.3649	-.4746	-.3320	-.4183	-.3166		
-.600	-.6438	-.5399	-.3769	-.5232	-.3759		
-.450	-.1052	-.8334	-.4485	-.3504	-.3359		
-.300	-1.0087	-.1.3163	-.6947	-.4022	-.3300		
-.150	-.0078	-.5971	-.9527	-.7166	-.2056		
.150	-.0588	-.1479	-.2837	-.3178	.1076		
.300	-.1682	-.2377	-.3436	-.3789	-.3295		
.450	-.1121	-.2917	-.3133	-.2938	-.2651		
.600	.0661	-.1605	-.1950	-.2028	-.2035		
.750	-.0551	-.0364	-.0902	-.0206	-.0976		
.900	-.1863	.0639	.0952	.1220	.0100		

MACH ( 1 ) =	.155	ALPHA ( 2 ) =	- .015	RNL =	1.200	MACH =	.165
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## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C							
-.900	-.0972	-.1488	-.1788	-.0946	-.0386		
-.750	-.2165	-.3462	-.2367	-.3536	-.1929		
-.600	-.3576	-.2669	-.1933	-.1449	-.2287		
-.450	-.6987	-.4924	-.1838	-.1110C7	-.1621		
-.300	-.4781	-.7566	-.3545	-.0996	-.0992		
-.150	.1255	-.1750	-.5746	-.2736	.1266		
.150	-.0895	-.2331	-.4397	-.5637	-.0624		
.300	-.1616	-.3531	-.4370	-.5186	-.4383		
.450	-.132*	-.3254	-.3466	-.3492	-.3374		
.600	-.0200	-.1664	-.1937	-.2147	-.2246		
.750	-.0693	-.0312	-.0755	-.0106	-.0820		
.900	-.1591	.0617	.1173	.1451	.0106		

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## TABULATED SOURCE DATA - CASTE

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CA57-B B16C5F1 J4D W3T10 WING TOTAL SURFACE

(RADWOS)

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.965 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0535	-.0780	-.2017	-.1311	-.0950
-.750	-.0421	-.1707	-.2303	-.2965	-.1434
-.600	-.1026	-.0649	-.1276	-.0227	-.1433
-.450	-.3709	-.1823	.0562	.0580	-.0216
-.300	-.1390	-.3395	-.0650	.1543	.0600
-.150	2442	.1073	-.2804	.1025	.3331
-.150	-.1400	-.4867	-.5876	-.6969	-.4302
.300	-.2522	-.5119	-.4816	-.6429	-.6011
.450	-.1831	-.4132	-.3657	-.4028	-.4621
.600	-.0501	-.2006	-.2301	-.2663	-.3286
.750	-.0947	-.0276	-.0919	-.0313	-.1380
.900	-.1679	.0480	-.0044	.1162	-.0489

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.985 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	.1275	.0391	-.1605	-.1194	-.0431
-.750	.0965	.0127	-.1590	-.2222	-.1387
-.600	.0823	.1270	-.0209	.1167	-.0391
-.450	-.1393	.0358	.1452	.2112	.0653
-.300	.0793	-.0919	.1557	.3169	.1709
-.150	.3451	.2813	-.0996	.3376	.4564
.150	-.2008	-.6974	-.6570	-.1132	-.8479
.300	-.2943	-.5247	-.5497	-.7610	-.7747
.450	-.2055	-.4416	-.4308	-.4794	-.5477
.600	-.0702	-.2346	-.2644	-.3059	-.4443
.750	-.0901	-.0723	-.1259	-.0370	-.2542
.900	-.1766	.0370	.0493	.1413	-.1664

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.970 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

-.900	.2223	.1231	-.0416	-.0057	-.3067
-.750	.2193	.1290	-.0057	-.1172	
-.600	.2196	.2540	.5731		-.123
-.450	.0177	.1741	.1817		.384
-.300	.595	.595	.595		.2609

C-57-E B16CSF1 J40 W0TE10 WING TOTAL SURFACE

(RDW009)

MACH (1) = .163    ALPH A (5) = 14.970  
 SECTION (1) WING  
 21/B .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C	.4391	.3976	.0170	.4352	.4568
.150	- .2368	- 1.1863	- .7497	- 1.2786	- 1.6718
.300	- .3452	- .7065	- .5854	- .8291	- 1.1490
.450	- .2166	- .5297	- .5155	- .4929	- .9096
.600	- .0720	- .2779	- .3195	- .2926	- .6460
.750	- .1168	- .1686	- .1946	- .0582	- .5330
.900	- .1948	.0116	.0372	.0659	- .3787

MACH (1) = .165    ALPH A (6) = 19.965    ANL = 1.200    MACH = .165

## DEPENDENT VARIABLE CP

X/C	.2735	.1268	.1746	.3655	.0374
.900	- .2917	.1957	.0068	.2717	- .1231
.750	.3135	.3211	.0799	.1936	.0617
.600	.1199	.2906	.0332	.3775	.2082
.450	.3374	.2140	.1108	.5502	.3263
.300	.1150	.4642	.1373	.6365	.5112
.150	- .2878	- .5510	- .7665	- 1.4283	- 1.1402
.000	.4238	.8790	.6590	.3186	- 1.0828
.450	- .2545	- .7053	- .7780	- 1.1001	- 1.0258
.600	- .1569	- .4902	- .7124	- 1.0035	- .9437
.750	- .2459	- .2959	- .5233	- .4875	- .8806
.900	- .2658	- .1270	- .3140	- .4442	- .8571



DATE 04 OCT 74

TABULATED SOURCE DATA - CHARTS

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CHARTS

SECTION ( 1 ) WING

(RDW/M0) ( 12 NOV 73 )

## REFERENCE DATA

BEDF	=	4.1120 94.171.	ZMP	=	45.5940 IN.
LREF	=	19.2300 IN.	TRP	=	.0000 IN.
MREF	=	37.9350 IN.	ZMP	=	-.4050 IN.
SCALE	=	.0405			

MACH ( 1 ) = .165 ALPH<sub>A</sub> ( 1 ) = -.015 ANL = 1.200 MACH = .165

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.800	-.1393	-.2037	-.1770	-.1904	-.0937
-.750	-.3010	-.3932	-.2646	-.3818	-.2668
-.600	-.3244	-.4482	-.3116	-.2746	-.3293
-.450	-.6121	-.6797	-.5436	-.5009	-.3093
-.300	-.8720	-.1.0776	-.5675	-.3460	-.2848
-.150	.0242	-.4808	-.7844	-.6028	-.1336
.150	-.6356	-.1.464	-.2861	-.3324	.1125
.300	-.1.5550	-.2314	-.3502	-.3039	-.3343
.450	-.0926	-.3062	-.3050	-.2965	-.2757
.600	.0035	-.1.462	-.1.721	-.2058	-.2028
.750	-.0517	-.0755	-.0616	-.0262	-.0982
.800	-.1.8116	-.0534	-.0428	-.1.104	.0051

MACH ( 1 ) = .165 ALPH<sub>A</sub> ( 2 ) = -.015 ANL = 1.200 MACH = .165

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.800	-.0754	-.1461	-.1212	-.0906	-.1173
-.750	-.1.961	-.2.964	-.1.954	-.3.049	-.1.740
-.600	-.2.937	-.2.304	-.1.990	-.1.112	-.2.003
-.450	-.5935	-.4.084	-.1.266	-.0786	-.1.441
-.300	-.6240	-.6134	-.2.663	-.0836	-.0641
-.150	.1.408	-.1.224	-.4.510	-.2.126	.1.907
.150	-.0948	-.2.603	-.4.370	-.5.722	-.0634
.300	-.1.971	-.3.191	-.4.346	-.5.249	-.4.560
.450	-.1.338	-.3.427	-.3.478	-.3.514	-.3.455
.600	-.0266	-.1.626	-.1.629	-.2.181	-.2.268
.750	-.0.717	-.0.977	-.0.647	-.0.271	-.0.949
.800	-.1.625	-.0.510	-.0.447	-.2.16	.0.045

(RDM10)

## CASH-B B16CF1 JAO WATE10 WING TOTAL SURFACE

MACH ( 1 ) = .165    ALPHA ( 3 ) = 4.970    RNL = 1.200    MACH = .165

SECTION ( 1 )WING  
DEPENDENT VARIABLE CP

X/C					
- .900	.0347	-.0063	.2003	-.1360	-.1179
- .750	-.0352	-.1338	-.1900	-.2781	-.1464
- .600	-.0479	-.0172	-.1022	-.0017	-.1169
- .450	-.2933	-.1223	.1022	.0669	-.0169
- .300	-.0611	-.2374	-.0182	.1881	.0683
- .150	.2541	.1416	-.1177	.1698	.3586
.150	-.1471	-.5173	-.5735	-.6764	-.4365
.300	-.2577	-.5223	-.5673	-.6499	-.6170
.450	-.1633	-.4071	-.3605	-.4091	-.47e1
.600	-.0383	-.1992	-.2133	-.2604	-.5343
.750	-.0959	-.0708	-.0896	-.0452	-.1637
.900	-.1720	-.0638	-.0198	.1041	-.0256

MACH ( 1 ) = .165    ALPHA ( 4 ) = 9.980    RNL = 1.200    MACH = .165

SECTION ( 1 )WING  
DEPENDENT VARIABLE CP

X/C					
- .900	.1479	.0273	-.1507	-.1491	-.0992
- .750	-.1053	.0141	-.1771	-.2279	-.1541
- .600	.1174	-.1480	-.0212	.0995	-.0412
- .450	-.0950	.0768	.1647	.2146	.0573
- .300	.1051	-.0179	.1680	.3982	.1643
- .150	.3473	.2897	-.0341	.3593	.4499
.150	-.2050	-.7191	-.7035	-.1143	-.9154
.300	-.3203	-.5282	-.5875	-.7855	-.8165
.450	-.8146	-.4432	-.4283	-.4992	-.6986
.600	-.0762	-.2567	-.2835	-.3240	-.4821
.750	-.1119	-.1190	-.1592	-.0602	-.2813
.900	-.1699	.0192	.00111	.1145	-.2045

MACH ( 1 ) = .165    ALPHA ( 5 ) = 14.970    RNL = 1.200    MACH = .165

SECTION ( 1 )WING  
DEPENDENT VARIABLE CP

X/C					
- .900	.2291	.0864	-.0396	-.0984	-.0318
- .750	.8164	-.1130	.0068	-.1711	-.1273
- .600	.2311	.2458	.0848	.1864	.0203
- .450	.0367	.1750	.1876	.3358	.1429
- .300	.2440	.0869	.1393	.4825	.2169

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## TABULATED SOURCE DATA - CASE 7

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CAST-E B14C5F1 JAO WATE10 WING TOTAL SURFACE

(ADMM10)

MACH ( 1 ) = .165

ALPHA ( 5 ) = 14.970

SECTION 11 WING

DEPENDENT VARIABLE CP

21/8 .00000 .33400 .52000 .66300 .8730

X/C

-.130	.1336	.3626	-.0245	.3071	.4616
.150	-.2316	-.1.2296	-.7594	-1.2544	-1.8722
.300	-.3636	-.7013	-.6436	-.8092	-.1.2276
.450	-.2281	-1.0092	-.5621	-.5032	-.8981
.600	-.0900	-.3209	-.3658	-.3623	-.7393
.750	-.1328	-.1189	-.2460	-.1386	-.6344
.900	-.2127	-.0277	-.0054	-.0355	-.5126

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.950 RNL = 1.200 MACH = .165

SECTION 11 WING

DEPENDENT VARIABLE CP

21/8 .00000 .33400 .52000 .66300 .8730

X/C

-.900	.3093	.1013	-.1706	-.3726	-.0376
-.750	-.2964	.1687	-.0331	-.3231	-.1154
-.600	-.3291	.3235	.0696	.1436	.0768
-.450	.1634	.5168	.0794	.2980	.2116
-.300	.3743	.2757	.1319	.5200	.3315
-.150	.5161	.4936	.2151	.7038	.5170
.150	-.2924	-1.5429	-.7432	-1.3564	-1.1223
.300	-.4352	-.9181	-.7143	-1.3442	-1.0571
.450	-.2956	-.7350	-.8133	-1.1201	-1.0011
.600	-.1617	-.4976	-.7106	-1.0109	-0.9446
.750	-.2540	-.3335	-.5453	-.5425	-.0763
.900	-.2601	-.2304	-.2242	-.4763	-.6653

## CAA37-B B10CSF1 J40 WTE10 WING TOTAL SURFACE

(ADWW11) (12 NOV 73)

## REFERENCE DATA

WING	4,4120 SQ.FT.	XMAP =	43.3940 IN.	BETA =	.000
LEEF	.16,2500 IN.	YMAP =	.0000 IN.	M/B =	.125
REEF	.37,9350 IN.	ZMAP =	.4050 IN.	BOFLAP =	.000
SCALE	.0403			ELEVN =	.000

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.010 RH = 1.200 MACH = .165

## SECTION ( 1 ) WING

Z/E = .0000 .3340 .5200 .6630 .8730

X/C	Z/E	ALPHA ( 1 )	RH	MACH
- .900	-.1704	-.1686	-.1431	-.1111
- .750	-.2462	-.2221	-.2000	-.3160
- .600	-.3249	-.2837	-.2503	-.2022
- .450	-.6270	-.4774	-.2028	-.2030
- .300	-.8242	-.6905	-.3603	-.2122
- .150	.6870	-.3158	-.5660	-.3960
.150	-.9337	-.1507	-.2637	-.3403
.300	-.16C8	-.2405	-.3127	-.3301
.450	-.0910	-.2925	-.3183	-.3000
.600	-.0044	-.1565	-.1605	-.1987
.750	-.0500	-.0316	-.0913	-.0150
.900	-.17C1	.0148	.0166	.1176

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.025 RH/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

Z/E = .0000 .3340 .5200 .6630 .8730

X/C	Z/E	ALPHA ( 2 )	RH/L	MACH
- .900	-.0930	-.1175	-.0974	-.0773
- .750	-.1555	-.2350	-.1545	-.2700
- .600	-.1798	-.1665	-.1294	-.0770
- .450	-.1254	-.2977	-.0576	-.1770
- .300	-.2055	-.4667	-.1491	-.0336
- .150	.159C	-.0668	-.3961	-.0477
.150	-.0112	-.2715	-.4028	-.2559
.300	-.2112	-.5251	-.4001	-.5203
.450	-.132C	-.3466	-.3527	-.456C
.600	-.1348	-.1867	-.2165	-.3553
.750	-.0735	-.0321	-.0462	-.2364
.900	-.1558	.C173	.0514	.0223



## CAST-2

(ADW11)

## 010CSF1 J40 WATE16 WING TOTAL SURFACE

MACH ( 1 ) = .165 ALPH A ( 3 ) = 4.965 RNL = 1.200 MACH = .165

## SECTION : 1) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .3200 .6630 .8730

X/C	.000	.0160	.0591	.1998	.0965	.1217
- .750	-.0210	-.1013	-.1850	-.2161	-.1435	
- .600	.0203	.0177	-.0157	.0549	-.1049	
- .450	-.2076	-.0814	.1406	.1398	.0045	
- .300	-.0170	-.1216	.0387	.2479	.0319	
- .150	.2713	.1833	-.1113	.2298	.3968	
.150	-.1394	-.4884	-.5111	-.8609	.4505	
.300	-.2547	-.5121	-.4675	-.6457	.6187	
.450	-.1795	-.4564	-.3179	-.4147	-.4630	
.600	-.0530	-.1926	-.2176	-.2563	.3362	
.750	-.0892	-.0669	-.1069	-.0345	-.1565	
.900	-.1658	-.0364	-.0355	.1101	-.0361	

MACH ( 1 ) = .165 ALPH A ( 4 ) = 9.975 RNL = 1.200 MACH = .165

## SECTION : 1) WING

## DEPENDENT VARIABLE CP

X/C	.000	.0214	.1709	.0590	.0610
- .750	.1051	.0163	-.1872	-.1339	-.1551
- .600	.1594	.1606	-.0101	.1464	-.0344
- .450	-.0306	.0691	.1507	.2418	.0708
- .300	.1620	.0401	.1861	.4006	.1810
- .150	.3504	.3125	.0220	.4558	.4766
.150	-.2068	-.7976	-.6559	-.1394	-.9610
.300	-.3246	-.5579	-.5833	-.8071	-.8435
.450	-.2123	-.4752	-.4349	-.5153	-.7119
.600	-.0817	-.2473	-.2394	-.3350	-.4984
.750	-.1115	-.1244	-.1461	-.0664	-.2972
.900	-.1673	-.1600	.0164	.1256	-.2125

MACH ( 1 ) = .165 ALPH A ( 5 ) = 14.960 RNL = 1.200 MACH = .165

## SECTION : 1) WING

## DEPENDENT VARIABLE CP

X/C	.000	.0616	.0807	.0958	.0694
- .750	.2037	.0248	-.0237	-.1161	-.1127
- .600	.2604	.2436	.0454	.1530	.0192
- .450	.0844	.1926	.0515	.2414	.1371
- .300	.2940	.1768	.1963	.4846	.2526

CA57-B B16C3F1 J40 WATEL WING TOTAL SURFACE

(RDWMA1)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.981

SECTION ( 1 ) WING

21/B .0000 .3340 .5200 .6630 .8730

DEPENDENT VARIABLE CP

X/C	.150	.4302	.4037	.1106	.6494	.4634
	.150	-.2610	-1.2203	-.7477	-1.3596	-2.5247
	.300	-.3176	-.7462	-.6524	-.9682	-2.1486
	.450	-.2351	-.9031	-.5435	-.5751	-1.5939
	.600	-.0980	-.3118	-.3307	-.3661	-1.0665
	.750	-.1288	-.1692	-.2140	-.1278	-7297
	.900	-.2124	-.0930	-.0256	-.0048	.5570

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.975 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C	.150	.2429	.0447	-.1940	-.1549	-.1325
	.150	.2610	.1156	-.1233	-.0965	-.1423
	.300	.3415	.2875	-.0998	.2051	.0437
	.450	.1950	.3067	-.0226	.2691	.1703
	.600	.4044	.3103	.1691	.4103	.2610
	.750	.5143	.4966	.4084	.7818	.4945
	.900	-.2921	-1.5318	-.7426	-.2985	-1.0581

DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

DEPENDENT VARIABLE CP

X/C	.150	.300	.450	.600	.750	.900
	-.2921	-.1202	-.2834	-.1651	-.2480	-.2862
	-.3148	-.7757	-.6985	-.5069	-.7565	-.1990
	-.7164	-.11523	-.1.1406	-.0590	-.5914	-.4748
	-.1.1523	-.0265	-.1.1406	-.9619	-.9050	-.9032
	-.0363	-.0265	-.0265	-.9619	-.9050	-.9032



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TABULATED SOURCE DATA - QAS378

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SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

(RDYH12)

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

SECTION (1) WING

MACH (1) = .165 ALPHA (3) = 4.980

DEPENDENT VARIABLE CP

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## TABULATED SOURCE DATA - CASTS

PAGE 199

MACH ( 1 ) = .165

ALPHA ( 5 ) = 15.010

(RDW12)

SECTION ( 1 ) WING

CASE-B B16CSF1 J40 W07516 WING TOTAL SURFACE

X/C

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

-.150	.3782	.2971	-.0404	.4332	.4761
.150	-.2558	-1.2066	-.7602	-1.2613	-1.6247
.300	-.3687	-.7413	-.6143	-.6985	-1.0725
.450	-.2260	-.5750	-.4940	-.5162	-.8095
.600	-.0682	-.3102	-.3030	-.3056	-.5710
.750	-.1133	-.1697	-.1679	-.0687	-.4245
.900	-.1677	.0192	.0212	.1236	-.3473

MACH ( 1 ) = .165

ALPHA ( 6 ) = 19.990

DEPENDENT VARIABLE CP

SECTION ( 1 ) WING

RDW12

21/8 .0000 .3340 .5200 .6630 .8730

X/C					
-.900	.1169	-.0132	-.2693	-.2697	-.3276
-.750	.1287	-.0006	-.1046	-.1614	-.1597
-.600	-.1715	.1772	-.0006	.2269	.0581
-.450	-.0405	.1453	-.1611	.3663	.2023
-.300	.2111	.1042	.0634	.5461	.3340
-.150	.4701	.4041	-.0579	.5601	.5242
.150	-.2763	-.1598	-.7633	-.2867	-.2258
.300	-.4413	-.8759	-.6421	-.1.2222	-.1.0613
.450	-.2676	-.7114	-.7107	-.1.0480	-.1.0112
.600	-.1258	-.4617	-.5637	-.9273	-.9877
.750	-.2026	-.2956	-.4566	-.3773	-.8625
.900	-.2324	-.0779	-.1630	-.2752	-.6570

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TABULATED SOURCE DATA - CASTB

PAGE 200

QA57-B (RDW#13) ( 12 NOV 73 )

QA57-B B16CSF1 JAD WATE10 WING TOTAL SURFACE

## REFERENCE DATA

BREF	=	4.4180 33.FT.	XMAP	=	43.5940 IN.	BETA	=	.0000
LREF	=	19.2900 IN.	YMAP	=	.0000 IN.	H/B	=	.2866
ZREF	=	37.6350 IN.	ZMAP	=	-.4050 IN.	ELEVON	=	.0000
SCALE	=	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.005 ANL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8750

X/C

-.900	-.11337	-.1161	-.0907	-.0151	-.1657
-.750	-.1634	-.2054	-.1736	-.2695	-.2924
-.600	-.2049	-.1889	-.1623	-.1476	-.2913
-.450	-.4581	-.4105	-.1461	-.1390	-.2311
-.300	-.4483	-.6507	-.2453	-.1383	-.1795
-.150	.0650	-.2415	-.4385	-.3235	-.0653
.150	-.0365	-.1281	-.3023	-.3205	.1307
.300	-.1331	-.2194	-.3256	-.3629	.3707
.450	-.0752	-.2722	-.2953	-.2726	.3792
.600	.0107	-.1434	-.1661	-.1740	-.2142
.750	-.0360	-.0331	-.0639	-.0728	-.0008
.900	-.1572	-.3725	-.1001	-.1306	.0570

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.020 ANL = 1.200 MACH = .165

## SECTION ( 2 ) WING

## DEPENDENT VARIABLE CP

-.900	-.0739	-.0837	-.0846	-.0367	-.1526
-.750	-.11103	-.1804	-.1326	-.2682	-.2687
-.600	-.1295	-.1088	-.0918	-.0753	-.2220
-.450	-.3664	-.3064	-.0432	-.0293	-.1500
-.300	-.3034	-.4927	-.1449	-.0066	-.0456
-.150	.1292	-.0819	-.2646	-.0846	.1846
.150	-.0846	-.2436	-.4557	-.5717	.0802
.300	-.1944	-.3088	-.4215	-.5082	-.4669
.450	-.1258	-.3236	-.3478	-.3400	-.4327
.600	-.0195	-.1649	-.1941	-.2036	-.2471
.750	-.0643	.0053	-.0716	.0427	-.0314
.900	-.1502	.0654	.0930	.1250	.0399



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## TABULATED SOURCE DATA - CASTS

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MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.995 RNL = 1.200 MACH = .165

(ADW13)

SECTION ( 1 ) WING  
DEPENDENT VARIABLE CP

X/C						
-.900	-.0274	-.0636	-.1467	-.0312	-.1657	
-.750	-.0460	-.1316	-.1933	-.2003	-.2269	
-.600	-.0363	-.0234	-.0618	.0324	-.1490	
-.450	-.2540	-.1775	.0767	.1190	.0282	
-.300	-.1374	-.2697	.0217	.2034	.0851	
-.150	-.2086	-.0952	-.1631	.1789	.3295	
.150	-.1453	-.4148	-.5911	-.8585	-.4154	
.300	-.2558	-.4400	-.4924	-.6423	-.6277	
.450	-.1802	-.3948	-.3663	-.3973	-.6433	
.600	-.0372	-.2250	-.2100	-.2349	-.2860	
.750	-.0897	-.0479	-.0852	.0039	-.1180	
.900	-.1628	.0414	.0008	.1025	-.0082	

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.985 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING  
DEPENDENT VARIABLE CP

X/C						
-.900	.0149	-.0329	-.1310	-.0519	-.1943	
-.750	.0184	-.0592	-.2182	-.1238	-.1943	
-.600	.0583	.0690	-.1080	.1237	.0818	
-.450	-.1565	-.0621	.0764	.22 <sup>c</sup>	.053 <sup>c</sup>	
-.300	.0147	-.1316	.1796	.35 <sup>c</sup>	.1725	
-.150	.2911	-.2234	-.0480	.3887	.4281	
.150	-.2003	-.7305	-.6566	-.11201	-.8866	
.300	-.3205	-.5353	-.5613	-.7730	-.8017	
.450	-.2210	-.4829	-.4463	-.4903	-.5757	
.600	-.0765	-.2526	-.2708	-.3184	-.4539	
.750	-.1145	-.0734	-.1257	-.0245	-.2499	
.900	-.1907	.0126	.0159	.0909	-.1406	

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.970 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING  
DEPENDENT VARIABLE CP

X/C						
-.900	.0607	-.0154	-.0787	-.744	-.2627	
-.750	.0745	-.0276	-.0491	-.75	-.1521	
-.600	.1379	.1372	.3375	.4	-.1552	
-.450	-.0609	.0522	.25 <sup>c</sup>	.2	.146	
-.300	.1471	.019.	.177 <sup>c</sup>		-.140	

(ADW13)

CAST-B B18C5F1 J40 WTE18 WING TOTAL SURFACE

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.970

SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .6730

DEPENDENT VARIABLE CP

X/C					
-.150	.3711	.3151	.0316	.5767	.4566
-.150	-.2546	-.1204	-.1784	-.1,3105	-.1,6397
.300	-.3693	-.17907	-.5989	-.5050	-.1,1666
.450	-.2278	-.6053	-.5069	-.5433	-.6871
.600	-.0860	-.3159	-.3186	-.3343	-.6770
.750	-.1216	-.1731	-.1976	.0162	-.5418
.900	-.2052	-.0021	-.0033	.0723	-.4280

MACH ( 1 ) = .165 ALPHA ( 6 ) = 20.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .6730

DEPENDENT VARIABLE CP

X/C					
-.900	.0734	-.0346	-.1587	-.1830	-.3693
-.750	.1119	-.0176	-.0504	-.1270	-.1827
-.600	.2112	.1626	.0773	.1655	.0165
-.450	.0259	.1489	-.1013	.2643	.1615
-.300	.2731	.1485	.0807	.4802	.2927
-.150	.4639	.4141	.1081	.7392	.4915
.150	-.2814	-.1,4765	-.7656	-.1,1829	-.1,0762
.300	-.4189	-.8718	-.6386	-.1,1819	-.9944
.450	-.2726	-.7326	-.7344	-.1,1193	-.9978
.600	-.1324	-.4999	-.6594	-.1,0211	-.9915
.750	-.2175	-.2950	-.3210	-.4859	-.8913
.900	-.2719	-.0936	-.2168	-.3773	-.6782

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TABULATED SOURCE DATA - OA 57B

PAGE 203

CA57-B B16C5F1 J40 WATE16 WING TOTAL SURFACE (RCWMA14) ( 12 NOV 73 )

## REFERENCE DATA

%OFF	%	4.0120N 92.5°	YHAP	=	43.5940 IN.	BETA	=	.000	PIN/P = 1.300
LREF	=	19.2300 IN.	THAP	=	.0000 IN.	M/B	=	.286	DDFLAP = .000
BREF	=	37.9350 IN.	ZHAP	=	-.4050 IN.	ELEVON	=	.000	
SCALE	=	.0403							

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (1) = -4.010 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

SECTION (1)WING

DEPENDENT VARIABLE CP

T+18 .0000 .3340 .5200 .6630 .8730

Y/C

-.900	-.0374	-.1060	-.0616	-.0434	-.1580
-.750	-.1631	-.2459	-.2090	-.3045	-.2978
-.600	-.2780	-.2256	-.1938	-.1838	-.2862
-.450	-.5655	-.4933	-.2156	-.1655	-.2530
-.300	-.5561	-.7227	-.3352	-.1763	-.2046
-.150	.0558	-.2831	-.5347	-.3836	-.0924
.150	-.0387	-.1229	-.2859	-.2961	-.1507
.300	-.1449	-.2078	-.3162	-.3552	-.3144
.450	-.0766	-.2552	-.2808	-.2578	-.0680
.600	-.0202	-.1305	-.1561	-.1620	-.2736
.750	-.0360	.0820	-.0587	.0817	-.0737
.900	-.1497	.0958	.1257	.1468	.0374

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (2) = .005 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

SECTION (2)WING

DEPENDENT VARIABLE CP

T+18 .0000 .3340 .5200 .6630 .8730

Y/C

-.900	-.0279	-.0845	-.0528	-.0560	-.1408
-.750	-.1093	-.2060	-.1616	-.2664	-.2521
-.600	-.1934	-.1285	-.1121	-.0935	-.2223
-.450	-.4388	-.3881	-.0932	-.0458	-.1573
-.300	-.3334	-.5461	-.1781	-.0109	-.0507
-.150	.1249	-.1085	-.3566	-.1282	.1699
.150	-.0795	-.2335	-.4421	-.5461	-.0477
.300	-.1910	-.2963	-.4101	-.4904	-.4178
.450	-.1191	-.3108	-.3371	-.3304	-.1114
.600	-.0135	-.1521	-.1797	-.1932	-.3175
.750	-.0567	.0470	-.0579	.0553	-.5632
.900	-.1440	.0854	.1151	.1352	-.143

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TABULATED SOURCE DATA - CASTB

PAGE 204

CA57-B B16C5F1 J40 WATE10 WING TOTAL SURFACE

MACH ( 1 ) = .165	ALPHA ( 3 ) = 4.980	R/N/L = 1.200	MACH = .165
SECTION ( 1 ) WING			

21/8 .0000 .3340 .5200 .6630 .8730					
X/C					

.900 .0166 -.0711 -.1606 -.0975 -.1546					
-.750 -.0479 -.1435 -.2191 -.2283 -.2343					
-.600 -.0825 -.0297 -.1051 -.0095 -.1514					
-.450 -.3283 -.2133 .0460 .0844 .0330					
DEPENDENT VARIABLE CP					
.300 -.2071 -.3193 -.0003 .1759 .0613					
-.150 -.2075 .0749 -.1673 .1412 .3182					
-.1416 -.4103 -.5741 -.6404 -.3647					
.300 -.2616 -.4376 -.4811 -.6305 -.5511					
.450 -.1742 -.3933 -.3672 -.3906 -.2548					
.600 -.0532 -.2137 -.2021 -.2354 -.3712					
.750 -.0912 -.0266 -.0857 .0083 -.1291					
.900 -.1606 .0424 .0072 .0977 -.0315					

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.975 R/N/L = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730					
X/C					

.900 .0700 -.0203 -.1684 -.1232 -.1916
-.750 .0280 -.0652 -.2455 -.1774 -.1820
-.600 .0167 .0643 -.0667 .0557 -.0713
-.450 -.1970 -.0663 .0907 .1946 .0593
-.300 -.0364 -.1544 .1680 .3184 .1843
-.150 .2975 .2113 -.0482 .3413 .4329
.150 -.2007 -.7213 -.6704 -.0996 -.8295
.300 -.3106 -.5407 -.5641 -.7738 -.7315
.450 -.2000 -.4580 -.4406 -.4939 -.5447
.600 -.0758 -.2414 -.2633 -.3131 -.4718
.750 -.1030 -.0289 -.1312 -.0280 -.2356
.900 -.1701 .0425 .0176 .0609 -.1552

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.970 R/N/L = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730					
X/C					

.900 .0963 .0042 -.1443 -.1354 -.2380
-.750 .0862 -.0114 -.1970 -.1438 -.1582
-.600 .1151 .1431 -.0401 .1651 .0096
-.450 -.1079 .0351 .1209 .3124 .1434
-.300 .1020 -.0062 .2146 .4383 .2593



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TABULATED SOURCE DATA - CASTB

PAGE 205

SECTION 11 WING

CA57-B 816C-1 JAO WATE10 WING TOTAL SURFACE

(DDVM14)

MACH ( 1 ) = .165

ALPHA ( 5 ) = 14.970

SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
-.150	.3652	.3166	.0369	.4590	.4732
.150	-.2404	-.1624	-.7644	-.12680	-.6156
.300	-.3711	-.7199	-.6043	-.8683	-.1298
.450	-.2300	-.5913	-.4932	-.5216	-.8698
.600	-.0721	-.2934	-.3059	-.3171	-.6679
.750	-.1174	-.1236	-.1963	-.0527	-.5284
.900	-.2069	-.0032	-.0022	.0951	-.3999

MACH ( 1 ) = .165

ALPHA ( 6 ) = 20.010

RN/L = 1.200 MACH = .165

SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
-.150	.1116	-.0504	-.2148	-.3663	-.3415
-.750	.1351	-.0005	-.1994	-.2353	-.1637
-.600	.1911	.1617	.0064	.1527	.0509
-.450	.0042	.1559	-.1166	.3216	.1996
-.300	.2411	.1416	.1073	.5224	.3268
-.150	.4687	.4123	.0464	.6891	.5200
.150	-.2855	-.15257	-.7630	-.13085	-.1810
.300	-.4219	-.8124	-.6634	-.11900	-.0736
.450	-.2690	-.7291	-.7268	-.1.0633	-.0795
.600	-.1338	-.4107	-.6020	-.9602	-.9992
.750	-.2746	-.2379	-.4963	-.4095	-.8814
.900	-.2706	-.0855	-.2175	-.3235	-.6411

## CMA57-B B16C5F1 J40 W07E10 WING TOTAL SURFACE

(RDW15) (12 NOV 73)

## REFERENCE DATA

BREF = 4.4120 83. FT. XMPP = 47.5940 IN.  
 LREF = 19.2300 IN. YMPP = .0000 IN.  
 BREF = 37.9390 IN. ZMPP = -.4050 IN.  
 SCALE = .0405

MACH (1) = .165 ALPHA (1) = -3.965 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
-.900	-.1087	-.1123	-.0822	-.0060	-.1495
-.750	-.1634	-.2057	-.1777	-.2831	-.2877
-.600	-.2105	-.1930	-.1668	-.1409	-.2886
-.450	-.4517	-.4084	-.1590	-.1292	-.2393
-.300	-.4491	-.6511	-.2335	-.1393	-.1796
-.150	.0592	-.2411	-.4397	-.3192	-.0534
.150	-.0462	-.1239	-.2959	-.3100	-.1544
.300	-.1546	-.2199	-.3023	-.3634	-.3144
.450	-.0787	-.2618	-.2513	.2618	-.0713
.600	.0113	-.1375	-.1619	-.1700	-.3645
.750	-.0349	.0323	-.0005	.0817	-.0691
.900	-.1508	.0786	.0986	.1296	.0210

MACH (1) = .165 ALPHA (2) = .015 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
-.900	-.0760	-.0863	-.0521	-.0371	-.1423
-.750	-.1063	-.1727	-.1271	-.2562	-.2679
-.600	-.1301	-.1072	-.0939	-.0748	-.2217
-.450	-.3695	-.3104	-.0327	-.0243	-.1436
-.300	-.2981	-.4675	-.1281	.0092	-.0436
-.150	-.1267	-.0766	-.2646	-.0886	.1930
.150	-.0654	-.2461	-.4574	-.5681	-.6842
.300	-.1990	-.3027	-.3901	-.5077	-.4193
.450	-.1297	-.3204	-.3446	-.3339	-.1391
.600	-.0210	-.1657	-.1920	-.2082	-.3514
.750	-.0585	.0254	-.0162	.0412	-.0793
.900	-.1541	.0676	.0840	.1209	.0044



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## TABULATED SOURCE DATA - GA57B

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MACH ( 1 ) = .165    ALPHA ( 3 ) = 5.000    RNL = 1.200    MACH = .165  
 SECTION ( 1 ) WING    DEPENDENT VARIABLE CP

(RDW15)

X/C	.0000	.0220	.0420	.0630	.0730
-.900	-.0227	-.0622	-.1475	-.0321	-.1516
-.750	-.0441	-.1228	-.1962	-.1913	-.2279
-.600	-.0531	-.0171	-.0851	.0420	-.1441
-.450	-.2473	-.1793	.0871	.1271	-.0194
-.300	-.1362	-.2735	.0412	.2120	.0913
-.150	-.2122	.0995	-.1121	.1837	.3380
.150	-.1390	-.4015	-.5623	.8356	-.3801
.300	-.2596	-.4237	-.4659	.6333	-.5275
.450	-.1676	-.3796	-.3576	-.3866	-.2402
.600	-.0508	-.2070	-.1970	-.2323	-.4684
.750	-.0814	-.0259	-.0608	.0047	-.1352
.900	-.1547	.0512	.0055	.1035	-.0404

MACH ( 1 ) = .165    ALPHA ( 4 ) = 10.015    RNL = 1.200    MACH = .165  
 SECTION ( 1 ) WING    DEPENDENT VARIABLE CP

X/C	.0000	.03340	.05200	.0630	.0730
-.900	.0241	-.0241	-.1562	-.0326	-.1912
-.750	.0239	-.0395	-.2277	-.1110	-.1938
-.600	-.0593	.0163	-.0922	-.1347	-.0756
-.450	-.1500	-.0558	.1005	.2336	.0586
-.300	.0177	-.1304	-.2066	.3624	.1796
-.150	.2941	.2298	-.0172	-.3924	.4446
.150	-.1997	-.7194	-.6453	-.1001	-.8642
.300	-.3225	-.5254	-.5375	-.7656	-.7242
.450	-.2080	-.4720	-.4325	-.4915	-.5492
.600	-.0732	-.2396	-.2569	-.3176	-.5429
.750	-.1012	-.0443	-.1232	-.0093	-.2914
.900	-.1787	.0419	.0160	.1074	-.1804

X/C	.0000	.0340	.05200	.0630	.0730
-.900	.0651	-.0599	-.0402	-.0402	-.2642
-.750	.0653	-.2177	-.2192	-.1112	-.1741
-.600	-.1409	.1494	-.0229	.71	-.0162
-.450	-.0764	-.0774	-.0281	-.11	-.11
-.300	-.1076	-.1074	-.0281	-.11	-.11
-.150	-.0764	-.0774	-.0281	-.11	-.11
.150	-.0764	-.0774	-.0281	-.11	-.11
.300	-.1076	-.1074	-.0281	-.11	-.11
.450	-.0764	-.0774	-.0281	-.11	-.11
.600	-.1076	-.1074	-.0281	-.11	-.11
.750	-.0764	-.0774	-.0281	-.11	-.11
.900	-.1076	-.1074	-.0281	-.11	-.11

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TABULATED SOURCE DATA - CAST 8

PAGE 208

CAST 8 B16C5F1 JAO WING TOTAL SURFACE

(ADYMA1)

MACH (1) = .165

ALPHA (5) = 14.973

SECTION (1) WING

2/18 .0000 .3340 .3200 .6630 .6730

DEPENDENT VARIABLE CP

X/C	.3767	.3213	.0629	.5836	.4711
-.190	-.2493	-1.1611	-.7705	-1.2949	-1.7249
-.150	-.3797	-.7298	-.5799	-.8978	-.1514
.300	-.2224	-.9985	-.4698	-.5369	-.9150
.450	-.0804	-.9037	-.5104	-.3287	-.8085
.600	-.1148	-.1353	-.1994	.0359	.6070
.750	-.2019	-.0002	-.0033	.0830	-.4341
.900					

MACH (1) = .165 ALPHA (6) = -.0000 RNL = 1.200 MACH = .165

SECTION (1) WING

2/18 .0000 .3340 .3200 .6630 .6730

DEPENDENT VARIABLE CP

X/C	.0766	.0421	-.1715	-.1659	-.3126
-.900	-.1134	-.0111	-.0550	-.1112	-.2009
-.750	.2135	.1732	.0722	.1720	.0163
-.600					
-.450	.0320	.1422	-.1158	.2751	.1601
-.300	.2726	.1396	.1028	.4981	.2544
-.150	.4655	.4144	.1067	.7577	.4975
.150	-.2718	-.1497	-.7342	-.1141	-.0323
.300	-.4246	-.8623	-.6176	-.1538	-.0013
.450	-.2587	-.1485	-.7353	-.1116	-.0514
.600	-.1303	-.4661	-.6556	-.0045	-.9954
.750	-.2096	-.2652	-.5057	-.5033	-.8934
.900	-.2662	-.0894	-.2462	-.4169	-.6657



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TABULATED SOURCE DATA - CASTE

PAGE 209

CA97-E B100SF1 J40 WATE10 WING TOTAL SURFACE

## REFERENCE DATA

SREF	R	4.4120 84 FT.	XWEP	Z	43.5940 IN.	BETA	=	.000	PIM/P	=	1.500
REF	-	4.2200 IN.	XWEP	Z	.0000 IN.	H/B	=	.286	BOFLAP	=	-16.000
SREF	R	37.9350 IN.	ZWEP	Z	-.4050 IN.	ELEVON	=	.000			
SCALE	Z	.0403									

MACH (1) = .165 ALPHA (1) = -3.985 AN/L = 1.200 MACH = .165

## SECTION (1) WING

21/8 0000 .3340 .3200 .6630 .8730

## X/C

-.900	-.1626	-.1682	-.1154	-.0780	-.1356
-.750	-.2301	-.3179	-.2700	-.3279	-.2866
-.600	-.3445	-.2337	-.2333	-.2281	-.3131
-.450	-.5765	-.6142	-.5135	-.2136	-.2844
-.300	-.5861	-.5184	-.4504	-.2366	-.2293
-.150	.0312	-.3524	-.6657	-.4790	-.1471
.150	-.0435	-.1134	-.2780	-.2873	-.1434
.300	-.1659	-.2077	-.2814	-.2615	-.3439
.450	-.0835	-.2567	-.2612	-.2539	-.3032
.500	.0165	-.1316	-.1525	-.1615	-.2156
.750	-.0345	.0127	-.0179	.0525	-.0058
LOC	-.1433	.1027	.1686	.1644	.0596

MACH (1) = .165 ALPHA (2) = .015 AN/L = 1.200 MACH = .165

## SECTION (1) WING

21/8 0000 .3340 .3200 .6630 .8730

## X/C

-.900	-.1603	-.1328	-.1043	-.0715	-.1241
-.750	-.1892	-.2687	-.2206	-.2912	-.2556
-.600	-.2497	-.1416	-.1333	-.1326	-.2446
-.450	-.5546	-.4866	-.1698	-.0931	-.1673
-.300	-.4230	-.1037	-.3310	-.0645	-.0846
-.150	-.1057	-.1616	-.5162	-.2151	-.1125
.150	-.0804	-.2312	-.4421	-.5120	-.0525
.300	-.2177	-.2692	-.3703	-.4250	-.4867
.450	-.3114	-.57	-.3456	-.3283	-.3796
.600	-.0193	-.1596	-.1851	-.2057	-.2653
.750	-.0600	.0289	-.0463	.0276	-.0504
LOC	-.1436	.0667	.1453	.1276	.0391

## PARAMETRIC DATA

BETA	=	.000
H/B	=	.286
ELEVON	=	.000

## DEPENDENT VARIABLE CP

## SECTION (1) WING

## X/C

MACH ( 1 ) = .165 SECTION ( 1 )WING

(RDW16)

CA57-B B16CAF1 J40 WTE10 WING OTAL SURFACE

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.1216	-.1135	-.2247	-.0986	-.1599
-.750	-.1100	-.2035	-.2632	-.2490	-.2183
-.600	-.1461	-.0302	-.1367	-.0320	-.1638
-.450	-.7998	-.2867	-.0178	.0494	-.0659
-.300	-.2326	-.4354	-.1076	.1234	.0551
-.150	.1901	.0214	-.2755	.0707	.3075
.150	-.1463	-.4080	-.5863	-.8516	-.3666
.300	-.2743	-.4198	-.4635	-.5808	-.6426
.450	-.1728	-.3802	-.3651	-.3997	-.4975
.600	-.0590	-.2255	-.2185	-.2425	-.3177
.750	-.0883	-.0378	-.0794	-.0124	-.1217
.900	-.1431	.0468	.0321	.0820	-.0354

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10,000 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

21/8 .0630 .3340 .5200 .6630 .8730

X/C

-.900	-.0858	-.0152	-.2553	-.0035	-.1844
-.750	-.0475	-.1346	-.2940	-.1709	-.1753
-.600	-.0421	.0486	-.1384	.0812	-.1020
-.450	-.2736	-.1436	.0424	.1668	.0527
-.300	-.0838	-.2536	.0829	.2716	.1587
-.150	.2714	.1734	-.1329	.2857	.4229
.150	-.2174	-.7400	-.6416	-.1098	-.8117
.300	-.3544	-.5329	-.5417	-.7129	-.8020
.450	-.2120	-.4713	-.4475	-.4831	-.6208
.600	-.0904	-.2571	-.2397	-.3132	-.4017
.750	-.1142	-.0370	-.1171	-.0464	-.2415
.900	-.1593	.0230	.0258	.0636	-.1596

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15,010 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0349	-.0245	-.2270	-.1190	-.2209
-.750	.0328	-.0482	-.2526	-.1235	-.1286
-.600	.0685	.1428	-.0683	.1725	-.0204
-.450	-.1544	-.0124	.0829	.2975	.1266
-.300	.0730	-.0747	.1699	.4000	.2513



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TAPPLATED SOURCE DATA - Q3378

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SECTION ( 1 )WING

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.010

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	2/B	.0000	.3340	.5200	.6630	.8730
-.150	.3712	.2891	-.0241	.4204	.4644	
.150	-.2484	-1.1475	-.7657	-1.2753	-1.5443	
.300	-.3952	-.6939	-.5823	-.8184	-1.0738	
.450	-.2249	-.5771	-.4893	-.5332	-.8895	
.600	-.0814	-.2967	-.3021	-.3280	-.6463	
.750	-.0993	-.1528	-.1714	.0280	-.4653	
.900	-.1633	.0161	.0200	.0862	-.3744	

MACH ( 1 ) = .165 ALPHA ( 6 ) = 20.000 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C	2/B	.0000	.3340	.5200	.6630	.8730
-.900	-.0318	-.0460	-.12458	-.2636	-.2894	
-.750	.0834	-.2283	-.1284	-.1349	-.1076	
-.600	.1587	.1955	-.0268	.2330	.0592	
-.450	-.0530	.1271	-.1326	.3825	.2001	
-.300	.2061	.0740	.1306	.5401	.334	
-.150	.4617	.3894	-.0871	.5838	.5293	
.150	-.2816	-.5124	-.7865	-.1.3722	-.1.3060	
.300	-.4499	-.8675	-.6365	-.1.2263	-.1.2252	
.450	-.2830	-.7036	-.6670	-.9954	-.1.1146	
.600	-.1290	-.4426	-.5590	-.8575	-.1.0790	
.750	-.1734	-.2473	-.4364	-.3167	-.9313	
.900	-.2249	-.0820	-.1493	-.2529	-.8703	

(RDYN16)

(RDYN16)

## REFERENCE DATA

SREF =	4.5120	IN.	XHAP =	43.5940	IN.
LREF =	19.2500	IN.	YHAP =	.0000	IN.
BREF =	37.9350	IN.	ZHAP =	-.4050	IN.
SCALE =	.0405				

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (1) = -3.975 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

## X/C

-.900	-.1439	-.1256	-.0784	-.0463	-.1321
-.750	-.2095	-.2740	-.2282	-.2953	.2824
-.600	-.2912	-.2153	-.2321	-.1917	-.2970
-.450	-.5733	-.5209	-.2769	-.1823	-.2645
-.300	-.5521	-.7902	-.3067	-.2004	-.2135
-.150	.0446	-.73035	-.5865	-.4054	-.0992
.150	-.0420	-.1184	-.2846	-.3017	.1425
.300	-.1706	-.2086	-.2900	-.2974	-.3534
.450	-.0637	-.2580	-.2870	-.2599	-.3011
.600	.0136	-.1332	-.1613	-.1775	-.2426
.750	-.0394	.0770	-.0820	.0477	-.0113
.900	-.1485	.0844	.1174	.1247	.0538

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (2) = .005 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730					
X/C					
-.900	-.1280	-.1086	-.0639	-.0635	-.1242
-.750	-.1615	-.2352	-.1776	-.2607	-.2452
-.600	-.2093	-.1025	-.1342	-.1052	-.2330
-.450	-.4700	-.4170	-.1450	-.0621	-.1706
-.300	-.3900	-.6060	-.2117	-.0340	-.0655
-.150	.1131	-.1297	-.3639	-.1558	.1544
.150	-.0857	-.2376	-.4419	-.5511	-.0475
.300	-.2196	-.5006	-.3633	-.4317	-.4775
.450	-.1246	-.3133	-.3406	-.3326	-.3595
.600	-.0259	-.1611	-.1655	-.2104	-.2850
.750	-.0655	.0193	-.0758	.0213	-.0491
.900	-.1408	.0805	.1093	.1165	.0360

## PARAMETRIC DATA

BETA =	.000	PIN/P =	1.300
H/B =	.286	BDFLAP =	-.18.000
ELEVON =	.000		

(RDYH17) (112 NOV 73)





CA57-B B16C5F1 JAO WSTE16 WING TOTAL SURFACE (RDYN16) ( 12 NOV 73 )

## REFERENCE DATA

SREF	4.4120	SA.FT.	XMAP	=	43.5940 IN.	BETA	=	.600	PTN/P	=	1.000
LREF	19.2500	IN.	1MAP	=	.0000 IN.	H/B	=	.286	BOFLAP	=	-10.000
BALY	37.8350	IN.	2MAP	=	-.4050 IN.	ELEVON	=	.000			
SCALE	=	.0005									
MACH ( 1 ) =	.165		ALPHA ( 1 ) =	=	.3.990	RNL	=	1.200	MACH	=	.165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C											
-.900	-.1655		-.1208	=	-.1004	=	.0242	=	-.1322		
-.750	-.1925		-.2163	=	-.1795	=	.2662	=	.2699		
-.600	-.2213		-.1457	=	-.1688	=	-.1478	=	.2805		
-.450	-.4612		-.4202	=	-.2049	=	-.1484	=	.2513		
-.300	-.4640		-.6890	=	-.2020	=	-.1643	=	.1838		
-.150	.0606		-.2451	=	-.4879	=	-.3402	=	.0377		
.150	-.0363		-.1243	=	-.2986	=	-.3310	=	.1386		
.300	-.1779		-.2073	=	-.3049	=	-.2906	=	.3672		
.450	-.0919		-.2673	=	-.3042	=	-.2806	=	.2796		
.600	.0121		-.1499	=	-.1774	=	-.1956	=	.2468		
.750	-.0452		.0113	=	-.0900	=	.0477	=	.0207		
.900	-.1501		.0520	=	.0841	=	.10C9	=	.0424		

MACH ( 1 ) = .165 ALPHA ( 2 ) = .035 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C											
-.900	-.1467		-.1015	=	-.0780	=	-.0603	=	-.1490		
-.750	-.1534		-.2202	=	-.1520	=	-.2596	=	.2608		
-.600	-.1577		-.0680	=	-.1173	=	-.0897	=	.2385		
-.450	-.3886		-.3290	=	-.1012	=	-.0496	=	.1729		
-.300	-.3269		-.5238	=	-.1468	=	-.0188	=	.0689		
-.150	.1137		-.0985	=	-.2999	=	-.1191	=	.1925		
.150	-.0932		-.2546	=	-.4642	=	-.5795	=	.0665		
.300	-.2374		-.3087	=	-.4127	=	-.4114	=	.0386		
.450	-.1399		-.3224	=	-.3552	=	-.3524	=	.3609		
.600	-.0340		-.1894	=	-.2192	=	-.2157	=	.3776		
.750	-.0871		-.0417	=	-.1171	=	-.1121	=	.1724		
.900	-.1535		C449	=	.2711	=	-.1419	=	.1419		

## SECTION ( 1 ) WING

MACH ( 1 ) = .165    ALPHA ( 3 ) = 3,000    RNL = 1,200    MACH = .165

## DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	- .1093	- .0896	- .1609	- .0684	- .1706	
- .750	- .0886	- .1721	- .2058	- .1918	- .2076	
- .600	- .0619	.0233	- .1058	.0213	- .1687	
- .450	- .2758	- .1919	.0901	.1121	-.0291	
- .300	- .1590	- .3160	.0148	.1974	.0745	
- .150	.2021	.0817	- .1676	.1593	.3655	
- .150	- .1472	- .4059	- .5968	-.8551	.3887	
.300	- .2983	- .4223	- .1793	-.5482	-.6455	
.450	- .1765	- .3847	- .3738	-.3989	-.4532	
.600	- .0642	- .2228	- .2217	-.2509	-.3599	
.750	- .1040	- .0764	- .0966	-.0052	-.1346	
.900	- .1546	.0162	-.0083	.0785	-.0350	

MACH ( 1 ) = .165    ALPHA ( 4 ) = 10,000    RNL = 1,200    MACH = .165

## DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	- .0609	- .0557	- .1734	-.0548	- .1954	
- .750	- .0154	-.0939	- .2299	-.1121	-.1793	
- .600	.0354	.1102	-.0986	.1201	-.1080	
- .450	- .1711	-.0743	.1041	.2270	.0499	
- .300	.0026	-.1669	.1776	.3511	.1680	
- .150	.2877	.2119	-.0703	.3637	.4652	
.150	-.2070	-.7222	-.6521	-.11137	-.8640	
.300	-.3586	-.5202	-.5614	-.6964	-.8080	
.450	-.2092	-.4816	-.4373	-.4745	-.5976	
.600	-.0661	-.2613	-.2779	-.3207	-.4716	
.750	-.1168	-.0992	-.1421	-.0525	-.2885	
.900	-.1698	-.0643	.0022	.0697	-.2081	

MACH ( 1 ) = .165    ALPHA ( 5 ) = 14,995    RNL = 1,200    MACH = .165

## DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	-.0191	-.0480	-.0678	-.0863	-.2335	
- .750	.0492	-.0456	-.0513	-.0831	-.1416	
- .600	.1292	.1970	.0153	.1669	-.0273	
- .450	-.0682	.0439	-.0350	.2719	.1167	
- .300	-.1449	-.0590	.1764	.4681	.2395	

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TABULATED SOURCE DATA - CANTAB

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CANTAB SOURCE DATA - CANTAB

(RDW16)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.995

## SECTION ( 1 ) WING

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.995 RNL = 1.200 MACH = .165

2/1/8 .0000 .3340 .5200 .6630 .6730

X/C

-.150	.3760	.3144	.0699	.5736	.4796
-.150	-.2376	-.11592	-.7565	-.1.2911	-.1.7739
.300	-.14097	-.1080	-.6102	-.6337	-.1.2393
.450	-.2281	-.6126	-.4726	-.5434	.9813
.600	-.0751	-.3007	-.3142	-.3320	-.7284
.750	-.1109	-.1.484	-.2094	-.0036	-.5578
.900	-.1771	-.0074	-.0066	.0429	-.4356

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.995 RNL = 1.200 MACH = .165

2/1/8 .0000 .3340 .5200 .6630 .6730

X/C

-.900	-.0125	-.0690	-.1.635	-.1.690	-.3126
-.750	.0739	-.0213	-.0349	-.0920	-.1.338
-.600	.1935	.2134	.0853	.1.774	.0130
-.450	.0191	.1.436	-.0891	.2763	.1644
-.300	.2614	.1.260	.1100	.5052	.2936
-.150	.4641	.4003	.0716	.7328	.5036
.150	-.2791	-.1.4504	-.7401	-.1.2015	-.1.1949
.300	-.4645	-.6485	-.6417	-.1.3332	-.1.1209
.450	-.2774	-.7379	-.6899	-.1.1706	-.1.0121
.600	-.1365	-.4808	-.5993	-.1.0371	-.1.0897
.750	-.2017	-.2901	-.4460	-.4166	-.9384
.900	-.2332	-.0787	-.2094	-.3297	-.7268

CA57-B B16CSF1 J40 WATE10 WING TOTAL SURFACE (RDW19) (12 NOV 73)

## REFERENCE DATA

BREF =	4.4120 SQ FT.	XMAP =	43.5940 IN.	BETA =	.000	PTM/P =	1.500
LREF =	19.2500 IN.	YMAP =	.0000 IN.	M/B =	.286	BDFLAP =	20.000
BREF =	37.9350 IN.	ZMAP =	-.4050 IN.	ELEVON =	.000		
SCALE =	.0405						

## PARAMETRIC DATA

SECTION ( 1 ) WING							
DEPENDENT VARIABLE CP							
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	-3.985	RNL =	1.200	MACH =	.165
21/8	.0000	.3340	.5200	.6630	.8730		
X/C							
-.900	.1309	-.0410	-.0874	-.0694	-.1584		
-.750	-.1115	-.1645	-.2304	-.3219	-.3056		
-.600	-.2922	-.1479	-.2211	-.1993	-.2923		
-.450	-.6007	-.5157	-.3179	-.1676	-.2665		
-.300	-.5290	-.8449	-.3498	-.2140	-.2102		
-.150	.0495	-.3432	-.6358	-.4370	-.0566		
.150	-.0463	-.1228	-.2860	-.3072	-.1532		
.300	-.1625	-.1952	-.3013	-.2433	-.3515		
.450	-.0977	-.2545	-.2955	-.2653	-.2633		
.600	.0013	-.1468	-.1619	-.1636	-.2018		
.750	-.0503	.0212	-.0534	.0735	.0017		
.900	-.2164	.0523	.1488	.1591	.0658		
MACH ( 1 ) =	.165	ALPHA ( 2 ) =	-.010	RNL =	1.200	MACH =	.165
21/8	.0000	.3340	.5200	.6630	.8730		
X/C							
-.900	.1441	-.0325	-.0906	-.0776	-.1604		
-.750	-.0750	-.1628	-.1982	-.2926	-.2730		
-.600	-.2104	-.0537	-.1290	-.1086	-.2355		
-.450	-.4939	-.3967	-.1827	-.0693	-.1845		
-.300	-.3673	-.6704	-.2397	-.0505	-.0707		
-.150	.1093	-.1625	-.4441	-.1606	.1939		
.150	-.0976	-.2589	-.4571	-.5653	-.0629		
.300	-.2316	-.3091	-.4110	-.3887	-.4633		
.450	-.1465	-.3224	-.3638	-.3509	-.3563		
.600	-.0491	-.1946	-.2059	-.2109	-.2553		
.750	-.1048	-.0499	-.0735	.0403	-.0395		
.900	-.2274	.0614	.1201	.1398	.0381		



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TABULATED SOURCE DATA - CASTB

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## CART-8 B16CSF1 J40 WATE10 WING TOTAL SURFACE (INFOV10)

MACH (1) = .165 ALPHA (3) = 4.975 RNL = 1.200 MACH = .165

## SECTION (1) WING

21/8 .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.2109	-.0080	-.094	-.1030	-.1735
-.750	.0110	-.1035	-.2282	-.2320	-.2133
-.600	.0612	-.0939	.0093	-.1503	
-.450	-.3318	-.1941	.0560	.0812	-.0352
-.300	-.1941	-.3878	-.0338	.1636	.0818
-.150	-.2080	.0488	-.2194	.1168	.3710
.150	-.1492	-.3989	-.5665	-.8262	-.3659
.300	-.2635	-.4207	-.4756	-.4675	-.6278
.450	-.1715	-.3610	-.3735	-.3828	-.4527
.600	-.0735	-.2222	-.2069	-.2204	-.3024
.750	-.1226	-.0993	-.0840	.0217	-.0942
.900	-.2216	.0362	.0175	.1080	-.0059

MACH (1) = .165 ALPHA (4) = 9.975 RNL = 1.200 MACH = .165

## SECTION (1) WING

21/8 .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.2486	.0239	-.1948	-.1203	-.2099
-.750	.0642	-.0305	-.2282	-.1716	-.1890
-.600	.0123	-.1444	-.0709	.0960	-.1032
-.450	-.2195	-.0673	.0874	.1890	.0608
-.300	-.0505	-.2051	.1-29	.2972	.1679
-.150	.2685	.1927	-.1070	.3041	.4651
.150	-.2197	-.7321	-.6453	-.1036	-.8433
.300	-.3725	-.5615	-.5644	-.5703	-.7949
.450	-.2252	-.4984	-.4599	-.4279	-.6068
.600	-.1065	-.2677	-.2768	-.2668	-.4547
.750	-.1590	-.1415	-.1378	-.0227	-.2419
.900	-.2390	.0013	.0160	.0748	-.1562

MACH (1) = .165 ALPHA (5) = 14.990 RNL = 1.200 MACH = .165

## SECTION (1) WING

21/8 .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.3025	.0707	-.1499	-.1478	-.2554
-.750	.1517	.5223	-.1756	.1555	-.1566
-.600	.1270	.2352	-.0753	-.0744	-.0253
-.450	-.6377	.0343	.1135		.1304
-.300	-.1111	-.0227	.1-2-		2471

(RDW418)

## CA37-B B10C5F1 J40 W7E18 WING TOTAL SURFACE

MACH ( 1 ) = .165 ALPHA ( 3 ) = 14.990

## SECTION ( 1 ) WING

X/C	21/8	.0000	.3340	.5200	.6630	.6730
-1.90	.3910	.3192	.0009	.4324	.5035	
.150	-.2463	-1.1603	-.7959	-1.2969	-1.5351	
.300	-.4127	-.7243	-.6440	-.6548	-.0614	
.450	-.2441	-.0210	-.4965	-.4894	-.8610	
.600	-.0945	-.3193	-.3473	-.3617	-.6361	
.750	-.1412	-.2045	-.2167	-.0208	-.5309	
.900	-.2591	-.0074	.0129	.0590	-.4006	

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.980 ANL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	21/8	.0000	.3340	.5200	.6630	.6730
- .900	.3284	.0447	-.2770	-.3665	-.4037	
-.750	.1670	.0705	-.1153	-.2017	-.1857	
-.600	.1936	.2528	-.0174	.1910	.0061	
-.450	-.0033	.1813	-.1637	.3645	.1721	
-.300	.2167	.1158	.1054	.5369	.3071	
-.150	.4617	.3900	-.1055	.6101	.5135	
.150	-.3124	-.16000	-.0388	-.3299	-.2492	
.300	-.4981	-.9148	-.7144	-.2662	-.2180	
.450	-.3204	-.7769	-.7346	-.12046	-.11505	
.600	-.1649	-.5113	-.6698	-.1.0291	-.1.1357	
.750	-.2333	-.3446	-.0242	-.4642	-.9901	
.900	-.3316	-.1096	-.2402	-.4073	-.7643	

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TABULATED SOURCE DATA - CA578

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CA57-E 810CSF1 JAO WATE18 WING TOTAL SURFACE (ADYR2D) (112 NOV 73)

## REFERENCE DATA

REF	4.4120 63. FT.	XMAP	Z	43.5940 IN.
LREF	.19.2300 IN.	YMAP	=	.0000 IN.
W	.31.6330 IN.	ZMAP	=	-.4050 IN.
SCALE	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.4.000 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C

-.900	-.1334	-.0477	-.0651	-.0244	-.1517
-.750	-.0938	-.1603	-.1893	-.2863	-.2938
-.600	-.2496	-.1223	-.1629	-.1584	-.2869
-.450	-.5398	-.4420	-.2555	-.1541	-.2470
-.300	-.5117	-.7607	-.2671	-.1704	-.1903
-.150	-.0375	-.2981	-.5581	-.3669	-.0134
.150	-.0432	-.1235	-.2621	-.2963	.1534
.300	-.1772	-.2050	-.3085	-.0102	-.3554
.450	-.0947	-.2574	-.3039	-.1963	-.2591
.600	.0018	-.1534	-.1631	-.1523	-.2165
.750	-.0547	-.0046	-.0697	.0884	.0002
.900	-.2099	-.0733	-.1169	.1486	.0576

MACH ( 1 ) = .165 ALPHA ( 2 ) = .000 RNL = 1.200 MACH = .165

SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C

-.900	-.1553	-.0264	-.0469	-.0481	-.1553
-.750	-.0430	-.1467	-.1405	-.2544	-.2594
-.600	-.1742	-.0229	-.1044	-.0813	-.2313
-.450	-.4292	-.3238	-.1239	-.0315	-.1557
-.300	-.3692	-.5622	-.1601	-.0059	-.0484
-.150	.0223	-.1370	-.3597	-.1274	.2454
.150	.0921	-.2524	-.4591	-.5679	.0616
.300	-.2270	-.2931	-.4032	-.1692	-.4818
.450	-.1370	-.3131	-.3593	-.2623	-.3426
.600	-.0438	-.1914	-.2094	-.2029	-.2754
.750	-.0964	-.0447	.0754	.0621	-.0348
.900	-.2137	.0575	.1063	.1302	.0477

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(ADV20)

MACH (1) = .165    ALPHAS (3) = 4.975    RNL = 1.200    MACH = .165  
 SECTION (1) WING  
 DEPENDENT VARIABLE CP

	21/0	.0000	.3340	.5200	.6630	.8730
L/C						
- .900	.2066	-.0193	-.1399	-.1009	-.1870	
- .750	.0311	-.0967	-.1914	-.2211	-.2266	
- .600	-.0597	.0691	-.0890	.0070	-.1579	
- .450	-.2988	-.1141	.0961	.1034	-.0277	
- .300	-.1636	-.204	.0140	.1927	.0774	
- .150	.2109	.3657	-.1661	.1555	.3798	
.150	-.1522	-.4241	-.5950	-.8521	-.3650	
.300	-.2892	-.4333	-.4796	-.3512	-.6449	
.450	-.1876	-.3956	-.3787	-.3417	-.4540	
.600	-.0790	-.2418	-.2301	-.2370	-.3358	
.750	-.11255	-.1251	-.0947	.0332	-.1193	
.900	-.2329	.C154	-.0075	.1087	-.0201	

MACH (1) = .165    ALPHAS (4) = 9.990    RNL = 1.200    MACH = .165  
 SECTION (1) WING  
 DEPENDENT VARIABLE CP

	21/0	.3340	.5200	.6630	.8730	
L/C						
- .900	.2637	.0397	-.1266	-.1220	-.2157	
- .750	.0932	-.0169	-.1655	-.1667	-.1915	
- .600	.0515	-.1753	-.0039	.1056	-.1066	
- .450	-.1751	-.0216	.1166	.2128	.0625	
- .300	-.0126	-.1445	.1964	.3355	.1776	
- .150	.3051	.2414	-.0499	.3489	.4755	
.150	-.2041	-.7006	-.6415	-.0993	-.8754	
.300	-.3664	-.5370	-.5746	-.5219	-.8093	
.450	-.2149	-.4876	-.4497	-.4627	-.6009	
.600	-.0936	-.2541	-.2744	-.2768	-.4559	
.750	-.1530	-.1576	-.1477	.0C12	-.2689	
.900	-.2491	.0020	.0144	.0907	-.1772	

MACH (1) = .165    ALPHAS (5) = 14.995    RNL = 1.200    MACH = .165  
 SECTION (1) WING  
 DEPENDENT VARIABLE CP

	21/0	.0000	.3340	.5200	.6630	.8730
L/C						
- .900	.3066	.0538	-.1110	-.1559	-.2652	
- .750	.1648	C169	-.1339	-.1393	-.1644	
- .600	.1139	.2343	.0C72	.1603	-.0229	
- .450	-.0663	.0722	.1486	.3191	.1227	
- .300	.1525	.0131	.1463	.4432	2447	



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TABULATED SOURCE DATA - QA57B

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CART-E B16CF1 340 WATE10 WING TOTAL SURFACE (ADMM20)

MACH 1.11 = .165 ALPHA 1.5 = 14.995

SECTION 111WING

DEPENDENT VARIABLE CP

2170 .0000 .3340 .5200 .6630 .8730

X/C

-.150	.3624	.3599	.0239	.4453	.4860
-.150	-.2674	-.2182	-.7687	-.12364	-.16064
.300	-.4126	-.6947	-.6997	-.7251	-.10998
.450	-.2483	-.8625	-.3220	-.5034	-.9397
.600	-.1127	-.3382	-.3464	-.721	-.7033
.750	-.1446	-.2155	-.2324	-.5218	-.5743
.900	-.1626	-.0422	.0056	.0375	.4254

MACH 1.11 = .165 ALPHA 1.5 = 19.990

DEPENANT VARIABLE CP

2170 .0000 .1140 .5200 .6630 .8730

X/C

-.400	.3477	.0386	-.2212	.3824	.5794
-.750	2164	.0744	-.1045	-.2349	-.1705
-.600	-.2235	.2810	.0011C	.1529	.0217
-.450	.0474	.2173	.0040	.3168	.2007
-.300	.2691	.803	.1592	.5240	.3255
-.150	.4761	.4234	.0942	.7096	.5331
.150	-.3062	-.1592	-.7764	-.13198	-.11949
.300	-.4818	-.9062	-.7944	-.13328	-.12048
.450	-.3086	-.7541	-.8154	-.1140	-.1243
.600	-.1647	-.4919	-.6336	-.10256	-.11336
.750	-.2269	-.3427	-.5038	-.4579	-.10019
.900	-.3231	-.1214	-.2294	-.3150	-.7349

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TABULATED SOURCE DATA - Q3STB

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QA57-B B16C5F1 J40 WTE10 WING TOTAL SURFACE

(ADW21) ( 12 NOV 73 )

## REFERENCE DATA

BREF =	4.4120 SQ.FT.	XHWP =	43.5940 IN.
LREF =	19.2100 IN.	YHWP =	.0000 IN.
BREF =	37.8350 IN.	ZHWP =	.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.975 ROLL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730
X/C					
- .900	.0119	-.0728	-.0887	-.0517	-.1597
- .750	-.1043	-.1813	-.1535	-.2455	-.2795
- .600	-.1690	-.0963	-.1484	-.1233	.2833
- .450	-.3959	-.3591	-.1839	-.1256	.2426
- .300	-.4121	-.6249	-.1826	-.1180	-.1654
- .150	.0679	-.2432	-.4619	-.3020	.0229
.150	-.0463	-.1284	-.2788	-.3306	1.241
.300	-.1868	-.2281	-.3445	-.1322	-.3632
.450	-.0958	-.2786	-.3122	-.2311	-.2712
.600	-.0022	-.1524	-.1780	-.1719	.2444
.750	-.0583	-.0483	-.0909	.0741	-.0089
.900	-.1886	.0223	.0750	.1134	.0522

MACH ( 1 ) = .165 ALPHA ( 2 ) = .025 ROLL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730
X/C					
- .900	.0463	-.0476	-.0387	-.0288	-.1616
- .750	-.0506	-.1455	-.1037	-.2364	-.2557
- .600	-.0912	-.0089	-.0808	-.0602	-.2344
- .450	-.3191	-.2564	-.0756	-.0111	-.1599
- .300	-.2666	-.4528	-.0908	.0198	.0391
- .150	.1531	-.0963	-.2620	-.0876	.2797
.150	-.0847	-.2507	-.4325	-.5812	-.0944
.300	-.2328	-.3068	-.4202	-.2603	-.4966
.450	-.1413	-.3392	-.3655	-.2905	-.3379
.600	-.0379	-.1637	-.2136	-.2109	-.2341
.750	-.0933	-.0838	-.0857	.0613	-.0469
.900	-.1967	.0193	.0749	.1277	.0411

## PARAMETRIC DATA

BETA =	.000	PTR/P = 1.000
H/B =	.286	BDPLP = 20.000
ELEVON =	.000	

DATE 08 OCT 74

## TABULATED SOURCE DATA - CAA578

PAGE 225

(RDW21)

MACH ( 1 ) = .165    ALPHA ( 3 ) = 5.00    RNL = 1.200    MACH = .165  
 SECTION ( 1 ) WING

21/8    .0000    .3340    .5200    .5630    .6730  
 DEPENDENT VARIABLE CP

X/C

-.900	.1132	-.0234	-.1535	-.0697	-.2041
-.750	.0057	-.1089	-.1861	-.1868	-.2162
-.600	-.0042	.0119	-.0703	.0561	-.1692
-.450	-.2054	-.1219	.1135	.1329	-.0207
-.300	-.1138	-.1249	.0370	.2261	.0886
-.150	.2111	.0938	-.1413	.1976	.3913
.150	-.1545	-.4101	-.5789	-.8407	-.3988
.300	-.3089	-.4523	-.5087	-.4136	.6435
.450	-.1726	-.4045	-.3733	-.3780	.4606
.600	-.0795	-.2309	-.2449	-.2239	-.3472
.750	-.1377	-.1634	-.1066	.0367	-.1203
.900	-.2029	-.0046	-.0311	.1046	-.0376

MACH ( 1 ) = .165    ALPHA ( 4 ) = 10.00    RNL = 1.200    MACH = .165

SECTION ( 1 ) WING

21/8    .0000    .3340    .5200    .6630    .8730  
 DEPENDENT VARIABLE CP

X/C

-.900	.1673	.0029	-.1508	-.0842	-.2344
-.750	.0772	-.0301	-.2010	-.1206	-.2089
-.600	.0869	.1556	-.0528	.1307	-.1116
-.450	-.1107	-.0174	.1177	.2381	.0453
-.300	.0338	.0424	.1834	.3704	.1695
-.150	.2956	.2240	-.0642	.3890	.4715
.150	-.2311	-.7325	-.6559	-.11367	-.8918
.300	-.3885	-.5560	-.6466	-.5045	-.9256
.450	-.2145	-.4938	-.4675	-.3157	-.6064
.600	-.0895	-.2748	-.2901	-.3137	-.4888
.750	-.1517	-.1876	-.1521	-.0097	-.2919
.900	-.2379	-.0552	-.0186	.0836	-.2137

MACH ( 1 ) = .165    ALPHA ( 5 ) = 15.015    RNL = 1.200    MACH = .165

SECTION ( 1 ) WING

21/8    .0000    .3340    .5200    .6630    .8730  
 DEPENDENT VARIABLE CP

-.900	.2272	.0261	-.0869	-.2994	-.2678
-.750	.1461	.0259	-.0375	.1002	-.1837
-.600	.1806	.2360	.0356	.1514	-.0402
-.450	-.0096	.0987	-.0224	-.0007	.1205
-.300	.1797	.1803	.1647	.4421	.2313

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TABULATED SOURCE DATA - CAA57B

PAGE 226

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15 015

CMA57-B

B10C5F1 JAO WOTE10 WING TOTAL SURFACE

(ADVM21)

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .6730

X/C

-.150	.3694	.3391	.0744	.6016	.4732
.150	-.2484	-.1970	-.7579	-.1.3265	-.1.6632
.300	-.4199	-.7346	-.6717	-.3089	-.1.1327
.450	-.2385	-.6134	-.5320	-.1.2860	-.9365
.600	-.0946	-.3179	-.3223	-.3205	-.7159
.750	-.1450	-.2121	-.1956	-.0203	-.5931
.900	-.2486	-.0151	-.0379	.0648	-.4476

MACH ( 1 ) = .165 ALPHA ( 6 ) = 20 005 RVAL = 1.200 MACH = .165

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .6730

X/C

-.900	.2576	.0042	-.2164	-.1407	-.3803
-.750	.1774	.0370	-.1647	-.0517	-.1944
-.600	.2453	.2618	-.0579	.2231	-.0088
-.450	.0769	.2024	-.0578	.3186	.1682
-.300	.2961	.2973	.2319	.4908	.2866
-.150	.4998	.4257	.2825	.7485	.5069
.150	-.2946	-.1.5275	-.7682	-.1.2941	-.1.1950
.300	-.4890	-.9197	-.7443	-.1.3612	-.1.1390
.450	-.3030	-.7564	-.8505	-.1.2446	-.1.0968
.600	-.1544	-.5067	-.6928	-.1.1022	-.1.1530
.750	-.2395	-.3415	-.3455	-.4587	-.9968
.900	-.3107	-.0992	-.2579	-.3497	-.7456



## CA57-B 816CF1 JAO WATE10 WING TOTAL SURFACE (RDW22)

MACH ( 1 ) = .165 ALPHAB ( 3 ) = 5.005 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C	.2293	.0159	.2063	.1622	.2150
-.900					
-.750	.0263	-.0658	-.2410	-.2760	-.2596
-.600	-.0453	.0888	-.1078	-.0256	-.1943
-.450	-.2975	-.1525	.0412	.0728	-.0532
-.300	-.0927	-.0930	-.0209	.1747	.0474
-.150	-.2367	.1245	-.2255	.1553	.3513
.150	-.1626	-.5033	-.5989	-.6918	-.4250
.300	-.5111	-.5236	-.5001	.4741	.6657
.450	-.2116	-.4841	-.3994	-.4228	-.4694
.600	-.0867	-.2440	-.2552	-.2560	-.3580
.750	-.1552	-.1365	-.1329	.0176	-.1371
.900	-.2766	.0132	-.0567	.0844	-.0440

MACH ( 1 ) = .165 ALPHAB ( 4 ) = 9.995 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C	.577	.1105	-.0860	-.1599	-.2466
-.900					
-.750	.1955	.0976	-.0886	-.2139	-.2214
-.600	.1388	.2527	.0297	.0987	-.1161
-.450	-.0706	.0631	.1468	.2136	.0350
-.300	.1061	.1345	.1757	.3254	.1556
-.150	.3603	.2898	-.0549	.3613	.552
.150	-.2146	-.6764	-.6791	-.1418	-.9094
.300	-.3740	-.5628	-.5843	-.6490	-.8193
.450	-.2249	-.5030	-.4680	.5155	.6093
.600	-.1009	-.2795	-.2918	-.2979	-.4823
.750	-.597	-.1940	-.1613	-.0022	-.2922
.900	.673	.0152	-.0026	.0850	-.2162

MACH ( 1 ) = .165 ALPHAB ( 5 ) = 14.995 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C	.4662	.2024	-.0635	-.1707	-.3017
-.900					
-.750	.3209	.2246	.0366	.1602	-.1852
-.600	.2619	.3553	.1138	.2108	-.5258
-.450	.0737	.1923	.2322	.3455	.1162
-.300	.2376	.2564	.2214	.4489	.2431

==

DATE OF OCT 74 TABULATED SCALAR DATA - CASTB

## SECTION ( 1 ) WING C457-B B165SF1 JAC WTE10 WING TOTAL SURFACE (RDYR22)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.995

## SECTION ( 1 ) WING DEFENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.5630	.6730
-.150	.4463	.3949	-.1483	.2172	.4760	
-.150	-.2758	-.12547	-.7904	-.12663	-2.07	A
.300	-.4355	-.7751	-.6873	-.8166	-1.2537	
.450	-.2531	-.6275	-.5920	-.5283	-1.0363	
.600	-.1211	-.3528	-.3637	-.3342	-.7597	
.750	-.1652	-.2446	-.2545	-.1436	-.6017	
.900	-.2641	-.0078	-.0335	-.0493	-.4542	

MACH ( 1 ) = .165 ALPHA ( 6 ) = 20.015 RNL = 1.200 MACH =

## SECTION ( 1 ) WING DEFENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.5630	.6730
-.900	.5318	.2216	-.1469	-.3975	-.3645	
-.750	.4083	.2908	-.0009	-.2817	-.1764	
-.600	.3688	.4355	.0834	.1511	.0315	
-.450	.1985	.3394	.1265	.3376	.1906	
-.300	.3680	.4224	.2221	.5008	.3005	
-.150	.5346	.4972	.0575	.7131	.5284	
.150	-.3097	-.6400	-.8699	-.14457	-.2378	
.350	-.4919	-.9175	-.7560	-.13996	-.1765	
.450	-.3385	-.8058	-.9790	-.12232	-.1122	
.600	-.1841	-.5415	-.7649	-.10940	-.1.1112	
.750	-.2522	-.3744	-.5745	-.5244	-.9925	
.900	-.3402	-.1026	-.3494	-.5344	-.7277	



DATE 08 OCT 74 TABULATED SOURCE DATA - CA378

CA37-B B16C5F1 JAO WSTE18 WING TOTAL SURFACE

(NDVW23)

MACH ( 1 ) = .165 ALPHAI ( 3 ) = 4.985 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEFINITION VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	.2511	.0012	-.1405	-.1474	-.2170	
- .750	.0647	-.0573	-.1849	-.2459	-.2475	
- .600	.0046	.1245	-.0604	.0056	-.1843	
- .450	-.2173	-.0742	.0934	.1090	-.0274	
- .300	-.0441	.0016	.0423	.2226	.0619	
- .150	.2671	.1588	-.1372	.2442	.3789	
.150	-.1663	-.5210	-.5891	-.6677	-.4453	
.300	-.3136	-.5459	-.5077	-.5115	-.6737	
.450	-.2009	-.4861	-.3937	-.4208	-.4655	
.600	-.0924	-.2424	-.2438	-.2607	-.3518	
.750	-.1516	-.1579	-.1386	.0056	-.1528	
.900	-.2622	.0046	-.0639	.0771	-.0108	

MACH ( 1 ) = .165 ALPHAI ( 4 ) = 10.000 RNL = 1.200 MACH = .65

## SECTION ( 1 ) WING

DEFINITION VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	.3666	.0961	-.0545	-.1450	-.2597	
- .750	.2061	.0934	.0365	-.1995	-.2224	
- .600	.1650	.2674	.0320	.0969	-.1286	
- .450	-.0292	.0080	.1586	.2300	.0339	
- .300	-.1374	.1975	.1956	.3577	.1598	
- .150	.3620	.5041	-.0054	.4034	.4606	
.150	-.2222	-.7298	-.7211	-.1612	-.9761	
.300	-.3651	-.5605	-.5835	.6802	-.8373	
.450	-.23.2	-.4994	-.5103	-.5349	-.6329	
.600	-.1092	-.2879	-.3131	-.3266	-.5260	
.750	-.1716	-.2051	-.1691	-.0196	-.3163	
.900	-.2619	-.0064	-.0130	.0714	-.2376	

MACH ( 1 ) = .165 ALPHAI ( 5 ) = 15.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEFINITION VARIABLE CP

X/C	21/8	.0005	.3340	.52.5	.6630	.8735
- .900	.4593	.1741	-.0376	-.1767	-.3	
- .750	.3219	.1957	.477	-.121	-.1763	
- .600	.2613	.5610	.1152	-.111	-.5316	
- .450	1.	21.1	2.23	1.	2.34	
- .300	1.117	3.117	3.117	1.	3.117	



DATE 08 OCT 74

TABULATED SOURCE DATA - GA37P

PAGE 233

CAE7-P B10CSF1 J45 WATE10 WING TOTAL SURFACE

(ROW24) ( 12 NOV 73 )

## REFERENCE DATA

X/C					
-	=	.4120	.6471.	ZMXP =	.43.3940 IN.
REF	=	10.2300	IN.	IMPF =	.0600 IN.
BREF	=	57.9350	IN.	ZMXP =	-.4950 IN.
SCALE	=	.0405			

MACH ( 1 ) = .165

ALPHA ( 1 ) =

- 3.380

RNL =

1.200

MACH =

1.0

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C						
-.900	=	.0392	-.1390	-.1386	-.0932	-.1910
-.750	=	.1853	-.2692	-.2326	-.3170	-.3264
-.600	=	.2940	-.1915	-.243	-.1862	-.3310
-.450	=	.5285	-.4634	-.2954	-.1940	-.2933
-.300	=	.1594	-.4602	-.3354	-.2135	-.2213
-.150	=	.0702	-.3313	-.5607	-.2814	-.0125
.150	=	.0603	-.1579	-.297	-.3590	.1174
.300	=	.2019	-.2568	-.3412	-.2721	.3901
.450	=	.1067	-.3041	-.3356	-.2877	.2026
.600	=	.0146	-.1753	-.1964	-.1956	.0254
.750	=	.0703	-.1145	-.1300	.0480	.0332
.900	=	.1923	.0450	.0534	.0842	

MACH ( 1 ) = .165

ALPHA ( 2 ) =

.025

RNL =

1.200

MACH =

.5

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C						
-.100	=	.0448	-.0584	-.08942	-.0700	-.1826
-.750	=	.0866	-.1948	-.1427	-.2564	-.214
-.600	=	.1145	-.0278	-.1556	.0701	-.562
-.450	=	.3453	-.2705	-.1049	-.0243	-.1547
-.300	=	.2414	-.1773	-.1308	.0357	.0489
-.150	=	.1734	-.0594	-.2776	-.0382	.2713
.150	=	.0969	-.2764	-.4576	-.6116	-.1155
.300	=	.2531	-.3587	-.4338	-.1956	.5132
.450	=	.1448	-.3556	-.3442	-.1541	.5624
.600	=	.0316	-.1554	-.2217	-.2257	-.3554
.750	=	.1950	-.1743	-.1743	-.1743	-.1743
.900	=	.2120	-.1742	-.1742	-.1742	-.1742

## PARAMETRIC DATA

BETA	=	.000	PIN/P =	1.000
R.E	=	.125	BDFLAP =	20.000
ELEVAN	=	.000		

DATE 06 OCT 74

TABULATED SOURCE DATA - CA370

PAGE 234

## CA370 810CSF JAO WATE10 WING TOTAL SURFACE (RDW24)

MACH ( 1 ) = .95 ALPHA ( 3 ) = 4.995 ANL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	.1601	-.0028	-.1508	-.1035	-.2229
-.750	.0582	-.0594	-.1624	-.1610	-.2329
-.600	.0657	.172	-.0446	.0601	-.1798
-.450	-.1396	-.0517	.1143	.1543	-.0150
-.300	.0169	.0568	.0619	.2747	.0607
-.150	-.2718	.1819	-.1061	.2958	.4039
.150	-.1627	-.5141	.6055	.9067	-.4657
.300	-.3189	-.5365	-.5111	.5385	-.6836
.450	-.1989	-.4872	-.3998	.4340	-.4769
.600	-.0901	-.2444	.2534	.2779	-.3332
.750	-.1516	-.1773	-.1389	.0014	-.1610
.900	-.2353	.0170	-.0519	.0762	-.0279

MACH ( 1 ) = .165 ALPHA ( 4 ) = 10.005 ANL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	.2673	.0700	-.0681	-.1081	-.2447
-.750	.1873	.0643	-.0822	-.1326	-.2169
-.600	.1990	.2591	.0446	.1271	-.1122
-.450	.0336	.1025	.1622	.2359	.0372
-.300	.1688	.2284	.1793	.4243	.1680
-.150	.3571	.3160	.0186	.5103	.4902
.150	-.2295	-.7332	-.5783	-.1459	-1.0000
.300	-.3817	-.765	-.5706	-.7133	-.8494
.450	-.2216	-.4936	-.5046	-.5361	-.6454
.600	-.1143	-.2889	-.2977	-.3197	-.5286
.750	-.1602	-.2130	-.1765	-.0194	-.3567
.900	-.2323	-.0033	-.0119	.0726	-.2309

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.995 ANL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

-.900	.3564	.1301	-.0437	-.0625	-.2781
-.750	.2970	.1506	-.0003	.0737	-.1707
-.600	.3089	.3423	.0680	.1639	-.0177
-.450	.1571	.2222	.0903	.2696	.1111
-.300	.3227	.3387	.2267	.5097	.2190

DATE 06 SEPT 74

ELEVATED SURFACE DATA - LAYER

A.E. 235

SECTION (1) WING P:66571 .4C WING18 WING TOTAL SURFACE (RDVW24)

MACH (.1) = .163 ALPHAL 0.0 = 14.163

SECTION (1) WING

DEPENDENT VARIABLE CP

X/C	1.00	.4493	.4196	.0620	.6661	.4730
.150	-2752	-1.2519	-7333	-1.3526	-2.6536	
.300	-4486	-6036	-6486	-6691	-1.7511	
.450	-2559	-6305	-6231	-5227	-1.3152	
.600	-1217	-3505	-3504	-3536	-7536	
.750	-1684	-2647	-2550	-1226	-6279	
.900	-2311	-0167	-2219	-0346	-4563	

MACH (.1) = .163 ALPHAL 0.0 = 19.995

PWL = 1.200 MACH = 1.000

SECTION (1) WING

DEPENDENT VARIABLE CP

X/C	1.00	.4168	.1322	.1551	.1369	.4259
.150	.3627	.1974	.2553	.2226	.2015	
.300	.3936	.4042	C375	.2466	.0035	
.450	.2642	.3339	.1363	.2571	.1572	
.600	.4374	.4645	.3968	.4276	.2498	
.750	.5327	.5125	.4406	.6059	.5290	
.900	.3143	-1.6137	-7841	-1.4207	-1.1594	
.450	.3593	.8074	.7365	-1.2537	-1.1141	
.600	.1920	.5665	.8792	-1.2219	-1.1242	
.750	.2113	.4312	.7834	-1.1223	-1.0590	
.900	.3212	.4297	.6561	-1.6653	-1.5948	

## CAST-E B16C5F1 J40 WTE10 WING TOTAL SURFACE

(RDW25) (112 NOV 73)

## REFERENCE DATA

SREF =	4.4120 IN.	XMAP =	43.5940 IN.	BETA =	.000
LREF =	19.2300 IN.	YMAP =	.0000 IN.	M/B =	.125
SREF =	37.9350 IN.	ZMAP =	-.4050 IN.	ELEVON =	.000
SCALE =	.0405				

MACH (1) = .165 ALPHA (1) = 10.000 ANGL = 1.200 MACH = .165

## SECTION (1) WING

DEFINITION VARIABLE CP

21/E .0000 .3340 .5200 .6630 .8730

x/c					
- .900	1.2255	.3437	.0356	.0697	-.3607
- .750	.5402	.4637	.3210	-.0203	-.4168
- .600	.4371	.5639	.2790	.1407	-.3606
- .450	.2846	.4066	.19134	.1410	-.1922
- .300	.4136	.3256	.2295	.1870	-.0424
- .150	.4748	.4743	.0922	.2946	.3453
.150	-.2316	-.7477	-.7165	-.1146	-.6949
.300	-.5876	-.5474	-.5707	-.6271	-.8216
.450	-.2392	-.5225	-.4693	-.4945	-.6284
.600	-.1076	-.2693	-.3037	-.3133	-.4427
.750	-.1537	-.0463	-.2082	-.0045	-.2824
.900	-.2452	-.0358	-.0492	-.0584	-.2491

MACH (1) = .165 ALPHA (2) = 14.965 ANGL = 1.200 MACH = .165

## SECTION (1) WING

DEFINITION VARIABLE CP

x/c					
- .900	1.1548	.5446	.2213	.2129	-.4336
- .750	.7001	.6325	.4339	.1158	-.3676
- .600	.5294	.6429	.2291	.2066	-.2568
- .450	.4001	.4648	.1221	.124	-.0719
- .300	.4663	.4139	.2754	.2361	.0743
.150	.5164	-.44	-.1412	.4272	.4006
.150	-.2776	-.1236	-.7619	-.1959	-.23465
.300	-.4597	-.7510	-.6391	-.6626	-.14431
.450	-.2117	-.6567	-.5773	-.5111	-.13028
.600	-.1263	-.3610	-.4505	-.3745	-.0384
.750	-.1769	-.1712	-.3507	-.1579	-.7150
.900	-.2620	-.0118	-.0172	-.0715	-.4556



DATE 08 OCT 74

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INTRODUCTION

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18 CIRCUITS

## CASE 7-B 316C3F1 J40 W/TE18 WING TOTAL SURFACE

(DDM426) (112 NOV 73)

## REFERENCE DATA

SREF =	4.4120 82 FT.	XPR =	= 43.5940 IN.
LREF =	.19.2300 IN.	YPR =	= .0000 IN.
BREF =	.37.9350 IN.	ZPR =	= -.4050 IN.
SCALE =	.0403		

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

2/B .0000 .3340 .5200 .6630 .8730

X/C

-.900	.6412	.2576	.0464	.1030	-.3759
-.750	.4967	.3706	.2523	-.0097	.3950
-.600	.4304	.5367	.1859	.1052	-.3702
-.450	.3035	.4081	.2913	.1221	-.2027
-.300	.4463	.3638	.2747	.2131	-.0509
-.150	.4692	.4814	.0982	.3512	.3316
.150	-.2356	-.7342	-.6850	-.1147	-.9466
.300	-.3977	-.5609	-.5656	-.5923	-.8221
.450	-.2283	-.4947	-.4627	-.4836	-.8308
.600	-.1029	-.2874	-.3105	-.3133	-.4645
.750	-.1580	-.1214	-.2181	-.0072	-.2955
.900	-.2192	.0150	.0424	.0511	-.2629

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.985 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

2/B .0000 .3340 .5200 .6630 .8730

X/C

-.900	.9116	.4113	.1832	.1848	-.4540
-.750	.6756	.5314	.3883	.0857	-.3826
-.600	.5063	.6111	.2011	.1796	-.2385
-.450	.3925	.4762	.3327	.1919	-.0546
-.300	.4955	.4340	.2957	.2303	.0875
-.150	.5360	.5305	-.1736	.3672	.4032
.150	-.2931	-.1293	-.7696	-.12008	-.2.6638
.300	-.4507	-.7719	-.6586	-.6681	-.1.7203
.450	-.2829	-.6262	-.5944	-.5164	-.1.4559
.600	-.1245	-.3608	-.4063	-.3722	-.9987
.750	-.1704	-.2163	-.3186	-.1707	-.1.7909
.900	-.2429	-.0167	-.0371	-.1003	-.4585



MACH (1) =	.165	ALPHA (3) =	20.005	RN/L =	1.200	MACH =	.165
SECTION (1) WING				DEPENDENT VARIABLE CP			
21/8	.0000	.3340	.5200	.6630	.8730		
X/C							
-.900	1.0624	.4758	.1717	.0959	-.4953		
-.750	.7698	.5825	.4076	-.0031	-.3085		
-.600	.5636	.5610	.3227	.2159	-.0596		
-.450	.4637	.5403	.3975	.2093	.1186		
-.300	.5659	.5030	.2659	.2543	.2423		
-.150	.5943	.5783	-.1545	-.0612	.4685		
.150	-.3687	-1.7165	-.8756	-1.7391	-1.2289		
.300	-.4958	-.9133	-.7922	-1.2956	-1.1728		
.450	-.4167	-.8073	-.9153	-1.0869	-1.1279		
.600	-.2225	-.6120	-.7412	-.9451	-1.0309		
.750	-.2935	-.5835	-.6445	-.5666	-.8542		
.900	-.3061	-.0938	-.3170	-.5092	-.6787		

(RDY44261)

## CA37-B B16CSF1 J40 WATE10 WING TOTAL SURFACE

(INDW27) (12 NOV 73)

## REFERENCE DATA

BREF =	4.4120 Sq. FT.	XWEP =	43.3940 IN.	BETA =	.000
LREF =	19.2300 IN.	ZWEP =	.0000 IN.	H/B =	.125
BREF =	37.9350 IN.	ZWEP =	-.4050 IN.	BDFLAP =	.000
SCALE =	.0405			ELEVON =	

MACH ( 1 ) = .165 ALPHA ( 1 ) = 9.985 ANVL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

x/c					
-.900	.2381	.1339	.0414	.0163	-.4381
-.750	.2857	.1643	.1381	.1965	.4126
-.600	.3063	.4084	.0929	.0116	.2940
-.450	.2508	.2882	.2620	.0820	-.1343
-.300	.4413	.3494	.2114	.3325	-.0056
-.150	.4553	.4570	.0807	.4625	.3717
.150	-.2804	-.7158	-.6956	-.1146	-1.0029
.300	-.3775	-.5495	-.5624	-.5987	-.8744
.450	-.2457	-.5059	-.5051	-.5103	-.6782
.600	-.1046	-.2900	-.3163	-.3276	-.4912
.750	-.1385	-.1616	-.2237	-.0193	-.3747
.900	-.2056	.0035	.0174	.0274	-.3470

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.985 ANVL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c					
-.900	.4629	.2410	.0655	.0433	-.4120
-.750	.4790	.3103	.2285	-.0424	-.3092
-.600	.4910	.4984	.2263	.0702	-.2192
-.450	.3822	.3729	.2895	.1424	-.0152
-.300	.5056	.4283	.2292	.2915	.1077
-.150	.5213	.5236	.0286	.5095	.3991
.150	-.3035	-.9839	-.7324	-.1192	-.6596
.300	-.3791	-.7091	-.6099	-.5914	-.5928
.450	-.2642	-.5670	-.5463	-.4498	-.3676
.600	-.1369	-.3714	-.4047	-.3839	-.8247
.750	-.1946	-.2771	-.3782	-.1876	-.7616
.900	-.2267	-.0444	-.1228	-.1202	-.4429



DATE 08 OCT 74

TABULATED SOURCE DATA - CASTE

FILE 241

MACH = 1.1) = .165    ALPHAS = 19.23C    ANL = 1,200    MACH = .165  
 CASTE DATA - E16.3F1 JAC 48TE16 WING TOTAL SURFACE

(RDV427)

## SECTION 1: WING

## DEPENDENT VARIABLE CF

X/C	21/0	.0000	.3340	.5200	.6630	.8730
.900	.6238	.2936	.0240	-.0004	-.4018	
.750	.5917	.3666	.2609	-.0228	-.2339	
.600	.5483	.5594	.3292	.2421	-.1457	
.450	.4552	.4335	.4126	.3470	.0417	
.300	.5659	.4825	.2266	.3701	.2024	
.150	.5845	.5726	-.0386	.0328	.4646	
.150	.3326	-1.5821	-.8226	-1.7489	-1.1612	
.300	-.4253	-.8309	-.7798	-1.3119	-1.1476	
.450	-.4379	-.6373	-.8530	-1.0809	-1.0912	
.600	-.2768	-.6046	-.7376	-.9670	-.9927	
.750	-.3152	-.4047	-.6431	-.5471	-.8195	
.900	-.2958	-.1580	-.3849	-.5294	-.6325	

## REFERENCE DATA

BREF =	4.1120 64 FT.	XMAP =	43.5940 IN.	
LREF =	19.2300 IN.	YMAP =	.0000 IN.	
BREF =	37.9350 IN.	ZMAP =	-.4050 IN.	
SCALE =	.0405			

$$\text{MACH } (-1) = .165 \quad \text{ALPHA } (-1) = -.025 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730
X/C					
-.900	-.1549	-.2477	.0430	.0659	.0229
-.750	-.2198	-.1220	.1555	.1967	.0812
-.600	-.4927	-.2076	-.125	-.0486	.0466
-.450	-.8050	-.5678	.2548	-.0917	-.1124
-.300	-.7604	-.7612	.3162	-.1531	-.1156
-.150	.0425	-.4691	-.7937	-.3967	.0451
.150	-.0857	-.2280	-.3271	-.4112	.0509
.300	-.1939	-.2562	-.4080	-.2283	-.5342
.450	-.1454	-.3702	-.3646	-.4301	-.5908
.600	-.0382	-.2891	-.3465	-.3942	-.4246
.750	-.0937	-.3513	-.4305	-.6685	-.3007
.900	-.2169	-.1482	-.0731	-.0314	-.1174

$$\text{MACH } (-1) = .165 \quad \text{ALPHA } (-2) = -.020 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730
X/C					
-.900	-.1017	-.0793	.3038	.1758	-.0387
-.750	-.1146	.0346	.2767	.3230	.1274
-.600	-.2140	.0038	.0305	.0829	.0977
-.450	-.4969	-.2685	-.0556	.0888	.0144
-.300	-.3306	-.3326	-.1155	.0848	.0462
-.150	.1613	-.1016	-.3764	-.0389	.3098
.150	-.1336	-.3848	-.5006	-.7046	-.1761
.300	-.2309	-.3538	-.5006	-.3852	-.6875
.450	-.1944	-.4295	-.4376	-.5204	-.6848
.600	-.0645	-.3343	-.3801	-.4405	-.5320
.750	-.1176	-.3273	-.3939	-.6431	-.3584
.900	-.1862	-.1612	-.1297	-.0702	-.1796

(RDW-28) ( 12 NOV 73 )

## PARAMETRIC DATA

BETA =	.000	PIN/P =	1.500
H/B =	.125	BDFLAP =	-.10.000
ELEVON =	15.000		



DATE 09 OCT 74

TABULATED SOURCE DATA - OA57B

PAGE 243

MACH ( 1 ) = .165 ALPHAS ( 3 ) = 4.950 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

OA57-B 816CF1 J40 WATE10 WING TOTAL SURFACE

(RDW/M28)

MACH ( 1 ) = .165 ALPHAS ( 4 ) = 9.965 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

OA57-B 816CF1 J40 WATE10 WING TOTAL SURFACE

(RDW/M28)

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C

- .900	.0013	.0699	.2702	.2533	-.0861
- .750	.0592	.2224	.3436	.4326	.1381
- .600	.0274	.2174	.1314	.1988	.1313
- .450	-.1938	.0015	.1774	.2411	.0921
- .300	-.0319	-.0173	.1410	.3047	.1325
- .150	-.2002	.1764	-.1191	.2942	.4384
.150	-.2089	-.5587	-.6496	-1.0077	-.6200
.300	-.3182	-.5375	-.5929	-.5482	-.8795
.450	-.2362	-.5139	-.4732	-.6080	-.8305
.600	-.1043	-.3866	-.3916	-.4801	-.6671
.750	-.1514	-.3404	-.3142	-.6131	-.5021
.900	-.1687	-.1690	-.2205	-.1625	-.3352

MACH ( 1 ) = .165 ALPHAS ( 4 ) = 9.965 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C

- .900	.1056	.2426	.4799	.2552	-.1394
- .750	.2122	.3677	.4186	.4335	.1409
- .600	.1841	.3516	.2269	.2835	.1727
- .450	-.0007	.1663	.2756	.3584	.1610
- .300	.1485	.1916	.3295	.4251	.2290
- .150	.3748	.3177	.3496	.4483	.4979
.150	-.3044	-.6971	-.7391	-1.2590	-1.2115
.300	-.3692	-.6200	-.6198	-.7105	-.1232
.450	-.2626	-.5684	-.5493	-.6705	-1.0110
.600	-.1306	-.4037	-.4328	-.5232	-.7934
.750	-.1543	-.3469	-.4074	-.5820	-.6620
.900	-.1629	-.1750	-.0916	-.1382	-.5016

MACH ( 1 ) = .165 ALPHAS ( 5 ) = 14.965 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C

- .900	.1663	.3124	.4071	.2316	-.1413
- .750	.3018	.4336	.5074	.5363	.1981
- .600	.2900	.4301	.2756	.3655	.2503
- .450	.1144	.2607	.2332	.4729	.2505
- .300	.2653	.3097	.2724	.5742	.3387

(ADW420)

## CA57-B B16C SF1 J40 WTE10 WING TOTAL SURFACE

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.965

SECTION ( 1 ) WING  
21/8 .0000 .3340 .5200 .6630 .8730  
X/C

X/C	-1.50	-1.40	-1.30	-1.20	-1.10
	.4614	.4183	.3138	.4925	.5140
	-.3667	-.8653	-.8248	-.3013	-.7741
	-.4293	-.7955	-.7666	-.7731	1.9679
	-.3092	-.7043	-.6548	-.6245	1.5779
	-.6000	-.1616	-.4763	-.5226	-.5501
	-.7500	-.1793	-.3609	-.3203	-.4412
	-.9000	-.1924	-.2454	-.2262	-.2549

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990 RNU/L = 1.200 MACH = .165  
SECTION ( 1 ) WING  
21/8 .0000 .3340 .5200 .6630 .8730  
X/C

X/C	-1.900	-.1929	.3063	.2524	-.1231	-.1027
	-.750	.3531	.4601	.4951	.4107	.2550
	-.600	.3704	.4817	.2406	.2943	.3149
	-.450	.2233	.3767	.2021	.4270	.3216
	-.300	.3753	.4305	.2358	.6035	.4094
	-.150	.5396	.5077	.2864	.7462	.5564
	-.150	-.4064	-.13077	-.8626	-.3979	-.12220
	-.300	-.5155	-.8816	-.9419	-.3016	-.1672
	-.450	-.4674	-.9203	-.0191	-.1979	-.1503
	-.600	-.2898	-.6969	-.8842	-.1339	-.0576
	-.750	-.3056	-.5078	-.6535	-.8040	.9546
	-.900	-.2729	-.3758	-.5445	-.7103	-.7679



## CAST-E B16C5F1 J40 W07E18 WING TOTAL SURFACE

(RDW29) (12 NOV 73)

## REFERENCE DATA

SREF = 4,4120 SQ.FT. XMRP = 43,5940 IN.  
 LREF = 19,2300 IN. YMRP = 00:30 IN.  
 CBL = 37,9350 IN. ZMRP = .4030 IN.  
 SCALE = .0403

MACH (1) = .165 ALPHA (1) = -.010 RNL = 1,200 MACH = .165

## SECTION (1) WING

X/C	.0000	.3340	.5200	.6630	.8730	DEPENDENT VARIABLE CP
- .900	-.1371	-.2512	.1345	.0715	-.0458	
- .750	-.1708	-.1203	.1436	.1639	.0765	
- .600	-.3561	-.1709	-.0094	-.0346	.0314	
- .450	-.6077	-.3879	-.1735	-.0535	-.1049	
- .300	-.6438	-.5283	-.2198	-.0779	-.0844	
- .150	-.0103	-.3461	-.6271	-.2746	.0982	
.150	-.0897	-.4128	-.3609	-.4280	-.0075	
.300	-.2267	-.2473	-.4193	-.1553	-.5401	
.450	-.1416	-.3492	-.3403	-.4423	-.5876	
.600	-.0374	-.2938	-.3612	-.4110	-.4173	
.750	-.1033	-.1545	-.4324	-.7405	-.3037	
.900	-.2068	-.2410	-.1358	-.0527	-.1394	
MACH (1) = .165 ALPHA (2) = -.030 RNL = 1,200 MACH = 165						
MACH (1) = .165 ALPHA (2) = -.030 RNL = 1,200 MACH = 165						

X/C	.0000	.3340	.5200	.6630	.8730	DEPENDENT VARIABLE CP
- .900	-.0435	-.0338	.1706	.1640	-.0612	
- .750	-.0554	.0534	.2093	.2968	.1130	
- .600	-.1346	.G508	.0481	.0848	.0867	
- .450	-.3378	-.2006	-.0101	.1101	.0223	
- .300	-.2635	-.1863	-.0363	.1342	.0493	
- .150	.1787	-.0363	.2398	.G350	.3284	
.150	-.1361	-.5767	-.5315	-.7136	-.1765	
.300	-.2829	-.3420	-.5022	-.3067	-.6938	
.450	-.1916	-.4123	-.4027	-.5117	-.6774	
.600	-.0625	-.3213	-.3846	-.4425	-.5221	
.750	-.1247	-.1428	-.3900	-.6796	-.3723	
.900	-.1876	-.2931	-.1615	-.0919	-.1549	

SECTION (1) WING  
DEPENDENT VARIABLE CP

SECTION (1) WING  
DEPENDENT VARIABLE CP

SECTION (1) WING  
DEPENDENT VARIABLE CP

## TABULATED SOURCE DATA - CASTS

(ADN29)

MACH ( 1 ) = .165    ALPHAS ( 3 ) = 4.965    RNL = 1.200    MACH = .165  
 SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

	21/8	.0000	.3340	.5200	.6630	.8730
X/C						
- .900	.0848	.0907	.1473	.1743	.0963	
- .750	.1044	.2304	.3218	.3864	.1319	
- .600	.0847	.2497	.1369	.1794	.1158	
- .450	-.11130	.0613	.2250	.2595	.0954	
- .300	.0226	.0825	.2204	.3563	.1277	
- .150	.2953	.2142	-.0474	.3582	.4510	
- .150	-.2411	-.8097	-.8655	-.0098	-.6432	
.300	-.3434	-.5300	-.5657	-.7400	-.6999	
.450	-.2247	-.5204	-.4546	-.5920	-.8107	
.600	-.1005	-.3533	-.3847	-.4717	-.6376	
.750	-.1508	-.1663	-.276	-.6390	-.6869	
.900	-.1634	-.2603	-.1639	-.1511	-.3246	

MACH ( 1 ) = .165    ALPHAS ( 4 ) = 9.965    RNL = 1.200    MACH = .165

## DEPENDENT VARIABLE CP

	21/8	.0000	.3340	.5200	.6630	.8730
X/C						
- .900	.1727	.2202	.2420	.1620	-.1499	
- .750	.2270	.3363	.3731	.4192	.1596	
- .600	.2201	.3651	.2223	.2441	.1566	
- .450	.0464	.1910	.2689	.3636	.1533	
- .300	.1865	.2622	.3511	.4730	.2248	
- .150	.3830	.3426	.0872	.4650	.4941	
.150	-.3097	-.9011	-.7778	-.1.2686	-.1.2758	
.300	-.3943	-.5861	-.6204	-.5578	-.1.1259	
.450	-.2584	-.5528	-.5422	-.6663	-.1.0163	
.600	-.1270	-.3633	-.4128	-.5017	-.6213	
.750	-.1694	-.0681	-.3937	-.5472	-.6803	
.900	-.1675	-.3121	-.1736	-.1411	-.5234	

MACH ( 1 ) = .165    ALPHAS ( 5 ) = 14.950    RNL = 1.200    MACH = .165

## DEPENDENT VARIABLE CP

	21/8	.0000	.3340	.5200	.6630	.8730
X/C						
- .900	.2132	.2736	.2521	.1563	-.1277	
- .750	.3033	.3927	.3964	.4308	.2083	
- .600	.3132	.4297	.2646	.2881	.2545	
- .450	.1587	.2914	.2087	.3956	.2500	
- .300	.3088	.3610	.2491	.5622	.3359	





CA37-B B16C5F1 J40 407E10 WING TOTAL SURFACE

(ADVMSO) ( 12 NOV 73 )

## REFERENCE DATA

$MACH =$	4.4120	ft.	$X_{REF} =$	43.3940 in.
$L_{REF} =$	19.2300	in.	$Y_{REF} =$	.0000 in.
$B_{REF} =$	37.9350	in.	$Z_{REF} =$	-.4050 in.
SCALE =	.0405			

MACH ( 1 ) = .165    ALPHAI ( 1 ) = -.04000    RNUL = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6650 .8730

x/c					
-.900	-.1216	-.0958	.0159	.0507	-.0630
-.750	-.1409	-.0708	.0636	.1530	.0774
-.600	-.1913	-.0700	-.0547	-.0141	.0440
-.450	-.4170	-.2466	-.0644	.0198	-.0873
-.300	-.4415	-.3220	-.1335	-.0055	-.0716
-.150	-.1046	-.2178	-.4560	-.1643	.1550
.150	-.0764	-.3124	-.3609	-.4370	-.0103
.300	-.2284	-.2701	-.4285	-.1262	-.5553
.450	-.1360	-.3527	-.3443	-.4442	-.5975
.600	-.0314	-.2939	-.3522	-.4145	-.4052
.750	-.0962	-.2813	-.4261	-.7583	-.3155
.900	-.1887	-.2876	-.1206	-.0469	-.1551

MACH ( 1 ) = .165    ALPHAI ( 2 ) = -.020    RNUL = 1.200    MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

x/c					
-.900	-.0581	-.0059	.0461	.1147	-.0799
-.750	-.0630	.0463	.1316	.2672	.1065
-.600	-.0436	.0821	.0425	.0782	.0824
-.450	-.2586	-.1128	.0427	.1201	.0227
-.300	-.1687	-.0976	.0114	.1779	.0546
-.150	-.1919	.0079	-.1794	.0902	.3479
.150	-.1304	-.4619	-.5255	-.7259	-.1978
.300	-.2920	-.3704	-.5101	-.2726	.6940
.450	-.1693	-.4132	-.4186	-.5198	-.6870
.600	-.0646	-.3365	-.3860	-.4542	-.5207
.750	-.1316	-.2843	-.3754	-.6947	-.3723
.900	-.1635	-.2628	-.1557	-.0948	-.1804



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TABULATED SOURCE DATA - CA370

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MACH ( 1 ) = .165    ALPH<sub>A</sub> ( 3 ) = 4.965    R<sub>L/L</sub> = 1.200    MACH = .165  
 SECTION ( 1 ) WING  
 21/8 .0000 .3340 .3200 .6630 .6730  
 DEPENDENT VARIABLE CP

X/C  
 -.900 .0268 .0077 .0461 .1113 -.1100  
 -.750 .0740 .2009 .2346 .3623 .1169  
 -.600 .1237 .2382 .0953 .1683 .1046  
 -.450 -.0600 .0690 .2273 .2683 .1008  
 -.300 .0726 .1242 .2329 .4000 .1195  
 -.150 .2915 .2298 .0900 .3931 .4623  
 .150 -.2320 .6556 .6763 -.0286 .6729  
 .300 -.5688 -.5609 -.5843 -.4642 -.6673  
 .450 -.2316 -.5413 -.4708 -.6075 -.6145  
 .600 -.1072 -.5628 -.5969 -.4922 -.6478  
 .750 -.1627 -.2639 -.3593 -.6908 -.5147  
 .900 -.1736 -.2561 -.1879 -.1492 -.3560

MACH ( 1 ) = .165    ALPH<sub>A</sub> ( 4 ) = 9.965    R<sub>L/L</sub> = 1.200    MACH = .165  
 SECTION ( 1 ) WING  
 21/8 .0000 .3340 .3200 .6630 .6730  
 DEPENDENT VARIABLE CP

X/C  
 -.900 .1096 .1750 .0949 .1442 -.1615  
 -.750 .1950 .2921 .2846 .3777 .1361  
 -.600 .2548 .3496 .1673 .2143 .1529  
 -.450 .0952 .1935 .2519 .3130 .1569  
 -.300 .2288 .2834 .3594 .5187 .2319  
 -.150 .3609 .3548 .1194 .5926 .4527  
 .150 -.2879 -.5582 -.7313 -.12746 -.13374  
 .300 -.4124 -.5912 -.6213 -.6517 -.1324  
 .450 -.2568 -.5832 -.5416 -.6824 -.0164  
 .600 -.1187 -.3831 -.4037 -.5139 -.6500  
 .750 -.1787 -.3021 -.3865 -.5574 -.7079  
 .900 -.1655 -.2627 -.1268 -.1417 -.5437

MACH ( 1 ) = .165    ALPH<sub>A</sub> ( 5 ) = 14.960    R<sub>L/L</sub> = 1.200    MACH = .165  
 SECTION ( 1 ) WING  
 21/8 .0000 .3340 .3200 .6630 .6730  
 DEPENDENT VARIABLE CP

X/C  
 -.900 .1573 .2194 .1553 .1260 -.1559  
 -.750 .2601 .3086 .2814 .3601 .1782  
 -.600 .3293 .3913 .2135 .2561 .2283  
 -.450 .1889 .2816 .1801 .1164 .2223  
 -.300 .3462 .3601 .2870 .5606 .3133



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## TABULATED SOURCE DATA - CASTE

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CAST-B 816CF1 J40 WATE10 WINE TOTAL SURFACE

(RDW30)

MACH (1) = .165 ALPH (5) = 14.980

## SECTION (1) WINE

## DEPENDENT VARIABLE CP

M/C	.0000	.3340	.5600	.6630	.6730
-1.150	.4611	.4376	.0735	.7236	.5000
-1.150	-.3312	-1.4602	-.6155	-1.1604	-2.4793
-1.300	-.4506	-.6016	-.7373	-.6304	-2.3783
-1.450	-.2965	-.7193	-.6706	-.6543	-2.1294
-1.600	-.1604	-.4357	-.5126	-.5125	-1.5408
-1.750	-.1879	-.3365	-.5491	-.4486	-1.4094
-1.900	-.1916	-.3100	-.5022	-.2319	-.6656

MACH (1) = .165 ALPH (5) = 19.980 RFL = 1.200 MACH = .165

## SECTION (1) WINE

## DEPENDENT VARIABLE CP

M/C	.0000	.3340	.5600	.6630	.6730
-1.900	.1643	.1929	.0363	.0025	-.1628
-1.750	-.2929	.3043	.2233	.3947	.1806
-1.600	.3907	.4275	.1197	.2167	.2502
-1.450	.2702	.3671	.1345	.3129	.2799
-1.300	.4464	.4861	.3015	.4110	.3520
-1.150	-.5138	.5246	.4711	.8277	.5510
-1.000	-.3792	1.9683	-.7508	-.4636	-.1406
-2000	-.5165	-.8403	-.8326	-.2263	-.1078
-450	-.4242	-.9149	-.9535	-.1669	-.1110
-4000	-.2821	-.6436	-.8774	-.1213	-.0702
-7500	-.3217	-.5004	-.6562	-.9276	-.9605
-9000	-.3071	-.4476	-.5194	-.6807	-.6526

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TABULATED SOURCE DATA - CAA578

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CA\*7-B 816C5F1 140 WING TOTAL SURFACE

## REFERENCE DATA

SALI	4.4120	.54111.	XMP1P	=	43.5940 IN.	BETA	=	.000
LREF	19.2300	IN.	YMP1P	=	.0000 IN.	1/B	=	1.500
BREF	37.9350	IN.	ZMP1P	=	-.0050 IN.	ELC1A	=	-10.000
SCALE	=	.0405						15.000

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

DEPENDENT VARIABLE CP

X/C

-.900	1.0420	.8594	1.2512	.6653	-.3496
-.750	.7552	.8439	.9808	.1677	-.0493
-.600	.5623	.7471	.3767	.4246	-.0101
-.450	.4442	.5906	.7266	.3665	-.0921
-.300	.4625	.4576	.3944	.2994	-.0427
-.150	.4777	.5353	.2116	.3791	.3255
.150	-.3289	-.7911	-.7578	-.1.2336	-.1.1617
.300	-.4486	-.6086	-.6295	-.6320	-.1.1023
.450	-.2774	-.5705	-.5235	-.5964	-.9864
.600	-.1323	-.3732	-.4044	-.4291	-.7948
.750	-.1755	-.3333	-.3936	-.2789	-.6493
.900	-.1778	-.1096	-.0509	-.1240	-.4723

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.005 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

DEPENDENT VARIABLE CP

-.900	.9927	.6853	1.1403	1.0248	-.3963
-.750	-.7814	.4529	.9338	.1037	.0191
-.600	.5969	.7639	.1921	.1619	.0417
-.450	.4757	.5917	.3519	.2312	.0289
-.300	.4647	.4515	.2634	.2125	.1492
-.150	.5477	.5544	-.2284	.1028	.4106
.150	-.3468	-.3409	-.8134	-.1.3158	-.6604
.300	-.4625	-.8261	-.7357	-.6874	-.1.9406
.450	-.2764	-.6756	-.6386	-.6281	-.1.6463
.600	-.1340	-.4064	-.5078	-.5007	-.1.1532
.750	-.1753	-.3061	-.5055	-.3982	-.1.3140
.900	-.1945	-.2053	-.2166	-.2585	-.8445

12 NOV 72  
12 NOV 72

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TABULATED SOURCE DATA - OA978

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(ADW31)

CA57-B B10C5F1 J40 W07E10 WING TOTAL SURFACE

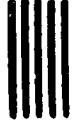
MACH ( 1 ) = .165 ALPHA ( 3 ) = 20.000 RNL = 1.200 MACH = .165

SECTION ( 11 ) WING

Z/ZB .0000 .3740 .3200 .6630 .8730

DEPENDENT VARIABLE CP

X/C	Coefficient	Coefficient	Coefficient	Coefficient	Coefficient
-.900	.9910	.9804	.9006	.8239	-.3528
-.750	.7951	.9504	.8798	.5548	.0536
-.600	.845C	.7770	.4200	.2349	.1321
-.450	.5274	.6252	.3745	.3170	.1690
-.300	.5501	.3252	.2629	.2636	.2174
-.150	.6151	.6104	-.1709	-.0901	.4730
.150	-.3679	-.6570	-.8026	-.6652	-.1.2424
.300	-.5382	-.9645	-.8205	-.4277	-.1.1774
.450	-.3843	-.8596	-.9794	-.1.2336	-.1.1259
.600	-.2574	-.6301	-.8495	-.1.1136	-.1.0603
.750	-.3128	-.4494	-.7916	-.8547	-.9620
.900	-.2830	-.3163	-.4912	-.7167	-.8228



DATE OF CCT 7.

TABULATED SOURCE DATA - CASE 7B

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CA57-E B16C5F1 JAO WTE10 WING TOTAL SURFACE (ROW32) ( 12 NOV 73 )

## REFERENCE DATA

SREF	=	4.4120 83 FT	XHARF	=	43.5940 IN.
LREF	=	19.2300 IN.	RHARF	=	.0000 IN.
BREF	=	37.9350 IN.	ZHARF	=	-.4050 IN.
SCALE	=	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = 9.990 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/B .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.6991	.6922	.7370	.6648	-.3319
-.750	.6865	.7600	.7772	.2292	-.0580
-.600	.5335	.7072	.3025	.3772	-.0441
-.450	.4374	.5756	.4834	.2648	-.1201
-.300	.5121	.4765	.4228	.3376	-.0849
-.150	.4837	.5397	.2164	.4754	.3208
.150	-.2949	-.7906	-.7315	-1.2195	-1.1752
.300	-.4347	-.6067	-.6084	-.6239	-1.1137
.450	-.2680	-.5576	-.5192	-.5886	-.9803
.600	-.1243	-.3583	-.3897	-.4128	-.8003
.750	-.1632	-.1765	-.3723	-.2781	-.6712
.900	-.1699	-.2019	-.0844	-.1191	-.5015

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.970 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/B .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.7604	.7137	.6151	.6724	-.4565
-.750	.7476	.7603	.7643	.3919	-.0327
-.600	.5761	.7228	.3621	.0825	-.0071
-.450	.4704	.5749	.3769	.2372	.0230
-.300	.5647	.4841	.3094	.2106	.1510
-.150	.5431	.5627	.1155	.1045	.4123
.150	-.3510	-.1.3108	-.7750	-1.3683	-2.6579
.300	-.4833	-.6167	-.7215	-.6781	-2.0458
.450	-.2782	-.6613	-.6169	-.5908	-1.6822
.600	-.1412	-.3791	-.4650	-.4950	-1.1419
.750	-.1192	-.2138	-.4553	-.3577	-1.2426
.900	-.1950	-.2437	-.2264	-.2236	-.8222

## PARAMETRIC DATA

BETA	=	.000
R/P	=	.039
E: E <sub>7</sub> , A <sub>7</sub>	=	15.000

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TABULATED SOURCE DATA - OA57B

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SECTION ( 1 ) WING	MACH ( 1 ) = .165	ALPHA ( 3 ) = 19.960	RN/L = 1.200	MACH = .165	TOTAL SURFACE		
					CA57-B	B16C5F1	J40 WATE10
21/0	.0000	.3340	.5200	.6630	.8730		
X/C							
-.900	.9061	.7177	.5338	.5031	-.3694		
-.750	.7739	.7760	.7376	.3936	.0116		
-.600	.6202	.7431	.4884	.2213	.1101		
-.450	.592	.6123	.4482	.1259	.1690		
-.300	.5968	.5483	.3215	.2745	.2731		
-.150	.6107	.5091	.1116	.0194	.4679		
.150	-.5844	-.6329	-.8137	-.6318	-.12389		
.300	-.5406	-.9337	-.6115	-.4034	-.11905		
.450	-.3871	-.8704	-.9911	-.12501	-.11424		
.600	-.2684	-.6437	-.8318	-.10005	-.0651		
.750	-.3265	-.3646	-.7933	-.6940	-.9713		
.900	-.2945	-.3635	-.5117	-.7165	-.8466		

(RDW32)

## CA57-B B16C5F1 J40 WATE10 WING TOTAL SURFACE

(RDYH33) (12 NOV 73)

## REFERENCE DATA

BREF = 4.4120 82.FT.  
 'OFF = 16 2200 IN.  
 BREF = 37.9350 IN.  
 SCALE = .0405

MACH (1) = .165 ALPHA (1) = 10.005 RNL = 1.200 MACH = 1.6

SECTION (1) WING

X/C .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C	.2901	.5178	.4116	.3181	.4931
-.900	.5288	.5000	.5453	.2684	-.3296
-.750	.5254	.5036	.3357	.2198	-.3355
-.600	.5940	.4308	.3976	.1788	-.1505
-.450	.5115	.4209	.3842	.4081	-.0198
-.300	.4722	.5217	.2013	.5519	.3541
-.150	.2899	.8004	-.7578	-.1.2571	-.1.2919
.300	.4363	.6148	-.6239	-.6399	-.1.1536
.450	.2642	.5638	-.5103	-.6207	-.1.0360
.600	.1270	.3524	-.4066	-.4381	-.8647
.750	.1693	.1228	-.3996	-.3988	-.7322
.900	.1708	.2624	-.0905	-.1240	-.5745

MACH (1) = .165 ALPHA (2) = 14.990 RNL = 1.200 MACH = 1.6

SECTION (1) WING

X/C	.3767	.5486	.3750	.2969	-.4483
-.900	.5468	.5667	.5358	.2628	-.766
-.750	.5446	.5854	.4004	.2735	-.5-2
-.600	.4523	.4378	.3930	.2666	-.1539
-.450	.5341	.4489	.2657	.2550	.1212
-.300	.5700	.5453	-.0002	.1435	.4517
-.150	.3183	.1.3103	-.7679	-.1.5-84	-.2.6405
.300	.4835	.6299	-.6954	-.6-12	-.2.5114
.450	.2770	.6797	-.6253	-.7-1	-.2.1357
.600	.1473	.3872	-.4732	-.5722	-.1.4579
.750	.1495	.1493	-.4104	-.5214	-.1.4525
.900	.1552	.3228	-.2547	-.2174	-.1.4525

## PARAMETRIC DATA

BETA = .0000 FTNP = 1.000  
 M,B = .039 PCFLAP = -.1R 000  
 ELEV, Y = 15.000

MACH (1) = .165    ALPHA (3) = 20.015    AN/L = 1.200    MACH = .165  
 SECTION (1)WING    037-B    B16CSF1 :40 WING10 WING TOTAL SURFACE

(RDVW33)

## SECTION (1)WING

037-B    B16CSF1 :40 WING10 WING TOTAL SURFACE

## DEPENDENT VARIABLE CP

21/8 .00000 .3340 .9200 .6630 .6730

X/C

-.900	.4764	.5761	.3533	.2311	-.3306
-.750	.6266	.6063	.5674	.2812	-.0202
-.600	.5919	.6309	.4606	.3475	.0792
-.450	.5133	.5018	.5219	.4271	.0377
-.300	.5972	.5206	.3615	.4179	.1367
-.150	.5944	.5937	-.0615	.0868	.4539
.150	-.3531	-.16063	-.7969	-.5016	-.12144
.300	-.5431	-.9926	-.8019	-.4282	-1.1795
.450	-.4019	-.9042	-1.0599	-1.2489	-1.1413
.600	-.2859	-.6627	-.9150	-1.1366	-1.0726
.750	-.3412	-.4904	-.7982	-.9057	-.9817
.900	-.3000	-.35562	-.5250	-.7172	-.8387



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TABULATED SOURCE DATA - OA378

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OA57-B B16C5F1 J40 WOTE10 WING TOTAL SURFACE

(RDW34) ( 12 NOV 73 )

## REFERENCE DATA

LREF = .4120 IN.  
LREF = 19.2300 IN.  
BREF = 37.9350 IN.  
SCALE = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .0000 -.3340 .5200 .6630 .9730  
DEPENDENT VARIABLE CP

X/C

-.900	-.0962	-.0438	.1086	.0859	-.0354
-.750	-.1152	.0423	.1766	.2203	.0892
-.600	-.2644	-.0589	.0298	.0127	.0594
-.450	-.5342	-.3442	-.1474	-.0301	-.0935
-.300	-.4938	-.4902	-.1536	-.0330	-.0741
-.150	-.0648	-.2846	-.5874	-.2707	.0969
.150	-.0723	-.1668	-.3060	-.3977	.0368
.300	-.2184	-.2735	-.4123	-.1220	-.5167
.450	-.1362	-.3483	-.4076	-.4154	-.5771
.600	-.0324	-.2897	-.3143	-.3669	-.4037
.750	-.0872	-.1746	-.4185	-.6201	-.2613
.900	-.1778	-.1177	-.0321	-.0091	-.1075

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C .0000 -.3340 .5200 .6630 .9730  
DEPENDENT VARIABLE CP

X/C

-.900	-.0654	-.0131	.1068	.1505	-.0425
-.750	-.0662	.1080	.2906	.3150	.1245
-.600	-.1609	.0450	.0579	.0688	.1927
-.450	-.4287	-.2276	-.0473	.0861	.0035
-.300	-.3447	-.3127	-.0761	.1082	.0368
-.150	.1364	-.1077	-.3629	-.0395	.3097
.150	-.1086	-.2765	-.4708	-.6695	-.1982
.300	-.2726	-.3566	-.4911	-.2119	-.6738
.450	-.1769	-.4059	-.4528	-.4951	-.6745
.600	-.0600	-.2988	-.3354	-.4137	-.5157
.750	-.1163	-.1158	-.3812	-.3146	-.3190
.900	-.1711	-.1971	-.0790	-.1441	-.1655

## TABULATED SOURCE DATA - OA378

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.995 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

.0000 .3340 .5200 .6630 .8730

X/C

X/C	.-900	-.0132	-.0073	.0397	.2333	.0602
	-.750	.0033	.1567	.2946	.4167	.1570
	-.600	-.0466	.1591	.0634	.1780	.1468
	-.450	-.2878	-.0721	.1153	.2136	.1054
	-.300	-.1674	-.1040	.0859	.3047	.1443
	-.150	.2255	.0616	-.1613	.2116	.4240
	.150	-.1935	-.4576	-.6137	.9675	.5644
	.300	-.3476	-.4939	-.5648	.4279	.6352
	.450	-.2177	-.4900	-.4850	.5669	.7847
	.600	-.0838	-.3447	-.3550	.4576	.6203
	.750	-.1390	-.0935	-.3278	.5935	.4080
	.900	-.1648	-.2919	-.1290	.0980	-.2807

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.995 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

.0000 .3340 .5200 .6630 .8730

X/C

X/C	-.900	-.0046	.0767	.0399	.3047	-.1064
	-.750	.0707	.2269	.3095	.4970	.1800
	-.600	.01606	.2290	.1152	.2692	.1617
	-.450	-.1639	.0568	.1992	.3237	.1661
	-.300	-.0193	.0376	.2871	.4258	.2154
	-.150	.3137	.2347	-.0323	.4204	.5038
	.150	-.2787	-.7598	-.6828	-.1291	-.1065
	.300	-.4020	-.5940	-.6243	-.6072	-.0227
	.450	-.2446	-.5325	-.5333	-.6496	-.9447
	.600	-.1051	-.3602	-.3874	-.5101	-.7497
	.750	-.1589	-.2350	-.3357	-.6415	-.6024
	.900	-.1645	-.2897	-.1350	-.0665	-.5367

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.975 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

.0000 .3340 .5200 .6630 .8730

X/C

X/C	-.900	.0070	.1192	.0683	.2853	-.1180
	-.750	.1345	.2873	.3075	.5502	.2074
	-.600	-.1447	.2966	.1019	.3595	.2463
	-.450	-.0748	.1244	.2100	.4347	.2398
	-.300	.1132	.1954	.2289	.5362	.3210



(ADWNA4)

CALC-EF1 JAO DATE16 WING TOTAL SURFACE

(ADN134)

MACH ( 1) = .165 ALPHA ( 5) = 14.975

SECTION ( 1) WING DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
- .150	.3953	.3330	.0230	.5063	.4944
.150	-.3299	-1.2777	-.6392	-1.3944	-1.9341
.300	-.4390	-.7651	-.6800	-.7492	-1.3714
.450	.2881	-.7032	-.5691	-.6701	-1.2337
.600	-.1237	-.4146	-.4361	-.4633	-1.1403
.750	-.1488	-.2739	-.4316	-.4825	-1.0735
.900	-.2004	-.3304	-.1862	-.1240	-.8808

MACH ( 1) = .165 ALPHA ( 6) = 19.990 RNL = 1.200 MACH = 1.157

SECTION ( 1) WING DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
- .900	.0164	.1237	.0869	.1186	-.1571
-.750	.1762	.3015	.3758	.5457	.2297
-.600	.2282	.3316	.1090	.3646	.2929
-.450	.0411	.2707	-.0754	.4582	.3012
-.300	.2363	.3324	.1639	.6233	.3935
-.150	.4630	.4343	.0997	.7101	.5521
.150	-.3454	-.6129	-.7908	-1.2155	-1.0981
.300	-.5194	-.9704	-.7198	-1.2011	-1.0835
.450	-.3347	-.8563	-.8686	-1.2024	-1.0770
.600	-.1068	-.6030	-.8157	-1.0886	-1.0886
.750	-.2293	-.5084	-.8110	-.8850	-.9936
.900	-.2738	-.4507	-.5007	-.7311	-.9057

## TABULATED SOURCE DATA - CASTE

(REV035) (12 NOV 73)

## REFERENCE DATA

BALI = 4,4120 SQ.FT. XMAP = 43,3940 IN.  
 LREF = 10,2300 IN. YMAP = .0000 IN.  
 BREF = 37,9350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH (1) = .165 ALPHA (1) = -.3993 RNL = 1,200 MACH = .165

## SECTION (1)WING

2/18 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0751	-.0455	.0531	.0605	-.0395
-.750	-.0793	.0083	.1722	.2120	.0690
-.600	-.1925	.0550	.0300	.0010	.0593
-.450	-.4564	-.2381	-.0903	-.0151	-.0796
-.300	-.4505	-.4167	-.1015	.0043	-.0614
-.150	.0834	-.2316	.5122	-.2036	.1310
.150	-.0579	-.1637	.3196	-.4050	.0205
.300	-.2138	-.2640	-.4087	-.0900	-.5224
.450	-.1264	-.3445	-.4074	-.4061	-.5743
.600	-.0170	-.2786	-.3157	-.3709	-.3927
.750	-.0760	-.1947	-.4018	-.6384	-.2640
.900	-.1726	-.2061	-.0375	-.0116	-.1074

MACH (1) = .165 ALPHA (1) = -.020 RNL = 1,200 MACH = .165

## SECTION (1)WING

2/18 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0544	-.0043	.0510	.1333	-.0494
-.750	-.0503	.0626	.2124	.3062	.1259
-.600	-.1209	.0313	.0431	.C778	.1031
-.450	-.3619	-.1632	-.0043	.0946	.0120
-.300	-.3060	-.2500	-.0476	.1551	.0496
-.150	.1408	-.0766	-.2904	.0174	.3251
.150	-.1098	-.2814	-.4827	-.6740	-.1767
.300	-.2738	-.3592	-.4937	-.2418	-.6451
.450	-.1712	-.4134	-.4629	-.4922	-.2737
.600	-.0632	-.3070	-.3437	-.4106	-.4981
.750	-.1116	-.2000	-.3685	-.6025	-.5187
.900	-.1651	-.2216	-.0890	-.0553	-.1702



## CASTE-E B16C5F1 J40 WTE10 WING TOTAL SURFACE

(RDWMA5)

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (3) = 4.900 \quad \text{RNL} = 1.200 \quad \text{MACH} = .165$$

## SECTION 1: WING

DEFFICIENT VARIABLE CP

2/18 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0164	.0168	-.0061	.1868	-.0677
-.750	.0395	.1492	.2612	.3924	.1526
-.600	-.0057	.1593	.0647	.1851	.1694
-.450	-.2262	-.0130	.1538	.2248	.1030
-.300	-.1291	-.0545	.1404	.3440	.1401
-.150	.2330	.1241	-.1261	.2975	.4472
.150	-.2009	-.1613	-.6191	-.9628	-.5667
.300	-.3500	-.5232	-.5763	-.4168	.8390
.450	-.2175	-.5057	-.4857	-.5678	.7958
.600	-.0883	-.3481	-.3492	-.4468	.6050
.750	-.1411	-.2224	-.2854	-.6107	.4226
.900	-.1643	-.2465	-.1779	-.1025	.2957

MACH (1) = .165 ALPHA (4) = 9.985 RNL = 1.200 MACH = .165

SECTION 1: WING

DEFFICIENT VARIABLE CP

2/18 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0089	.0647	.0081	.1643	-.1338
-.750	.0614	.2037	.2508	.4319	.1536
-.600	.0762	.2264	.1057	.2173	.1500
-.450	-.1370	.0742	.2029	.3068	.1605
-.300	-.0021	.0930	.2998	.4474	.2002
-.150	.3040	.2335	.0010	.4454	.4929
.150	-.2705	-.7781	-.7285	-.2498	-1.1637
.300	-.4258	-.5967	-.6547	-.6107	-1.0559
.450	-.2768	-.5637	-.5575	-.5683	-.9666
.600	-.1229	-.3702	-.4044	-.5245	-.7806
.750	-.1634	-.2600	-.3559	-.6489	-.6343
.900	-.1932	-.2872	-.1650	-.0947	-.5622

MACH (1) = .165 ALPHA (5) = 14.970 RNL = 1.200 MACH = .165

## SECTION 1: WING

DEFFICIENT VARIABLE CP

2/18 .0000 .3340 .5200 .6630 .8730
X/C
-.900 .0416 .1151 .0763 .1432 -.1417
-.750 .1538 .2649 .2560 .4492 .2514
-.600 .1765 .3248 .1440 .479 .2479
-.450 .2246 .1623 .1616 .2401 .2401
-.300 .1555 .2122 .2122 .3133 .3133

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TABULATED SOURCE DATA - QA570

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MACH ( 1 ) = .165

ALPHA ( 5 ) = 14.970

(ADWW35)

SECTION ( 1 ) WING

X/C

X/C	.0000	.3340	.6620	.6630	.8750
-.150	-.4002	.3462	-.1280	.4782	.5028
.150	-.3031	-.1260	-.8130	-.14030	-.23841
.300	-.4657	.6145	.7102	-.7524	-.14244
.450	-.2732	-.7104	-.6343	-.6745	-.11637
.600	-.1216	-.4181	-.4565	-.4716	-.10245
.750	-.1564	-.2971	-.4567	-.4087	-.9389
.900	-.1656	-.3261	-.2018	-.2239	-.7870

MACH ( 1 ) = .165

ALPHA ( 6 ) = 19.980

RNL = 1.2000 MACH = .165

SECTION ( 1 ) WING

X/C

X/C	.0000	.3340	.6620	.6630	.8750
-.900	.0279	.0876	.0259	-.1308	-.1598
-.750	.1842	.2619	.5014	.3901	.2228
-.600	.2462	.3282	.1157	.2614	.2883
-.450	.0722	.2691	-.0177	.3607	.2967
-.300	.2763	.3549	.1766	.6100	.3893
-.150	.4653	.4428	.2191	.7507	.5465
.150	-.3508	-.1384	-.7790	-.1203	-.1.0644
.300	-.5284	-.9586	-.7177	-.1.1286	-.1.0506
.450	-.3361	-.8340	-.6596	-.1.1591	-.1.0573
.600	-.1950	-.6590	-.8195	-.1.0664	-.1.0536
.750	-.2545	-.5299	-.8145	-.9050	-.9765
.900	-.2658	-.4893	-.4322	-.7452	-.8985

MACH ( 1 ) = .165

ALPHA ( 5 ) = 14.970

(ADWW35)

SECTION ( 1 ) WING

X/C

X/C	.0000	.3340	.6620	.6630	.8750
-.150	-.4002	.3462	-.1280	.4782	.5028
.150	-.3031	-.1260	-.8130	-.14030	-.23841
.300	-.4657	.6145	.7102	-.7524	-.14244
.450	-.2732	-.7104	-.6343	-.6745	-.11637
.600	-.1216	-.4181	-.4565	-.4716	-.10245
.750	-.1564	-.2971	-.4567	-.4087	-.9389
.900	-.1656	-.3261	-.2018	-.2239	-.7870

MACH ( 1 ) = .165

ALPHA ( 6 ) = 19.980

RNL = 1.2000 MACH = .165

X/C

X/C	.0000	.3340	.6620	.6630	.8750
-.900	.0279	.0876	.0259	-.1308	-.1598
-.750	.1842	.2619	.5014	.3901	.2228
-.600	.2462	.3282	.1157	.2614	.2883
-.450	.0722	.2691	-.0177	.3607	.2967
-.300	.2763	.3549	.1766	.6100	.3893
-.150	.4653	.4428	.2191	.7507	.5465
.150	-.3508	-.1384	-.7790	-.1.203	-.1.0644
.300	-.5284	-.9586	-.7177	-.1.1286	-.1.0506
.450	-.3361	-.8340	-.6596	-.1.1591	-.1.0573
.600	-.1950	-.6590	-.8195	-.1.0664	-.1.0536
.750	-.2545	-.5299	-.8145	-.9050	-.9765
.900	-.2658	-.4893	-.4322	-.7452	-.8985

## CAST-E B16C5F1 JAO WATE10 WING TOTAL SURFACE

(ACMWS6) (12 NOV 73)

## REFERENCE DATA

SUPER	A	4.4120 83.FT.	WHP	=	43.5940 IN	
A/C	16.250 IN.	WHP	=	.0000 IN		
BREF	37.950 IN.	WHP	=	-4050 IN		
SCALE	.005					
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	-3.995	RNL =	1.200	MACH = .165
SECTION ( 1 )WING				DEPENDENT VARIABLE CP		
21/8	.0000	.3340	.5200	.6630	.8730	
X/C						
- .900	- .1074	- .0467	- .0091	.0723	- .0562	
- .750	- .0976	- .0127	.0891	.1635	.0811	
- .600	- .1202	- .0322	- .0249	- .0031	.0466	
- .450	- .3512	- .1806	- .0399	- .0020	- .0766	
- .300	- .5685	- .3655	- .0790	.0450	- .0597	
- .150	- .0859	- .1698	- .4146	- .1426	.1575	
.150	- .0645	- .1695	- .3273	- .4118	.0040	
.300	- .2274	- .2751	- .4084	- .0732	.5332	
.450	- .1353	- .3544	- .4201	- .4171	.5486	
.600	- .0253	- .2867	- .3178	- .779	- .4032	
.750	- .0861	- .2225	- .4016	- .6613	- .2737	
.900	- .1936	- .2321	- .0332	- .0248	- .1197	

## PARAMETRIC DATA

BETA	=	.000	PINP	=	1.000	
H/B	=	.285	EDFLAP	=	-16.000	
ELEVN	=	15.000				
MACH ( 1 ) =	.165	ALPHA ( 2 ) =	- .010	RNL =	1.200	MACH = .165
SECTION ( 1 )WING				DEPENDENT VARIABLE CP		
21/8	.0000	.3340	.5200	.6630	.8730	
X/C						
- .900	- .0824	- .0098	- .0002	.0880	- .0560	
- .750	- .0517	.0504	.1319	.2475	.1121	
- .600	- .0571	.0556	.0313	.0560	.0458	
- .450	- .2717	- .1228	.0282	.0956	.0132	
- .300	- .2580	- .2027	- .0057	.1747	.0469	
- .150	- .1417	- .0554	- .2043	.6606	.3339	
.150	- .1142	- .2661	- .4912	.6870	.1500	
.300	- .2864	- .3723	- .0558	- .2374	.6775	
.450	- .1607	- .4225	- .1767	.4973	.6724	
.600	- .0663	- .3253	- .3472	.4159	.4995	
.750	- .1223	- .2577	- .3661	.6150	.3295	
.900	- .1603	- .1770	- .0873	- .6642	- .1765	

## CAA97-B B16C3F1 J40 WATE10 WING TOTAL SURFACE (ADW036)

MACH ( 1 ) = .165 ALPH ( 3 ) = 5.005 RNL = 1.200 MACH = .165

## SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
- .900	-.0356	.0226	-.0363	.1353	-.0962
- .750	.0193	.1271	.2032	.3397	.1323
- .600	.0427	.1482	.0390	.1500	.1237
- .450	-.1604	-.0036	.1751	.2428	.0635
- .300	-.0743	-.0323	.1690	.3791	.1135
- .150	.2790	.1359	.0646	.3293	.4581
.150	-.2143	-.4615	-.6428	-.9136	-.5681
.300	-.3585	-.5136	-.5921	-.4150	-.6551
.450	-.2243	-.5096	-.4894	-.5736	-.8075
.600	-.0942	-.3392	-.3593	-.4521	-.6004
.750	-.1422	-.3646	-.2808	-.6175	-.4362
.900	-.1734	-.1720	-.1766	-.1085	-.3292

MACH ( 1 ) = .165 ALPH ( 4 ) = 9.980 RNL = 1.200 MACH = .165

## SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
- .900	.0023	.0680	.0076	.1167	.1351
- .750	.0787	.1944	.1760	.3709	.1526
- .600	.1280	.2348	.0346	.2136	.1444
- .450	-.0583	.0089	.2109	.3304	.1537
- .300	.0613	.1138	.3225	.5026	.2049
- .150	.3094	.2577	.0388	.5205	.5168
.150	-.2992	-.7663	-.7037	-.1.2421	-.1.1722
.300	-.4270	-.6032	-.6447	-.5969	-.1.0590
.450	-.2536	-.3773	-.3635	-.6621	-.9581
.600	-.1154	-.3778	-.4026	-.3150	-.7679
.750	-.1643	-.3014	-.3646	-.6122	-.6556
.900	-.1750	-.2224	-.1330	-.0956	-.5522

MACH ( 1 ) = .165 ALPH ( 5 ) = 15.005 RNL = 1.200 MACH = .165

## SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
- .900	.0270	.0851	.0613	.0414	-.1436
- .750	.1334	.2184	.2158	.3402	.1679
- .600	.2041	.2890	.1219	.2326	.2231
- .450	.0222	.1785	.0515	.3207	.2253
- .300	.1924	.2791	.2618	.5696	.3037

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.005

GA57-E B16CF1 JAO WTE10 WING TOTAL SURFACE  
(RDWWS)

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

MACH ( 1 )	RN/L	ALPHA ( 5 )	RN/L	MACH = .165
21/B	.0000	.3340	.5200	.6630 .8730
X/C				
-.150	.3981	.3516	.0710	.6733 .4908
.150	-.3778	-.1528	-.8139	-.14506 -2.6341
.300	-.5087	-.8129	-.6832	-.7866 -1.6190
.450	-.2917	-.7045	-.6271	-.6841 -1.3115
.600	-.1282	-.4248	-.45 0	-.4704 -1.0029
.750	-.1613	-.3292	-.4495	-.4461 -.9572
.900	-.1839	-.2570	-.1202	-.1708 -.8258

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.985 RN/L = 1.200 MACH = .165

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

MACH ( 1 )	RN/L	ALPHA ( 6 )	RN/L	MACH = .165
21/B	.0000	.3340	.5200	.6630 .8730
X/C				
-.900	.0130	.0677	-.0424	-.0447 -.1796
-.750	.1435	.1957	.1728	.3311 .1822
-.600	.2555	.3003	.1194	.2450 .2387
-.450	.0959	.2603	-.0657	.3173 .2540
-.300	-.3008	.3442	.1470	.5921 .3375
-.150	.4758	.4398	.2388	.7603 .5172
.150	-.3712	-.14946	-.7278	-.2317 -.9884
.300	-.5807	-.9538	-.6965	-.1.060 -1.0059
.450	-.3784	-.6504	-.6546	-.1.0691 -.9996
.600	-.1797	-.6142	-.8222	-.9991 -1.0072
.7*	-.2580	-.5654	-.6282	-.9165 -.9340
.9C..	.2878	-.4264	-.4767	-.7942 -.8525

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TABULATED SOURCE DATA - QAS70

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(RDW37) (112 NOV 73)

## REFERENCE DATA

BREF	=	4,4120 IN. FT.	XMAP	=	43,1940 IN.
LREF	=	18,2300 IN.	YMAP	=	.0000 IN.
BREF	=	57,9350 IN.	ZMAP	=	-.4050 IN.
SCALE	=	.0403			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.995 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

2 / B .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	-.2663	-.12648	-.3323	-.3763	-.3718
-.750	-.3436	-.4715	-.5684	-.6232	-.3969
-.600	-.4165	-.3956	-.4458	-.4368	-.4438
-.450	-.7284	-.6188	-.3639	-.3513	-.4761
-.300	-.6165	-.6983	-.3717	-.3132	-.4077
-.150	.0069	-.4371	-.8215	-.5025	-.2604
.150	-.0026	-.0553	-.1549	-.1686	-.1198
.300	-.1163	-.1220	-.1986	-.1590	-.2128
.450	-.0172	-.1384	-.1451	-.0729	-.1154
.600	.0824	.0268	.0384	.0720	.0605
.750	.0268	.2108	.2832	.3763	.1899
.900	-.0623	.2215	.2330	.2623	.0926

MACH ( 1 ) = .165 ALPHA ( 2 ) = .000 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C					
-.900	-.2919	-.2302	-.2627	-.4480	-.3016
-.750	-.3417	-.5069	-.6283	-.6731	-.3177
-.600	-.3733	-.3416	-.4048	-.3814	-.3527
-.450	-.6772	-.5614	-.3121	-.2534	-.3719
-.300	-.4875	-.5753	-.2313	-.1834	-.2649
-.150	.0666	-.2770	-.6704	-.3134	.0034
.150	-.0456	-.1620	-.3023	-.3684	-.2914
.300	-.1560	-.2005	-.2851	-.2759	-.3038
.450	-.0467	-.1807	-.1861	-.1268	-.1474
.600	.0431	.0067	.0204	.0633	.0758
.750	.0020	.1766	.2611	.3282	.2087
.900	-.0744	.2282	.2219	.2585	.1324

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TABULATED SOURCE DATA - Q378

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CARTESIAN COORDINATES

SECTION ( 1 ) WING

(RDW37)

MACH ( 1 ) = .165	ALPHA ( 3 ) = 5.000	RNL = 1.200	MACH = .165
.00000	.3340	.5200	.6630 .8730

DEFINITION VARIABLE CP

X/C				
-.900	-.2681	-.1826	-.2193	-.3060
-.750	-.2738	-.4387	-.6757	-.6992
-.600	-.2467	-.2261	-.3978	-.2663
-.450	-.5317	-.3915	-.1714	-.1018
-.300	-.2970	-.3523	-.1875	.0119
-.150	-.1609	-.0525	-.4084	.0046
.150	-.1177	-.3014	-.4545	-.6632
.300	-.2365	-.3193	-.3586	-.4149
.450	-.1025	-.2436	-.12112	-.1870
.600	.0166	-.0190	-.0243	.0163
.750	-.0195	.1240	.1090	.2471
.900	-.0844	.1737	.1271	.2222

MACH ( 1 ) = .165	ALPHA ( 4 ) = 9.985	RNL = 1.200	MACH = .165
.00000	.3340	.5200	.6630 .8730

DEFINITION VARIABLE CP

X/C				
-.900	-.2370	-.1551	-.2435	-.3242
-.750	-.1960	-.3784	-.7412	-.7016
-.600	-.1344	-.1293	-.3703	-.1403
-.450	-.4078	-.2220	-.0294	.0238
-.300	-.1244	-.1351	.1091	.1796
-.150	-.2481	-.1160	-.2426	.2609
.150	-.2184	-.6230	-.5337	-.6886
.300	-.2750	-.4591	-.4231	-.5246
.450	-.1346	-.3381	-.2823	-.2671
.600	-.0159	-.1073	-.0898	-.0482
.750	-.0381	.0816	.0537	.2122
.900	-.0998	.1553	.1307	.2228

MACH ( 1 ) = .165	ALPHA ( 5 ) = 15.000	RNL = 1.200	MACH = .165
.00000	.3340	.5200	.6630 .8730

DEFINITION VARIABLE CP

X/C				
-.900	-.1913	-.1344	-.2594	-.3594
-.750	-.1217	-.3200	-.7696	-.6987
-.600	-.0243	-.0333	-.3493	-.0214
-.450	-.2766	-.0938	-.0057	.1157
-.300	.0235	.0368	.2591	.1133

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TABULATED SOURCE DATA - CA57B

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MACH ( 1 ) = .165

ALPHA ( 5 ) = 15.000

(ADW57)

SECTICN ( 1 )WING

DEPENDENT VARIABLE CP

21/P .0000 .3340 .5200 .6630 .8730

X/C

-.150	.3425	.2355	-.0996	.4327	.4642
-.150	-.2471	-.1239	-.6444	-.12650	-.12143
.300	-.3316	-.6026	-.4817	-.6700	.6169
.450	-.1567	-.4579	-.3320	-.3468	-.4546
.600	-.0212	-.1607	-.1326	-.1141	.6257
.750	-.0359	.0593	.0595	.2819	-.3014
.900	-.1144	.1903	.1146	.2225	-.0946

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.995 RNL = 1.200 MACH = .165

SECTICN ( 1 )WING

DEPENDENT VARIABLE CP

21/P .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.1762	-.1828	-.3232	-.4918	-.4465
-.750	-.0641	-.3048	-.7927	-.7622	.6983
-.600	-.0706	.0387	-.2813	.0345	-.0183
-.450	-.1780	.0315	-.1400	.2759	.0501
-.300	-.1594	.1533	-.1118	.4565	.1973
-.150	.4362	.3424	-.1759	.5088	.5369
.150	-.2770	-.14100	-.6834	-.1.0722	-.9373
.300	-.3857	-.7603	-.5190	-.1.1137	-.8866
.450	-.2059	-.5655	-.5259	-.9018	-.8526
.600	-.0776	-.2902	-.3741	-.6577	-.8149
.750	-.1262	-.0700	-.1352	-.1345	.6358
.900	-.1790	.0848	-.0440	-.1146	-.5692

CAS7-B B16C5F1 140 WTE10 WING TOTAL SURFACE

(RDW36) (12 NOV 73)

## REFERENCE DATA

$x_{ref}$	=	4.4120 IN.	$x_{mp}$	=	43.5940 IN.
LREF	=	19.2300 IN.	$r_{mp}$	=	.0000 IN.
BREF	=	37.9350 IN.	$z_{mp}$	=	-.4050 IN.
SCALE	=	.0405			
MACH (1) =	.165	ALPHA (1) =	-.3.990	RNL =	1.200 MACH = .165

## SECTION 1 (1) WING

## DEPENDENT VARIABLE CP

$x/c$					
-.900	-.2583	-.2693	-.3251	-.3757	-.3817
-.750	-.3493	-.4809	-.5750	-.5942	-.4052
-.600	-.4349	-.3886	-.4398	-.4206	-.4287
-.450	-.7176	-.6211	-.3703	-.3581	-.4845
-.300	-.6147	-.7035	-.3932	-.3105	-.4117
-.150	-.0051	-.4401	-.7985	-.4796	-.2427
.150	-.0175	-.0645	-.1520	-.1659	-.1180
.300	-.1165	-.1304	-.2112	-.1681	-.2207
.450	-.0216	-.1425	-.1523	-.0824	-.1255
.600	.0666	.0160	.0365	.0746	.0636
.750	.0204	.2014	.2730	.3699	.1829
.900	-.0904	.2169	.2199	.2517	.0842
MACH (1) =	.165	ALPHA (2) =	.025	RNL = 1.200 MACH = .165	

## SECTION 1 (1) WING

## DEPENDENT VARIABLE CP

$x/c$					
-.900	-.12551	-.22262	-.2576	-.4336	-.2925
-.750	-.2957	-.4425	-.5424	-.6131	-.3558
-.600	-.3338	-.2971	-.3621	-.3263	-.3134
-.450	-.5948	-.8846	-.2589	-.2199	-.3359
-.300	-.4502	-.5091	-.2845	-.1406	-.2360
-.150	.0718	-.2349	-.5793	-.2205	.0615
.150	-.0539	-.1742	-.3161	-.3851	-.5227
.300	-.1709	-.2218	-.3653	-.2921	-.3143
.450	-.0527	-.1917	-.2015	-.1380	-.1527
.600	.0298	-.0063	.0109	.0561	.0690
.750	-.0111	.1512	.2346	.7153	.2065
.900	-.0777	.2145	.2020	.2433	.1358



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TABULATED SOURCE DATA - GA57B

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GA57-B 816C5F1 J40 WTE16 WING TOTAL SURFACE

(RDW38)

MACH ( 1) = .165 ALPHA ( 5) = 14.985

SECTION 11 WING

DEFINITION VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C					
-.150	.3417	.2537	-.0361	.4743	.4952
.150	-.2429	-1.0590	-.6532	-1.0729	-1.2191
.300	-.3435	-.6114	-.5086	-.6422	-.6414
.450	-.1573	-.4506	-.3295	-.3540	-.5211
.600	-.0	.5	-.1655	-.1338	-.1147
.750	-.0458	.0501	.0534	.2657	-.3166
.900	-.1315	.1622	.1029	.2056	-.1183

MACH ( 1) = .165 ALPHA ( 6) = 19.990 RNL = 1.200 MACH = .155

SECTION 11 WING

DEFINITION VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C					
-.900	-.1261	-.1582	-.2736	-.4531	-.4394
-.750	-.0198	-.2420	-.6282	-.6854	-.6886
-.600	.1042	.0630	-.1570	.0471	.0378
-.450	-.0809	.0687	-.1910	.2588	.0555
-.300	.2008	.1969	.1582	.5033	.2063
-.150	.4396	.3644	-.1114	.6084	.5432
.150	-.2695	-.14081	-.6697	-.10254	-.9421
.300	-.3913	-.7666	-.5441	-.11438	-.9106
.450	-.2015	-.5732	-.5564	-.9101	-.8738
.600	-.0694	-.3139	-.4076	-.7110	-.8098
.750	-.1364	-.08	-.1771	-.1824	-.6394
.900	-.1655	.0	-.0697	-.1490	-.5617

REFERENCE DATA		CAST-B B16C-SF1 J40 WSTE10 WING TOTAL SURFACE			PARAMETRIC DATA		
LEEF =	.4.4120 SQ FT.	XLEAP =	.43.594G IN.		BETA =	.000	PTHP = 1.000
LEEF =	.10.2500 IN.	YLEAP =	.0000 IN.		H/B =	.286	BDFLAP = -10.000
LEEF =	.37.9350 IN.	ZLEAP =	-.4050 IN.		ELEVON =	-15.000	
SCALE =	.0405						
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	-.3.990	RNL =	1.200	MACH =	.165
SECTION ( 1 ) WING							
DEPENDENT VARIABLE CP							
21/8	.0000	.3340	.5200	.6630	.8730		
X/C							
-.900	-.26668	-.2763	-.2680	-.2684	-.3483		
-.750	-.3114	-.3369	-.4244	-.5157	-.3626		
-.600	-.3204	-.3203	-.3616	-.3563	-.4010		
-.450	-.5885	-.5042	-.2779	-.3053	-.4495		
-.300	-.5168	-.6084	-.3188	-.2121	-.3792		
-.150	.0150	.3747	-.6972	-.4198	-.1668		
.150	-.0103	-.0671	-.1715	-.1687	-.1397		
.300	-.1220	-.1350	-.2176	-.1855	-.2354		
.450	-.0167	-.1563	-.1445	-.0841	-.1267		
.600	.0660	.0128	.0260	.0639	.0588		
.750	.0150	.1976	.2693	.3653	.1633		
.900	-.1025	.2116	.2230	.2510	.0957		
MACH ( 1 ) =	.165	ALPHA ( 2 ) =	.000	RNL =	1.200	MACH =	.165
SECTION ( 1 ) WING							
DEPENDENT VARIABLE CP							
21/8	.0000	.3340	.5200	.6630	.8730		
X/C							
-.900	-.2457	-.2282	-.2153	-.3137	-.2812		
-.750	-.2711	-.3456	-.4437	-.5485	-.5392		
-.600	-.2584	-.2531	-.2991	-.2795	-.2963		
-.450	-.4969	-.4143	-.2050	-.1981	-.3286		
-.300	-.3792	-.4416	-.2301	-.0617	-.2270		
-.150	.0769	-.1920	-.5078	-.1722	.0697		
.150	-.0548	-.1781	-.3278	-.3959	-.3422		
.300	-.1764	-.2221	-.3122	-.3083	-.3308		
.450	-.0533	-.1956	-.2063	-.1500	-.1613		
.600	.0253	-.0110	.0035	.0491	.0622		
.750	-.0110	.1487	.2257	.5034	.1936		
.900	-.0965	.2078	.1961	.2285	.1363		

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## CA57-B B16CSF1 JAO WATE10 WING TOTAL SURFACE (ADYNS)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.000

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
-1.90	.3384	.2568	-.0127	.5431	.5113	
-1.50	-.2221	-1.0693	-.6362	-1.0783	-1.2263	
-1.00	-.3471	-.8094	-.4912	-.6476	-.6612	
-0.50	-.1381	-.4699	-.3440	-.3631	-.5561	
.000	-.0303	-.1750	-.1349	-.1182	-.6788	
.750	-.0502	.0436	.0477	.2619	-.3031	
.900	-.1347	.1663	.1002	.2021	-.1262	

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.995 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
-1.900	-.1107	-.1558	-.2380	-.3441	-.4615	
-1.750	-.0204	-.2103	-.4433	-.4684	-.6675	
-1.600	.1146	.0608	-.0329	.0921	.0485	
-1.450	-.0507	.0536	-.2932	.1990	.0307	
-1.300	.2210	.1913	.2561	.5570	.1976	
-1.150	.4270	.3593	.0526	.6771	.5298	
-1.000	-.2622	-1.3916	-.6398	-.9685	-.9297	
-0.800	-.4002	-.7679	-.5028	-.10561	-.6682	
-0.500	-.2073	-.5891	-.5583	-.9772	-.8632	
-0.000	-.0944	-.3399	-.4324	-.7942	-.8173	
.750	-.1690	-.1030	-.1608	-.2736	-.6797	
.900	-.1966	.0475	-.0629	-.1760	-.5982	

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## TABULATED SURFACE DATA - CASTE

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CA57-E B16C5F1 J40 W07E18 WING TOTAL SURFACE

(ADW40) ( 12 NOV 73 )

## AEROFLOW DATA

WIND	4.4120 \$3.F1.	XMAP =	43.5940 IN.
LATR	16.2300 IN.	YMAP =	.0000 IN.
BREF	37.9350 IN.	ZMAP =	-.4050 IN.
SCALE	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = 9.995 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/0	.0000	.3340	.5200	.6630	.8730
------	-------	-------	-------	-------	-------

## DEPENDENT VARIABLE CP

X/C					
-.900	1.0162	-.3433	-.2721	-.4036	-.3981
-.750	.2715	-.2964	-.6722	-.7779	-.4419
-.600	.3254	.2724	-.0079	-.2455	-.3119
-.450	.1053	.2449	.1922	-.1092	-.2726
-.300	.3529	.3179	.2824	.2471	-.1177
-.150	.4334	.4149	.0020	.2616	.3211
.150	-.2622	-.6652	-.6781	-.9738	-.1.0783
.300	-.3435	-.4771	-.5000	-.5987	-.6110
.450	-.5913	-.3953	-.3436	-.3238	-.3061
.600	-.0634	-.1626	-.1684	-.1381	-.1480
.750	-.1087	.0141	.0223	.1982	-.0539
.900	-.2101	.0958	.0517	.1299	-.0390

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/0	.0000	.3340	.5200	.6630	.8730
------	-------	-------	-------	-------	-------

## DEPENDENT VARIABLE CP

X/C					
-.900	.9840	-.1137	-.1435	-.1920	-.4443
-.750	.4943	.0096	-.4090	-.6036	-.4270
-.600	.4011	.3847	-.0053	-.0934	-.1923
-.450	.2064	.3131	.2170	.0025	-.1764
-.300	.3925	.4000	.3444	.3315	-.0226
-.150	.4963	.4728	.0668	.4105	.4230
.150	-.2834	-.1.1726	-.6742	-.1.1201	-.1.2159
.300	-.3735	-.6383	-.5415	-.6556	-.6359
.450	-.2017	-.4753	-.3621	-.5229	-.5450
.600	-.0626	-.2162	-.2221	-.1.17	-.4105
.750	-.0949	-.0151	-.0146	-.0152	-.0152
.900	-.2792	-.1152	-.0224	-.1.17	-.1.17

## PARAMETRIC DATA

BETA	= .000	PTN/P = 1.500
H/E	= .039	BDFLAP = -16.000
ELEVON	= -15.000	

SECTION (1) WING	MACH = 1.12 = .165	ALPHA (3) = 10.990	CAST-B B16C5F1 JAO WTE10 WING TOTAL SURFACE (DOWNWAD)			
			AN/L = 1.200	MACH = .165	DEPENDENT VARIABLE C°	
X/C	21/8	.0000	.3340	.5200	.4630	.8730
-1.000	1.0130	-.0100	-.1632	-.2708	-.5727	
-1.750	.5336	.1640	-.2763	-.5472	-.6408	
-1.600	.4722	.4754	.1009	-.0219	-.0801	
-1.450	.2997	.3867	.2443	.1716	-.0045	
-1.300	.4930	.4739	.2932	.4561	.1580	
-1.150	.5636	.5319	-.1697	.3690	.4560	
-1.900	.3258	-.13480	-.6901	-.12429	-.1.1365	
-1.300	-.4467	-.7712	-.6522	-.9746	-1.0517	
-1.450	-.2907	-.6630	-.6934	-.6649	-1.0023	
-1.600	-.1727	-.4450	-.5973	-.6526	-.9092	
.750	-.2233	-.1679	-.3110	-.3205	-.5700	
.900	-.2676	.0151	-.1766	-.2474	-.5035	



(ADMM1) (12 NOV 73)

## REFERENCE DATA

WRF	4.4120 IN.	XMAP = 43.3940 IN.	BETA = .000	PIN/P = 1.300
WRY	.00 PWN IN.	YMAP = 0.000 IN.	M/B = .039	BOFLAP = -10.000
WZT	37.8350 IN.	ZMAP = -.0050 IN.	ELEON = -15.000	
SCALE	.0403			
MACH (1) =	.165	ALPHA (1) = 10.000	AVUL = 1.000	MACH = .165

## SECTION 1 WING

			DEPENDENT VARIABLE CP	
21/8	.0000	.3340	.5200	.6630 .6730
X/C				
- .900	.3917	-.3067	.2903	-.3604 -.3152
- .750	.1600	-.2055	.6150	-.7627 -.3745
- .600	.3022	.2211	-.0629	-.1637 -.2946
- .450	.1137	.2251	.1750	-.0710 -.2153
- .300	.3643	.3136	.3161	.3212 -.1028
- .150	.4391	.4202	.0433	.3516 .3451
.150	-.2456	-.6547	-.6340	-.9614 -.1.0862
.300	-.3439	-.4793	-.4902	-.5925 -.5919
.450	-.1112	-.3914	-.3406	-.3236 -.3022
.600	-.0599	-.1797	-.1800	-.1340 -.1624
.750	-.1079	.0099	.0173	.1956 -.0674
.900	-.1997	.0932	.0937	.1397 -.0436
MACH (1) =	.165	ALPHA (2) = 14.995	AVUL = 1.000	MACH = .165

## SECTION 1 WING

			DEPENDENT VARIABLE CP	
21/8	.0000	.3340	.5200	.6630 .6730
Y/C				
- .900	.5277	-.1502	-.1643	-.2215 -.4370
- .750	.3999	-.3397	-.3792	-.5627 -.3696
- .600	.3983	.3579	-.0824	-.0333 -.1291
- .450	.2394	.3083	.1937	.0029 -.1448
- .300	.4391	.4100	.3217	.3635 .0157
- .150	.5025	.4834	.1001	.5161 .4249
.150	-.2798	-.1.1786	-.6639	-.1.1524 -.3620
.300	-.3694	-.6617	-.5224	-.5577 -.0056
.450	-.1970	-.4866	-.3925	-.3554 -.7865
.600	-.0691	-.2229	-.2258	-.1320 -.845
.750	-.1163	-.0294	-.0012	-.746 -.2456
.900	-.254	.1104	.0245	-.1622 -.1540

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## TABULATED SOURCE DATA - CASTE

(ADYMA1)

MACH 1 (1)	.165	ALPHA (3)	20.000	AN/L	= 1.200	MACH = .165
<b>SECTION 1 (1) MACH</b>						
2.170	.0000	.3340	.3200	.6630	.6730	DEPENDENT VARIABLE CP
						X/C
-.9000	.7083	-.0845	-.2045	-.2965	-.5351	
-.750	.5050	.1026	-.2969	-.5210	-.6945	
-.600	.4747	.4487	.1193	-.0250	-.0417	
-.450	.3285	.3796	.2667	.0612	.0315	
-.300	.3115	.4799	.3518	.3467	.1808	
-.150	.3634	.5408	-.1905	.3367	.4616	
.150	-.2250	-1.9370	-.6823	-1.5148	-1.1300	
.300	-.4625	-.7883	-.6660	-1.0898	-1.0452	
.450	-.3036	-.6776	-.7216	-.9212	-.9990	
.600	-.1844	-.4435	-.5918	-.7213	-.9367	
.750	-.2325	-.1884	-.3316	-.5549	-.7017	
.900	-.2731	.0047	-.1909	-.2694	-.6310	



## REFERENCE DATA

SREF = 4.4120 80. FT. XMAP = 43.5940 IN.  
 LREF = 19.2360 IN. YMAP = .0000 IN.  
 DREF = 37.0350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 ANL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	21/0	.0000	.3340	.5200	.6630	.8730
- .900	-.0598	-.2802	-.2375	-.4936	-.3096	
- .750	.0017	-.3335	-.5434	-.8286	-.4859	
- .600	.2358	.1374	-.1796	-.2449	-.2324	
- .450	.0751	.1245	.1137	-.0918	-.2049	
- .300	.3765	.2606	.3146	.3590	-.0935	
- .150	.4149	.3939	.0142	.4157	.3841	
.150	-.2464	-.6508	-.6390	-.9686	-.1106	
.300	-.3495	-.4898	-.4935	-.8061	-.5975	
.450	-.1811	-.4017	-.3500	-.3312	-.3131	
.600	-.0595	-.1755	-.1760	-.1390	-.1879	
.750	-.1043	.0081	.0225	.2136	-.0976	
.900	-.1990	.0901	.0538	.1286	-.0549	

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.995 ANL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	21/0	.0000	.3340	.5200	.6630	.8730
- .900	.2163	-.1032	-.1914	-.2762	-.4493	
- .750	.2431	-.1304	-.3553	-.5193	-.5793	
- .600	.3701	-.2886	-.0409	-.0203	-.1042	
- .450	.2419	.2592	.1211	.1008	-.0649	
- .300	.4456	.3811	.3109	.5153	.0606	
- .150	.4825	.4739	.1031	.6257	.495	
.150	-.2944	-.11884	-.6749	-.12010	-.13244	
.300	-.4090	-.6979	-.5287	-.7544	-.2319	
.450	-.2098	-.5224	-.4244	-.4967	-.5007	
.600	-.0632	-.2532	-.2577	-.2636	-.0842	
.750	-.1306	-.0524	-.0321	.0041	-.6827	
.900	-.2087	.1005	.0076	.0266	-.4105	

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TABULATED SOURCE DATA - CA57B

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MACH (.1) = .165 ALPHAS (3) = 20.000 AN/L = 1.200 MACH = .165

CA57-B B16C5F1 J40 WATE16 WING TOTAL SURFACE

(RDVNA42)

SECTION (1) WING		DEPENDENT VARIABLE CP			
27/8	.0000	.3340	.5200	.6630	.6730
X/C					
-.900	.3301	-.1495	-.2124	-.3530	-.5289
-.750	.3803	-.0190	-.2779	-.4900	-.7694
-.600	.4607	.3536	.0749	.0376	.0072
-.450	.3395	.3254	.2472	.1476	.0412
-.300	.5196	.4643	.2955	.4435	.2022
-.150	.5566	.5406	-.0776	.5581	.4460
.150	-.3269	-1.5205	-.6914	-1.5135	-1.0917
.300	-.4721	-.8274	-.6610	-1.2054	-1.0255
.450	-.2966	-.6881	-.7479	-1.0339	-.9832
.600	-.1709	-.4466	-.6299	-.8613	-.9421
.750	-.2319	-.1616	-.3327	-.4069	-.7321
.900	-.2653	.0036	-.2072	-.3227	-.6711

REFERENCE DATA							PARAMETRIC DATA		
SREF	4.4120 84.FT.	XMAP	=	43.5940 IN.	BETA	=	.000	PTN/P = 1.500	
LREF	19.2300 IN	YMAP	=	.0000 IN.	M/B	=	.125	BOFLAP = -16.000	
BREF	37.9350 IN.	ZMAP	=	-.4050 IN.	ELEVON	=	-15.000		
SCALE	.0405								
MACH ( 1 ) = .165		ALPHA ( 1 ) = -3.990		RNL = 1.200	MACH = .165				
SECTION ( 1 )WING									
21/8	.0000	.3340	.5200	.6630	.8730				
DEPENDENT VARIABLE CP									
X/C									
-.900	-.3276	-.5286	-.4924	-.6083	-.5638				
-.750	-.7374	-.6392	-.8016	-.9361	-.5907				
-.600	-.9678	-.8177	-.8167	-.7179	-.6331				
-.450	-.13828	-.11432	-.7507	-.6471	-.7184				
-.300	-.12518	-.13485	-.6979	-.4443	-.6160				
-.150	-.0696	-.83337	-.1.2741	-.8238	-.4841				
.150	-.02902	-.11094	-.2035	-.2134	-.1547				
.300	-.1580	-.17732	-.2597	-.2214	-.2796				
.450	-.0399	-.1951	-.2065	-.1327	-.1727				
.600	-.0285	-.0357	-.0175	.0237	.0119				
.750	-.0227	.1607	.2594	.3362	.1255				
.900	-.1689	.1524	.1561	.1981	-.0016				
MACH ( 1 ) = .165	ALPHA ( 2 ) = .010	RNL = 1.200	MACH = .165						
SECTION ( 1 )WING									
21/8	.0000	.3340	.5200	.6630	.8730				
DEPENDENT VARIABLE CP									
X/C									
-.900	-.3516	-.4072	-.4022	-.5100	-.4087				
-.750	-.6146	-.5173	-.7825	-.8422	-.4475				
-.600	-.6076	-.5516	-.6088	-.5C84	-.4456				
-.450	-.9418	-.7007	-.4660	-.3786	-.4695				
-.300	-.6098	-.7534	-.3814	-.11112	-.3560				
-.150	.0684	-.3506	-.8264	-.3677	-.0509				
.150	-.0790	-.2177	-.3616	-.4337	-.3496				
.300	-.1924	-.2483	-.3416	-.3376	-.3623				
.450	-.0602	-.2276	-.2363	-.1775	-.1886				
.600	.0031	-.0452	-.0251	.0215	.0341				
.750	-.0360	.1264	.2041	.2853	.1593				
.900	-.1339	.175A	.1575	.2035	.0435				



OA37-B B16C5F1 J40 W8TE10 WING TOTAL SURFACE

(RDWMA3)

MACH ( 1 ) = .165    ALPH A ( 5 ) = 14.960

## SECTION ( 1 ) WING

21/0 .0000 .3340 .5200 .6630 .8730

MACH ( 1 ) = .165    ALPH A ( 6 ) = 20.005

21/0 .0000 .3340 .5200 .6630 .8730

X/C

X/C

X/C	-1.90	.3076	.3027	-.0352	.4480	.4821
	.190	-.2745	-.1000	-.6861	-.0908	-.1.2375
	.300	-.3611	-.6344	-.5317	-.6558	-.7009
	.450	-.1770	-.4709	-.3599	-.3325	-.5752
	.600	-.0433	-.1653	-.1621	-.1152	-.7230
	.750	-.0655	.0228	.0390	.2072	-.3086
	.900	-.1525	.1631	.0751	.1427	-.1249

X/C

X/C

X/C	-1.90	.0584	-.1720	-.3000	-.4958	-.4714
	-.750	.0422	-.1767	-.6653	-.7794	-.6832
	-.600	.1760	.1516	-.1954	.0465	.0368
	-.450	-.0453	.1298	-.0013	.3103	.0395
	-.300	.2653	.2544	.1603	.6054	.2085
	-.150	.4856	.4139	-.2162	.4807	.5219
	.150	-.3220	-.1.4626	-.6743	-.1.1357	-.0115
	.300	-.4196	-.7789	-.5514	-.1.1538	.9763
	.450	-.2309	-.5859	-.5799	-.8821	-.9233
	.600	-.1125	-.3513	-.4570	-.6763	-.8725
	.750	-.1504	-.1117	-.2161	-.2031	-.6694
	.900	-.2043	.0486	-.1047	-.1.627	-.5961

SECTION ( 1 ) WING

21/0 .0000 .3340 .5200 .6630 .8730

X/C

DEPENDENT VARIABLE CP

X/C

DEPENDENT VARIABLE CP

X/C

(TROYWA4) ( 12 NOV 73 )

## REFERENCE DATA

SREF = 4.4120 83.FT. XMRP = 43.5940 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 BCALC = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.005 RN/L = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C  
 -.900 -.2733 -.4406 -.4665 -.5177 -.5100  
 -.750 -.5806 -.5244 -.6667 -.7316 -.5208  
 -.600 -.7648 -.6250 -.6397 -.6034 -.5562  
 -.450 -.10984 -.9357 -.5873 -.5360 -.6227  
 -.300 -.1.0362 -.1.1194 -.1.5503 -.3042 -.5389  
 -.150 -.02710 -.6864 -.1.0720 -.6556 -.3715  
 .150 -.0386 -.1.069 -.2047 -.2243 -.1655  
 .300 -.1555 -.1.759 -.2573 -.2105 -.2722  
 .450 -.0500 -.1.849 -.2002 -.1253 -.1611  
 .600 .0372 -.0248 -.0062 .0305 .0247  
 .750 -.0150 .1.612 .2420 .3406 .1442  
 .900 -.1480 .1.724 .1.008 .2039 .0283

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RN/L = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C  
 -.900 -.2655 -.3463 -.3597 -.4627 -.3804  
 -.750 -.5952 -.4837 -.6538 -.7525 -.4047  
 -.600 -.4729 -.4319 -.5113 -.4342 -.4043  
 -.450 -.7674 -.6501 -.6565 -.3115 -.4194  
 -.300 -.5263 -.6054 -.3446 -.0463 -.3066  
 -.150 -.0915 -.2733 -.6916 -.2690 .0176  
 .150 -.0761 -.2179 -.3733 -.4466 -.3752  
 .300 -.2034 -.2330 -.3570 -.3247 -.3627  
 .450 -.0777 -.2259 -.2337 -.1795 -.1924  
 .600 .0078 -.0422 -.0280 .0176 .0337  
 .750 -.0363 .1206 .1943 .2832 .1753  
 .900 -.1240 .1629 .1637 .2009 .0910

PARAMETRIC DATA

BETA = .000 PTN/P = 1.300  
 M/B = .125 2DFLAP = -16.000  
 ELEVON = -15.000

DATE 08 OCT - 4

TABULATED SOURCE DATA - CASTB

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(ADVN44)

CA57-B B16C5F1 J40 W07E18 WING TOTAL SURFACE.

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.975 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.2042	-.2433	-.2866	-.3677	-.2667
-.750	-.3423	-.4107	-.6843	-.7560	-.5326
-.600	-.2333	-.2186	-.3646	-.2573	-.2719
-.450	-.4638	-.3610	-.1156	-.0933	-.2541
-.300	-.1836	-.2378	-.1003	.2271	-.0877
-.150	.2061	.0336	-.3332	.1064	.2794
.150	-.1696	-.4130	-.4898	-.1106	-.7020
.300	-.2763	-.4190	-.3933	-.4486	-.5028
.450	-.1260	-.3298	-.2423	-.2300	-.2540
.600	-.0193	-.0973	-.0703	-.0307	.0135
.750	-.0664	-.0705	-.0376	.2118	.1448
.900	-.1312	.1392	.0641	.1731	.0961

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.990 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0955	-.2068	-.2647	-.3495	-.2682
-.750	-.1800	-.3166	-.6286	-.7308	-.3296
-.600	-.0438	-.0595	-.3015	-.1267	-.1604
-.450	-.2523	-.1468	.0013	.0490	-.1329
-.300	.0217	-.0021	.2408	.3758	.0086
-.150	.3071	.2021	-.1211	.3428	.4109
.150	-.2395	-.6392	-.5940	-.9675	-.1.0619
.300	-.3263	-.4376	-.4739	-.5835	-.5818
.450	-.1612	-.3776	-.3301	-.3256	-.2809
.600	-.0336	-.1440	-.1414	-.1197	-.1276
.750	-.0731	.0444	.0313	.2096	-.0131
.900	-.1530	.1200	.0637	.1904	.0120

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.985 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730
X/C
-.900 -.0341 -.1614 -.2189 -.2775 -.3215
-.750 -.0257 -.4254 -.4795 -.5122 -.5515
-.600 1.054 .0726 -.1135 -.0033 -.0033
-.450 -.0904 .0094 -.0617 -.0617 -.1217
-.300 -.0112 -.659 .3617 -.1217 -.1217

## CASTE - B16C5F1 JAO W@TE10 WING TOTAL SURFACE

(RDYNA44)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.985

SECTION ( 1 ) WING

21/8 .00000 .3340 .5200 .6630 .8730

DEPENDENT VARIABLE CP

X/C					
-.150	.4069	.3326	.0176	.4947	.5015
-.150	-.2781	-.0813	-.6174	-.1.1054	-.1.2650
.300	-.3156	.6320	-.5191	-.6730	-.7600
.450	-.1668	-.4639	-.3554	-.3469	-.6243
.600	-.0356	-.1805	-.1629	-.1123	-.7574
.750	-.0777	.0156	.0326	.1912	-.3103
.900	-.1576	.1522	.0692	.1314	-.1423

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.985

SECTION ( 1 ) WING

21/8 .00000 .3340 .5200 .6630 .8730

DEPENDENT VARIABLE CP

X/C					
-.900	.1091	-.1687	-.2987	-.4784	-.4944
-.750	.0750	-.1632	-.5863	-.5841	-.5825
-.600	.2063	.1561	-.1216	.0200	.0381
-.450	.0206	.1623	-.1607	.2110	.0653
-.300	.3085	.2937	.1006	.6155	.2104
-.150	.4886	.4260	-.0530	.6577	.5019
.150	-.2986	-.1.680	-.5882	-.1.1346	-.1.0614
.300	-.4579	-.7645	-.5729	-.1.1921	-.9861
.450	-.2699	-.5995	-.6103	-.9599	-.9493
.600	-.1387	-.3773	-.4960	-.7397	-.8902
.750	-.1619	-.1.297	-.2.39	-.2.07	-.7032
.900	-.2152	.0274	-.1.241	-.2.376	-.6369

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TABULATED SOURCE DATA - CASTS

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CAST-8 816CSF1 J40 WATE16 WING TOTAL SURFACE

(RDW45) ( 12 NOV 73 )

## REFERENCE DATA

SREF	4.4120 83.FT.	XMAP	=	43.5940 IN.	BETA	=	.000	PTM/P	=	1.000
LREF	19.2300 IN.	YMAP	=	.0000 IN.	H/B	=	.125	BOFLAP	=	-16.000
CHL	.9330 IN.	ZMAP	=	-.4050 IN.	ELEVON	=	-15.000			
SCALE	.0403									
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	=	-4.015	RNL	=	1.200	MACH	=	.165

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.3323	-.3883	-.3800	-.3744	-.4477
-.750	-.4831	-.4106	-.4875	-.6644	-.4505
-.600	-.5068	-.4606	-.4820	-.4941	-.4779
-.450	-.8012	-.7033	-.4297	-.4286	-.5373
-.300	-.7673	-.8446	-.4191	-.1850	-.4701
-.150	.0073	-.5116	-.6595	-.5151	-.2595
.150	-.0425	-.1180	-.2285	-.2491	-.1931
.300	-.1661	-.1622	-.2693	-.2262	-.2861
.450	-.0574	-.1957	-.2117	-.1365	-.1641
.600	.0281	-.0297	-.0133	.0281	.053
.750	-.0236	-.1561	.2563	.3339	.1471
.900	-.1657	.1669	.1798	.2079	.0416

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.3762	-.2690	-.5303	-.3403	-.3494
-.750	-.4503	-.3824	-.4594	-.5393	-.6572
-.600	-.3326	-.3360	-.4019	-.3557	-.3586
-.450	-.2741	-.6142	-.5323	-.2634	-.2561
-.300	-.4009	-.4872	-.3801	-.2742	-.0031
-.150	.1061	-.2015	-.5598	-.1927	.0701
.150	-.0782	-.2210	-.3732	-.4559	-.3920
.300	-.2077	-.2377	-.3016	-.3454	-.3370
.450	-.3731	-.0826	-.2365	-.2397	-.1856
.500	.0062	-.0450	-.0269	.0156	.0352
.750	-.0383	.1148	.1467	.1839	.2771
.900	.1745	-.1411	.1755	.1561	.1987

CA37-B B16CSF1 J40 W/E18 WING TOTAL SURFACE (RDVWAS)

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.950 RNL = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	- .2274	- .2563	- .2928	- .3003	- .2640	
- .750	- .3303	- .3654	- .6322	- .7137	- .3316	
- .600	- .1556	- .1656	- .3111	- .1935	- .2674	
- .450	- .3683	- .3329	- .0707	- .0449	- .2291	
- .300	- .1274	- .1945	- .0736	.2629	- .0805	
- .150	- .2183	.0630	- .2951	.1994	.3034	
.150	- .1701	- .4019	- .5031	- .7194	- .7196	
.300	- .2942	- .4183	- .4002	- .4520	- .5077	
.450	- .1369	- .3349	- .2472	- .2259	- .2558	
.600	- .0076	- .0907	- .0860	- .0345	.0056	
.750	- .0710	.0753	.0347	.2126	.1419	
.900	- .1567	.1351	.0649	.1782	.0912	

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.960 RNL = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	- .1268	- .2080	- .2874	- .2453	- .2588	
- .750	- .1745	- .3009	- .6010	- .6072	- .2435	
- .600	- .5054	- .0503	- .2594	- .0363	- .1363	
- .450	- .1937	- .1343	.0247	.1036	- .1122	
- .300	.0654	.0185	.2607	.1416	.0106	
- .150	.3C3:	.2179	.0972	.4005	.4429	
.150	- .2495	- .6362	- .6043	- .9330	- .1044	
.300	- .3439	- .4537	- .4741	- .45907	- .5740	
.450	- .1618	- .3797	- .3309	- .3130	- .2626	
.600	- .0409	- .1441	- .1314	- .1066	- .1212	
.750	- .0741	.0439	.0465	.2321	.0348	
.900	- .1599	.1155	.0652	.2146	.0005	

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.940 RNL = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	- .0059	- .1611	- .2816	- .2426	- .3295	
- .750	- .0253	- .1930	- .5617	- .4903	- .2226	
- .600	.1315	.0747	- .1921	.0485	.0204	
- .450	- .0497	.0153	.0084	.1585	-.0063	
- .300	.2122	.1753	.3159	.5401	.1327	

## CR57-B B16CF1 JAO WTE16 WING TOTAL SURFACE

(AD07445)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.945

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.3950	.3298	.0163	.3822	.4872
.150	-.2378	-.0847	-.6912	-.1513	-.3119
.300	-.4172	-.6354	-.5128	-.6902	-.8149
.450	-.1964	-.5030	-.3750	-.3698	-.6845
.600	-.0951	-.2047	-.1812	-.1278	-.0216
.750	-.1112	-.0123	-.0294	-.1871	-.3309
.900	-.1705	-.1247	-.0581	-.1429	-.1703

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.970 ANVL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

X/C	.6261	.0037	-.1775	-.2820	-.4069
-.900	-.0054	.0004	-.1695	-.4566	-.5194
-.750	-.2206	.2296	.1468	.0349	.0652
-.600	.1971	.0750	.1556	.2956	.1601
-.450	-.3000	-.3347	.3003	.2419	.6060
-.300	-.4658	-.4865	.4345	.0576	.7431
-.150	-.3597	-.3290	-.4139	-.6643	-.1058
.300	-.9658	-.5168	-.8005	-.5533	-.1259
.450	-.9458	-.2902	-.6219	-.6322	-.10474
.600	-.1199	-.1156	-.3906	-.5117	-.6629
.750	-.0631	-.1636	-.1345	-.2491	-.3276
.900	-.7233	-.2250	.0285	-.1509	-.2492

CA37-B 010C5F1 J41 WATE10 WING TOTAL SURFACE

(DYNAMIC) (12 NOV 73)

## REFERENCE DATA

BREF	=	4,4120 SQ FT.	XWEP	=	43,5940 IN.
LREF	=	19,2300 IN.	YWEP	=	.0003 IN.
GREF	=	37,0350 IN.	ZWEP	=	.4050 IN.
SCALE	=	.6405			

MACH (1) = .165 ALPHAI (1) = -.0005 ANL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C					
-.900	-.1568	-.1623	-.0063	.0686	-.0807
-.750	-.2465	-.1596	-.0620	.1693	.0191
-.600	-.3440	-.2921	-.2127	.0314	.0664
-.450	-.6952	-.0754	-.0709	.1126	-.0759
-.300	-.3649	-.4496	-.2179	.1718	-.1211
-.150	.0849	-.2995	-.5598	.2684	-.1643
.150	.0760	-.1658	-.3463	.4841	-.4766
.300	-.2310	-.2925	-.4308	.4115	-.6185
.450	-.1569	-.3829	-.4442	.4359	-.5184
.600	-.0403	-.3179	-.3372	.4126	.6441
.750	-.0916	-.3724	-.9315	.4662	.3870
.900	-.2014	-.0820	-.0877	.0159	.0900

MACH (1) = .165 ALPHAI (2) = -.010 ANL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C					
-.900	-.1256	-.6490	.0074	.1388	-.0966
-.750	-.1441	.0065	.0360	.2768	.0640
-.600	-.1432	-.1172	-.0266	.0662	.1259
-.450	-.4568	-.0481	.0209	.0411	.0269
-.300	-.2114	-.1613	-.0613	.0538	-.0006
-.150	.1616	-.0092	-.1638	.0138	.0711
.150	-.1242	-.3114	-.5032	-.7039	-.7818
.300	-.2661	-.3882	-.5174	-.6255	-.7752
.450	-.1961	-.4362	-.4874	-.5212	.6691
.600	-.0717	-.5348	-.3512	-.4367	-.6819
.750	-.1139	-.3227	-.6598	-.4542	-.4356
.900	-.1635	-.1221	-.1072	-.0622	-.1413

## PARAMETRIC DATA

BETA	=	.000	FTN/P	=	1,300
M/B	=	.125	EOF LAP	=	-16,000
ELEV0	=	15,000			



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(ADWMA6)

TABULATED SOURCE DATA - CAST8

## CAST-B B16CSF1 J41 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (3) = 4.985 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .6730

X/C

-.900	.0225	.0780	.0384	.1574	-.1290
-.750	.0314	.1736	.2009	.3592	.0970
-.600	.0739	.0847	.1238	.2180	.1847
-.450	-.1613	.1761	.2299	.2040	.1037
-.300	.0620	.1104	.2501	.2664	.1016
-.150	.2949	.2405	.0753	.35 2	.2332
.150	-.2316	-.6079	-.6425	-1.0575	-1.2197
.300	-.3631	-.6164	-.3857	-.7803	-.9884
.450	-.2306	-.5437	-.5922	-.5961	-.7988
.600	-.0946	-.3457	-.3549	-.4902	-.7590
.750	-.1436	-.3091	-.4553	-.4630	-.5747
.900	-.1593	-.0430	-.1267	-.1516	-.2657

MACH (1) = .165 ALPHA (4) = 9.970 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .6730

X/C

-.900	.0821	.2123	.1009	-.0308	-.1790
-.750	.1734	.3060	.2600	.2940	.1185
-.600	.2249	.2122	.1948	.2266	.2473
-.450	.0058	.3241	.3583	.2291	.1653
-.300	.2299	.2717	.3673	.3926	.1883
-.150	.3844	.3656	.1710	.3665	.3374
.150	-.2846	-.1617	-.1668	-.12871	-.1.6266
.300	-.4111	-.6101	-.6329	-.8953	-.1.1604
.450	-.2591	-.5602	-.5408	-.6433	-.9360
.600	-.1042	-.3687	-.3830	-.1099	-.8407
.750	-.1572	-.3286	-.3474	-.1433	-.7291
.900	-.1679	-.0906	-.0521	-.1254	-.4416

MACH (1) = .165 ALPHA (5) = 14.960 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .6730

X/C

-.900	.1540	.2652	.1175	-.0636	-.2325
-.750	.2590	.3992	.3013	.2605	.1668
-.600	.5196	.2890	.2604	.2238	.3454
-.450	.1197	.4186	.3524	.2655	.2574
-.300	.3417	.3578	.3903	.4750	.3243

## CAST 8 BLOCKSF1 J41 WTE10 WING TOTAL SURFACE (ADWMA6)

MACH (1) = .165	ALPHA (5) = 14.960	CAST 8 BLOCKSF1 J41 WTE10 WING TOTAL SURFACE		
SECTION (1) WING		DEPENDENT VARIABLE CP		
21/8 .0000	.3340	.5200	.6630	.8730
X/C				
-1.90 -4377	.4902	-.0602	.2745	.4408
-1.50 -.3225	-1.3415	-.8203	-1.4469	-2.5187
-1.00 -.4015	-.8468	-.7283	-.9923	-2.1809
-0.50 -.2942	-.7167	-.6608	-.6798	-2.0303
.000 -.1335	-.4342	-.4905	-.5404	-1.6226
.750 -.1643	-.3955	-.8492	-.3936	-1.2293
.900 -.1915	-.1724	-.1367	-.2786	-.6242
MACH (1) = .165	ALPHA (6) = 19.960	ANVL = 1.200	MACH = .165	
SECTION (1) WING				
21/8 .0000	.3340	.5200	.6630	.8730
X/C				
-1.90 -.2012	.2820	.0465	-.1124	-.2130
-1.750 .5216	.4524	.4041	.2301	.1825
-1.500 .3960	.3462	.2008	.2417	.3534
-1.450 .2162	.4953	.3916	.2906	.3074
-1.300 .4493	.4394	.1921	.3730	.3676
-1.150 .5342	.5276	.1110	.1426	.4677
-1.000 -.3467	-1.6653	-.6183	-1.4587	-1.1514
-0.500 -.5193	-1.0104	-.7986	-1.3671	-1.1265
-0.150 -.3511	-.6698	-.9885	-1.2120	-1.1036
.600 -.2218	-.5949	-.6561	-1.0936	-1.0766
.750 -.2665	-.3777	-.2168	-.8925	-.9741
.900 -.2812	-.2611	-.4525	-.7144	-.8929

(RDW47) (12 NOV 73)

CAST-B B16C5F1 J41 W7E10 WING TOTAL SURFACE

## REFERENCE DATA

BASE =	4,4120 82.FT.	XMRP =	43.5940 IN.
LREF =	19.2300 IN.	1MRF =	.0000 IN.
WNL =	.0000 IN.	ZMRP =	-.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.010 ANGL = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT VARIABLE CP

X/C					
-.830	-.1680	-.0962	-.0407	.0306	-.0912
-.750	-.1986	-.0945	-.0330	.1420	.0123
-.600	-.2314	-.2144	-.1270	.0239	.0690
-.450	-.4668	-.1482	-.0069	.0506	-.0667
-.300	-.4283	-.3623	-.1448	.0259	-.1058
-.150	.1064	.2240	.4729	.2033	.1260
.150	-.0834	-.1986	-.3628	-.4741	-.4962
.300	-.2404	-.3013	-.4447	-.4915	-.6343
.450	-.1551	-.3802	-.4493	-.4574	-.5927
.600	-.0468	-.3243	-.3146	-.4201	-.6384
.750	-.0938	-.3719	-.6871	-.5121	-.4012
.900	-.1984	-.0749	-.0109	-.0659	-.1036

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.015 ANGL = 1.200 MACH = .165

SECTION ( 1 )WING DEPENDENT VARIABLE CP

X/C					
-.900	-.1358	-.0278	.0013	.1115	-.0932
-.750	-.1151	.0461	.0634	.2726	.0546
-.600	-.0614	-.0851	.5097	.1001	.1319
-.450	-.3026	.0093	.0764	.0961	.0416
-.300	-.1337	-.1016	-.0332	.2182	.0132
-.150	.2058	.0295	-.1428	.0445	.0952
.150	-.1194	-.3099	-.5103	-.7575	-.8016
.300	-.2961	-.3870	-.5097	-.6561	-.7832
.450	-.1983	-.4369	-.6822	-.5111	-.6708
.600	-.0603	-.3282	-.3375	-.4448	-.6090
.750	-.1180	-.3155	-.5964	-.4625	-.4444
.900	-.1820	-.0920	-.1357	-.0653	-.1474

## CAAA7-B F16C3F1 J41 WTE18 WING TOTAL SURFACE (RADWAT)

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.935 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	-.1070	.0722	.0132	.1068	-.1256
-.750	.0294	.1714	.1777	.3565	.0933
-.600	.1122	.0783	.1103	.2361	.1767
-.450	-.0845	-.1987	.2469	.2537	.1164
-.300	.1031	.1173	.2898	.4210	.1162
-.150	.3023	.2540	.0885	.3250	.2466
.150	-.2261	-.6079	-.6519	-.1.0707	-1.2391
.300	-.3620	-.6137	-.5881	-.1.9228	-.9789
.450	-.2336	-.5363	-.5015	-.6021	-.8000
.600	-.0993	-.3466	-.3677	-.4838	-.7444
.750	-.1490	-.3188	-.4819	-.4524	-.6025
.900	-.1698	-.0252	-.1233	-.1672	-.3117

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.965 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.0710	.1703	.0635	.0171	-.1804
-.750	.1487	.2823	.2283	.3495	.0984
-.600	.2405	.1931	.2103	.3196	.2516
-.450	.0632	.3292	.3387	.3537	.1661
-.300	.2473	.2774	.3097	.5328	.2138
-.150	.3862	.3692	.2116	.5188	.3690
.150	-.2662	-.7529	-.7421	-.1.3C16	-1.6485
.300	-.4239	-.6262	-.6172	-.8964	-.1.1642
.450	-.2590	-.5717	-.5541	-.6476	-.9573
.600	-.1106	-.3702	-.3812	-.5121	-.8438
.750	-.1675	-.3538	-.6629	-.4614	-.7479
.900	-.1753	-.0967	-.0748	-.1300	-.4792

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.985 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.1376	.2085	.1078	-.1.807	-.2289
-.750	.2323	.3475	.2921	.2900	.1391
-.600	.3244	.2699	.2799	.1752	.3286
-.450	.1650	.4192	.2818	.2475	.2633
-.300	.3372	.3658	.2090	.4305	.3321



DATE 08 OCT 74

TABULATED SOURCE DATA - OA47B

PAGE 295

CAST-B 816C5F1 J41 WATE16 WING TOTAL SURFACE

(REVW47)

MACH (1) = .165 ALPHA (1) = 14.985

SECTION (1) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.150	.4575	.4519	.1073	.4040	.4418
.150	-.3165	-.3269	-.6160	-1.5065	-2.1908
.300	-.4824	-.6423	-.7116	-1.0591	-2.0125
.450	-.2985	-.7178	-.6712	-.7228	-.0771
.600	-.1426	-.4112	-.4612	-.5647	-.6882
.750	-.1702	-.4178	-.8354	-.4271	-.12507
.900	-.2035	-.1599	-.1805	-.2967	-.9266

MACH (1) = .165 ALPHA (1) = 19.980 RNL = 1.200 MACH = .165

SECTION (1) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	.1626	.2118	.0460	-.2652	-.2013
-.750	-.7832	-.3892	.3041	.1991	.1570
-.600	.3914	.3237	.2714	.2419	.3460
-.450	.2531	.4670	.3949	.3741	.7148
-.300	.4326	.4429	.3998	.4874	.3823
-.150	.5332	.5266	.1606	.5492	.4746
.150	-.5391	-.16254	-.7835	-.5083	-.11077
.300	-.5218	-.9904	-.8029	-.12736	-.1.0842
.450	-.3373	-.8488	-.9564	-.1.1634	-.1.0854
.600	-.2124	-.5916	-.8086	-.1.0846	-.1.0555
.750	-.2623	-.5686	-.1.1400	-.8617	-.9603
.900	-.2866	-.2536	-.4444	-.7286	-.8859

## CAST-B B15C5F1 J41 WATE10 WING TOTAL SURFACE

## REFERENCE DATA

SREF = 4.4120 SD.FT. XMRP = 43.5940 IN.  
 LREF = 19.2300 IN. IMRP = .0000 IN.  
 BREF = 37.9350 IN. ZMRP = -.4050 IN.  
 SCALE = .0403

MACH ( 1 ) = .165 ALPHA ( 1 ) = .000 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

## X/C

-.900	.7831	.6983	.7401	.1782	-.3977
-.750	.6303	.7270	.5757	.0233	-.3017
-.600	.5406	.5510	.2996	.2285	-.1607
-.450	.4178	.6264	.4174	.1894	-.1970
-.300	.5278	.5406	.3904	.3478	-.0726
-.150	.4902	.5315	.2591	.3023	.1433
.150	-.2650	-.7833	-.7237	-.12001	-1.5481
.300	-.4070	-.5863	-.1757	-.8120	-1.1211
.450	-.2509	-.5212	-.4971	-.5793	-.9531
.600	-.1056	-.3668	-.3594	-.4244	-.7970
.750	-.1482	-.3066	-.3923	-.3024	-.6456
.900	-.1602	-.0331	-.3339	-.0984	-.5910

MACH ( 1 ) = .165 ALPHA ( 2 ) = 4.990 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

## X/C

-.900	.7737	.6875	.6761	.4933	-.4410
-.750	.6886	.7272	.6141	.5195	-.2122
-.600	.5607	.5127	.2548	.1036	-.0526
-.450	.4477	.6340	.4076	.1399	.0350
-.300	.5493	.5402	.1090	.1890	.1612
-.150	.5228	.5518	.0647	.0233	.3248
.150	-.3162	-.3147	-.7661	-.3958	-2.5556
.300	-.4627	-.1817	-.6556	-.9339	-2.0813
.450	-.2844	-.6534	-.5923	-.6025	-.9875
.600	-.1441	-.3912	-.4247	-.5281	-.5139
.750	-.1636	-.3595	-.5231	-.3859	-.1434
.900	-.2087	-.1001	-.1673	-.2590	-.8658

## PARAMETRIC DATA

BETA = .000 PTN/P = 1.300  
 M/B = .039 BDFLAP = -16.000  
 ELEVON = 15.000



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TABULATED SOURCE DATA - QAS7B

PAGE 297

CA57-B B16C5F1 J41 WATE18 WING TOTAL SURFACE

(RDVMA)

MACH (.1) = .165 ALPHA (.3) = 10.000 RNL = 1.200 MACH = .165

SECTION (1) WING

DEPENDENT VARIABLE CP

X/C					
-.900	.8444	.6941	.5986	.5162	-.1048
-.750	.7135	.7310	.5716	-.3462	-.1217
-.600	.5912	.5476	.3107	.1569	.0952
-.450	.4932	.6617	.4530	.2109	.1110
-.300	.3918	.5682	.3110	.2227	.2177
-.150	.5147	.5878	.2528	.0142	.3701
.150	-.3804	-.6163	-.7949	-.1.5824	-1.1694
.300	-.6100	-.9755	-.7848	-.1.3838	-1.1528
.450	-.4150	-.8293	-.9788	-.1.1900	-1.1132
.600	-.2157	-.6273	-.7832	-.1.0933	-1.0579
.750	-.3067	-.5635	-.8584	-.9087	-.9460
.900	-.2971	-.2496	-.4585	-.7165	-.8493

DATE 03 OCT 74

TABULATED SOURCE DATA - CAA37B

PAGE 298

CA37-B B16C5F1 J41 W0719 WING TOTAL SURFACE

(RDW49)

( 12 NOV 73 )

## REFERENCE DATA

SAEF =	4.4120	83.FT.	XMAP =	43.5940 IN.	BETA =	.000	PTN/P =	1.000
LREF =	19.2300	IN.	YMAP =	.0000 IN.	H/B =	.039	BDFAP =	-16.000
BREF =	37.9350	IN.	ZMAP =	-.4030 IN.	ELEVON =	15.000		
SCALE =	.0405							

MACH ( 1 ) = .165    ALPHA ( 1 ) = -.005    RNU/L = 1.200    MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C								
- .900	.3463	.5154	.3845	.3219	-.3819			
- .750	.3998	.5366	.6063	.1671	-.3495			
- .600	.4872	.4413	.3491	.2070	-.1406			
- .450	.3967	.5774	.4136	.1800	-.1695			
- .300	.5026	.5080	.3896	.3802	-.0498			
- .150	.4767	.5167	.2778	.3830	.1386			
.150	-.2802	-.7480	-.7249	-.2086	-1.5723			
.300	-.4672	-.6079	-.5877	-.8192	-1.1334			
.450	-.2584	-.5146	-.5134	-.5733	-.9432			
.600	-.1111	-.3593	-.3758	-.4359	-.8018			
.750	-.1523	-.3226	-.3754	-.3034	-.6659			
.900	-.1635	-.0469	-.0459	-.0939	-.3910			

MACH ( 1 ) = .165    ALPHA ( 2 ) = 4.995    RNU/L = 1.200    MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C								
- .900	.4398	.5250	.3626	.2766	-.4127			
- .750	.5507	.5676	.5242	.1249	-.2321			
- .600	.5226	.4297	.3767	.1615	-.0924			
- .450	.4440	.5904	.4394	.2233	.0274			
- .300	.5293	.5136	.1372	.2253	.1758			
- .150	.5148	.5367	-.0302	.0345	.3367			
.150	-.3503	-.12720	-.7645	+.4114	-2.5406			
.300	-.5804	-.7965	.6738	-.455	-2.1446			
.450	-.3348	-.6252	-.5842	-.6275	-2.0503			
.600	-.1288	-.4281	-.4930	-.5196	-1.5740			
.750	-.1523	-.5779	-.5050	-.3861	-1.2283			
.900	-.1929	-.1115	-.2045	-.2550	-.7939			



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TABULATED SOURCE DATA - CA37B

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MACH ( 1 ) = .165	ALPHA ( 3 ) = 9.995	RVAL = 1.200	MACH = .165
SECTION ( 1 ) WING			
Z/R6	.0000	.3340	.5200
DEPENDENT VARIABLE CP			
X/C			
- .9000	.5149	.5272	.5084
- .7500	.5689	.6038	.5390
- .6000	.5702	.4855	.4007
- .4500	.4958	.5232	.4803
- .3000	.5799	.5493	.5464
- .1500	.5703	.5768	.2269
.150	.41116	-1.5637	.8045
.300	.6882	-.9718	.7626
.450	-.4210	-.8227	-.0104
.600	-.2366	-.6252	-.8794
.750	-.3050	-.5589	-.8434
.900	-.2864	-.2378	-.5108
			-.7487
			-.0393

(RDV449)

(RDW930) (12 NOV 73)

## REFERENCE DATA

CA57-B B16C3F1 J41 WATE10 WI... TOTAL SURFACE

## PARAMETRIC DATA

REFD = 4,4120 3A FT.  
 LREF = 18.2300 IN.  
 BREF = 37.9350 IN.  
 SCALE = .0405

MACH ( 1 ) = .165    ALPHA ( 1 ) = -.010  
 MACH ( 2 ) = .165    ALPHA ( 2 ) = -.020

SECTION ( 1 ) WING

SECTION ( 2 ) WING

SECTION ( 3 ) WING

SECTION ( 4 ) WING

SECTION ( 5 ) WING

SECTION ( 6 ) WING

SECTION ( 7 ) WING

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SECTION ( 244 ) WING

DATE 08 OCT 74

TABULATED SOURCE DATA - OAS78

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CASS7-E B16CF1 J41 WTE10 WING TOTAL SURFACE (RDW50)

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.985 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C .0000 .3340 .5200 .6630 .8730  
DEPENDENT VARIABLE CP

X/C	.0000	.0737	.0076	-.0126	.1987	-.1109
-.750	-.0034	.1160	.1462	.3088	.0784	
-.600	-.0134	.0152	.0536	.1994	.1993	
-.450	-.2684	.0700	.1616	.2074	.1163	
-.300	-.0324	.0003	.2125	.3770	.1171	
-.150	-.2363	.1398	.0050	.2662	.2333	
.150	-.2323	-.4770	-.6416	-1.0138	-1.1613	
.300	-.3599	-.5194	-.5759	-.7585	-.9388	
.450	-.2269	-.4912	-.5033	-.5768	-.7620	
.600	-.0816	-.3660	-.3549	-.4696	-.6808	
.750	-.1225	-.2908	-.2794	-.4757	-.5354	
.900	-.1630	-.1094	-.1908	-.1232	-.3000	

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.975 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C .0000 .3340 .5200 .6630 .8730  
DEPENDENT VARIABLE CP

X/C	.0000	.1333	.0900	-.0094	.2347	-.1567
-.750	-.0633	.1970	.1554	.2166	.0926	
-.600	-.0906	.1015	.1009	.2849	.2317	
-.450	-.1497	.1934	.2998	.2860	.1843	
-.300	-.0934	.1396	.3912	.4695	.1961	
-.150	.3205	.2700	.1211	.3938	.3497	
.150	-.2747	-.7604	-.7449	-.1.2911	-.1.5804	
.300	-.4349	-.5923	-.6423	-.8807	-.1.1087	
.450	-.2558	-.5328	-.5589	-.6549	-.9071	
.600	-.1026	-.1833	-.4117	-.5306	-.7759	
.750	-.1565	-.3581	-.3922	-.6947	-.6710	
.900	-.1672	-.0946	-.0963	-.2116	-.4667	

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.980 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

X/C	.0000	.3340	.5200	.6630	.8730
-.900	.1630	.1460	.0450	.1774	-.1753
-.750	-.1164	.2609	.2245	.3450	.1492
-.600	-.1126	.1682	.2050	.2734	.3186
-.450	-.0601	.2923	.2931	.5074	.2452
-.300	.2117	.2127	.1544	.4847	.3055

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TABULATED SOURCE DATA - CASTE

PAGE 302

## CA37-E B16C5F1 J41 WATE10 WING TOTAL SURFACE

(REVNW50)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.980

## SECTION 11)WING

21/8 .0000 .3340 .5200 .6630 .8730

## DEFINENT VARIABLE CP

X/C	.9955	.5623	-.0843	.3493	.4557
-.150	-.3467	-1.2683	-.8075	-1.4078	-2.5545
.300	-.5372	-1.7995	-.6975	-.9243	-.4005
.450	-.5242	-.6914	-.6533	-.6615	-.1226
.600	-.1273	-.4436	-.5229	-.4997	-.0302
.750	-.1249	-.4063	-.4984	-.3445	-.9894
.900	-.1706	-.1632	-.1765	-.2455	-.7150

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.980 RNL = 1.200 MACH = .165

## SECTION 11)WING

21/8 .0000 .3340 .5200 .5630 .8730

## DEFINENT VARIABLE CP

X/C	.1756	.1152	-.1073	-.0825	-.2177
-.950	-.510	.2820	.1785	.3783	.1701
-.500	.2421	.2231	.2040	.3258	.3572
-.150	.0173	.3741	.3176	.4121	.2920
.300	.3297	.3450	.2654	.5647	.3644
.450	4.754	4.446	.5956	.4209	.4844
.600	-.4.215	-.1.5123	-.7038	-.1.3176	-.0.169
.750	-.1.207	-.6.563	-.7482	-.1.1213	-.3.287
.900	-.3.712	-.8.986	-.9373	-.1.4950	-.1.0160
6.1	-.1.113	-.6.022	-.9.423	-.1.452	-.1.0370
7.0	-.2.143	-.8.769	-.8.957	-.2.274	-.9.335
.9.-0	-.2.825	-.2.721	-.5.701	-.9.255	-.6.614

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TABLED INACCUE DATA - 14376

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REF. 7-B 810CSF1 J41 M7E10 WING TOTAL SURFACE  
 (ADJUST) ( 12 Nov 73 )

REFERENCE DATA						PARAMETRIC DATA		
WIND	.1180 60 FT	WIND	43.940 IN.	BETA	.000	PIN/P	1.000	
UREF	19.2500 IN.	WIND	.0000 IN.	M/B	.286	DDFLAP	-16.000	
BREF	37.9350 IN.	WIND	-.4050 IN.	ELEVON	15.000			
SCALE	.0403							
MACH ( 1 ) = .165	ALPHA ( 1 ) = -.015	ANVL = 1.200	MACH = .165					

SECTION 11 WING  
 DEPENDENT VARIABLE CP

2/76 .0000 .3340 .5200 .6630 .6730

X/C					
-.900	-.0632	-.0528	-.0223	.0667	-.0864
-.750	-.1290	-.0348	.0026	.1828	.0343
-.600	-.1131	-.1198	-.0556	.0724	.0727
-.450	-.3761	-.0648	.0267	-.0269	-.0716
-.300	-.3364	-.2870	-.1172	.0688	-.0751
-.150	.0932	-.1690	-.4065	-.1797	-.1130
.150	-.0707	-.1685	-.3239	-.4456	-.4897
.300	-.2428	-.2887	-.4101	-.4492	-.6036
.450	-.1373	-.3540	-.4641	-.4395	-.5616
.600	-.0346	-.2965	-.3313	-.4007	-.6019
.750	-.0935	-.3454	-.4466	-.5010	-.4517
.900	-.1779	-.0753	-.1316	-.0533	-.1268

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.005 ANVL = 1.200 MACH = .165

SECTION 11 WING  
 DEPENDENT VARIABLE CP

X/C					
-.900	-.0265	-.0024	.0032	.0983	-.0915
-.750	-.0723	.0627	.0420	.2586	.0602
-.600	-.0723	-.1255	.0143	.0786	.1466
-.450	-.2891	.0387	.0590	.0789	.0084
-.300	-.1810	-.1395	-.0096	.1991	.0205
-.150	.1828	-.0241	-.1674	.0274	.0933
.150	-.1063	-.2802	-.4925	-.1720	-.7665
.300	-.2616	-.3701	-.5053	-.6065	-.7657
.450	-.1661	-.3991	-.4942	-.5081	.6467
.600	-.0316	-.3169	-.3469	-.4308	.6438
.750	-.1033	-.3062	-.3884	-.4697	-.4621
.900	-.1634	-.0602	-.1326	-.0922	-.1922

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TABULATED SOURCE DATA - CASTE

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MACH (1) = .165    ALPH (3) = 4.985    RNL = 1.200    MACH = .165  
 SECTION 1: WING  
 21/8 .0000 .3340 .5200 .6630 .8730  
 X/C

MACH (1) = .165    ALPH (4) = 9.990    RNL = 1.200    MACH = .165  
 SECTION 1: WING  
 21/8 .0000 .3340 .5200 .6630 .8730  
 X/C

MACH (1) = .165    ALPH (5) = 14.970    RNL = 1.200    MACH = .165  
 SECTION 1: WING  
 21/8 .0000 .3340 .5200 .6630 .8730  
 X/C

CASE-B    B16C5F1 J41 WTE10 WING TOTAL SURFACE  
 (RDWHS1)

DEPENDENT VARIABLE CP

SECTION 1: WING

X/C

MACH (1) = .165    ALPH (3) = 4.985    RNL = 1.200    MACH = .165  
 SECTION 1: WING  
 21/8 .0000 .3340 .5200 .6630 .8730  
 X/C

MACH (1) = .165    ALPH (4) = 9.990    RNL = 1.200    MACH = .165  
 SECTION 1: WING  
 21/8 .0000 .3340 .5200 .6630 .8730  
 X/C

MACH (1) = .165    ALPH (5) = 14.970    RNL = 1.200    MACH = .165  
 SECTION 1: WING  
 21/8 .0000 .3340 .5200 .6630 .8730  
 X/C

CASE-B    B16C5F1 J41 WTE10 WING TOTAL SURFACE  
 (RDWHS1)

DEPENDENT VARIABLE CP



DATE 08 OCT 74

TABULATED SOURCE DATA - OA97B

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MACH 1 (1) = .165 ALPHAS ( 5 ) = 14.970

(ADWMS1)

## SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.3950	.3654	.0568	.5686	.4437
.190	-.3490	-.1221	-.8101	-.14635	-2.6381
.300	-.5680	-.7993	-.6882	-.9738	-1.4079
.450	-.3153	-.5018	-.6462	-.6556	-1.1874
.600	-.1357	-.4501	-.4808	-.4817	-1.0358
.750	-.1461	-.4037	-.4543	-.3479	-1.0041
.900	-.1722	-.1674	-.1761	-.2260	-.7453

MACH 1 (1) = .165 ALPHAS ( 6 ) = 19.960 RFL = 1.800 MACH = .165

## SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.1366	.0668	-.0534	-.1669	-.2013
-.300	.1279	.2654	.1728	.3126	.1382
-.600	.2441	.1769	.2068	.2781	.3330
-.450	.0675	.3129	.2939	.3247	.2862
-.300	.3200	.3233	.2013	.4443	.3629
-.190	.4692	.4458	.0847	.3109	.4872
.150	-.4009	-.15382	-.7593	-.14381	-1.0893
.300	-.5982	-.9662	-.7465	-.11717	-1.0566
.450	-.3761	-.8196	-.9164	-.11567	-1.0410
.600	-.1960	-.3768	-.7956	-.11372	-1.0071
.750	-.2739	-.5028	-.7696	-.7811	-.9209
.900	-.2794	-.2653	-.5336	-.7594	-.8432

CA57-E 816C5F1 141 WATE10 WING TOTAL SURFACE

(RDYMS2) (112 NOV 73)

## REFERENCE DATA

	WREF = 4,4120 IN. FT.	XMAP = 43,5940 IN.	BETA = .000	PIN/P = 1,300
WAEF = 16,2300 IN.	YMAP = .0000 IN.	H/B = .246	EDFLAP = -16,000	
SAREF = 37,0350 IN.	ZMAP = -.4050 IN.	ELEVON = .0000		
SCALE = .0405				
MACH (1) = .165	ALPHA (1) = -4,005	RNL = 1,200	MACH = .165	

## SECTION 11 WING

21/E .0000 .3340 .5200 .6630 .6730

X/C	- .000	- .0322	- .1294	- .1263	- .1414	- .1777
	- .750	- .2239	- .2567	- .2190	- .3520	- .3833
	- .600	- .2980	- .3769	- .2473	- .2082	- .3471
	- .450	- .6281	- .3279	- .1171	- .1907	- .3530
	- .300	- .4364	- .4745	- .2363	- .3649	- .2393
	- .150	.0566	- .3065	- .5703	- .377	- .2975
MACH (1) = .165	ALPHA (1) = -4,005	RNL = 1,200	MACH = .165			

## DEFINITION OF

## PARAMETRIC DATA



X/C	- .95	- .7251	- .1033	- .0840	- .0950	- .1557
	- .750	- .1754	- .2261	- .1606	- .3169	- .3543
	- .600	- .2127	- .3102	- .1910	- .116	- .2799
	- .450	- .5066	- .2365	- .0473	- .0761	- .2436
	- .300	- .2934	- .3222	- .1322	- .3979	- .1019
MACH (1) = .165	ALPHA (1) = -4,005	RNL = 1,200	MACH = .165			

## DEFINITION OF

## DEFINITION OF

C:\37-B B16CF1 J41 V\\*\\*\\_18 WING TOTAL SURFACE (RDWWS2)

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.993 AN/L = 1.200 MACH = .165

## SECTION ( 1 )WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C					
-.900	.0073	-.0762	-.0659	-.1076	-.1762
-.750	-.1063	-.1619	-.1587	-.3053	-.3344
-.600	-.1012	-.2162	-.0986	-.0087	-.2035
-.450	-.3740	-.1124	.0861	.0676	-.1345
-.300	-.1237	-.1165	.1004	.3096	.0343
-.150	.2073	.0710	-.0620	.1746	.1525
.150	-.1939	-.3956	-.3409	-.8358	-.8937
.300	-.3578	-.4296	-.4686	-.5816	-.7288
.450	-.2305	-.3778	-.3631	-.3885	-.6718
.600	-.0292	-.2313	-.1942	-.2297	-.3406
.750	-.0792	-.1139	-.1003	-.0354	-.1530
.900	-.1306	.0540	-.0011	.0825	-.0230

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.990 AN/L = 1.200 MACH = .165

## SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C					
-.900	.0617	-.0449	-.1086	-.0944	-.2144
-.750	-.0372	-.1004	-.1221	-.2314	-.2961
-.600	-.0022	-.1370	-.0726	.0897	-.1366
-.450	-.2530	.0169	.2004	.1676	.0285
-.300	.0216	.0439	.3461	.4127	.1209
-.150	.2862	.2036	.0432	.3382	.2638
.150	-.2781	-.6931	-.6646	-.1.1032	-.1.2993
.300	-.4116	.5335	-.5538	-.7293	-.9147
.450	-.2356	-.4426	-.4611	-.4711	-.7554
.600	-.0665	-.2723	-.2826	-.2973	-.4316
.750	-.1010	-.1526	-.1437	-.0901	-.2644
.900	-.1496	.0455	.0068	.0346	-.1211

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.990 AN/L = 1.200 MACH = .165

## SECTION ( 1 )WING

DEPENDENT VARIABLE CP

X/C					
-.900	.0676	-.0371	-.0760	-.1372	-.2644
-.750	.0237	-.0416	-.0992	-.1628	-.2480
-.600	.0946	-.0634	-.0323	.1735	.0421
-.450	-.1572	.1425	.12402	.2^r^2	.0715
-.300	1.45	1.535	.2912	.4193	.2070

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TABULATED SOURCE DATA - CAS7B

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MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.990

SECTION ( 1 ) WING

CA37-B B16CSF1 J41 WOTE10 WING TOTAL SURFACE

21/B .0000 .3340 .5200 .6630 .8730

X/C

-.150	.3668	.3070	.0999	.3963	.3843
-.150	-.2986	-.11486	-.7677	-1.2760	-1.6099
.300	-.4937	-.7501	-.6166	-.8303	-.0865
.450	-.2705	-.5836	-.5045	-.5163	-.0554
.600	-.0835	-.3180	-.3162	-.3197	-.6939
.750	-.1123	-.1786	-.1998	-.0461	-.5432
.900	-.1521	.0268	-.0128	.0768	-.3128

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C

-.900	.1296	-.0458	-.1507	-.3198	-.3698
-.750	.0744	.0106	-.1203	-.2203	-.2545
-.600	.1648	.0841	.0663	.2220	.0265
-.450	-.0517	.2515	.2572	.3377	.1656
-.300	.2846	.2629	.2468	.5505	.3021
-.150	.1583	.4154	.0273	.4299	.4670
.150	-.3662	-.14991	-.7662	-1.2837	-.2177
.300	-.5502	-.8868	-.6892	-1.3366	-.11384
.450	-.3461	-.6862	-.7707	-1.0350	-.0982
.600	-.1565	-.4245	-.5848	-.9292	-.0195
.750	-.2226	-.3182	-.5517	-.3505	-.8341
.900	-.2217	-.1024	-.2148	-.2953	-.5819



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TABULATED SOURCE DATA - OA37B

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OA37-B B16C5F1 J41 W07E18 WING TOTAL SURFACE

(ADWMS) ( 12 NOV 71 )

## REFERENCE DATA

SREF	*	4.4120 SQ.FT.	XMAP	=	43.5940 IN.	BETA	=	.000
LREF	*	19.2300 IN.	YMAP	=	.0000 IN.	H/B	=	.1000
BREF	*	37.9350 IN.	ZMAP	=	-.4030 IN.	ELEVON	=	-10.000
SCALE	*	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = -4.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730
X/C					
-.900	-.1248	-.1125	-.1075	-.1206	-.1746
-.750	-.2023	-.1874	-.1770	-.3056	-.3740
-.600	-.2388	-.2701	-.1947	-.1670	-.3371
-.450	-.4820	-.2863	-.0801	-.1605	-.3357
-.300	-.4060	-.4353	-.2143	-.0311	-.2267
-.150	.0611	-.2701	-.5494	-.3044	-.2702
.150	-.0466	-.1228	-.2533	-.3220	-.3102
.300	-.1628	-.2108	-.3210	-.3267	-.4279
.450	-.0928	-.2340	-.3367	-.2677	-.6245
.600	.0072	-.1638	-.1625	-.1126	-.2777
.750	-.0554	-.0737	-.1268	-.0170	-.0997
.900	-.1483	.3554	.0773	.1006	.0104

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.005 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730
X/C					
-.900	-.0866	-.0666	-.0723	-.1032	-.1607
-.750	-.1660	-.1504	-.1343	-.2908	-.3565
-.600	-.1582	-.2697	-.1399	-.1024	-.2812
-.450	-.4011	-.1970	-.0365	-.0548	-.2333
-.300	-.2768	-.2873	-.0907	-.1246	-.0992
-.150	-.1277	-.1	-.3013	-.0721	-.0357
.150	-.0869	-.2369	-.4218	-.5783	-.5664
.300	-.2422	-.3092	-.4165	-.4733	-.5717
.450	-.1390	-.3110	-.3615	-.3365	-.7111
.600	-.0190	-.1663	-.1637	-.1455	-.3053
.750	-.0719	-.0912	-.1235	-.0364	-.1431
.900	-.1446	.0494	.0607	.0926	.0058

(RDW453)

CA57-B B16C5F1 J41 W0TE10 WING TOTAL SURFACE

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.985 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

21/B .0000 .3340 .5200 .6630 .8730

## DEPENDENT VARIABLE CP

X/C -.900 -.0518 -.0770 -.0802 -.0963 -.1901

-.750 -.1041 -.1076 -.1554 -.2120 -.3371

-.600 -.0692 -.2248 -.0870 .0180 -.2106

-.450 -.2971 -.0883 -.1086 -.0855 -.1428

-.300 -.1222 -.0987 -.1144 .3198 .0278

-.150 .2011 .0764 -.0762 .1727 .1523

.150 -.1635 -.4084 -.5643 -.8579 .9302

.300 -.3139 -.4354 -.4890 -.6124 -.7540

.450 -.1897 -.3829 -.3876 -.3910 -.7433

.600 -.0602 -.2379 -.2035 -.1843 -.3633

.750 -.1083 -.1187 -.1146 -.0593 -.1825

.900 -.1510 .0364 -.0249 .0362 -.0397

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.980 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C -.900 .0048 -.0498 -.0778 -.0579 -.2210

-.750 -.0370 -.0628 -.1218 -.1882 -.2967

-.600 .0273 -.2173 -.0121 .1309 -.1376

-.450 -.1693 .0302 .1960 .2148 -.0267

-.300 .0279 .0496 .3626 .4366 .1240

-.150 .2854 .2095 .0652 .0581 .2618

.150 -.2748 -.5970 -.6533 -.1099 -.13216

.300 -.4213 -.5289 -.5569 -.7317 -.9208

.450 -.2290 -.4630 -.4578 -.4729 -.7879

.600 -.0641 -.2752 -.2534 -.2618 -.496

.750 -.1002 -.1332 -.1584 -.0989 -.2738

.900 -.1640 .0276 .0026 .0246 -.1346

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.985 RNL = 1.200 MACH = .165

## SECTION ( 1 )WING

## DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C -.900 .0643 -.0417 -.0586 -.0665 -.2655

-.750 .0263 -.0101 -.0916 -.1142 -.2443

-.600 .1146 .0596 .0730 .2330 -.0211

-.450 -.0864 .1409 .2146 .3186 .0837

-.300 .1572 .1745 .2924 .5384 .2190

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TABULATED SOURCE DATA - CA57B

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MACH ( 1 ) = .165    ALPHA ( 5 ) = 14.985  
 SECTION ( 1 ) WING  
 CA57-B    B16CSF1 J41 WATE16 WING TOTAL SURFACE (RDWWS3)

## DEPENDENT VARIABLE CP

X/C	.3600	.5163	.1543	.5475	.3975
-1.50	.3047	-1.1431	-.7266	-1.2737	-1.6231
-1.50	-.3004	-.7088	-.5917	-.6184	-.0938
-1.50	-.2454	-.5737	.5031	-.5170	.8825
-1.50	-.0930	-.3106	-.2931	-.3039	-.7162
-1.50	-.1403	-.1232	-.2104	-.0493	-.5497
.500	-.1567	.0206	-.0074	.0746	-.3138

MACH ( 1 ) = .165    ALPHA ( 6 ) = 19.980    RNL = 1.200    MACH = .165

## DEPENDENT VARIABLE CP

X/C	.0918	.0700	-.1344	-.2060	-.4011
-1.50	.0655	.0111	-.1326	-.1292	-.2684
-1.50	-.1859	.0956	.0546	.2882	.0145
-1.50	-.0095	.2438	.1986	.3804	.1451
-1.50	-.3275	.2794	.1980	.5236	.2946
-1.50	.4479	.4058	.1163	.5192	.4534
-1.50	-.3516	-1.5120	-.7508	-1.2748	-1.1385
-1.50	-.5093	-.6691	-.6741	-1.2512	-.1.0927
-1.50	-.3268	-.6671	-.7724	-1.1603	-.0480
.500	-.1693	-.4383	-.6207	-.9532	-.0222
.750	-.2735	-.1939	-.5746	-.4220	-.8472
.900	-.2430	-.1029	-.2722	-.3321	-.6421

(RDWV4) ( 12 NOV 73 )

CAA37-B B16C5F1 J41 W07E16 WING TOTAL SURFACE

## REFERENCE DATA

SREF	=	4.4120 SQ.FT.	XMAP	=	43.5940 IN.
LREF	=	19.2300 IN.	1MAP	=	.0000 IN.
BREF	=	37.935C IN.	2MAP	=	-.4050 IN.
SCALE	=	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	.4968	.1640	.0213	-.0256	-.4351
-.750	.3611	.3226	.2202	-.2017	-.5676
-.600	.4627	.3481	.2152	.0572	-.3722
-.450	.2755	.3239	.3374	.0417	-.2516
-.300	.4761	.4746	.3699	.3548	-.0629
-.150	.4826	.5016	.2240	.2937	.1330
.150	-.2640	-.6990	-.7026	-.1007	-.13582
.300	-.4304	-.5458	-.5426	-.7096	-.9437
.450	-.2475	-.4522	-.4404	-.4555	-.7960
.600	-.0726	.2775	-.2957	-.3059	-.4651
.750	-.1039	.0040	-.2500	-.0575	-.3018
.900	-.1707	.0250	.0157	.0755	-.2181

MACH ( 1 ) = .163 ALPHA ( 2 ) = 15.005 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	.6422	.3333	.1471	.0205	.4169
-.750	.5522	.4755	.2873	-.0479	-.4032
-.600	.5187	.4194	.1914	.0583	-.0826
-.450	.3620	.5745	.3471	.0874	-.0265
-.300	.5168	.5132	.2527	.3153	.1427
-.150	.5280	.5373	.0343	.4113	.3245
.150	-.3189	-.12277	-.7237	-.3030	-2.6393
.300	-.5421	-.7316	-.5812	-.7713	-1.6507
.450	-.2856	-.5608	-.4996	-.4843	-1.1705
.600	-.1099	-.3190	-.3600	-.3363	.8774
.750	-.1547	-.0151	-.3557	-.1249	-.6450
.900	-.1717	.0060	-.0355	.0419	-.3749



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TABULATED SOURCE DATA - CASTS

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		CAST-8			WING TOTAL SURFACE			(ADYNMSA)				
		B16CSF1	J41	WTE16	WING	TOTAL	SURFACE					
MACH	( 1 )	=	.165	ALPHA ( 3 )	=	19.995	RNL	=	1.200	MACH	=	.165
SECTION ( 1 ) WING												
21/8	.0000	.	.3340	.	.5200	.	.6630	.	.6730	.	.	DEPENDENT VARIABLE CP
X/C												
-.900	.7885	.	.3953	.	.1559	.	.1242	.	.4653	.	.	
-.750	.6268	.	.5504	.	.3272	.	.1206	.	.3542	.	.	
-.600	.5840	.	.4870	.	.1934	.	.1177	.	.0154	.	.	
-.450	.4292	.	.6195	.	.4319	.	.1629	.	.0965	.	.	
-.300	.5743	.	.5451	.	.2258	.	.2242	.	.2541	.	.	
-.150	.5820	.	.5822	.	.1521	.	.0542	.	.3935	.	.	
.150	-.3806	-	.15857	-	.7440	-	.1274	-	.2475	.	.	
.300	-.5681	-	.8903	-	.7422	-	.1373	-	.11797	.	.	
.450	-.3907	-	.7184	-	.6373	-	.1.0912	-	.1.1428	.	.	
.600	-.2351	-	.5196	-	.7069	-	.9628	-	.0599	.	.	
.750	-.2918	-	.2123	-	.6701	-	.3394	-	.8945	.	.	
.900	-.2616	-	.1268	-	.3584	-	.4765	-	.6365	.	.	

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TABULATED SOURCE DATA - CA378

CA37-B B10C5F1 J41 WATE16 WING TOTAL SURFACE

(RDW455) ( 12 NOV 73 )

## REFERENCE DATA

BREF =	4.4120 SQ.FT.	XHYP =	43.9840 IN.	BETA =	.000	PTN/P =	1.300
LREF =	19.2300 IN.	YHYP =	.0000 IN.	H/B =	.039	BDFLAP =	-10.000
BREF =	37.9750 IN.	ZHYP =	.4050 IN.	ELEVON =	.000		
SCALE =	.0405						

MACH ( 1 ) = .165 ALPHA ( 1 ) = 9.990 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C							
-.900	.4560	.1806	.0206	-.0414	-.4293		
-.750	.3631	.3235	.2536	-.1925	-.5584		
-.600	.4550	.3405	.1946	.0589	-.3675		
-.450	.2695	.5191	.3411	.0480	-.2499		
-.300	.4686	.4681	.3773	.3597	-.0616		
-.150	.4806	.4972	.2378	.2964	.1354		
.300	-.2476	-.7010	-.7010	-.10935	-.13651		
.450	-.4233	-.5423	-.5441	-.7133	-.9426		
.600	-.2304	-.4554	-.4371	-.4565	-.7972		
.750	-.0710	-.2717	-.2697	-.3078	-.4667		
.900	-.1175	.0169	-.3267	-.0677	-.2992		

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.990 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C							
-.900	.6221	.3284	.1219	.0194	-.4136		
-.750	.5487	.4729	.2957	-.0590	-.4077		
-.600	.5144	.4115	.1779	.0646	-.0626		
-.450	.3575	.5708	.3703	.0852	-.0246		
-.300	.5103	.5083	.2356	.3050	.1343		
-.150	.5261	.5354	.0635	.3794	.3255		
.300	-.3113	-.12323	-.7159	-.13056	-.2639		
.450	-.5231	-.7172	-.6035	-.1577	-.6586		
.600	-.2686	-.5664	-.5219	-.4157	-.1943		
.750	-.1093	-.3034	-.3445	-.3546	-.8548		
.900	-.1564	.0080	-.2297	-.1194	-.6619		



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TABULATED SOURCE DATA - Q3378

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CASE-B 816C5F1 J41 WOTE16 WING TOTAL SURFACE

(RDW455)

MACH ( 1 ) = .165    ALPHAI ( 3 ) = 19.995    AN/L = 1.2000 MACH = .165

SECTION 111MPC

DEPENDENT VARIABLE CP

X/C					
- .900	.7702	.3928	.1416	-.1462	-.4722
- .775	.8204	.5377	.3231	-.1279	-.3593
- .600	.5572	.4865	.1609	.1209	-.0231
- .450	.4244	.6134	.4192	.1466	.0893
- .300	.5660	.5349	.2330	.2096	.2467
- .150	.5157	.5741	-.1634	-.0543	.3824
.150	-.3779	-1.6122	-.7635	-1.7649	-1.2344
.300	-.5794	-.9022	-.7510	-1.3425	-1.1794
.450	-.3695	-.7196	-.8621	-1.1056	-1.1583
.600	-.2412	-.5261	-.7240	-.9412	-1.0398
.750	-.2820	-.1639	-.6250	-.5676	-.8905
.900	-.2641	-.1338	-.4349	-.4410	-.6423

## CALCULATED SOURCE DATA - CASTE

(ADW56) (12 NOV 73)

## REFERENCE DATA

SREF = 4.4120 93.FT. XMAP = 43.5940 IN.  
 LREF = 19.2300 IN. YMAP = .0000 IN.  
 BREF = 37.8350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH (1) = .165 ALPHA (1) = 10.005 RNL = 1.200 MACH = .165

## SECTION (1)WING

DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
- .900	.1451	.0944	.0449	-.0568	-.4433
- .750	.2147	.1643	.2335	-.1681	-.5156
- .600	.3941	.2267	.1993	-.0270	-.2663
- .450	.2664	.4831	.3212	.0482	-.1659
- .300	.4601	.4451	.5922	.4021	-.0369
- .150	.4737	.4656	.2557	.3788	.1669
.150	-.2531	-.6975	-.6952	-1.0889	-1.3774
.300	-.4500	-.5560	-.5427	-.7201	-.9539
.450	-.2271	-.4497	-.4434	-.4606	-.8174
.600	-.0795	-.2636	-.2776	-.3087	-.4645
.750	-.1224	.0126	-.1287	-.0781	-.3213
.900	-.1642	-.0119	-.0902	.0613	-.2465

MACH (1) = .165 ALPHA (2) = 15.020 RNL = 1.200 MACH = .165

## SECTION (1)WING

DEPENDENT VARIABLE CP

X/C	.0000	.3340	.5200	.6630	.8730
- .900	.3353	.2042	.0823	.0012	-.4059
- .750	.3760	.2993	.3536	-.0349	-.3572
- .600	.4656	.3134	.1961	.1228	-.0373
- .450	.3630	.5330	.3569	.1310	.0046
- .300	.4934	.4818	.2393	.3261	.1611
- .150	.5114	.5163	.1092	.4315	.3410
.150	-.5307	-.12374	-.7214	-1.3199	-2.6486
.300	-.5297	-.7529	-.5962	-.7871	-1.6683
.450	-.2506	-.5347	-.5434	-.5225	-1.2715
.600	-.1239	-.3146	-.3500	-.3457	-.8439
.750	-.1603	-.0561	-.2190	-.1409	-.6057
.900	-.1708	-.0049	-.1633	-.0504	-.4106



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TABULATED SOURCE DATA - DATA

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DATA7-B B16CF1 J41 WATE10 WING TOTAL SURFACE  
(ADDITIONAL)

MACH (.1) = .165 ALPHA (.3) = 20.015 AN/L = 1.260 MACH = .165

SECTION 11 WING

X/C						
- .900	.4710	.2470	.0370	-.0971	-.4423	
- .750	.4935	.3006	.3666	-.0338	-.3275	
- .600	.5326	.3823	.2957	.1939	-.0232	
- .450	.4332	.5822	.4517	.2295	.1066	
- .300	.5623	.5237	.2772	.2762	.2686	
- .150	.5753	.5753	-.1541	-.0196	.4067	
.150	-.3614	-.15839	-.7532	-.7199	-.1.2145	
.300	-.5701	-.9062	-.7516	-.1.3522	-.1.1820	
.450	-.3646	-.7255	-.6700	-.1.1257	-.1.1590	
.600	-.2332	-.5216	-.7176	-.9764	-.1.0443	
.750	-.2773	-.2696	-.5991	-.5771	-.8952	
.900	-.2337	-.1295	-.4007	-.4503	-.6437	

DEPENDENT VARIABLE CP

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TABULATED SOURCE DATA - CASTB

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## CAST-B B16C3F1 J41 WATE18 WING TOTAL SURFACE

## REFERENCE DATA

SREF	=	4.4120 IN. FT.	XHWP	=	43.3940 IN.
LREF	=	18.2300 IN.	ZHWP	=	.0000 IN.
BREF	=	37.9350 IN.	ZHWP	=	-.4050 IN.
SCAL_	=	.0405			
MACH ( 1 ) =		.165	ALPHA ( 1 ) =		-.3.960

## SECTION ( 1 ) WING

## DEFINITION VARIABLE CP

Z/E	.0000	.3340	.5200	.6630	.8730
X/C					
- .900	-.1166	-.2525	-.2956	-.2315	-.2429
- .750	-.4183	-.5829	-.3352	-.4624	-.4657
- .600	-.5310	-.6411	-.4200	-.5242	-.4285
- .450	-.9128	-.5960	-.2620	-.3065	-.4465
- .300	-.7682	-.7735	-.3190	-.1223	-.3169
- .150	.0273	-.4663	-.7386	-.4221	-.3948
.150	-.0626	-.1553	-.2913	-.3646	-.3334
.300	-.2050	-.2433	-.3617	-.3619	-.4675
.450	-.1141	-.2608	-.3763	-.3173	-.5857
.600	-.3.01	-.1949	-.1619	-.2422	-.3253
.750	-.5653	-.5338	-.1558	-.0550	-.1510
.900	-.1788	.0701	.0205	.0113	-.0312

MACH ( 1 ) = .65 ALPHA ( 2 ) = -.005 ANL = 1.200 MACH = .165

## DEFINITION VARIABLE CP

Z/E	.0000	.3340	.5200	.6630	.8730
X/C					
- .900	-.0526	-.1578	-.1326	-.1379	-.1675
- .750	-.2978	-.2855	-.2382	-.3116	-.3984
- .600	-.2827	-.4346	-.2439	-.1576	-.2150
- .450	-.6082	-.2980	-.0999	-.1125	-.2658
- .300	-.3324	.3552	-.1569	.0980	-.1338
- .150	-.1492	-.12C2	-.3113	-.1159	-.0717
.150	-.1132	-.2618	-.4370	-.6056	-.5824
.300	-.2666	-.3253	-.43C4	-.4875	-.5913
.450	-.15C4	-.3246	-.3993	-.3662	-.7330
.600	-.2133	-.2023	-.25C4	-.298C	-.3268
.750	-.C4C6	-.C471	-.1274	-.C354	-.1576
.900	-.1572	.C696	.C254	.C493	-.2155

|||||

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TABULATED SURFACE DATA - QAS7B

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SECTION ( 1 ) WING		MACH ( 1 ) = .165		ALPHA ( 3 ) = 4.960		RNL = 1.200	MACH = .165	GA37-B B16CF1 J41 WTE10 WING TOTAL SURFACE (830457)	
						DEPENDENT VARIABLE CP			
X/C									
.900	.0321	-.1107	-.0959	-.1436	-.2171				
.750	-.1451	-.1593	-.1669	-.3351	-.3688				
.600	-.0684	-.1987	-.1147	-.0203	-.2355				
.450	-.3350	-.0519	-.1101	.0712	-.1669				
.300	-.0411	-.0479	-.1395	.3219	.0056				
.150	2.139	1.880	-.0435	1.865	1.337				
.150	-.2417	-.5233	-.5806	-.8931	-.9703				
.300	-.3696	-.5322	-.5015	-.8276	-.7712				
.450	-.2156	-.4363	-.3941	-.4243	-.7865				
.600	-.450	-.2335	-.2305	-.2948	-.5937				
.750	-.0644	-.0697	-.1297	-.0753	-.1934				
.900	-.1565	.0651	-.0451	.0656	-.0597				
		MACH ( 1 ) = .165		ALPHA ( 3 ) = 9.995		RNL = 1.200	MACH = .165		
SECTION ( 1 ) WING						DEPENDENT VARIABLE CP			
X/C									
.900	.1701	-.0217	-.0657	-.1723	-.2474				
.750	.0124	-.0225	-.0777	-.2642	-.3160				
.600	.0979	.0132	-.0631	.0969	-.1453				
.450	-.1368	.1450	.2366	.1694	-.0423				
.300	.1403	.1508	.3755	.4245	.1067				
.150	.3434	.2689	.1399	.3476	.2780				
.150	-.2473	-.6833	-.6864	-.1.025	-.1.3627				
.300	-.4054	-.5372	-.5797	-.7501	-.9404				
.450	-.2151	-.4485	-.4523	-.4934	-.8166				
.600	-.0638	-.2650	-.2216	-.3509	-.4780				
.750	-.1138	-.1159	-.1654	-.0869	-.2931				
.900	-.1585	.0176	-.0242	.0850	-.1511				
		MACH ( 1 ) = .165		ALPHA ( 3 ) = 15.020		RNL = 1.200	MACH = .165		
SECTION ( 1 ) WING						DEPENDENT VARIABLE CP			
X/C									
.900	.2103	.0400	-.0367	-.1056	-.3021				
.750	.1374	.1081	.0297	-.3236	-.2690				
.600	.2159	.1728	.1246	.0563	-.0051				
.450	-.0129	.2662	.2778	.1733	.0795				
.300	.2148	.2590	.1454	.4144	.2201				

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TABULATED SOURCE DATA - CASE 78

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CASE 7-B B1605F1 J41 WTE10 WING TOTAL SURFACE

MACH (.1) = .165 ALPHA (.5) = 15.020

SECTION (1) WING

21/8 .0000 .3340 .5200 .6630 .6730

DEPENDENT VARIABLE CP

X/C	Cp	Cp	Cp	Cp	Cp
-.190	.4260	.4056	-.1956	.2223	.3876
.190	-.3164	-.1675	-.7556	-1.2417	-.7927
.300	-.4979	-.7242	-.6206	-.7633	-.1396
.450	-.2352	-.5854	-.5436	-.5017	-.9037
.600	-.0986	-.3194	-.3626	-.3269	-.6973
.750	-.1390	-.1394	-.2315	-.1226	-.5223
.900	-.1630	.0103	-.1297	-.0336	-.3336

MACH (.1) = .165 ALPHA (.6) = 19.995

SECTION (1) WING

21/8 .0000 .3340 .5200 .6630 .6730

DEPENDENT VARIABLE CP

X/C	Cp	Cp	Cp	Cp	Cp
-.900	.3994	.0514	-.1640	-.4680	-.3982
-.750	.2156	.1721	.0017	-.3666	-.2567
-.600	.3158	.1940	.1548	.1319	.0615
-.450	.1174	.3916	.3510	.3263	.1533
-.300	.3960	.2980	.3605	.5411	.3115
-.150	.5060	.4634	.0625	.4372	.4492
.150	-.5332	-.15590	-.7611	-.14350	-.12864
.300	-.5398	-.8839	-.7176	-.35646	-.11959
.450	-.3470	-.6901	-.6176	-.1163	-.11592
.600	-.1857	-.4311	-.6675	-.9637	-.04873
.750	-.2662	-.5141	-.0567	-.4637	-.6986
.900	-.2412	-.1292	.3453	-.3543	-.6286

CAST-B S13C5F1 J41 WOTEST WING TOTAL SURFACE

(RDY450) : 12 NOV 73 )

## REFERENCE DATA

	SREF = 4.41-.3 BA.FT.	XMAP = 43.5940 IN.	BETA = .000	PTN/P = 1.000
LREF = 10.2300 IN.	(MAP = .0600 IN.	M/B = .125	BDFLAP = -10.000	
CALI = .37.9350 IN.	ZMAP = -.4050 IN.	ELEVON = .000		
SCBLE = 0403				

MACH ( 1 ) = .165 ALPHA ( 1 ) = -3.995 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .3000 .3340 .3200 .6630 .8730

X/C	MACH ( 1 ) = .165 ALPHA ( 2 ) = -.605 RNL = 1.200 MACH = .165
-.900	-.2025 -.1966 -.1991 -.1784 -.2143
-.750	-.3546 -.3431 -.2572 -.3598 -.4273
-.600	-.4138 -.4902 -.2977 -.2191 -.3818
-.450	-.6757 -.6098 -.1901 -.2285 -.3930
-.300	-.6458 -.6644 -.2335 -.0629 -.2694
-.151	.0546 -.3633 -.3652 -.3223 -.3074
.150	-.0601 -.1469 -.2144 -.3501 -.3466
.300	-.2043 -.2410 -.3560 -.3595 -.4651
.450	-.1002 -.2700 -.3693 -.3664 -.5987
.600	-.0048 -.1719 -.1893 -.2244 -.3271
.750	-.0543 -.0468 -.1432 -.0392 -.1465
.900	-.1686 .1046 .0277 .0221 -.0140

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

X/C	MACH ( 1 ) = .165 ALPHA ( 2 ) = -.605 RNL = 1.200 MACH = .165
-.900	-.1129 -.1406 -.1120 -.1348 -.1695
-.750	-.2580 -.2447 -.1630 -.3271 -.3887
-.600	-.2102 -.3894 -.1604 -.1105 -.3002
-.450	-.4700 -.3074 -.0751 -.0445 -.2665
-.300	-.2851 -.2904 -.1253 .1248 -.1217
-.150	.1572 -.0828 -.2663 -.0765 -.0366
.150	-.1312 -.2633 -.4394 -.6001 -.6042
.300	-.2802 -.3352 -.4412 -.4967 -.5988
.450	-.1571 -.3237 -.4002 -.3681 -.7180
.600	-.0169 -.2020 -.1972 -.2555 -.3316
.750	-.0883 -.0777 -.1329 -.0443 -.1841
.900	-.1624 .0817 .0132 .0462 -.0170

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## TABULATED SOURCE DATA - QAS7B

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MACH ( 1) = .165	ALPHA ( 3) = 4.995	R/N/L = 1.200	MACH = .165	(RDWWS)
SECTION ( 1) WING	DEPENDENT VARIABLE CP			
21/B	.0000	.3340	.5200	.6630 .8730
X/C				
- .900	-.0113	-.0983	-.0916	-.1119 -.2031
- .750	-.1282	-.1449	-.1338	-.2778 -.3465
- .600	-.0246	-.2111	-.0568	.0472 -.2095
- .450	-.2338	-.0550	-.1299	.0984 -.1334
- .300	-.0160	-.0099	-.1728	.3451 .0154
- .150	.2622	.1699	-.0080	.2167 .1665
.150	-.2376	-.4962	-.5759	-.5717 -.9783
.300	-.3766	-.5238	-.5043	-.219 -.7732
.450	-.2029	-.4286	-.3913	-.4168 -.7667
.600	-.0134	-.2229	-.2152	-.2793 -.3657
.750	-.0912	-.0819	-.1247	-.0691 -.2014
.900	-.1571	.0601	-.0466	.0724 -.0544

CA57-B B16C5F1 J42 WTE10 WING TOTAL SURFACE (RDW55) (12 NOV 73)

## REFERENCE DATA

REF	X	Y	Z	WING	WINGP	WINGT	WINGP	WINGT	BETA	PIN/P
LREF	4.4120	.83	F							
LREF	=	10.2300	IN.	1WING	=	.0000	IN.			
BREF	=	37.9350	IN.	2WING	=	-.4030	IN.			
S ALR	=	.0405								

MACH ( 1 ) = .165    ALPHA ( 1 ) = -4.025    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6650	.8730				
- .900		-.2950	-.2050	-.2007	-.2298	-.2351				
- .750		-.3613	-.3383	-.4520	-.4820	-.4467				
- .600		-.4722	-.4535	-.4039	-.3916	-.4735				
- .450		-.6532	-.4736	-.3467	-.4490	-.4266				
- .300		-.7233	-.4260	-.4114	-.3516	-.2941				
- .150		-.0140	-.3982	-.5062	-.4556	-.3755				
.150		-.0645	-.1618	-.3507	-.5961	-.3604				
.300		-.2054	-.2301	-.6555	-.3967	-.4847				
.450		-.0185	-.4575	-.3370	-.3347	-.7213				
.600		-.0435	-.2176	-.2038	-.2397	-.3535				
.750		-.0497	-.1427	-.1069	-.0476	-.1251				
.900		-.1652	.0035	-.0306	.0753	-.0156				

MACH ( 1 ) = .165    ALPHA ( 2 ) = -.010    RNL = 1.200    MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C	21/8	.0000	.3340	.5200	.6650	.8730				
- .900		-.2634	-.1908	-.1367	-.1740	-.1928				
- .750		-.2936	-.2340	-.3604	-.3863	-.4017				
- .600		-.2822	-.2842	-.2317	-.2326	-.3537				
- .450		-.5997	-.2507	-.1438	-.2502	-.2821				
- .300		-.3425	-.1606	-.1483	-.1265	-.05				
- .150		.0999	-.1503	-.2345	-.1268	-.0576				
.150		-.1291	-.3263	-.6336	-.8253	-.6232				
.300		-.2705	-.3698	-.7545	-.5082	-.6161				
.450		-.0745	-.5143	-.4051	-.3681	-.7784				
.600		-.0703	-.2556	-.2372	-.2451	-.5626				
.750		-.0704	-.1484	-.1306	-.0555	-.1307				
.900		-.1499	-.0206	-.0503	.0666	-.0114				

## P. PARAMETER DATA



(REV459)

CA37-B B16C5F1 J42 WATE10 WING TOTAL SURFACE

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.980

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C	.000	.3190	.3104	.3751	.3302
.150	-.4074	-.7601	-.1689	-.5250	-2.1746
.300	-.6206	-.6940	-.1550	-.9984	-1.8141
.450	-.2629	-.7503	-.6938	-.7616	-1.5476
.600	-.1500	-.3789	-.4748	-.6532	-1.2817
.750	-.1299	-.2268	-.2487	-.2075	-.7595
.900	-.1419	-.0202	-.0298	-.1125	-.4715

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.975 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING DEPENDENT VARIABLE CP

X/C	.000	.3340	.3200	.6630	.8730
-.000	.0966	.0854	.0242	-.0577	-.4689
-.150	.2037	.1973	.1133	-.0143	-.3154
-.300	.2841	.2873	.2730	.2016	-.0205
-.450	.0997	.3351	.4031	.2358	.0862
-.600	.3620	.3959	.4222	.3920	.2126
-.750	.4556	.5899	.4229	.3971	.3104
-.900	-.4605	-.6507	-.9338	-.2.0923	-.1.7237
-.370	-.6937	-.1538	-.1.3280	-.1.6983	-.1.5436
-.450	-.3826	-.7732	-.9670	-.1.3718	-.1.4965
.600	-.2377	-.4168	-.8026	-.1.2640	-.1.4013
.750	-.1544	-.2590	-.2633	-.6360	-.1.1607
.900	-.1534	-.1248	-.0711	-.5224	-.8077

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TABULATED SOURCE DATA - CAA57B

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(RDW460) 12 NOV 73 1

## REFERENCE DATA

SREF =	4.4120 SQ.FT.	XMRP =	43.5940 IN.
LREF =	19.2300 IN.	YMRP =	.0000 IN.
BREF =	37.9350 IN.	ZMRP =	-.4050 IN.
SCALE =	.0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CF

X/C				
-.900	-.2497	-.2020	-.1754	-.2200
-.750	-.3274	-.3042	-.4592	-.4553
-.600	-.4224	-.4070	-.3577	-.3550
-.450	-.7536	-.4358	-.3114	-.4326
-.300	-.6763	-.3627	-.3845	-.3299
-.150	.0236	-.3589	-.4650	-.4186
.150	-.0554	-.1324	-.2913	-.5099
.300	-.1687	-.1963	-.5799	-.4020
.450	-.0370	-.4078	-.2899	-.2876
.600	-.0354	-.1866	-.1690	-.2038
.750	-.0484	-.1173	-.0914	-.0408
.900	-.1531	.0145	-.0128	.0827

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.010 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

X/C				
-.900	-.2463	-.1931	-.1279	-.1692
-.750	-.2583	-.2341	-.3463	-.3612
-.600	-.2356	-.2429	-.2176	-.2093
-.450	-.3228	-.2177	-.1183	-.2418
-.300	-.3165	-.1359	-.1231	-.1115
-.150	-.1991	-.1040	-.2096	-.1071
.150	-.1556	-.2954	-.3628	-.709
.300	-.2812	-.3366	-.6993	-.4603
.450	-.0932	-.4563	-.3392	-.3347
.600	-.0477	-.2238	-.2078	-.2222
.750	-.0525	-.1313	-.1136	-.0620
.900	-.1422	-.0123	-.0291	.0593

## PARAMETRIC DATA

BETA =	.000	PTN/P = 1.300
M/B =	.125	BDFLAP = -16.000
ELEVON =	.000	



DATE 09 OCT 74

TABULATED SOURCE DATA - CAA78

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## CAAA7-B B16C5F1 J42 WATE18 WING TOTAL SURFACE (RDW460)

MACH ( 1 ) = .165 ALPH A ( 3 ) = 4.963 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .9000	- .1792	- .1108	- .3840	- .1087	- .2122	
- .7500	- .1193	- .0996	- .2165	- .2449	- .3555	
- .6000	- .0561	- .0829	- .0500	- .0544	- .2562	
- .4500	- .3047	- .0160	.0534	- .0519	- .1581	
- .3000	- .0607	.0737	.0931	.0674	- .0347	
- .1500	.2009	.0906	.0311	.1262	.1263	
.1500	- .2528	- .4920	- .9659	- .104	.968	
.3000	- .3496	- .4590	- .8127	- .9995	- .7568	
.4500	- .1431	- .1219	- .4482	- .3877	- .7571	
.6000	- .0808	- .2689	- .2777	- .2731	- .3927	
.7500	- .0628	- .1559	- .1575	- .1198	- .1760	
.9000	- .1420	.0039	- .0528	- .0424	.0486	

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.960 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .9000	- .0556	- .0221	- .0250	- .0519	- .2319	
- .7500	.0079	.0220	- .0699	- .1319	- .3068	
- .6000	.0854	.0775	.0908	.0834	- .1589	
- .4500	- .1308	.1320	.1925	.0630	- .0495	
- .3000	.1047	.2177	.2367	.2016	.0873	
- .1500	.2829	.2187	.1654	.2907	.2503	
.1500	- .2914	- .7019	- 1.1290	- 1.2515	- 1.3090	
.3000	- .4531	- .6047	- .9357	- .7210	- .8452	
.4500	- .1752	- .6573	- .5479	- .5118	- .6346	
.6000	- .1214	- .3310	- .3184	- .3549	- .4574	
.7500	- .1322	- .1877	- .1709	- .0394	- .2993	
.9000	- .1552	.0196	- .0118	.0273	- .1688	

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.960 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

X/C	21/8	.0000	.3340	.5200	.6630	.8730
- .900	.0469	.0458	.0024	.0722	.3214	
- .750	.1267	.1250	.0212	.0508	.3140	
- .600	.1972	.1861	.1774	.1402	-.0561	
- .450	.0090	.2506	.3141	.1075	.0065	
- .300	.2519	.3216	.3356	.3191	.1679	

## CA57-B B16CSF1 J42 WATE16 WING TOTAL SURFACE

(RDW460)

$$\text{MACH } (-1) = .165 \quad \text{ALPHA } (-5) = 14.960$$

## SECTION (-1) WING

## DEPENDENT VARIABLE CP

X/C					
-.150	.3631	.3037	.2957	.3562	.3175
.150	-.3718	-.7929	-.1.1692	-1.4170	-2.6341
.300	-.5542	-.8459	-.1.0351	-.8587	-.81729
.450	-.2347	-.7015	-.6491	-.6701	-.4142
.600	-.1684	-.4017	-.4874	-.5732	-.1.3280
.750	-.1503	-.2303	-.2530	-.2388	-.6986
.900	-.1601	-.0499	-.0440	-.1704	-.3956

$$\text{MACH } (-1) = .165 \quad \text{ALPHA } (-6) = 19.965 \quad \text{ANL} = 1.200 \quad \text{MACH} = .165$$

## SECTION (-1) WING

## DEPENDENT VARIABLE CP

X/C					
-.900	.1002	.0764	-.0186	-.0877	-.4246
-.750	.2106	.1690	.0839	-.0358	-.3193
-.600	.3011	.2828	.2558	.1881	-.0212
-.450	.1245	.3465	.4073	.2262	.0826
-.300	.3678	.4069	.4213	.3917	.2124
-.150	.4640	.3957	.4262	.4173	.3214
.150	-.4330	-.8701	-.9622	-.2.0847	-.1.4249
.300	-.321	-.6438	-.1.2483	-.1.6222	-.1.3408
.450	-.3377	-.7640	-.9675	-.1.2767	-.1.2904
.600	-.2487	-.4545	-.7781	-.1.0124	-.1.1220
.750	-.1970	-.3051	-.3112	-.6792	-.86828
.900	-.1778	-.1545	-.1142	-.5453	-.6538

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## CA37-B B16CF1 J42 W07E16 WING TOTAL SURFACE

## REFERENCE DATA

SREF	2	4,4120 IN.FT.	XMAP =	43.3940 IN.	BETA =	.000	PTN/P =	1.000
LREF	=	19.2300 IN.	YMAP =	.6000 IN.	H/B =	.125	BOFLAP =	-16.000
WIND	=	.1.0350 IN.	ZMAP =	-.4050 IN.	ELEVON =	.000		
SCALE	=	.0405						
MACH (1) =	.165	ALPHA (1) =	-.030	RNL =	1.200	MACH =	.165	

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.2609	-.3164	-.1778	-.2238	-.2276
-.750	-.3334	-.2651	-.4535	-.4456	-.4285
-.600	-.3689	-.3735	-.3610	-.3463	-.4568
-.450	-.5692	-.4200	-.2991	-.4350	-.3964
-.300	-.6061	-.3233	-.3641	-.3087	-.2743
-.150	-.0365	-.3176	-.4409	-.4037	-.3322
.150	-.0457	-.1109	-.2643	-.4727	-.3420
.300	-.1617	-.1658	-.4659	-.3457	-.4570
.450	-.0194	-.3433	-.2532	-.2676	-.7285
.600	-.0241	-.1716	-.1602	-.1959	-.3468
.750	-.0591	-.1221	-.1013	-.0473	-.1053
.900	-.1576	-.0112	-.0367	.0769	.0060

MACH (1) = .165 ALPHA (1) = -.025 RNL = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.2647	-.3598	-.1544	-.1855	-.2022
-.750	-.2621	-.1978	-.3532	-.3634	-.3939
-.600	-.2003	-.2351	-.2228	-.2079	-.3646
-.450	-.4345	-.2190	-.1094	-.2327	-.2732
-.300	-.2942	-.1042	-.1180	-.1096	-.1310
-.150	-.1116	-.0900	-.2045	-.1773	-.0468
.150	-.1461	-.2699	-.5259	-.6748	-.5928
.300	-.2698	-.2978	-.5478	-.4192	-.5614
.450	-.0605	-.3613	-.3124	-.2954	-.1617
.600	-.0471	-.2022	-.1958	-.2140	-.3508
.750	-.0755	-.1333	-.1104	-.0972	-.1256
.900	-.1518	-.0415	-.0304	-.0368	-.0123

## CA57-B B16C3F1 J42 WTE10 WING TOTAL SURFACE (RDW61)

MACH ( 1 ) = .165 ALPHAF ( 1 ) = 4.960 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.1982	-.3432	-.1078	-.1207	-.2089
-.750	-.1275	-.0791	-.2160	-.2332	-.3560
-.600	-.0237	-.0699	-.0153	-.0461	-.2643
-.450	-.2353	-.0192	-.0599	-.0614	-.1569
-.300	-.0529	.0895	.0899	.0768	.0322
-.150	.2105	.1078	.0076	.1323	.1253
.150	-.2211	-.1641	-.8538	-.9164	-.9399
.300	-.3417	-.4148	-.6010	-.5153	-.7209
.450	-.1195	-.4424	-.4017	-.3535	-.7462
.600	-.0755	-.2276	-.2298	-.2532	-.3764
.750	-.0907	.1522	-.1299	-.1350	-.1707
.900	-.1560	-.0413	-.0427	-.0911	-.0439

MACH ( 1 ) = .105 ALPHAF ( 4 ) = 9.965 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.5835	-.2328	-.0700	-.0886	-.2321
-.750	-.0026	.0396	-.1024	-.1716	-.3669
-.600	.1019	.0621	.0628	.0621	.1726
-.450	-.0731	.1318	.199.	.0386	.0616
-.300	.1172	.2249	.2259	.1885	.0657
-.150	.2876	.2152	.1549	.2756	.2417
.150	-.2725	-.6674	-.1073	-.11608	-.1.2160
.300	-.4252	-.5277	-.5086	-.6425	-.7985
.450	-.1536	-.5560	-.4741	-.5206	-.5980
.600	-.1194	-.3249	-.3188	-.3424	-.4532
.750	-.1421	-.1936	-.1958	-.1668	-.3074
.900	-.1698	-.0053	-.0742	-.1373	-.1513

MACH ( 1 ) = .165 ALPHAF ( 5 ) = 14.965 RN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	.0070	-.2894	-.0535	-.1230	-.3196
-.750	.1609	.1168	-.0217	-.0904	.32
-.600	.2036	.1815	.1580	.1220	.5771
-.450	-.0510	.2461	.3117	.0558	.0556
-.300	.2474	.3163	.3260	.3CA4	.16CA



MACH (1) = .165 ALPHAS ( 51 ) = 14.965 (ADYMA61)

SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.0000	.1000	.2000	.3000	.4000	.5000	.6000	.7000	.8000	.9000	.9650
-1.150	.3668	.3093	.2942	.3557	.3215						
-1.10	-.3354	-.7348	-.10910	-.2949	-.5650						
-1.00	-.5162	-.6199	-.9002	-.7008	-.8360						
-0.90	-.2107	-.6318	-.6120	-.5665	-.2415						
-0.80	-.1708	-.5663	-.4635	-.5034	-.1917						
-0.750	-.1865	-.2616	-.3141	-.3411	-.6144						
-0.700	-.1929	-.0716	-.1108	-.3039	-.4026						

MACH (1) = .165 ALPHAS ( 6 ) = 19.970 RNL = 1.200 MACH = .165

SECTION 11 WING

DEPENDENT VARIABLE CP

X/C	.0000	.1340	.2000	.3200	.4200	.5200	.6630	.6730			
-.900	.0468	-.3679	-.0841	-.1507	-.3613						
-.750	.1817	.1706	.0362	.0693	.3333						
-.600	.2884	.2634	.2316	.1709	.0367						
-.450	.1450	.3360	.3950	.1819	.0603						
-.300	.3616	.3990	.6557	.3794	.2007						
-.150	.4972	.3897	.4196	.4246	.3307						
.150	-.4110	-.7978	-.6557	-.7325	-.0726						
.300	-.5628	-.5939	-.1130	-.1293	-.00392						
.450	-.3075	-.7123	-.8175	-.9362	-.9237						
.600	-.2733	-.5016	-.6625	-.8447	-.7235						
.750	-.2406	-.3893	-.4557	-.5120	-.6125						
.900	-.2334	-.2060	-.2237	-.4934	-.5132						

CAST7-B B10C5F1 J42 WATE10 WING TOTAL SURFACE (EDWNE2) (112 NOV 73)

## REFERENCE DATA

BREF	2	4.4120 IN. FT.	XMAP	=	43.9840 IN.	BETA	=	.000 PTN/P = 1.500
LREF	2	19.2300 IN.	YMAP	=	.0000 IN.	H/B	=	.039 BDFLAP = -10.000
SREF	2	37.9350 IN.	ZMAP	=	-.4030 IN.	ELEVON	=	.000
SCALE	2	.0405						
MACH ( 1 )	=	.165	ALPHA ( 1 )	=	10.000	RNL	=	1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730			
X/C								
- .900	.4557	.0657	.0469	-.0118	-.3461			
- .750	.5039	.2311	.0853	-.0984	-.4433			
- .600	.2695	.3200	.2352	.1080	-.2430			
- .450	.1169	.3170	.2866	.1097	-.2079			
- .300	.3374	.3255	.3092	.2427	-.0503			
- .150	.3640	.3082	.2534	.2899	.1526			
.150	-.2657	-.7653	-.12551	-.1.4311	-.4079			
.300	-.4449	-.7070	-.1.0613	-.86CP1	-.9322			
.450	-.1560	-.7124	-.6129	-.62*2	-.7093			
.600	-.1616	-.3805	-.3816	-.4374	-.5935			
.750	-.1754	-.2084	-.1679	-.0856	-.4101			
.900	-.1554	.0323	.0320	.0042	.2393			

MACH ( 1 ) = .165 ALPHA ( 2 ) = 14.980 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

## DEPENDENT VARIABLE CP

21/8	.0000	.3340	.5200	.6630	.8730			
X/C								
- .900	-.2261	-.4364	.5690	-.0310	.1130			
- .750	-.1369	.5110	.3868	12.2764	11.9656			
- .600	.4120	.4122	.4346	.3517	.2170			
- .450	3.028	.0579	.2567	.4270	.4050			
- .300	.2383	.4286	.4347	16.5150	10.2923			
- .150	.4535	.4546	.4112	.4154	.3483			
.150	-.3303	-.3343	-.85C3	-.1.2030	-.1.6995			
.300	-1.8644	-.5099	-.7270	-.6.5174	-.10.5136			
.450	-1.0172	-1.6814	-.1.2022	-.7517	-.7257			
.600	-.1683	-.1710	-.3922	-.5689	-.9000			
.750	-1.5166	-.1587	-.2083270.0000	-.12.9143				
.900	-.3329	-.1.0093	-.1544	-.5286	-.1196			



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TABULATED SOURCE DATA - CAST8

MACH (1) = .165    ALPHAI (3) = 19.595    AN/L = 1.200    MACH = .165  
SECTION 11 MING    DEPENDENT VARIABLE CP

X/C	1.000	.4340	.7200	.6630	.6730
- .900	.7390	-.1164	.1702	.0491	-.5227
- .750	.5948	.4887	.3477	.0337	-.3927
- .600	.4913	.3063	.4042	.2844	-.0194
- .450	.3592	.15102	.4801	.2962	.0191
- .300	.2148	.5102	.4924	-.0866	.0269
- .150	.5285	.4733	.5224	.3253	.2309
.150	-.3911	-.9003	-.9036	-2.6173	-1.7908
.300	-.5865	-.6404	-1.3711	-1.6410	-1.8426
.450	-.2343	-.7524	-1.0409	-1.5032	-1.5753
.600	-.2269	-.4036	-.7803	-1.4297	-1.5269
.750	-.1666	-.2609	-.2506	-.6707	-1.3249
.900	-.1265	-.1176	-.0376	-.5041	-.6710

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TABULATED SURFACE DATA - CA578

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CA57-P B16C5F1 142 WTE10 WING TOTAL SURFACE

## REFERENCE DATA

BALF	4.4120	Bd.F1.	ZNRP	=	43.3940 IN.
LREF	19.2300	IN.	BNRP	=	.0000 IN.
BREF	37.9350	IN.	ZNRP	=	-.4050 IN.
SCALE	.0405				

MACH (.1) = .165 ALFA (.1) = 9.995 ANL = 1.200 MACH = .165

SECTION 1 WING DEPENDENT VARIABLE CP

21/8	.0700	3340	.9200	.6630	.8730
X/C					
- .900	.2259	.0612	.0231	.0370	-.3-64
- .750	.2764	.2167	.0686	-.1113	-.4276
- .600	.3018	.3112	.2346	.1145	-.2328
- .450	.1379	.3147	.2659	.1079	-.1943
- .300	.3470	.3366	.3132	.2446	- C442
- .150	.3765	.3166	.2670	.3033	.1565
.150	- .2675	- .7441	- 1.1752	- 1.293	- 1.3055
.300	- .4273	- .6532	- .9646	- .7525	- 67.4
.450	- .1129	- .6990	- .5198	- .5631	-.6795
.600	- 1.469	- .5479	- .5465	- .4173	-.5662
.750	- 1.736	- .1968	- .1867	- .3676	- 3702
.900	- 1.045	.0229	.0032	-.1376	- 1.977

MACH (.1) = .165 ALFA (.2) = 1.965 ANL = 1.200 MACH = .165

SECTION 1 WING DEPENDENT VARIABLE CP

21/8	.0000	3340	.9200	.6630	.8730
X/C					
- .900	.4054	-.0731	.0682	-.0251	-.4265
- .750	.4651	.3562	.2229	-.0501	-.4382
- .600	.4263	.4162	.3310	.2156	-.1365
- .450	.2810	.4272	.3969	.2176	-.0597
- .300	.4327	.4379	.6146	.3403	.0327
- .150	.4435	.4289	.4136	.5651	2.4
.150	- .5312	- .0374	- 1.1386	- 1.6234	- 1.1937
.300	- 1.562	- .6033	- 1.1074	- 9.984	- 65.72
.450	- 1.0917	- .7220	- .7265	- 6241	- 5.272
.600	- 1.607	- .4069	- .5691	- 1.994	- 1.3537
.750	- 1.925	- .2476	- .2574	- 3.715	- 8.92
.900	- 1.0451	- .5615	- .5275	- 3.51	- 45.



(RDYMA3) ( 12 NOV 73 )

PARAMETRIC DATA

BETA	=	.000	PTN/P	=	1.300
M/B	=	.039	BOFLAP	=	-.16.000
ELEVON	=	.300			

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TABULATED SOURCE DATA - QAS7B

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DATE 08 OCT 74  
SECTION ( 1 ) WING  
X/CMACH ( 1 ) = .165  
ALPHA ( 3 ) = 20.000  
RN/L = 1.200 MACH = .165

(ADVANCE3)

X/C	C17-B	BIGC5F1	J42	WTE10	WING	TOTAL SURFACE
21/76	.0000	.3340	.5200	.6630	.8730	
	DEPENDENT VARIABLE CP					
-.900	.6085	-.1921	.0867	-.0127	-.4562	
-.750	.5622	.4357	.3034	.0127	-.3753	
-.600	.4956	.4904	.3813	.2670	-.0592	
-.450	.3661	.5019	.4711	.2794	.0273	
-.300	.5107	.5093	.4865	.4072	.0415	
-.150	.5243	.4707	.5217	.3640	.2486	
.150	-.3991	-.9032	-.9612	-.2,4687	-.1,4753	
.300	-.5972	-.6570	-.1,3491	-.1,7796	-.1,4213	
.450	-.1934	-.7930	-.1,0629	-.1,3911	-.1,3686	
.600	-.2554	-.4878	-.6147	-.1,1120	-.1,1872	
.750	-.2063	-.3361	-.3267	-.7633	-.9610	
.900	-.1632	-.1749	-.1671	-.5296	-.7233	

CA57-B B16C5F1 J42 WTE10 WING TOTAL SURFACE

(RDWWSA) ( 12 NOV 73 )

## REFERENCE DATA

BREF = 4.4120 IN. FT. XMRP = 43.5940 IN.  
 LREF = 19.2300 IN. YMRP = .0000 IN.  
 DREF = 37.9350 IN. ZMRP = -.4030 IN.  
 BCALC = .0405

MACH ( 1 ) = .165 ALPHA ( 1 ) = 10.000 RVAL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6630 .6730

X/C					
-.900	.0629	.0500	-.0156	-.0795	-.2798
-.750	.1354	.1637	.0305	-.1233	-.3862
-.600	.3111	.2729	.2162	.1134	.2182
-.450	.1931	.3054	.2754	.0878	-.1980
-.300	.602	.3553	.3047	.2445	-.0293
-.150	.3285	.3176	.2693	.3137	.1851
.150	-.2154	-.7102	-.10866	-.11497	-.11484
.300	-.4085	-.5845	-.8265	-.6118	-.6768
.450	-.0325	-.5650	-.4967	-.4599	-.6783
.600	-.1389	-.3397	-.3322	-.3713	-.5664
.750	-.1773	-.1928	-.2122	-.2154	-.3137
.900	-.1688	.0154	.0525	-.1890	-.1713

MACH ( 1 ) = .165 ALPHA ( 2 ) = 15.010 RVAL = 1.200 MACH = .165

## SECTION ( 1 ) WING

DEPENDENT , "TABLE CP

X/C					
-.900	.2175	-.0867	.0093	-.0749	-.3800
-.750	.3462	.2999	.1674	-.0513	-.3986
-.600	.4374	.3600	.3145	.2039	-.1367
-.450	.3222	.4152	.5833	.2034	-.0374
-.300	.4414	.4332	.4983	.3435	.0073
-.150	.4602	.4119	.4148	.3534	.2516
.150	-.5217	-.7967	-.0839	-.4716	-.4216
.300	-.4876	-.6335	-.9396	-.7695	-.3729
.450	-.0397	-.6544	-.6830	-.7233	-.12278
.600	-.1642	-.4118	-.5184	-.6564	-.11221
.750	-.2041	-.2679	-.3654	-.4553	-.7614
.900	-.2047	-.0909	-.1185	-.3982	-.5387



MACH (1) = .165	ALPHA (3) = 18.990	RNL = 1.200	MACH = .165	Q37-B B16CF1 J42 WATE18 WING TOTAL SURFACE						(ADN64)
				DEPENDENT VARIABLE CP						
SECTION 11 WING										
X/C										
-.900	.3480	-.2322	-.0031	-.1007	-.4177					
-,750	.4722	.3784	.2469	.0213	-.4019					
-,600	.5143	.4514	.3616	.2533	-.0969					
-,450	.4116	.4901	.4606	.2629	.0193					
-,300	.3209	.5106	.4822	.4053	.0590					
-,150	.5305	.4806	.5247	.4061	.2457					
,150	-,3749	-,6335	-,6430	-,7095	-,0370					
,300	-,3458	-,5934	-,1.355	-,1.3355	-,1.0092					
,450	-,1140	-,7432	-,8642	-,9715	-,9143					
,600	-,2645	-,5582	-,7112	-,8722	-,7738					
,750	-,2546	-,4569	-,5363	-,6541	-,7049					
,900	-,2241	-,2473	-,2957	-,6119	-,6341					

1  
11  
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TABULATED SOURCE DATA - OASIS

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CAST-E E16C5F1 J42 WATE10 WING TOTAL SURFACE

(ADVW65) ( 12 NOV 73 )

## REFERENCE DATA

WATER =	4.4120	S3.FT.	XMAP =	43.3940 IN.
LREF =	19.2300	IN.	IMAP =	.0000 IN.
BREF =	37.9350	IN.	ZMAP =	-.4050 IN.
SCALE =	.0405			

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.950 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.2534	-.1414	-.1841	-.2270	-.2257
-.750	.3471	-.3220	-.4428	-.4606	-.4185
-.600	-.4702	-.4277	-.3657	-.3657	-.4467
-.450	-.8336	-.4592	-.3714	-.4616	-.3984
-.300	-.6996	-.3926	-.3985	-.3316	-.2808
-.150	.0146	-.3869	-.4933	-.4322	-.3543
.150	.0638	-.1596	-.5395	-.5119	-.3388
.300	-.1838	-.2192	-.6402	-.3965	-.4708
.450	.0036	-.4361	-.3188	-.3222	-.7131
.600	-.0563	-.2082	-.1863	-.2168	-.3291
.750	-.0293	-.1195	-.0960	-.0390	-.1119
.900	-.1461	.0174	-.0135	.0870	-.0079

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.015 RNL = 1.200 MACH = .165

SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.2847	-.1427	-.1355	-.1650	-.1878
-.750	-.3044	-.2986	-.3557	-.3615	-.3623
-.600	-.2739	-.2539	-.2272	-.2189	-.3204
-.450	-.6060	-.2660	-.2595	-.1697	-.2827
-.300	-.3289	-.1230	-.1230	-.1416	-.1124
-.150	.0994	-.1140	-.2265	-.1143	-.0493
.150	-.1162	-.3131	-.6115	-.7248	-.6137
.300	-.2518	-.3629	-.7361	-.5207	-.6127
.450	-.0511	-.5143	-.3965	-.3545	-.7741
.600	-.0669	-.2458	-.2271	-.2367	-.3524
.750	-.0576	-.1369	-.1235	-.0495	-.1202
.900	-.1511	-.0249	-.0421	.0691	-.0037



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## TABLED SOURCE DATA - C457B

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C457-B B16CSF1 J1/2 WSTE10 WING TOTAL SURFACE

(ADJUSTS)

MACH ( 1 ) = .165 ALPHA ( 3 ) = 5.000 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

.0000 .3343 .5200 .6630 .8730

X/C

-.900	-.2208	-.0667	-.0937	-.1150	-.2214
-.750	-.1630	-.1446	-.2375	-.2632	-.3313
-.600	-.0788	-.0777	-.0688	-.0676	-.2291
-.450	-.3568	-.0413	-.0097	-.1056	-.1525
-.300	-.0746	.0755	.0777	.0611	.0050
-.150	.1945	.0821	-.0484	.1222	.1197
.000	-.1929	-.5212	-.10260	-.1.0018	-.1.0036
.300	-.3461	-.5127	-.0966	-.6500	-.7608
.450	-.0930	-.5755	-.5019	-.4138	-.7714
.600	-.1184	-.3066	-.3153	-.2919	-.4025
.750	-.1225	-.1837	-.1913	-.1222	-.1634
.900	-.1415	-.0064	-.0893	-.0330	-.0623

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.990 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

.0000 .3340 .5200 .6630 .8730

X/C

-.900	-.1141	.0013	.0070	-.0417	-.2413
-.750	-.0123	.0007	-.0740	-.1198	-.2794
-.600	.0720	.0948	.0876	.0727	-.1353
-.450	-.1585	.1428	.1796	.0084	-.0455
-.300	-.1132	.2284	.2324	.2067	.0811
-.150	.2828	.2189	.1590	.3029	.2550
.000	-.2673	-.7397	-.1.2062	-.1.2368	-.1.3606
.300	-.4272	-.6352	-.1.0251	-.7462	-.8876
.450	-.1106	-.6635	-.6006	-.3651	-.6713
.600	-.1359	-.3568	-.3543	-.4059	-.4634
.750	-.1527	-.1935	-.1694	-.0933	-.3217
.900	-.1339	.0400	.0342	.0291	-.1688

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.935 AN/L = 1.200 MACH = .165

## SECTION ( 1 ) WING

.0000 .3340 .5200 .6630 .8730

X/C

-.900	.0288	-.0134	.0136	-.1.650	-.3642
-.750	.1039	.1007	.0208	-.1.39	-.2806
-.600	.1699	.2193	.1099	-.1.77	-.3736
-.450	-.0064	.2659	.3007	-.1.51	.0157
-.300	.2496	.3357	.3411	...	.1563

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## TABULATED SOURCE DATA - OASPA

(800466) (112 NOV 73)

## REFERENCE DATA

SREF =	4,1120 IN.	XMAP =	43,1940 IN.	BETA =	.0000 PTN/P = 1,300
LREF =	19,2300 IN.	YMAP =	0,0000 IN.	H/B =	.125 00FLAP = -18.000
DALY =	37,9350 IN.	ZMAP =	-1,050 IN.	ELEVON =	.000
SCALE =	.0405				

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.003 RNL = 1,200 MACH = .165

## SECTION ( 1 ) WING

2/1/8 .0000 .3340 .5200 .6630 .6730

## X/C

-.900	-.2651	-.1393	-.1784	-.2233	-.2319
-.750	-.3491	-.3203	-.4483	-.4510	-.4148
-.600	-.4032	-.3961	-.3649	-.3491	-.4418
-.450	-.7193	-.4373	-.3596	-.5244	-.3974
-.300	-.6433	-.3268	-.3730	-.3116	-.2776
-.150	.0259	-.3436	-.4634	-.4108	-.3401
.150	-.0506	-.1393	-.2916	-.4548	-.3336
.300	-.1834	-.1973	-.5517	-.3665	-.4577
.450	-.0109	-.3982	-.2902	-.2966	-.7549
.600	-.0439	-.1663	-.1701	-.2009	-.3301
.750	-.0441	-.1060	-.0921	-.0425	-.1061
.900	-.1478	.0162	-.0121	.0794	-.0116

MACH ( 1 ) = .165 ALPHA ( 2 ) = -.003 RNL = 1,200 MACH = .165

## SECTION ( 1 ) WING

2/1/8 .0000 .3340 .5200 .6630 .6730

## X/C

-.900	-.2660	-.1643	-.1339	-.1609	-.1963
-.750	-.2727	-.2530	-.3526	-.3567	-.5535
-.600	-.2134	-.2241	-.2280	-.1991	-.3201
-.450	-.5063	-.2203	-.1468	-.3270	-.2552
-.300	-.3029	-.0908	-.1185	-.0985	-.1056
-.150	.1126	-.0909	-.2099	-.0997	-.0439
.150	-.1033	-.2971	-.1904	-.6639	-.6033
.300	-.2508	-.3429	-.6555	-.4595	-.5877
.450	-.0363	-.4534	-.5559	-.3302	-.8035
.600	-.0754	-.2212	-.2080	-.2214	-.3527
.750	-.0703	-.1199	-.1134	-.0560	-.1240
.900	-.1460	-.0129	-.0275	.0488	-.0105

~~SECRET~~

MACH ( 1 ) = .165    ALPH<sub>A</sub> ( 3 ) = 4.985    AN/L = 1.200    MACH = .165

SECTION ( 1 )WING      DEPENDENT VARIABLE CP

X/C						
.900	-.2011	-.0904	-.0016	-.1116	-.2188	
-.750	-.1261	-.1428	-.2332	-.2585	-.3290	
-.600	-.0395	-.0611	-.0653	-.0561	-.2267	
-.450	-.2903	-.0172	.0254	-.1494	-.1465	
-.300	-.0516	.0946	.0920	.0759	.0020	
-.150	.2019	.0952	-.0306	.1316	.1230	
-.150	-.1871	-.4961	-.9632	.92	-.9701	
.300	-.3353	-.4711	-.7993	.59	-.7458	
.450	-.0820	-.5190	-.4482	-.3788	-.7737	
.600	-.1118	-.2704	-.2778	-.2754	-.3395	
.750	-.1254	-.1518	-.1593	-.1294	-.1773	
.900	-.1428	.0020	-.0523	-.0468	-.0549	

MACH ( 1 ) = .165    ALPH<sub>A</sub> ( 4 ) = 9.980    AN/L = 1.200    MACH = .165

SECTION ( 1 )WING      DEPENDENT VARIABLE CP

X/C						
.900	-.0907	.0026	-.0134	-.0495	-.2296	
-.750	.0139	.0038	-.0819	-.1305	-.2852	
-.600	.0936	.0931	.0787	.0527	-.1406	
-.450	-.1167	.1480	.1877	.0176	-.0470	
-.300	.1226	.2388	.2347	.2065	.0693	
-.150	.2686	.2221	.1612	.2955	.2900	
.150	-.2565	-.7175	-.11518	-1.1641	-1.2907	
.300	-.4094	-.6126	-.9255	-.6963	-.8400	
.450	-.1019	-.6302	-.5247	-.4987	-.5563	
.600	-.1445	-.3359	-.3274	-.3613	-.4539	
.750	-.1532	-.1783	-.1724	-.0581	-.3016	
.900	-.1475	.0300	.0031	.0266	-.1576	

MACH ( 1 ) = .165    ALPH<sub>A</sub> ( 5 ) = 14.980    AN/L = 1.200    MACH = .165

SECTION ( 1 )WING      DEPENDENT VARIABLE CP

X/C						
-.900	.0466	-.0363	.0139	-.0561	-.2909	
-.750	.1313	.1066	.0194	.0346	-.2620	
-.600	.2134	.2166	.1889	.1437	-.5578	
-.450	.0241	.2682	.3071	.1267	.0231	
-.300	.2393	.3414	.3404	.3338	.1496	

|||||

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TABULATED SOURCE DATA - CASTS

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## CAST-B

(ROWS)

## CAST-C

(ROWS)

## CAST-D

(ROWS)

## CAST-E

(ROWS)

## SECTION 1 WING

MACH = .165

(ROWS)

X/C	.000	.061	.123	.098	.075	.412
- .900	.2076	.1771	.0847	.0272	.2738	
- .750	.3062	.2868	.2607	.1970	.0310	
- .600	.1295	.3590	.3965	.2158	.0914	
- .450	.3712	.4175	.4195	.3926	.1553	
- .300	.4671	.4014	.4339	.4255	.3224	
- .150	.5672	.8448	.9622	.13977	.14142	
.300	.5782	.6637	.1.2434	.1.4617	.1.3152	
.450	.1718	.7505	.9692	.1.2473	.1.2770	
.600	.2153	.4513	.7752	.1.0094	.1.1174	
.750	.1749	.3236	.3196	.6925	.8616	
.900	.1603	.1474	.1316	.5388	.6299	

X/C	.000	.061	.123	.098	.075	.412
- .900	.2076	.1771	.0847	.0272	.2738	
- .750	.3062	.2868	.2607	.1970	.0310	
- .600	.1295	.3590	.3965	.2158	.0914	
- .450	.3712	.4175	.4195	.3926	.1553	
- .300	.4671	.4014	.4339	.4255	.3224	
- .150	.5672	.8448	.9622	.13977	.14142	
.300	.5782	.6637	.1.2434	.1.4617	.1.3152	
.450	.1718	.7505	.9692	.1.2473	.1.2770	
.600	.2153	.4513	.7752	.1.0094	.1.1174	
.750	.1749	.3236	.3196	.6925	.8616	
.900	.1603	.1474	.1316	.5388	.6299	

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## TABULATED SOURCE DATA - CASTB

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CAST-B B16C3F1 J42 WATE18 WING TOTAL SURFACE

## REFERENCE DATA

WREF	" 4120 84 FT.	XMAP =	43.5940 IN.
LREF	19.2300 IN.	YMAP =	.0000 IN.
BREF	37.9350 IN.	ZMAP =	-.4040 IN.
SCALE	= .0405		

MACH ( 1 ) = .165 ALPHA ( 1 ) = -.010 RNL = 1.200 MACH = .165

## SECTION 1: WING

## DEFINITION VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C				
-.900	-2928	-1677	-1757	.2215
-.750	-3428	-3121	-4372	-.4427
-.600	-3547	-3689	-.3822	-.3354
-.450	-3657	-4115	-.3408	-.5402
-.300	-5.53	-2853	-.3491	-.2897
-.150	0357	-3116	-.4418	-.3890
.150	-.0459	-1221	-.2745	-.4095
.300	-.1738	-1678	-.4527	-.3412
.450	0130	-3292	-.2517	-.2651
.600	-.0437	-1732	-.1624	-.1878
.750	-.6222	-1084	-.0991	-.0513
.900	-.1572	-.0029	-.0345	-.0735

MACH ( 1 ) = .165 ALPHA ( 2 ) = .015 RNL = 1.200 MACH = .165

## SECTION 1: WING

## DEFINITION VARIABLE CP

X/C				
-.500	-.2754	-.2C36	-.1622	-.1821
-.750	-.2755	-.2713	-.3705	-.3735
-.600	-.1920	-.2312	-.2464	-.2111
-.450	-.4211	-.2222	-.1492	-.3771
-.300	-.2792	-.0624	-.1168	-.0963
-.150	-.1043	-.0913	-.2111	-.1066
.050	-.1021	-.2859	-.5382	-.6070
.300	-.2405	-.3071	-.1400	-.4095
.450	-.0292	-.3769	-.3123	-.2898
.600	-.0797	-.2100	-.2001	-.2223
.750	-.0872	-.1256	-.1188	-.0958
.9	-.1566	-.0393	-.0494	-.0154

(12 NOV 73)

## PARAMETRIC DATA

ZETA	= .000	PIN/P = 1.000
H/B	= .125	BDFLAP = -16.000
ELEVON	= .000	

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TABULATED SOURCE DATA - CASTS

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MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.900 AN/L = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

		CASE-6	B10C5F1	J42	WTE10	WTE	TOTAL SURFACE
21/8	.0000	.3340	.1200	.6630	.6730		
M/C							
- .900	-.2107	-.1014	-.1075	-.1428	-.2257		
-.750	-.1311	-.1533	-.2480	-.2661	-.3299		
-.600	-.0281	-.0680	-.0780	-.0673	-.2315		
-.450	-.2270	-.0180	.0256	-.2265	-.1365		
-.300	-.0391	.1645	.0690	.0161	.0062		
-.150	.1978	.0925	-.0265	.1310	.1210		
.150	-.1802	-.4771	-.8878	-.8309	.9267		
.300	-.3136	-.1260	-.6823	-.5254	.7165		
.450	-.3709	-.1301	-.4022	-.3514	.7753		
.600	-.1140	-.2424	-.2421	-.2597	.3685		
.750	-.1393	-.1469	-.1391	-.1544	.1749		
.900	-.1555	-.0373	-.0462	-.1036	.0349		

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.970 AN/L = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

		CASE-6	B10C5F1	J42	WTE10	WTE	TOTAL SURFACE
21/8	.0000	.3340	.5200	.6630	.6730		
M/C							
- .900	-.0958	-.0186	-.0674	-.0977	-.2325		
-.750	-.0004	-.0179	-.1136	-.1513	-.2868		
-.600	.1102	.0613	.0615	.0416	-.1445		
-.450	-.0193	.1438	.1793	.0781	-.0572		
-.300	-.1227	.2400	.2260	.2003	.0618		
-.150	.2686	.2232	.1607	.2929	.2491		
.150	-.2499	-.6636	-.1.0646	-.1.0582	-.1.1979		
.300	-.4058	-.5425	-.7994	-.6290	-.7607		
.450	-.0937	-.5404	-.4767	-.4143	-.6210		
.600	-.1451	-.3137	-.3058	-.3205	-.4527		
.750	-.1711	-.1879	-.1949	-.1848	-.2913		
.900	-.1599	.0068	-.0736	-.1404	-.1420		

MACH ( 1 ) = .165 ALPHA ( 5 ) = 14.990 AN/L = 1.200 MACH = .165  
 SECTION ( 1 ) WING DEPENDENT VARIABLE CP

		CASE-6	B10C5F1	J42	WTE10	WTE	TOTAL SURFACE
21/8	.0000	.3340	.3200	.6630	.6730		
M/C							
-.900	.0113	-.0531	-.0365	-.1142	-.3093		
-.750	1.093	1.071	1.0256	1.15	1.2746		
-.600	-.2216	-.1941	1.052	1	1.2777		
-.450	2.61	2.623	2.41	2	2.65		
-.300	2.576	2.551	2.55	2	2.54		

## CASTE - 8166511 : 42 WATE10 WING TOTAL SURFACE

(ADW467)

MACH 1.11 = .165    ALPHA 1.51 = 14.990  
 SECTION 111111C  
 21/0 .0000 .3340 .5200 .6630 .8730

## DEFINITION VARIABLE CP

X/C    -.150    .3767    .3103    .3021    .5665    .3250  
 -.150    -.3047    -.7291    -.1072    -.1328    -.5460  
 -.300    -.4701    -.6157    -.8611    -.6839    1.6069  
 -.450    -.0920    -.6045    -.5966    -.5438    1.2254  
 -.600    -.1630    -.3674    -.4491    -.4873    1.1610  
 -.750    -.1759    -.2453    -.3004    -.3377    -.6462  
 -.900    -.1877    -.0514    -.0556    -.2966    -.3697

MACH 1.11 = .165    ALPHA 1.61 = 19.985    RPL = 1.2000    MACH = .165

## DEFINITION VARIABLE CP

SECTION 111111C  
 21/0 .0000 .3340 .5200 .6630 .8730

## DEFINITION VARIABLE CP

X/C    -.900    .6498    -.1748    -.0648    -.1344    -.3673  
 -.750    .1922    .1459    .5462    .0497    -.2604  
 -.600    .9051    .2564    .2335    .1689    -.0461  
 -.450    .1567    .5258    .3862    .1544    .0755  
 -.300    .3705    .4184    .4117    .3939    .1751  
 -.150    .4673    .4071    .4299    .4368    .3377  
 .150    -.3606    -.8041    -.9039    -.7192    1.0438  
 .300    -.5555    -.6121    -.6567    -.2276    -.9894  
 .450    -.1548    -.7017    -.8046    -.9288    -.9054  
 .600    -.2204    -.4672    -.6387    -.6369    -.7203  
 .750    -.2031    -.3717    -.4330    -.5646    -.6231  
 .900    -.2228    -.1694    -.2160    -.4908    -.4903



DATE 09 OCT 74

TABULATED SOURCE DATA - C4378

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CA57-B B10C5F1 J42 WTE10 WING TOTAL SURFACE

(ADVSS) (12 NOV 73)

## REFERENCE DATA

SURF #	4,4120 SF FT.	ZMAP =	43,3940 IN.	BETA =	.000
LREF #	19,2300 IN.	MAP =	.0000 IN.	M/B =	.266
WHL #	37,9350 IN.	ZMAP =	-.4050 IN.	BOFLAP =	-18,000
SCALE #	.0405			ELEVON =	.000
MACH (1) #	.165	ALPHA (1) =	-.4,000	AN/L =	1,200
				MACH =	.165

## SECTION (1) WING

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-1.1510	-1.1312	-1.053	-1.953	-1.747
	-.750	-.2042	-.2176	-.3246	-.3627	-.3506
	-.600	-.2909	-.3067	-.2514	-.2577	-.3597
	-.450	-.5720	-.3036	-.2327	-.4630	-.3149
	-.300	-.4213	-.2090	-.2590	-.2102	-.2113
	-.150	.0280	-.2480	-.3601	-.3173	-.2561
	.150	-.0514	-.1402	-.5113	-.5023	-.3506
	.300	-.1674	-.2603	-.6116	-.3997	-.4455
	.450	.0194	-.4310	-.2938	-.2962	-.7136
	.600	-.0453	-.1772	-.1620	-.1942	-.5044
	.750	-.0143	-.0826	-.07CA	-.0147	-.0876
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321
	.150	-.1020	-.2691	-.5947	-.7139	-.5844
	.300	-.2316	-.5454	-.7223	-.50e7	-.5787
	.450	-.0350	-.4e17	-.3776	-.3441	-.7786
	.600	-.2714	-.2355	-.2122	-.2217	-.3346
MACH (1) #	.165	ALPHA (1) =	.010	AN/L =	1,200	MACH = .165

## SECTION (1) WINC

.0000 .3340 .5200 .6630 .8790

## DEPENDENT VARIABLE CP

X/C	-1.900	-.2009	-.1616	-.1066	-.1364	-.1720
	-.750	-.1760	-.1824	-.2982	-.3268	-.3229
	-.600	-.2050	-.2429	-.1630	-.1621	-.2676
	-.450	-.4934	-.2072	-.1260	-.3904	-.2286
	-.300	-.2965	-.0630	-.1208	-.0746	-.6903
	-.150	.0920	-.1047	-.2051	-.1061	-.0321

MACH ( 1 ) = .165      ALPHA ( 3 ) = 3.000      RNL = 1.200      MACH = .165

(RDVNRB)

CA57-B B16C5F1 J42 W7E18 WING TOTAL SURFACE

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/B .0000 .3340 .5100 .6630 .8730

X/C	- .900	- .1609	- .1154	- .0913	- .1134	.1975
	- .750	- .1143	- .1200	- .2228	- .2511	.2992
	- .600	- .1005	- .1477	- .0706	- .0722	.2082
	- .450	- .3615	- .0725	.0089	.2862	.1280
	- .300	- .1448	.0340	.0115	.0705	.0048
	- .150	.1626	.0470	.0641	.1043	.1323
	.150	- .1782	- .5011	- .9689	- .9583	- .9471
	.300	- .3330	- .4607	- .8535	- .6188	- .7428
	.450	- .0767	- .5504	- .4783	- .3905	- .7644
	.600	- .1016	- .2864	- .2849	- .2625	- .3786
	.750	- .1981	- .1559	- .1677	- .1692	- .1611
	.900	- .1245	- .0085	- .0743	- .0003	- .0390

MACH ( 1 ) = .165      ALPHA ( 4 ) = 9.980      RNL = 1.200      MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C	- .900	- .1017	- .0297	.0076	.0380	.1989
	- .750	- .0328	- .0194	- .1049	- .1357	.2594
	- .600	.0032	- .0369	.0104	.0124	.1291
	- .450	- .2570	.0679	.1414	.1638	.0304
	- .300	.0079	.1875	.1950	.2051	.0922
	- .150	.2448	.1741	.0966	.2145	.2546
	.150	- .2528	- .6934	- .1839	- .1910	- .3164
	.300	- .4110	- .6119	- .9820	- .7308	- .8634
	.450	- .1023	- .6700	- .5893	- .175	.7303
	.600	- .1384	- .3372	- .5611	- .4011	.4530
	.750	- .1401	- .1617	- .1530	- .1177	.2593
	.900	- .1274	.0519	.0453	.0291	.1027

MACH ( 1 ) = .165      ALPHA ( 5 ) = 15.010      RNL = 1.200      MACH = .165

## SECTION ( 1 ) WING

DEPENDENT VARIABLE CP

21/B .0000 .3340 .5200 .6630 .8730

X/C	- .900	- .0440	- .0251	.0080	.0502	.2997
	- .750	.0212	.0326	.0165	.0850	.2532
	- .600	.0611	.0324	.1190	.0971	.0566
	- .450	- .1573	.1664	.2452	.0763	.0162
	- .300	.1340	.2107	.2401	.2734	.1428



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CA57-8 B16C-5F1 J42 W07E10 WING TOTAL SURFACE

(ADN066)

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.010

## SECTION ( 1 ) WING

21/8 .00000 .33400 .52000 .66300 .8730

## DEPENDENT VARIABLE CP

X/C					
-.150	.3149	.2480	.2254	.3586	.3197
.150	-.3216	-.7963	-.12509	-.14202	-.15964
.300	-.4637	-.6955	-.10952	-.8612	-.12893
.450	-.1089	-.7214	-.6347	-.6674	-.12328
.600	-.1697	-.38016	-.4502	-.5064	-.8501
.750	-.1437	-.2203	-.2212	-.1249	-.1902
.900	-.1649	-.0365	-.0168	-.0384	-.2468

MACH ( 1 ) = .165 ALPHA ( 6 ) = 19.990

## SECTION ( 1 ) WING

21/9 .00000 .33400 .52000 .66300 .8730

## DEPENDENT VARIABLE CP

X/C					
-.900	.0099	.0018	-.0136	-.1075	-.4287
-.750	.0875	.0923	.0049	-.0454	-.2807
-.600	.1713	.1196	.1949	.1608	-.0327
-.450	-.0372	.2730	.1362	.0560	.0794
-.300	.2648	.3632	.3766	.3852	.2014
-.150	.4095	.5383	.3641	.4333	.3579
.150	-.3840	-.8317	-.0930	-.8729	-.1.5738
.300	-.5736	-.6798	-.2940	-.4714	-.4445
.450	-.1430	-.7603	-.9260	-.1.2246	-.1.3875
.600	-.2137	-.4443	-.7876	-.1.1527	-.1.2299
.750	-.1520	-.2802	-.3199	-.5649	-.1.0141
.900	-.1663	-.1084	-.0829	-.5072	-.5832

## OA57-B 816C5F1 J42 WATE16 WING TOTAL SURFACE

(RDVW69) ( 12 NOV 73 )

## REFERENCE DATA

**REF** = 4,4120 SQ.FT.  
**LREF** = 19.2300 IN.  
**UREF** = 37.9350 IN.  
**SCALE** = .0405

**MACH ( 1 )** = .165    **ALPHA ( 1 )** = -3.990    **RNL** = 1,200    **MACH** = .165

SECTION ( 1 ) WING  
 21/B .0000 .3340 .5200 .6630 .8730  
 DEPENDENT VARIABLE CP

X/C	- .900	- .1867	.1523	- .1147	- .1513	- .1783
	- .750	- .1804	- .2103	- .3219	- .3461	- .3499
	- .600	- .2660	- .2965	- .2560	- .2600	- .3644
	- .450	- .5121	- .3901	- .2274	- .1984	- .3173
	- .300	- .4097	- .1681	- .2489	- .2980	- .2097
	- .150	.0337	- .2367	- .3531	- .3145	- .2543
	.150	- .0415	- .1192	- .2580	- .4377	- .3170
	.300	- .1587	- .1689	- .5233	- .3627	- .4312
	.450	.0273	- .3751	- .2641	- .2896	- .7247
	.600	- .0347	- .1579	- .1379	- .1730	- .3092
	.750	- .0221	- .0676	- .0559	- .0107	- .0790
	.900	- .1257	.0534	.0206	.1069	.0187

**MACH ( 1 )** = .165    **ALPHA ( 2 )** = .005    **RNL** = 1,200    **MACH** = .165

SECTION ( 1 ) WING  
 21/B .0000 .3340 .5200 .6630 .8730  
 DEPENDENT VARIABLE CP

X/C	- .900	- .1727	.1549	- .1041	- .1353	- .1677
	- .750	- .1459	- .1788	- .2856	- .3078	- .3186
	- .600	- .1804	- .2242	- .1828	- .1752	- .2816
	- .450	- .4285	- .1934	- .1107	- .1039	- .2253
	- .300	- .2863	- .0640	- .1052	- .0749	- .0885
	- .150	.0938	- .0931	- .1926	- .0937	- .0223
	.150	- .0923	- .2618	- .5398	- .6240	- .5569
	.300	- .2266	- .3074	- .6263	- .4463	- .5572
	.450	- .0139	- .4259	- .3264	- .3051	- .7908
	.600	- .0629	- .1942	- .1764	- .1906	- .3267
	.750	- .0501	- .0857	- .0825	- .0331	- .1073
	.900	- .1246	.0169	.0007	.0807	.0062

## PARAMETRIC DATA

**BETA** = .000    **PIN/P** = 1.300  
**H/B** = .286    **BDFLAP** = -16.000  
**ELEVON** = .000



## CA57-B 316CF1 J42 WTE1A WING TOTAL SURFACE (INDW65)

MACH ( 1) = .165 ALPHA ( 3) = 5.01° RNL = 1.200 MACH = .165

SECTION ( 1)WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6530 .8730

X/C

-.900	-.1391	-.1137	-.0796	-.1132	-.1946
-.750	-.0917	-.1173	-.2197	-.2463	-.2960
-.600	-.0769	-.1359	-.0779	-.0811	-.2127
-.450	-.3504	-.0631	.0231	.0031	-.1273
-.300	-.1366	.0676	.0540	.0650	.0012
-.150	.1616	.0555	-.0562	.1041	.1314
.150	-.1749	-.4762	-.9141	-.8825	-.9161
.300	-.5216	-.4401	-.7695	-.5621	-.7102
.450	-.0631	-.4961	-.4171	-.3516	-.7556
.600	-.0997	-.2482	-.2555	-.2453	-.3784
.750	-.1120	-.1275	-.1346	-.0976	-.1521
.900	-.1363	-.0048	-.0369	-.0162	-.0352

MACH ( 1) = .165 ALPHA ( 4) = 10.01° RNL = 1.200 MACH = .165

SECTION ( 1)WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6530 .8730

X/C

-.900	-.0914	-.0491	-.0237	-.0622	-.1974
-.750	-.0275	-.0369	-.1243	-.1610	-.2612
-.600	.0200	.0499	.0247	.0120	-.1421
-.450	-.2244	.0598	.1337	.1147	-.0401
-.300	.0034	.1628	.1820	.1892	.0841
-.150	.2441	.1666	.0635	.2584	.2439
.150	-.2451	-.6855	-.1521	-.1273	-.1274
.300	-.4115	-.5695	-.9037	-.6649	-.8208
.450	-.0915	-.6262	-.5263	-.4763	-.6808
.600	-.1335	-.3230	-.3350	-.3627	-.4346
.750	-.1541	-.1657	-.1679	-.078C	-.2538
.900	-.1418	.0335	.0092	.0381	-.1181

MACH ( 1) = .165 ALPHA ( 5) = 14.99° RNL = 1.200 MACH = .165

SECTION ( 1)WING DEPENDENT VARIABLE CP

21/8 .0000 .3340 .5200 .6530 .8730

X/C

-.900	-.0206	-.0280	-.0026	-.0560	-.2699
-.750	.0357	.0315	-.0304	-.1176	-.2472
-.600	.0996	.0396	.1148	.0838	-.0675
-.450	-.1227	.1746	.2475	.1590	.0167
-.300	.1370	.2773	.2793	.2635	.1439

MACH ( 1 ) = .165    ALPHA ( 5 ) = 14.990  
 SECTION ( 1 ) WING  
 21/8    .0000    .3340    .5200    .6630    .8730

## DEPENDENT VARIABLE CP

X/C	-1.150	.3162	.2555	.2322	.3887	.3249
	-.150	-.3089	-.7532	-.1.1771	-.1.3063	-.1.5212
	.300	-.4715	-.6562	-.9624	-.7489	-.1.3000
	.450	-.0840	-.6645	-.5943	-.5878	-.1.2469
	.600	-.1561	-.3632	-.4252	-.4779	-.7988
	.750	-.1500	-.2194	-.2300	-.1712	-.4367
	.900	-.1675	-.0362	-.0246	-.0858	-.2275

MACH ( 1 ) = .165    ALPHA ( 6 ) = 19.975    AN/L = 1.200    MACH = .165

## DEPENDENT VARIABLE CP

X/C	-.900	.0190	-.0194	-.0411	-.1282	-.3922
	-.750	.0883	.0717	-.0182	-.0986	-.2756
	-.600	-.1860	.1149	.1837	.1253	.0399
	-.450	-.0216	.2732	.3331	.2194	.0651
	-.300	.2597	.3604	.3677	.3710	.1979
	-.150	.4119	.3421	.3641	.4379	.3596
	.150	-.3631	-.6320	-.0332	-.1.7679	-.1.3365
	.300	-.5703	-.6524	-.1.1935	-.1.2933	-.1.2990
	.450	-.0760	-.7361	-.9015	-.1.1053	-.1.2056
	.600	-.2041	-.4469	-.7431	-.9596	-.1.0459
	.750	-.1720	-.3185	-.3235	-.3907	-.8305
	.900	-.1169	-.1366	-.1201	-.5177	-.5625

CAST-B B16C5F1 J42 WTE18 WING TOTAL SURFACE

(RDW69)

## CA37-B B16C5F1 J42 WATE18 WING TOTAL SURFACE

(RDW70) (12 NOV 73)

## REFERENCE DATA

BREF	3	4.4120 SQ.FT.	XMAP	=	43.5940 IN.
LREF	=	19.2300 IN.	YMAP	=	.0000 IN.
BREF	2	37.9330 IN.	ZMAP	=	-.4050 IN.
SCALE	=	.0005			

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (1) = -4.025 \quad \text{RN/L} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

## DEPENDENT VARIABLE CP

21/8 .0000 .3340 .3200 .6630 .8730

X/C	- .900	- .2061	- .1691	- .1201	- .1544	- .1633
	- .750	- .2080	- .2337	- .3298	- .3497	- .3518
	- .600	- .2401	- .2902	- .2674	- .2552	- .3695
	- .450	- .4134	- .2956	- .2278	- .2161	- .3223
	- .300	- .4075	- .1928	- .2602	- .2116	- .2142
	- .150	.0320	- .2318	- .3521	- .3136	- .2551
	.150	- .0354	- .0997	- .2243	- .3625	- .3108
	.300	- .1561	- .1480	- .4199	- .3163	- .4183
	.450	.0612	- .3044	- .2207	- .2386	- .7161
	.600	- .0394	- .1517	- .1326	- .1618	- .3091
	.750	- .0513	- .0869	- .0744	- .0254	- .0770
	.900	- .1262	.0303	- .0007	.0974	.0153

$$\text{MACH } (1) = .165 \quad \text{ALPHA } (2) = -.010 \quad \text{RN/L} = 1.200 \quad \text{MACH} = .165$$

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	- .900	- .1897	- .1774	- .1335	- .1492	- .1717
	- .750	- .1659	- .1952	- .2964	- .3133	- .3193
	- .600	- .1525	- .2234	- .1949	- .1817	- .2963
	- .450	- .3491	- .2002	- .1182	- .1250	- .2315
	- .300	- .2813	- .0584	- .1046	- .0139	- .0912
	- .150	.0973	- .0851	- .1959	- .1036	- .0271
	.150	- .0790	- .2336	- .4765	.5612	.5506
	.300	- .2173	- .2739	- .5044	- .3756	.5340
	.450	.0247	- .3504	- .2615	- .2603	- .7708
	.600	- .0579	- .1683	- .1613	- .1844	- .3256
	.750	- .0678	- .0907	- .0845	.0549	- .0992
	.900	- .1295	- .0107	- .3202	.0219	.0070

(RDYWT0)

CA57-B B16C5F1 J42 W07E10 WING TOTAL SURFACE

MACH ( 1 ) = .165 ALPHA ( 3 ) = 4.960 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.1536	-.1406	-.1053	-.1432	-.2046
-.750	-.1016	-.1346	-.2354	-.2584	-.3052
-.600	-.0562	-.1521	-.0979	-.0811	-.2204
-.450	-.2540	-.0768	.0139	.0205	-.1418
-.300	-.1289	.0672	.0458	.0579	-.0083
-.150	.1540	.0490	-.0599	.0964	.1308
.150	-.1699	-.4566	-.8302	-.7864	-.8706
.300	-.3015	-.3956	-.6439	-.4973	-.6837
.450	-.0215	-.4153	-.3697	-.3283	-.7611
.600	-.1039	-.2280	-.2225	-.2337	-.3549
.750	-.1098	-.1240	-.1187	-.1199	-.1525
.900	-.1386	-.0339	-.0351	-.0737	-.0366

MACH ( 1 ) = .165 ALPHA ( 4 ) = 9.960 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.1002	-.0753	-.0823	-.1332	-.2080
-.750	-.0310	-.0550	-.1549	-.1976	-.2723
-.600	.0337	-.0615	.0046	.0196	-.1483
-.450	-.1669	.0559	.1277	.0853	-.0566
-.300	.0088	.1861	.1734	.1766	.0735
-.150	.2360	.1664	.0632	.2543	.2407
.150	-.2365	-.6103	-.1.0366	-.1.0009	-.1.1605
.300	-.3876	-.5139	-.7744	-.6050	-.7701
.450	-.0423	-.5269	-.4602	-.4075	-.6294
.600	-.1419	-.2998	-.2906	-.3005	-.3913
.750	-.1462	-.1733	-.1703	-.1588	-.2489
.900	-.1522	.0107	-.0674	-.1105	-.1130

MACH ( 1 ) = .165 ALPHA ( 5 ) = 15.000 RNL = 1.200 MACH = .165

## SECTION ( 1 ) WING

21/8 .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0295	-.0501	-.0400	-.0999	-.2644
-.750	.0336	.0145	-.0724	-.1343	-.2467
-.600	-.1144	.0318	.0934	.0768	-.0700
-.450	-.0690	.1713	.2435	.1493	.0220
-.300	.1442	.2618	.2771	.2614	.1438



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TABULATED SOURCE DATA - OA37B

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OA37-B 816C5F1 J42 WATE10 WING TOTAL SURFACE

(RDWWT0)

MACH ( 1) = .165 ALPHA ( 5) = 15.000

SECTION ( 1) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .3200 .6630 .8730

X/C

-.150	.3197	.2931	.2257	.3610	.3344
-.150	-.2999	-.7153	-1.1116	-1.1738	-1.5243
-.300	-.4460	-.5937	-.8355	-.6366	-.13426
-.450	-.0449	-.5783	-.5538	-.4990	-1.1701
-.600	-.1556	-.3488	-.4095	-.4108	-.8442
-.750	-.1573	-.2185	-.2641	-.2816	-.5259
.900	-.1818	-.0417	-.0719	-.2329	-.2916

MACH ( 1) = .165 ALPHA ( 6) = 19.995

SECTION ( 1) WING

DEPENDENT VARIABLE CP

21/8 .0000 .3340 .3200 .6630 .8730

X/C

-.900	-.0133	-.0574	-.0795	-.1539	-.3456
-.750	.0784	.0444	-.0463	-.1266	-.2664
-.600	.1943	.0993	.1635	.1131	-.0430
-.450	.0240	.2615	.3218	.1991	.0607
-.300	.2625	.3583	.3601	.3574	.1957
-.150	.4138	.3369	.3578	.4363	.3691
.150	-.3450	-.7825	-.9507	-.1509	-1.1189
.300	-.5403	-.5969	-1.0365	-.0827	-1.0404
.450	-.0750	-.6742	-.7681	-.6680	-.9651
.600	-.1986	-.4632	-.6315	-.7567	-.8034
.750	-.1817	-.3522	-.3974	-.3414	-.6311
.900	-.1973	-.1814	-.1902	-.4279	-.4916

REFERENCE DATA							PARAMETRIC DATA			
MREF	4,4120 84.FT.	XMRP	=	43,5940 IN.	BETA	=	.000	PTN/P	=	1,300
LREF	19,2300 IN.	YMRP	=	.G300 IN.	H/B	=	.286	ELEVON	=	.000
GREF	37,9350 IN.	ZMRP	=	-.4050 IN.						
SCALE	.0405									
MACH ( 1 ) =	.165	ALPHA ( 1 ) =	-4.010	R/V/L =	1.200	MACH =	.165			
SECTION 1 WING										DEPENDENT VARIABLE CP
Z/R/B	.0000	.3340	.5200	.6630	.8730					
X/C										
- .900	-.1450	-.1472	-.1007	-.1644	-.1868					
- .750	-.1971	-.2183	-.3193	-.3661	-.3577					
- .600	-.2118	-.2982	-.2627	-.2665	-.3751					
- .450	-.5119	-.3009	-.2437	-.3943	-.3239					
- .300	-.4103	-.2126	-.2631	-.2210	-.2224					
- .150	-.0223	-.2484	-.3711	-.3311	-.2756					
.150	-.0110	-.1173	-.2700	-.4491	-.3216					
.350	-.1587	-.1592	-.5212	-.3501	-.4319					
.450	.0521	-.3710	-.2534	-.2732	-.7291					
.600	-.0423	-.1657	-.1454	-.1622	-.3191					
.750	.0055	-.0169	-.0635	-.0160	-.0923					
.900	.1254	.0446	.C127	.1112	.0134					
MACH ( 1 ) =	.165	ALPHA ( 2 ) =	.005	R/V/L =	1.200	MACH =	.165			
SECTION 1 WING										DEPENDENT VARIABLE CP
Z/R/B	.0000	.3340	.5200	.6630	.8730					
X/C										
- .900	-.1253	-.1296	-.0830	-.1153	-.1632					
- .750	-.1518	-.1708	-.2723	-.3054	-.3146					
- .600	-.1832	-.2269	-.1751	-.1545	-.2809					
- .450	-.4169	-.1885	-.1168	-.3165	-.2188					
- .300	-.2813	-.0626	-.0939	-.C672	-.2665					
- .150	.0695	-.0930	-.1973	-.0935	-.3277					
.150	-.0904	-.2616	-.5225	-.6331	-.5565					
.300	-.2153	-.3005	-.6173	-.4421	-.5513					
.450	.0230	-.4163	-.3331	-.2968	-.7779					
.600	-.0666	-.1942	-.1724	-.1900	-.3318					
.750	-.5142	-.0810	-.0771	-.0250	-.C945					
.900	.1048	.0243	.C056	.C950	.C133					



MACH (1) = .165 ALPH (3) = 4.990 RNL = 1.200 MACH = .165

(RDYMA71)

## SECTION (1) WING

DEFINITION VARIABLE CP

21/E .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0922	-.0977	-.0734	-.1031	-.1980
-.750	-.0955	-.1153	-.2158	-.2465	-.2928
-.600	-.0765	-.1385	-.0707	-.0624	-.2089
-.450	-.3122	-.0686	.0047	-.2492	-.1297
-.300	-.1338	.0658	.0546	.0723	-.0705
-.150	.1591	.0550	.0523	.1670	.1319
.150	-.1712	-.4745	-.6988	-.6755	-.9274
.300	-.3150	-.4530	-.7614	-.5556	-.7014
.450	.0154	-.4887	-.4181	-.3505	-.7545
.600	-.0992	-.2447	-.2399	-.2385	-.3765
.750	-.0777	-.1243	-.1293	-.0973	-.1358
.900	.0417	.0065	.0335	.0119	-.0351

MACH (1) = .165 ALPH (4) = 10.015 RNL = 1.200 MACH = .165

## SECTION (1) WING

DEFINITION VARIABLE CP

21/E .0000 .3340 .5200 .6630 .8730

X/C

-.900	-.0463	-.0340	-.0241	-.0614	-.1949
-.750	-.0274	-.0226	-.0202	-.1630	-.1644
-.600	.0219	-.0544	.0254	.0319	-.1371
-.450	-.2043	.0596	.1254	-.1843	-.0364
-.300	C:21	.1467	.1461	.1904	.0788
-.150	2391	.1666	.0038	.2601	.2423
.150	2461	-.6901	-.11307	-.11189	-.12459
.300	3944	-.5724	-.8657	-.6688	-.8214
.450	.0432	-.6129	-.5197	-.4700	-.6693
.600	-.1538	-.3168	-.3275	-.3470	-.4227
.750	-.1039	-.1549	-.1585	-.0699	-.2553
.900	.0245	.0361	-.0010	.0485	-.1157

MACH (1) = .165 ALPH (5) = 14.960 RNL = 1.200 MACH = .165

## SECTION (1) WING

DEFINITION VARIABLE CP

-.900	-.0026	-.0119	.0052	-.3622	-.2704
-.750	.0461	.0293	-.0435	-.0895	-.2385
-.600	-.1077	.0340	.1179	.1018	-.0612
-.450	-.0996	1.625	.2464	-.1032	.0176
-.300	1.417	2196	.2853	.2782	.1465

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165

SECTION (1) WING

MACH (1) = .165 ALPHA (5) = 14.000

CA37-B B12CC J42 WTE10 WING TOTAL SURFACE

MACH (1) = .165 ALPHA (6) = 19.970 RNL = 1.200 MACH = .165



## CA57-B B12C5 J42 WTE10 WING TOTAL SURFACE

(RDW72) (12 NOV 73)

## REFERENCE DATA

REFL N 4.4120 52 FT. XMAP = 43.5940 IN.  
 LREF = 10.2500 IN. YMAP = 0000 IN.  
 BREF = 37.9350 IN. ZMAP = -.4050 IN.  
 SCALE = .0405

MACH (1) = .165 ALPHA (1) = -3.985 AN/L = 1.200 MACH = .165

## SECTION 1) WING

## DEPENDENT VARIABLE CP

218 .0000 .3340 .5200 .6630 .8730

X/C	-1.000	-1.1777	-1.1504	-1.1058	-1.0530	-1.0190
	-.750	-.2034	-.2213	-.3139	-.3485	-.3559
	-.500	-.2296	-.2871	-.2681	-.2667	-.2716
	-.450	-.4034	-.2076	-.2400	-.0803	-.3153
	-.300	-.3903	-.1801	-.2503	-.1911	-.2190
	-.150	.0173	-.2229	-.3515	-.3194	-.2586
	.150	-.0204	-.0692	-.2269	-.3666	-.3115
	.300	-.1476	-.1330	-.1963	-.3089	-.4169
	.450	.0599	-.2929	-.2097	-.2253	-.7121
	.600	-.0240	-.1431	-.1270	-.1610	-.3103
	.750	-.0023	-.0681	-.0506	-.0170	-.0609
	.900	.1200	.0347	.0075	.1153	.0250

MACH (1) = .165 ALPHA (2) = .040 AN/L = 1.200 MACH = .165

## SECTION (1) WING

## DEPENDENT VARIABLE CP

X/C	-1.000	-1.1533	-1.1437	-1.1163	-1.1436	-1.1756
	-.750	-.1668	-.1870	-.2942	-.3224	-.3260
	-.500	-.1549	-.2302	-.1937	-.1637	-.2975
	-.450	-.3321	-.1929	-.11219	-.1167	-.2316
	-.300	-.2132	-.0546	-.1034	-.0834	-.0956
	-.150	.0073	-.0647	-.2543	-.1087	-.0311
	.150	-.0739	-.2962	-.4632	-.5676	-.5455
	.300	-.2138	-.2639	-.4467	-.3691	-.5375
	.450	.0359	-.3254	-.2620	-.2448	-.7954
	.600	-.1922	-.1672	-.1645	-.1473	-.3359
	.750	-.2292	-.2791	-.2784	-.2135	-.5948
	.900	-.1466	-.1724	-.2107	-.2105	-.0131

## PARAMETRIC DATA

BETA = .000 PTNP = 1.000  
 H/B = .286 ELEVON = .000  
 = .000

CARTER & SIEGEL SURFACE

(20w:2)

SECTION (1) MING	DEPENDENT VARIABLE CF				
	21/8	.00000	.33340	.52000	.66300
MACH (1) = .165	A(MACH (3)) = 4.955	RNL = 1.200	MACH = .165		
1/C					
- .900	- .1209	- .1182	- .0956	- .1377	- .2048
- .750	- .1048	- .1296	- .2360	- .2764	- .3126
- .600	- .0659	- .1541	- .2989	- .0919	- .2233
- .450	- .2500	- .0760	- .0550	- .0101	- .1391
- .300	- .1257	.0734	.0441	.0521	- .1154
- .150	- .1544	.0510	- .0630	.0505	.0275
.150	- .1637	- .4521	- .8272	- .7502	- .9726
.300	- .3613	- .3847	- .6326	- .5227	- .5615
.450	- .0136	- .4511	- .5821	- .5156	- .7654
.600	- .0972	- .2246	- .2217	.2340	- .1635
.750	- .2757	- .1169	- .1172	- .1243	- .1455
.900	- .0117	- .0117	- .0117	- .0117	- .0117

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- 900	- 2615	- 3519	- 3724	- 1268	- 2012
- 790	- 2617	- 2454	- 1472	- 1924	- 2675
- 600	- 2371	- 2162	2115	2255	- 1457
- 465	- 1435	- 5611	1172	1225	- 2457
- 371	- 2112	1932	1775	1421	- 1157
- 195	- 2517	1771	3543	2672	2425
- 195	- 2325	- 6367	- 12995	- 11732	- 11732
- 1	- 5942	- 642	- 7144	- 456	- 7631
- 413	- 5357	- 5136	- 450	- 3497	- 16199
- 613	- 1551	- 2649	- 2796	- 2963	- 3906
- 701	- 1161	- 1521	- 1562	- 1526	- 2414

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SECTION 11)WING

CA137-B B12C3 J42 WATE10 WING TOTAL SURFACE

(ADONIS72)

MACH (1) = .165 ALPH-A (1) = 14.995

SECTION 11)WING

CA137-B B12C3 J42 WATE10 WING TOTAL SURFACE

DEPENDENT VARIABLE CP

X/C	.190	.3256	.2606	.2331	.3634	.3366
	.150	.2803	.3583	-1.0518	-1.1661	-1.4771
	.300	.4442	.4745	-.8054	-.8216	-1.2877
	.450	.0424	.5668	-.5400	-.4978	-1.1745
	.600	-.1383	-.3336	-.3917	-.4106	-.6207
	.750	-.1263	-.2056	-.2495	-.2712	-.3241
	.900	.0202	-.0302	-.0662	-.2280	-.2827

MACH (1) = .165 ALPH-A (1) = 20.000 RNL = 1.000 MACH = 1.05

SECTION 11)WING

CA137-B B12C3 J42 WATE10 WING TOTAL SURFACE

DEPENDENT VARIABLE CP

X/C	.000	.3340	.3200	.6630	.8730
	.750	.0174	-.0459	-.0760	-.1582
	.867	.0594	.0458	-.0411	-.1054
	.950	.1990	.0699	.1685	.1116
	.0334	.2667	.5177	.2184	.0636
	.2603	.3603	.3621	.3626	.2028
	.4114	.7395	.3584	.4355	.3708
	.150	.5437	-.7857	-.9460	-.11647
	.300	-.5416	-.5901	-.0213	-.0548
	.450	-.0724	-.6630	-.7591	-.0554
	.600	-.2149	-.4643	-.6355	-.7583
	.750	.1591	-.3412	-.3934	-.5241
	.900	-.0019	-.1147	-.1901	-.4277